Purchase Planning Handbook

2017 Avionics Marketplace

There's **never been a better time** for an avionics upgrade. And the new stuff in the works will **blow your socks off.**



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ccording to the Aircraft Electronics Association's (AEA) year-end market report, total worldwide business and general aviation electronics sales for 2016 amounted to some \$2.2 billion, which was down 6.4% from 2015 figures. That was the lowest annual total since 2012, the first year the association began tracking sales. A little more than half of last year's billings came from "forward fit," that is installations in new production aircraft. And new aircraft deliveries were down in every category with the exception of an everso-slight uptick in turboprops.

By contrast, avionics retrofit sales reached new heights in 2016, continuing a positive four-year trend. Roughly twothirds of all avionics were sold in North America.

So, although overall sales of new business and general aviation aircraft are soft, operators are finding plenty of new technology reasons to retrofit their fleets. The following summary of news and developments among avionics manufacturers provides a surfeit of examples of why this is so.

Aspen Avionics

Albuquerque, New Mexico-based Aspen Avionics has earned a supplemental type certificate (STC) to interface its Evolution primary and multifunction displays (PFDs and MFDs) with Garmin's GTX 345 all-in-one transponder. Installing the GTX 345 with an Aspen EFD1000 PFD will allow aircraft owners to take full advantage of viewing ADS-B weather and traffic information directly in their line of sight, which is a good thing.

The 1090 MHz ADS-B Out feature provides users with access to dual-band ADS-B In traffic and subscription-free weather on Aspen Avionics' PFDs and MFDs, and combines a Mode S Extended Squitter (ES) transponder with an optional WAAS/GPS position source. The \$795 upgrade can be installed by any Aspen Avionics authorized dealer.

Aspen has also earned certification to interface its Evolution 1000 Pro PFD with the System 55X autopilot by Genesys Aerosystems. The STC will cover hundreds of light-single and twin-piston

Bombardier Global 7000 vision flight deck

aircraft. When installed, multiple altimeter setting adjustments are no longer required. The upgrade replaces separate monochrome displays, push buttons and panel knobs with a full-color display directly in the pilot's line of sight, saves panel space and can provide smooth, automatic level-offs with no altitude overshoot — just set the desired vertical speed and altitude on the PFD.

Integrating the System 55X autopilot with Aspen displays requires a software unlock and Analog Converter Unit 2 (ACU2). The software unlock is priced at \$1,995 for customers who already have an ACU2 installed. For those customers who require the unit, bundled pricing for the ACU2 and System 55X unlock is \$2,995.

Astronautics Corporation

Last year, the FAA selected Astronautics Corporation of America to research and develop a comprehensive approach to identify and assess potential cybersecurity threats as they relate to aircraft certification and continued operational



safety. The contract covers research to improve ways to identify and resolve such cyberthreats.

The project will include the development of an efficient process that identifies system security vulnerabilities and safety risks, including risk-mitigation information. The researched approach will support the FAA's development of aviation policies, regulations and training requirements to ensure flight safety and secure aircraft network systems from cyberattacks.

Astronautics will adapt its preexisting cybersecurity processes to support the implementation of the FAA aircraft system information security/ protection and safety risk assessment (SRA) framework. That framework will then be used to model aircraft communications addressing and reporting systems (ACARS), a digital data-link for transmission of short messages between aircraft and ground stations via air-band radio or satellite.

All work will be performed at the Astronautics headquarters in Milwaukee, using systems and software engineers from the current cyber team and inhouse development and cybersecurity labs.

As part of the process, Astronautics will collaborate with the FAA, FAAdesignated organizations, and aviation and information security partners with whom the company has established relationships.

The company's product areas include electronic PFDs, engine displays, mission computers, electronic flight bags (EFBs) and servers for airborne applications.

Astronics Max-Viz

East Aurora, New York-based Astronics Corp., through its subsidiary Astronics Max-Viz, recently announced that its Max-Viz 1200 EVS (enhanced vision system) for fixed- and rotary-wing aircraft has been certified to DO-160G environmental testing standards by the Radio Technical Commission for Aeronautics (RTCA).

The solid-state \$9,000 Max-Viz 1200 EVS requires no routine maintenance and features a low-power, uncooled thermal camera. The sensor image can be presented on any video-capable display that accepts composite video (RS-170) NTSC or PAL/analog signals. The 1.2lb. unit is compatible with a variety of display systems including Garmin's G500, 600 and 1000; Avidyne's R9; Bendix King's KMD-850; AvMap's EKP-V; Flight Displays' Flipper; various Rosen monitors; and EFBs.



Astronics Max-Viz 1200 EVS

With its infrared enhanced-vision thermal imaging system, the EVS enables pilots to see when flying day or night in smoke, haze and light fog. The EVS can work as an alternative to, or in tandem with, light-based night-vision goggle (NVG) technologies.

The Max-Viz 1200 EVS complements synthetic-vision displays, allowing pilots to see transient obstructions, like wildlife and construction barriers not in synthetic-vision databases. The system gives real-time confirmation of the operating environment, as well as supporting the approach-to-landing transition from IFR to VFR in marginal visual conditions. Astronics Max-Viz is an EVS supplier to aircraft manufacturers and to the retrofit market with over 40 STCs in fixed- and rotary-wing aircraft.

Avidyne

Melbourne, Florida-based Avidyne has been granted an Approved Model List-Supplemental Type Certificate (AML-STC) for its new IFD550 Navigator with integrated Attitude Reference Sensor (ARS).

The certification also includes FAA approval of Release 10.2 software that includes a host of new features including synthetic vision and two-way wireless connectivity with Avidyne's new IFD100 iPad app. Avidyne has also received approval for its new IFD545, IFD510 and IFD410 FMS/GPS systems.

The IFD550 is a full-featured FMS/ GPS/Nav/Com with all the same

functionality of Avidyne's current IFD540 but with the addition of an integrated ARS. This detects pitch-and-roll motion and enables the display of dynamic synthetic vision with full-motion 3-D "out-the-window" views as well as exocentric "in-trail" views of the aircraft and nearby terrain, obstacles and traffic. The IFD550 also gives pilots the ability to toggle synthetic vision off and view a traditional blue-over-brown attitude display, as well as an overlay of horizontal and vertical deviation indicators, a total velocity vector (TVV)/ flight path marker, and adjustable field of regard. The IFD550 has a list price starting at \$21,999.

Avidyne's Release 10.2 software is available as a field-loadable upgrade for existing IFD540 and IFD440 systems, giving these customers the ability to display synthetic vision views of the host aircraft, along with overlay of flight-plan, color-contoured terrain, obstacles, full-color 3-D traffic and terrain warnings. R10.2 also includes two-way wireless connection to Avidyne's IFD100 iPad app, wireless flight-plan transfer into the IFD, non-TSO TAWS functionality and support for European VFR (Bottlang) charts. In addition, it enables a 16-watt power output option on the



Avidyne IFD550

IFD440, incorporates improvements to Australian Published Holds, orbitaround-a-point circular holds, software enablement for the RDR2000 radar display on the IFD 5-Series, and more.

All new-production IFD 5-series and 4-series models are available now and will begin shipping immediately with R10.2 functionality. The 10.2 software upgrade for existing IFD540 and IFD440 units is available for download directly from the Avidyne website at no charge. Optionally, the software is available on a USB memory stick for \$150 from Avidyne. Costs do not include dealer labor to upgrade existing systems. Pricing for IFD models with synthetic vision starts at \$9,499.

Esterline CMC Electronics

Esterline Corp.'s latest cockpit avionics displays include a new overhead panel

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with touch-screen technology for the Gulfstream G500/G600 — the first of its kind— as well as Mason primary and secondary flight controls.

The Montreal-based company's utility control system touch-screen display technology is part of the Korry line of cockpit controls and displays, and is the first touch-screen control for overhead panels in civilian aviation. It replaces a number of separate display and switching components, while increasing flexibility and reliability. After pioneering this application for the business jet market, Esterline is making the technology available in a format easily adapted to a wider range of aircraft.

Meanwhile, Esterline's new CMA-6024 GPS sensor is a satellite-based augmentation system and ground-based augmentation system (SBAS/GBAS) CAT-I/II/III precision approach system. The sensor is an upgrade to the existing CMA-5024, with an embedded VHF data broadcast (VDB) receiver. It is also fully compliant with ADS-B and required navigation performance (RNP).

Moreover, according to CMC, the CMA-6024 is a plug-and-play, standalone unit requiring no specialized installation or integration support. It's available for forward-fit installation or as a retrofit, occupying the same space as the CMA-5024 it replaces.

The company's new PilotView CMA-1310 EFB and compact CMA-1525 aircraft information server work with CMC's new CMA-1310 or other tablets to secure connectivity with aircraft systems and wireless and satcom aircraftground communications. It's a tool that bridges the aircraft and the outside world.

The PilotView is offered as a standard option on the Bombardier Challenger 600 series; Dassault Falcon 900, 2000, 7X and 8X; Embraer Legacy 600/650; and Boeing Next-Generation 737s and BBJs.

FreeFlight Systems

CMD Flight Solutions, an engineering and certification company, recently obtained European Aviation Safety Agency (EASA) validation of FreeFlight Systems' AML-STC for the integration of the Waco, Texas, manufacturer's 1203C SBAS/GNSS sensor paired with Rockwell Collins TDR-94/94D transponders. The pairing is a cost-effective way to help aircraft owners meet any ADS-B mandate worldwide.



Freeflight AML-STC

Meanwhile, Signature TECHNICAir has purchased FreeFlight's ADS-B STC for turboprops for ADS-B compliance on Beechcraft King Air C90 models. This transfer of ownership will allow TECH-NICAir to modify and enhance the STC to meet the diverse needs of the King Air fleet while also expanding the offering to include the light jet market. The G1000 NXi is a faster, modernized and lighter avionics suite with expanded capabilities.

On the surface, the G1000 NXi's physical enhancements and display advancements incorporate faster processing power to support faster map rendering and smoother panning throughout the displays. The Olathe, Kansas, manufacturer says the displays initialize within seconds after start-up, providing immediate access to frequencies, flight-plan data and more. The system also incorporates contemporary animations and modernized design for improved readability. New LED backlighting increases display brightness and clarity, reduces power consumption and has improved dimming performance.

Garmin's Connext wireless cockpit



Garmin G500H

STC provides a fully compliant ADS-B solution to C90 operators.

Garmin International

The G1000 NXi, Garmin's successor to its popular G1000 integrated flight deck, features wireless cockpit connectivity, wireless aviation database updates using Garmin Flight Stream, enhanced situational awareness with SurfaceWatch, visual approaches and map overlay on the horizontal situation indicator (HSI). The FAA has granted STC approval for the G1000 NXi in the King Air 200 and 300/350. EASA approval is expected later this year.

Building on the G1000's success — some 16,000 are in use worldwide — the

connectivity unlocks more capabilities. Available as an option, Flight Stream 510 enables Database Concierge, the wireless transfer of aviation databases from the Garmin Pilot app on a mobile device to the G1000 NXi system. Flight Stream 510 also supports two-way flight



Garmin 5000 in Citation Longitude



Garmin G1000 NXi Hero

plan transfer — the sharing of traffic, weather, GPS information, backup attitude information and more — between the G1000 NXi and compatible mobile devices running Garmin Pilot or Fore-Flight mobile. Garmin's D2 Bravo and D2 Bravo Titanium aviator watches even sync with the app.

G1000 NXi equipped-aircraft are rule-compliant to meet FAA and EASA ADS-B requirements. The G1000 NXi also supports the display of various ADS-B In benefits, including traffic and subscription-free weather. FIS-B weather products include: NEXRAD, METARs, TAFs, PIREPs, winds, temps, NOTAMs, AIRMETs and SIGMETs, as well as exclusive traffic features such as Garmin's patented TargetTrend and TerminalTraffic features, and many more.



For new installations, the G1000 NXi is estimated to provide a weight savings of 250 lb. or more in King Air aircraft. New G1000 NXi installations also utilize a new, fully integrated and lightweight air data and attitude heading reference system (ADAHRS), streamlining the upgrade process. Garmin reports that King Air operators with an existing G1000 system can upgrade to the G1000 NXi with minimal aircraft downtime and disruption of the panel as the displays preserve the same footprint and connector, so panel modifications are not required. New G1000 NXi installations and display upgrades all come with a two-year warranty.

Genesys Aerosystems

MD Helicopters has selected Genesys' Integrated Display Units (IDUs) for three of its new helicopter models, the MD 902 Explorer, the MD 530G Scout Attack Helicopter and the all-new MD 6XX concept aircraft.

Genesys IDU 680 (left) and IDU450 PFD (below)



Meanwhile, the Mineral Wells, Texas, avionics manufacturer also joined with its Russian distributor, Heliatica, to announce validation by the Aviation Register of the Russian Federation for its FAA STC to install the Genesys HeliSAS stability augmentation system and autopilot onboard the Robinson Helicopter R44 and R66, and the Airbus Helicopters H125 family.

In addition to that validation, Heliatica also announced that it had recently completed the first HeliSAS installation on an Airbus H1300 helicopter. This work was done in cooperation with Heliswiss Iberica, a maintenance provider based in Barcelona, Spain, and a Genesys Aerosystems authorized distributor and service center.

Meanwhile, customers who purchase a new Genesys Aerosystems S-TEC 2100 digital flight control system by June 30, 2017, will receive a free (installation separate) Lynx NGT-9000 ADS-B transponder as well. The limited-time offer is meant to encourage twin-piston and turboprop aircraft owners to upgrade to an advanced digital autopilot and meet the upcoming ADS-B mandate at the same time.

Popular with operators of high-performance twins and turboprops, the S-TEC 2100's features include a solid-state three-axis digital flight control system, control wheel steering, IAS hold, GPS steering, heading preselect and hold PFD integration, altitude preselect and hold with autotrim.

The Lynx NGT-9000 touch-screen 1090ES/ADS-B transponder's features include a 978/1090 MHz dual-band receiver, L3 Lynx Tail patented flight ID, aircraft type and ground speed of other ADS-B traffic, a full-color, resistive touch-screen interface, full-color moving maps including TFRs, airports and NOTAMs, full-color graphical and textual weather displays, and a built-in WAAS/GPS requiring no external GPS connections. An embedded NextGen Active Traffic option eliminates the need for a separate box.

Honeywell Aerospace

Operators of several major business jet types will soon have an opportunity to replace outdated Laseref II and III navigation systems with the all-digital Laseref IV ring laser gyro-based inertial reference system (IRS).

This is significant because Honeywell will soon be ending aftermarket product support for Laseref II and III, both of

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Honeywell Primus Epic in Gulfstream G500

which have been out of production for several years.

Aircraft eligible for the upgrade include the Hawker 800, 800XP and 1000; Bombardier Challenger 600, 601-3A/-3R and Global Express; Cessna Citation X; Dassault Falcon 900A/B/C/EX; and Gulfstream GIV and GV.

The Phoenix manufacturer estimates that, depending on the platform, the upgrade will reduce aircraft weight by 25-75 lb. and mean time between failures (MTBFs) will show a 30% increase, thanks to Laseref IV's demonstrated 30,000 hr. This is the fourth-generation ring laser gyro-based IRS in the lightest 4 MCU rack-mountable package. Pricing for the replacement will be announced in the near future.

Meanwhile, the new Cessna Citation Hemisphere large-cabin business jet will feature Honeywell's Primus Epic integrated cockpit system. Honeywell says its transoceanic FMS, including SmartView for lower minimums, and precision inertial reference sensors should enable Hemisphere operators to reach destinations at reduced costs. In addition, the system will provide pilots with a conformal 3-D view of the outside world to improve situational awareness in any weather. And Honeywell's Connected Aircraft feature includes satcom connectivity as well as cockpit, cabin, and maintenance apps and services.

Other Primus Epic features for the Hemisphere include a SmartView synthetic vision system (SVS), IntuVue volumetric weather radar, airport 2-D and 3-D moving maps integrated with SmartView SVS for an "out-the-window, gate-to-gate" view of the airport, RNP, advanced LED large-format, high-resolution displays, touch-screen controls, and Aspire 300 satellite communications enabling simultaneous cockpit voice and data connectivity via the Iridium satellite system for safety services.

Optional JetWave cabin satellite communications deliver high-bandwidth global connectivity for passengers. And Honeywell is offering the latest cockpit safety technologies to Gulfstream operators to modernize their aircraft and enhance safety. STCs for enhanced features were recently received from the FAA for Gulfstream PlaneDeck cockpits on the GIV and GV. The latest suite of upgrades will increase crew situational awareness through the integration of synthetic vision with charts and maps, video capability and the XM ground-based weather displayed on PlaneDeck LCDs.

The suite of upgrades also adds TCAS symbology and XM weather information to PlaneDeck SV's enhanced moving map display as an additional overlay improving the flight crew's situational awareness. The Gulfstream GIV and GV join the GIV-SP, which was certified earlier, to be eligible for the latest cockpit upgrades.

L3 Technologies

To better reflect the company's breadth of offerings and scale, L-3 Communications has changed its name to L3 Technologies Inc. As part of its name change, L3 changed its email and website to http://www.L3T.com

Headquartered in New York City, L3 employs some 38,000 people worldwide as a provider of communication and electronic systems and products used on military, homeland security and commercial platforms. The company is also a prime contractor in aerospace sys-



L3 ACCS NXT

tems, security and detection systems, and pilot training. L3 reported 2015 sales of \$10.5 billion.

Meanwhile, ACSS, an L3 and Thales company, has developed the NXT-700, an ADS-B transponder for legacy corporate aircraft. This next-generation Mode S transponder will satisfy the DO-260B mandate for ADS-B on many legacy aircraft models. ACSS says it will reduce owner/operator costs, as well as downtime, because it is a 1/4 ATR short form-fit installation. The NXT-700 is designed for use on the following legacy aircraft models: Hawker 125-400, -600 and -700, and Beechcraft Hawker 400 SP/Beechjet; Bombardier CL-601-3A and 3R; Cessna CitationJet, Ultra, V, VII and 550; Dassault Falcon 10, 20, 50, 200, 900 and 900B; Gulfstream IIB, III and V; IAI Westwind 1124; and Learjet 35, 35A, 36 and 36A.

The transponder's configuration is compatible with current retrofit TCAS II 7.1 systems and may be able to leverage the aircraft's existing mounting rack and connectors for quick installation. Since no additional control heads are needed, the cockpit configuration will remain the same.

Rockwell Collins

The Cedar Rapids, Iowa-based manufacturer reports a number of recent Pro Line Fusion-related developments, including FAA certification for upgrades to King Air B200 and 350, and Citation CJ3 cockpits.

The King Air B200 series upgrade



Rockwell Collins Pro Line Fusion for B200

provides turnkey compliance with airspace modernization deadlines and transforms the panel with the first touch-screen PFD to be certified for operational use, as well as the largest widescreen PFDs available. Rockwell Collins is promoting the upgrade as enhancing aircraft since it involves the same iconbased, touch-screen technology found on new-production King Airs.

The avionics upgrade for King Air 350s has been expanded to include Rockwell Collins FMS navigation database updates and coverage under its Corporate Aircraft Service Program (CASP) at no additional charge for three years. Pro Line Fusion is designed to be easily updated with software upgrades, and to accommodate future technology enhancements, including the company's HGS-3500 head-up guidance system, EVS-3000 EVS and airport moving map.

Pro Line Fusion made its flying debut on the Citation CJ3 last August



Rockwell Collins Citation CJ3 flight deck

following implementation of the system in conjunction with Duncan Aviation. That upgrade also provides turnkey compliance with airspace modernization deadlines and transforms the panel with widescreen displays, high-resolution synthetic vision and touch-screen navigation.

The upgrades are FAA-certified and EASA-validated, and include ADS-B Out, synthetic vision, an updated FMS with localizer performance and approach procedures with vertical guidance (LPV/APV) and radius-to-fix (RF) legs, and the latest version of the Integrated Flight Information System (IFIS). Similar Pro Line 21 upgrade packages are in development for numerous other aircraft types, including Hawkers, Premiers and more.

Keys component of the Pro Line Fusion avionics system are the HGS-3500, the industry's first HGS developed for midsize and light business aircraft, and multi-spectral EVS-3000, which were certified on the Embraer Legacy 450 and Legacy 500 executive jets — the first such certification for both technologies. The systems bring transformative flight deck technology to the business aviation market segment to enhance pilot situational awareness and increased safety.

The HGS-3500 is designed with waveguide optics that couple with the Legacy 450's and 500's Pro Line Fusion avionics system. It comes standard with synthetic vision for even greater situational awareness, and can be upgraded with an enhanced vision option, enabled through the EVS-3000.

The manufacturer maintains that having synthetic and enhanced vision on the HGS sets the stage for a combined head-up vision system in the future, which will make for a full-time, augmented view of the outside world for enhanced situational awareness and approval for lower operating minima.

Universal Avionics

In an effort to help operators equip their aircraft for the FAA's NextGen mandate, Universal Avionics is extending three of its pricing incentive programs that had been set to expire at the end of 2016. The company's ADS-B Out incentive package program, and the SBAS-FMS upgrade incentive program for the Learjet 40/45/40XR/45XR and Citation Excel/XLS are now available through Dec. 31, 2017.

The ADS-B Out Incentive Package pairs Universal's SBAS-FMS with the



Universal family ADS-B

Rockwell Collins TDR-94(D) Mode S transponder to meet the upcoming NextGen ADS-B Out mandate. Unlike other standalone solutions, Universal's solution includes a TSO C146c-certified FMS, allowing operators to gain LPV as well as provide the necessary sensor requirements to meet data-link mandates like FANS 1/A+, Link 2000+ and FAA Data Comm. The Tucson, Arizona, manufacturer says it has taken a "buildingblock" approach to meeting mandates while adding real, long-term value to the aircraft.

Universal Avionics forward-fit FM-Ses have been featured on Learjets for over 30 years and Citations for more than two decades. Now, those aircraft operators can trade-in their existing FMS for a significant credit toward the purchase of a new, advanced capability SBAS-FMS. SBAS approach procedures like LPV offer several benefits over traditional GPS or ILS procedures.

The NextGen SBAS-FMS upgrade incentive program provides trade-in credit for competitor FMS or GPS systems, and the technology is the foundation for PBN requirements and ADS-B Out compliance.

On the Horizon

Avionics sales may be off temporarily, but avionics makers continue to advance the art and science of their products. The capabilities of near- and long-term forthcoming avionics underscore breakthrough advances in technology. Some examples:

▶ Rockwell Collins sees combined synthetic-infrared vision, blending weather information, personalized information displays, voice recognition, even pilot posture recognition as almost foregone conclusions.

► The NTSB is examining which of the many parameters recorded by flight recorders might most usefully be streamed back via satellite to ATC and flight operations.

► Teledyne Controls and GE Aviation have signed a strategic partnership that should simplify the flow of flight data off

> aircraft, and expand its value through GE's cloud-based platform.

► EFBs are beginning to offer real-time updates of operational information.

► Thales sees the convergence of big data, machine learning and connectivity in industry on the near horizon. And there are plenty more examples out there.

As we detailed in last December's special report, "The Internet of Airborne Things," the interrelation and interactivity of the digital cabin, flight deck, service providers and ATC are steadily increasing. Still, all the geewhizardry must be tempered with the realization that ultimately, the flight crew must be ready to demonstrate fundamental control under any circumstances because the "F" in MTBF is there for a reason. **BCA**



Expertise at the right place

Our Service Centre Network*

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*The ACJ Service Centre Network comes in addition to the Airbus worldwide support network

AIRBUS

VHF PANEL-MOUNT TRANSCEIVERS

Manufacturer	Model Channels Power Output (peak W)		Units/Weight (lb.)	Price	Remarks	
	TS0	Channel Display	Power Required	Size (in.)		
Garmin International 1200 E. 151st St. Olathe, KS 66062	GTR 225	760 w/25 kHz spac- ing; 2280 w/8.33 kHz spacing	10W or 16W	1/2.30	Not	
(800) 800-1020 (913) 397-8200 Fax: (913) 397-8282 www.garmin.com	C34e, C36e Class A; C40c, C128a, C169a Class 3, 4, 5, 6, C, E, H1, H2	LCD	9 - 33 V	6.25 x 1.65 x 10.4	provided by OEM	
	GNC 255	760 w/25 kHz spac- ing; 2280 w/8.33 kHz spacing	10W or 16W	1/3.02	Not provided	
	C37c C38c	LCD	9 - 33 V	6.25 x 1.65 x 10.4	by OEM	
Honeywell Aerospace	BendixKing KY 196A	760	16 nominal	1/3.2	\$5.264	Active/standby frequency; stored
BendixKing 9201 San Mateo Blvd NE	C37c C38c	LED	28 VDC	6.3 x 1.35 x 10.8	<i>4</i> 3,304	channels; LED, 28 V.
Albuquerque, NM 87113	BendixKing KY 197A	760	10 nominal	1/3.2	\$5.202	
(855) 250-7027 www.bendixking.com	C37c C38c	LED; nvm	14 VDC	6.3 x 1.35 x 10.8	φ 3,3 23	LED, INVIVI
	BendixKing KY 196B	2280	18 nominal	1/3.2	\$7.208	Active/standby frequency; stored
	C37c C38c	LED; nvm	28 VDC	6.3 x 1.35 x 10.8	φ1.290	channels; LED 28 VDC.

VHF PANEL-MOUNT TRANSCEIVERS

Manufacturer	Model	Channels	Power Output (peak W)	Units/Weight (lb.)	Price	Remarks
	TS0	Channel Display	Power Required	Size (in.)		
Honeywell Aerospace BendixKing						
Blvd. NE						
87113 (855) 250-7027	BendixKing KX 165-21	760 com; 200 nav	10 nominal	1/5.65	\$6.214	Nav/Com/GS/VR/LC CV 14V 760
www.bendixking.com	C37b C38c	LCD; non-volitile	28 VDC	6.25 x 2.05 x 10.16	φ 0 ,214	freq.
	BendixKing KX 165A-01	760 com; 200 nav	10 nominal	1/4.0	¢E 0.91	Nov(com 25 kHz; 28)/
	C37d; JTSO-2C37e; C38d; JTSO-2C38e	LCD; non-volitile	28 VDC	6.25 x 2.05 x 10.16	Ф <u></u> ,901	Nav/Com, 23 kmz, 26v
	BendixKing KX 165A-02	760 com; 200 nav	10 nominal	1/4.0	\$6,015	Nav/Com, 25 and 8.33 kHz; 28V
	C37d; JTSO-2C37E; C38d; JTSO-2C38e	LCD; non-volitile	28 VDC	6.25 x 2.05 x 10.16		

VHF REMOTE-MOUNT TRANSCEIVERS

Manufacturer	Model	Frequency Display	Xmit Power (W)	Units/Weight (lb.)	Price	Remarks
	TSO	Frequency Storage		Size or Form Factor	Power Required	
Aspen Avionics	ATX100	978 MHz In/Out		0.95 lb.	\$3,495	Includes installation kit; single-
Albuquerque, NM 87110 (505) 856-5034 Fax: (505) 314-5440 www.aspenavionics.com	TSO Rule Compliant ADS-B In/Out	_	10-40 VDC	5.0 x 5.75 x 1.7	_	band; meets ADS-B mandate below 18,000 ft.; ADS-B transceiver provides an ADS-B solution for aircraft equipped with a Mode A/C transponder and a WAAS GPS nav receiver.
	ATX100G	978 MHz In/Out		0.95 lb.	N/A	Includes installation kit; single- band; meets ADS-B mandate below 18,000 ft.; ADS-B trans- ceiver provides an ADS-B solu- tion for aircraft equipped with a Mode A/C transponder without a WAAS GPS nav receiver.
	TSO Rule Compliant ADS-B In/Out	—	10-40 VDC	5.0 x 5.75 x 1.7	—	
Honeywell Aerospace BendixKing Avionics 9201 San Mateo Blvd. NE	BendixKing KTR 908	gas discharge		2/4.3	\$16,291	Does not include KFS 598 con- trol head. 152 MHz and SECAL options available.
Albuquerque, NM 87113 (855) 250-7027 www.bendixking.com	C37c C38c	2 (9 channels); 118.0 - 151.975 MHz opt.	20	1.8 x 5.0 x 11.8	28 VDC	
Rockwell Collins 400 Collins Rd. NE Cedar Rapids, IA 52498	VHF-4000	CTL-22 gas discharge		2/4.7	see remarks	Built-in diagnostics; compat- ible only with CSDB or ARINC 429 controls. Options: 001 baseline: includes CTL-22. 101 adds 8.33: includes CTL-22. 201 adds Mode A/2 data; includes CTL-22. 301 adds 8.33 and Mode A/2 data: includes CTL-22C. Prices range from \$13,976 to \$21,892 (BCA estimate).
(319) 295-4085 Fax: (319) 295-2297 www.rockwellcollins.com	C37d C38d	8 frequencies; nvm	20	2.5 MCU	28 VDC	
	VHF-4000E	CTL-22C gas discharge		2/4.7	\$21,892- \$26,264*	Built-in diagnostics; compatible only with CSDB or ARINC 429 controls, Ontions: 101
	C37d C38d	8 frequencies; nvm	20	2.5 MCU	\$21,892- \$26,264*	adds 118.0-151.975 + 8.33; includes CTL-22C. 301 adds Mode A/2 data; includes CTL-22C. * <i>BCA</i> estimate.
	VHF-4000 Transceiver	gas discharge	20	24.7	\$68,620*	*BCA estimate.
		8.33/25 kHz		2/5 MCU	N/A	

HF TRANSCEIVERS

Manufashinan	Model	Frequency Range Xmit Power Units/Weight (lb.)		Price	Demontre	
Wallulacturer	TSO	Channels	(W)	Size or Form Factor	Power Required	Remarks
Honeywell Aerospace	HF-1050	2-29.999		4/29.9	\$77,680	Delivers 200 W PEP transmitter
Harbor Circle Phoenix, AZ 85034 (800) 601-3099 Fax: (602) 365-3343 www.aerospace.honeywell. com	East Sky Circle ix, AZ 85034 301-3099 02) 365-3343 c31d c32d 280,000 c32d		200 PEP (SSB)	KRX 1053 Receiver/Exciter: 5.56 lb; 10.8 x 3.1 x 5.0 KPA 1052 Power Amplifier: 6.67 lb; 12.7 x 7.2 x 1.8 KAC antenna coupler: 9.87 lb; 13.0 x 4.7 x 9.87	28 VDC	power and four squelch options. "Once tuned, always tuned" coupler capability provides <20 millisecond response. PS-440 controller provides 99 user-programmable channels, clarifier functional and coupler tune status.
Rockwell Collins 400 Collins Rd. NE Cedar Rapids, IA 52498	HF 9000 System	2 - 29.9999 SSB/AM AM data		3/27.5	\$99,980*	Fiberoptic interface; rapid-tune antenna coupler (40 millisec-
(319) 295-4085 Fax: (319) 295-2297 www.rockwellcollins.com	C31d C32d	280,000; 99 operator program- mable; 176 ITU r/t programmed	selectable power output 10.50, 175 PEP	controller: 2.625 x 5.75 x 5.85 transceiver: 7.625 x 5.55 x 12.60 antenna coupler: 7.6 x 3.8 x 13.0	28 VDC	ond computer training); BITE. Includes HF receiver/transmitter antenna coupler and radio- tuning unit. *Special order price and delivery.

HORIZONTAL SITUATION INDICATORS/COMPASS SYSTEMS

Manufacturar	Model	Gyro	Autopilet Outpute	Units/Weight (lb.)	Price	Domorko
Manufacturer	TSO	Slave Rate	Autophot Outputs	Form Factor	Power Required	Remarks
Astronautics 4115 N.Teutonia Ave. Milwaukee, WI 53209-6731 (414) 449-4000 www.astronautics.com	Roadrunner Electronic Flight Instrument (EFI) /none	see remarks	N/A	1/ 8.0 lb. / 4.65 x 4.98 x 1.65 in. electronics unit, 5.0 x 9.67 x 6.96 in. display head	— 28 VDC <50 Watts - optional 115 VAC 400 Hz primary power	Provides an upgrade for existing HSI/ ADI primary flight instruments. Capable of displaying weather, synthetic vision, terrain awareness and traffic informa- tion. interfaces with ARINC 429 input and output, differential analog, discrete interfaces, RS232 (bi-directional), synchro and resolver, direct output for TAWS aural alerts. Options include ARINC 453 and ARINC 568 inputs, and connectors matching legacy instru- ments. TSO approvals are planned.
	N/A	N/A		N/A	N/A	
Garmin International	GI 106B	—		1/1.4	\$2,599	
1200 E. 151st St. Olathe, KS 66062 (800) 800-1020 (913) 397-8200 Fax: (913) 397-8282 www.garmin.com	TSO C34e, C36e, C40c	_	14/28V	3.25/3.25/4.75	_	Course deviation indicator (CDI) with needle and glideslope.

HORIZONTAL SITUATION INDICATORS/COMPASS SYSTEMS

Manufacturer	Model	Gyro	Autonilot Outputs	Units/Weight (lb.)	Price	Remarks	
manufacturer	TSO	Slave Rate		Form Factor	Power Required	itemarks	
Honeywell Aerospace 1944 East Sky Harbor Circle				4/8	\$27,937		
Phoenix, AZ 85034 (800) 601-3099 Fax: (602-365-3343 (855) 250-7027 www.aerospace. honeywell.com	Phoenix, AZ 85034 KCS 55A Slaved 800) 601-3099 Compass szax: (602-365-3343 System 855) 250-7027 www.aerospace. noneywell.com	remote/3-deg. per minute	magnetic heading	3.375 in. x 3.375 in.	14-28 VDC	Includes KG 102A directional gyro, KMT 112 flux valve and KA 51B slaving unit and installation kits.	
	KI 825 Color EHSI/MFD	remote	extensive outputs; GPS selected	1/3.0	\$17,638	Integrated EHSI, AMLCD; arc mode; 360 mode; course map; interfaces	
	C113, C6d, C34e, C36e, C40C, C11a	_	discretes, EHSI-ready discretes 3 A	3 ATI	14 - 28 VDC	WX500 Stormscope. Priced for a new installation and a KCM 100.	
Sandel Avionics 2401 Dogwood Way Vista. CA 92081	SN3500 Primary Naviga- tion Display	remote		1/2.9	\$14,113	3-ATI Primary Navigation Display. Sunlight readable LED backlit display with 180 degree viewing angle and over	
Vista, CA 92081 (877) 726-3357 (760) 727-4900 Fax: (760) 727-4899 www.sandel.com	C113, C6d, C34e, C35d, C36e, C40e, C41d, C118, C119B	N/A	N/A	3 ATI	11-33 VDC; 33 W	10,000 hour MTBF. Combines HSI, RMI, color moving map and other fea- tures. Accepts synchro, stepper motor and ARINC 429 gyro inputs. Designed to work with a wide variety of digital and analog NAV, GPS/WAAS, DME, ADF and marker beacon receivers. Built-in LNAV roll-steering interface. Compatible with the WX-500 Stormscope. TACAN interface available. Optional interfaces for traffic (\$980), WSI datalink weather (\$980), Reversionary Attitude (\$980). High-vibration version \$17,714. NVIS compatible version \$21,895.	
	SN4500 Primary Navigation Display	remote		1/3.5	\$20,950	4-ATI Primary Navigation Display. Sunlight readable LED backlit display with 180 degree viewing angle and	
	C113, C6d, C34e, C36e, C40e, C41D, C118, C119B	N/A	analog	4 ATI	22-33 VDC; 40 W	HSI, RMI, color moving map and other features. Accepts synchro, stepper motor and ARINC 429 gyro inputs. Designed to work with a wide variety of digital and analog NAV, GPS/ WAAS, DME, ADF and marker beacon receivers. Built-in LNAV roll-steering interface. Compatible with the WX- 500 Stormscope. TACAN interface available. Optional interfaces for traffic (\$980), WSI datalink weather (\$980), Reversionary Attitude (\$980). High-vibration version \$23,800. NVIS compatible version \$27,050.	

AUTOMATIC DIRECTION FINDERS

	Model	Frequencies	Nav	Units/Weight (lb.)	Price	Remarks
Manufacturer	тео	Frequency Storage	Outputs	Sizo or Form Footor	Power Pequired	
	130	Display			rower keyuneu	
Honeywell Aerospace BendixKing	Bendix/King KR 87	200-1799 kHz		1/6.7 P/M	\$5,395	Times flight and approaches; slaved indicator and RMIs available as options. Incudes KA 44B antenna.
9201 San Mateo Blvd. NE		nvm	analog			
Albuquerque, NM 87113 (855) 250-7027 www.bendixking.com				6.25 x 1.3 x 11.23	11-33 VDC	
	C41c	gas discharge				
Rockwell Collins 400 Collins Rd. NE Cedar Rapids, IA 52498	ADF-4000	190-1799 kHz 2182 kHz emergency freq.	CSDB	3/6.4	\$18,272*	Built-in diagnostics; compatible only with CSDB or ARINC 429 controls; digital
(319) 295-4085		6 frequencies; nvm	ARINC 429			signal processing; dual antenna optional. Includes ANT-462A. *Dual system, \$36,728.
Fax: (319) 295-2297 www.rockwellcollins.com	C41d	gas discharge		2 MCU	28 VDC	

NAVIGATION RECEIVERS (PANEL- AND REMOTE-MOUNT)

Monufacturer	Model	Channel Display	Nav Outputs	Units/Weight (lb)	Price	Domoriko	
Manufacturer	TS0	Channel Storage	GS/MB	Size or Form Factor	Power Required	Remarks	
Honeywell Aerospace BendixKing Avionics 9201 San Mateo Blvd. NE	BendixKing KNR 634A	gas discharge	ARINC 429 CDI, HSI, RMI	2/6.5	\$42,317	Synchro-interface KNI 582 RMI	
Albuquerque, NM 87113 (855) 250-7027 www.bendixking.com	C40a, C36c, C34c, C35d	2 nav; nvm	_	3.0 x 5.0 x 10.0	28 VDC	standby frequencies.	
Garmin International 1200 E. 151st St. Olathe, KS 66062 (800) 800-1020 (913) 397-8200 Fax: ((913) 397-8282 www.garmin.com	GNC 255	LCD	ARINC 429	1/3.02	\$4,495	10W or 16W comm and 200 channel nav with VOR/LOC and G/S receiver.	
	C34e, C36e Class A; C40c, C128a, C169a Class 3, 4, 5, 6, C, E, H1, H2	Recall of frequency from database by facility name and type. Stores/ recalls 15 user defined frequen- cies. Stores/ recalls previous 20 frequencies used	CDI/HSI/EHSI/ EFIS	6.25 x 1.65 x 10.4	9 - 33V		
Rockwell Collins 400 Collins Rd. NE Cedar Rapids. JA 52498	VIR-4000	CTL-32 gas discharge	CSDB ARINC 429	2/3.9	See remarks	Special order item and pricing. Combines ADF and VOR/ILS/MKR	
(319) 295-4085 Fax: (319) 295-2297	C34e, C36e, C40c, C35d	6 frequencies; nvm	NA	2.5 MCU	28 VDC	receivers in a single package. Internal diagnostics capability.	
www.rockweiicollinis.colli	NAV-4500	CTL-32 gas discharge	CSDB ARINC 429	2/4.1	\$23,560*	Built-in diagnostics; compatible only with CSDB or ARINC 429 controls; digital signal processing: includes CTL	
	C34e, C36e, C40c, C35d	6 frequencies; nvm	NA	2.5 MCU	28 VDC	32 (\$4,904); meets Eurocontrol FM immunity standards only. RTU 4200, \$23,880. * <i>BCA</i> estimate.	
	NAV-4000	CTL-32 gas discharge	CSDB ARINC 429	2/4.7	\$32,532*	Built-in ADF; built-in diagnostics; compat- ible only with CSDB or ARINC 429 con-	
	C34e, C36e, C40c, C35d, C41d	6 frequencies; nvm	NA	2.5 MCU	28 VDC	trols; digital signal processing; Eurocontro FM immunity standards.*Configuration wil determine price.	

DISTANCE MEASURING EQUIPMENT

	Model	Channel	Nav Outputs	Units/Weight (lb.)	Price		
Manufacturer	TS0	Display	Power Outputs (peak W)	Size or Form Factor	Power Required	Remarks	
Honeywell Aerospace BendixKing Avionics 9201 San Mateo Blvd, NF	BendixKing KN62A	gas discharge	serial data	1/2.6 P/M	\$8,617	Includes antenna and installation kit; accepts remote channeling. Distance accuracy: ±0.1 nm nominal to 99 nm	
Albuquerque, NM 87113	-		100	6.3 x 1.3 x 12.3	11-33 VDC	±1.0 nm, 100 to 389 nm.	
(855) 250-7027 www.bendixking.com	BendixKing KN63	gas	serial data	2/3.6	\$13,760	Includes KDI 572 indicator, optional slaved indicator, Distance accuracy; ±0.1 nm nomi-	
	C66a	discharge	100	6.5 x 1.1 x 11.55	11-33 VDC	nal to 99 nm, ±1.0 nm, 100 to 389 nm.	
	BendixKing KDM 706A	gas discharge,	ARINC 429 ARINC 568	2/6.3	\$24,093	Includes KDI 572 indicator; optional slaved indicator;	
	C66b	indicators	250	3.0 x 5.25 x 12.8	28 VDC	kits/mounts not included.	
Rockwell Collins 400 Collins Rd. NE Cedar Rapids, IA 52498 (319) 295-4085	DME-4000	gas discharge	CSDB ARINC 429	2/4.4	\$20,100*	Tracks three channels simultaneously when linked to CTL-32, IND-42; decodes and displays station ident; digital signal processing; echo monitor; built-in diagnos- tics; includes IND-42. *BCA estimate.	
Fax: (319) 295-2297 www.rockwellcollins.com	C66c		300	2.5 MCU	28 VDC		

LONG-RANGE NAV/COMS

	Model		Inputs Units/Weight (Ib.)			
Manufacturer	System Type	Outputs	Size or Form Factor	Power Required	Remarks	
Avidyne 55 Old Bedford Rd. Lincoln, MA 01773 (781) 402-7400 (800) AVIDYNE Fax: (781) 402-7599 www.avidyne.com	TSO IFD 550	VHF Com; VOR-LOC-ILS; GPS; SBAS-WAAS, At-	1/8.5	\$21,999	FMS/GPS/NAV/COM system with integrated Attitude Reference Sensor (ARS). Egocentric and Exo- centric Synthetic Vision capability. Designed a drop-in replacement for the GNS530/W series naviga- tors, but with a larger display and touch-screen interface. 5.7-in.	
	GPS receiver with WASS (SBAS) capability, VOR/ ILS/ LOC receiver, VHF com, ARS	titude, WiFi, Bluetooth	2, 0.0		VGA (640 x 480) LED. 16 channel GPS/SBAS receiver with 1,000 user-defined waypoints/99 flight- plans. Includes Forward Looking Terrain Alerting (FLTA), Includes built In Bluetooth/WiFi capabil- ity. Optional TAWS-B, \$7,995; optional 16 W VHF transceiver, \$4,995 (28vdc aircraft only).	
	C34e, C36e, C40c, C110a, C113, C118, C146c, C147, C151b, C157, C165, C169a	-	4.60 x 6.3 x 11.0	11-33 VDC 10-watt VHF Com (16- watt option avail- able)	(\$1,49). Optional Wx Radar Inter- face -BendixKing RDR2000/2100 (\$3,999). Release 10.2 Software Upgrade for existing IFD540 and IFD440 units is available for download directly from the Avidyne website at no charge. Optionally, the software is avail- able on USB memory stick for \$150 from Avidyne. Costs do not include dealer labor to upgrade existing systems. All new-produc- tion IFD 5-Series and 4-Series models are available now and will begin shipping immediately with R10.2 functionality.	
	IFD545 GPS receiver with WASS (SBAS) capability, VOR/ ILS/ LOC receiver , ARS C34e, C36e, C40c, C110a, C113, C118, C146c, C147, C151b, C157, C165, C169a	GPS; SBAS-WAAS 	2.66 x 6.3 x 11.0	\$19,999 11-33 VDC	FMS/GPS Navigator with inte- grated Attitude Reference Sensor (ARS). Egocentric and Exocentric Synthetic Vision capability. Designed as a drop-in replace- ment for GNS500 navigators. 16 channel GPS/SBAS receiver with 1,000 user-defined waypoints/99 flightplans. Includes Forward Looking Terrain Alerting (FLTA). Includes built In Bluetooth/WiFi capability. Optional TAWS-B, \$7,995; optional 16 W VHF trans- ceiver, \$4,995 (28vdc aircraft only).	
	IFD 540	VHF Com;	1/8.5		FMS/GPS/NAV/COM system. Also	
	GPS receiver with WASS (SBAS) capability, VOR/	GPS SBAS-WAAS		\$15,999	ment for the GNS530 and 530W navigators but with a larger dis-	
	C34e, C36e, C40c, C110a, C113, C118, C146c, C147, C151b, C157, C165, C169a	_	4.60 x 6.3 x 11.0	11-33 VDC 10W VHF Com (16W option available)	piay and toucnscreen interface. 5.7-in. VGA (640 x 480) LED. 16-channel GPS/SBAS receiver with 1,000 user-defined way- points/99 flight plans. Includes Forward Looking Terrain Alerting (FLTA) and built-In Bluetooth/ Wi-Fi capability. Optional certified TAWS-B, \$7,995; optional 16 W VHF transceiver, \$4,995 (28VDC aircraft only).	
	IFD 510	GPS; SBAS-WAAS	1/8.0	\$15,995	FMS GPS Navigator. Also designed as a drop-in replace-	
	GPS receiver with WASS (SBAS) capability, VOR/ILS/LOC receiver C34e, C36e, C40c, C110a, C113, C118, C146c, C147, C151b, C157, C165, C169a	_	2.66 x 6.3 x 11.0	11-33 VDC	I ment for GNS500 navigators. 16-channel GPS/SBAS receiver with 1,000 user-defined way- points/99 flight plans. Includes Forward Looking Terrain Alerting (FLTA) and built-In Bluetooth/ Wi-Fi capability. Optional certified TAWS-B, \$7,995; optional 16 W VHF transceiver, \$4,995 (28VDC aircraft only).	

LONG-RANGE NAV/COMS

	Model	Inputs	Units/Weight (lb.)		
Manufacturer	System Type	Outrasta		Remarks	
	TS0	Outputs	Size or Form Factor		
Avidyne 55 Old Bedford Rd. Lincoln, MA 01773 (781) 402-7400 (800) AVIDYNE	IFD 410 GPS receiver with WASS (SBAS) capability,	GPS; SBAS-WAAS	1/6.0	FMS GPS Navigator. Also de- signed as a drop-in replacement for GNS400 navigators. 16-chan- nel GPS/SBAS receiver with 1,000 user-defined waypoints/99 flightplans. Optional Blue-tooth	
Fax: (781) 402-7599 www.avidyne.com	VOR/ILS/LOC receive C34e, C36e, C40c, C110a, C113, C118, C146c, C147, C151b, C157, C165, C169a	_	2.66 x 6.3 x 11.0	Alerting (FLTA) \$1,300. Optional Forward-Looking Terrain Alerting (FLTA) \$1,300 (\$2,000 for both) Optional certified TAWS-B \$7,995; optional 16 W VHF transceiver, \$4,995 (28VDC aircraft only).	
Esterline CMC Electronics 600 Dr. Frederik Philips Blvd. Montreal, Quebec Canada H4M 2S9	CMA-5024	ARINC 429 (complies with ARINC 743B/	1/5.5	Certified SBAS/LPV receiver, Certified SBAS/LPV receiver, fully compliant as an ADS-B and RNP navigation source; provides LP/LPV and SBAS LNAV/VNAV.	
(514) 748-3184 Fax: (514) 748-3100 www.cmcelectronics.ca	GPS receiver with SBAS and LPV	ARING 429)	0.5 × 8.5 × 2.5	growth to GBAS with built-in VDB currently in development, LPV/ GBAS stand-alone approach capability with CMA-5025 control head, Options include: Have	
	C145c Beta-3 C146c Delta-4	ARING 429	9.5 x 8.5 x 2.5	Quick and Doppler velocity radar emulation, meets or exceeds all FAR Part 25 requirements, opera- tion from -55c to +70C.	
	CMA-3024 GPS Receiver	ARINC 429	1/5.5	Low cost certified SBAS Receiver, fully compliant as an ADS-B and RNP navigation source. Ex- tremely reliable, MTBF > 48,000 hrs. Meets and exceeds all Part 25 requirements.	
	C145b Beta -3	ARINC 429	9.5 x 8.5 x 2.5		
Garmin 1200 E. 151st St.	GNC 255	RS-232	1/3.02		
Olathe, KS 66062 (800) 800-1020	VOR/ILS/LOC receiver, VHF com			10W or 16W com and 200 chan-	
(913) 397-8200 Fax: ((913) 397-8282 www.garmin.com	C34e, C36e Class A, C40c, C128a, C169a Class 3, 4 5, 6, C, E, H1, H2	ARINC 429, RS-232	6.25 x 1.65 x 10.4	Guideslope receiver	
	GTN 750	HSDB, ARINC 429, ARINC 453/708, RS-232, RS-422	1/9.3		
	VOR/ILS/LOC receiver, VHF com			Fully integrated GPS/NAV/COM/ MFD system. The unit's 6-in. high touchscreen provides access to high-resolution terrain map-	
	C34e, C35d Class A; C36e Class A, C40c, C63d Class BC, C74d Class A, C110a, C112c, C113 Class I, II, II, C118, C128a, C139, C146c Class 3, C147; among others		6.25 x 6.00 x 11.25	ping, graphical flight planning, geo-referenced charting, traffic display, multiple weather options, connectivity and more.	
	GTN 650	HSDB, ARNC 429, ARINC 453/708, RS-232	1/7.0		
	VOR/ILS/LOC receiver, VHF com	RS-422	2,	Full integrated system combines	
	C34e, C36e Class A, C40c, C74d Class A, C110a, C112c, C113 Class I, II, III, C118, C128a, C146c Class 3, among others	HSDB, ARINC 429	6.25 x 2.65 x 11.25	GPS, COM and NAV functions with MFD capabilties.	
	GSR 56	RS-232	1/2.51	The GSR 56 gives access to on-demand global weather information and text/voice com- munication through the Iridium stellite network.	
	Iridium weather datalink, text/voice communications	RS-232	2.08 x 6.96 x 12.96	TSO: 139	

LONG-RANGE NAV/COMS

Model	Inputs	Units/Weight (lb.)	Price			
Manufacturer	System Type				Remarks	
	TSO	Outputs	Size or Form Factor	Power Required		
Avidyne 55 Old Bedford Rd.	IFD 440	VHF com; VOR-LOC-ILS; GPS; SBAS-WAAS	1/6.6	\$11,999	FMS/GPS/NAV/COM system. Also designed as a drop-in	
Lincoln, MA 01773 (781) 402-7400 (800) AVIDYNE Fax: (781) 402-7599	GPS receiver with WASS (SBAS) capability, VOR/ ILS/LOC receiver, VHF com				replacement for the GNS430 and 430W navigators but with a larger display and touchscreen interface. 16-channel GPS/ SBAS receiver with 1,000 user defined waveninte (00	
www.avidyne.com	C34e, C36e, C40c, C110a, C113, C118, C146c, C147, C151b, C157, C165, C169a	_	2.66 x 6.3 x 11.0	11-33 VDC 10W VHF Com (16W option)	flight plans. Optional Bluetooth and Wi-Fi capability \$1,300. Optional Forward-Looking Terrain Alerting (FLTA), \$1,300 (\$2,000 for both). Optional certified TAWS-B \$7,995. Optional 16 W VHF transceiver, \$4,995 (28VDC aircraft only).	
Honeywell Aerospace	Laseref IV	ARINC 419/429	2/16.9	\$265,361*		
Harbor Circle Phoenix, AZ 85034 (800) 601-3099 Fax: (602) 365-3343 www.honeywell.com	Laser Gyro IRS	ARING 429/ASCB	49x76x131	28 VDC	All digital; 4 MCU, ARINC 704. *BCA estimate.	
	C4c, C5e, C6d		4.5 X 1.6 X 10.1	or 115 VAC		
	Laseref VI Micro IRS	ARINC 429	2/3.2	\$355,992*	Laseref VI Inertial Reference Unit with updated microprocessor,	
	Laser Gyro IRS	ARINC 429	6.5×6464	28 VDC	on-aircraft data load capability; HIGH Step II software for 100%	
	C4c, C5e, C6d, C129a	/		20.00	available RNP. *BCA estimate.	
	AH-1000 Attitude Header Reference Unit	ARINC 429 and discrete I/O	2/3.2	\$33, 334*	AH-1000 is a Microelectrome- chanical System (MEMS) attitude	
	MEMS AHRS				(AHRS) designed to serve as the AHRS for commercial aerospace	
	C3e, C4c, C5f, C6e (ETSO C3d, C4c, C5e, C6e)	ARINC 429	2.5 x 5.0 x7.8	28 VDC	primary and secondary attitude and heading systems. *BCA estimate.	
	KTR 2280		_	—	Functionality consists of a VHF communication transceiver that	
	Multi-Mode Digital Radio C37d, C38d, C40c, C36e, C34e, C41d	ARINC 429 Interface	_	27.5 VDC	can monitor two frequencies simultaneously, a VHF navigation receiver and an ADF receiver is an option enabled via software.	
Rockwell Collins					Dual-transceiver device that	
400 Collins Rd. NE Cedar Rapids, IA 52498 (319) 295-4085 Fax: (319) 295-2297 www.rockwellcollins.com	NxtLink ICS-120A Satcom Terminal	ARINC 429; audio tip/ ring; audio 4 wire; RS- 232; Discretes: RF, USB;	Single 2 MCU rack/6.3 lb.	\$38,022	Dual-transceiver device that combines a single channel of global voice and 2400 bps data service with a second Short Burst Data (SBD) channel in a single 2MCUL BIL The eventee provideo	
		mainenance port; remote SIM and configuration	15.98 x 2.33 x 7.75	28 VDC	the flight crew with an exclusive global voice channel and a dedicated data link channel to support ACARS, FANS messaging, ADS-C and CPDLC.	
	NxtLink ICS-220A	ARINC 429; audio tip/ ring; audio 4 wire; RS-	Single 2 MCU rack/6.3 lb.	\$48,714	Three-transceiver unit combines dual channels of global voice and 2400 bps data service with a third Short Burst Data (SBD) channel in a single 2MCU LRU.	
	Satcom Terminal	232; Discretes: RF, USB; mainenance port; remote SIM and configuration	15.98 x 2.33 x 7.75	28 VC	crew with an exclusive global voice channel and a dedicated data link channel to support ACARS, FANS messaging, ADS-C and CPDLC.	
	GPS-4000S	ARINC 429 (complies with	N/A	\$32,576*		
	GPS receiver w SBAS (WAAS) capability	ARINC 743A)			*BCA estimate.	
	C145A Class Beta-3	ARINC 429	2 MCU	28 VDC		

Model Modes		Modes	Units/Weight (lb.) Price		
Manufacturer	TSO	Power Output (W)	Size or Form Factor	Power Required	Remarks
ACSS, an L3 and Thales Company	Mode S RCZ-852	TCAS/Mode S/Fit ID control panel; RMU (Primus II radios)	5.0	\$78,409	Elementary and Enhanced Surveillance (ELS/EHS) and DO 260 compliant.
Phoenix. AZ 85027	C112	CTL92A/T/E	3.4 x 4.1 x 14.01	28 VDC	Certified on many regional and business jets.
(623) 445-7001 Fax: (623) 445-7000 www.acss.com	Mode S ATDL XS-950	TCAS/Mode S/Fit ID control panel; CTL92A/T/E	1/11.5	\$99,474	DO-260B and DO-181E compliant. Elemen- tary and enhanced surveilance (ELLS/EHS).
	C112		4 MCU	28 VDC	
	NTX-600 Mode S	TCAS/Mode S/Fit ID; control panel; RMU (Primus II radios)	1/5.0	\$51,418	D0-260B and D0-181E compliant. Elemen- tary and Enhanced Surveillance (ELS/EHS).
	C112d, C116b	115 VAC, 400 Hz or 28 VDC	3.4 x 4.1 14.01	NA	Selected for the Bombardier Q400.
	NXT-700	TCAS/Mode S/Fit ID; conrol panel	5.5	\$38,975	D0-2608 compliant; Elementary and Enhanced surveillance (ELS/EHS)
	C112d, C166b	500w max, 250W min	1/2 ATR Short	N/A	
	NXT-800 Mode S	TCAS/Mode S/Fit ID; control panel	1/8.6 (AC); 7.8 lb. (DC)	\$101,900	
	C112d, C166b	115 VAC, 400 Hz or 28 VDC	4 MCU	NA	
Avidyne Corp. 55 Old Bedford Rd. Lincoln, MA 01773	AXP340	Mode A/C/S with extended squitter; ADS-B OUT	1/2.98	N/A	Panel-mounted Class 1 Mode S Level 2 data- link transponder, with 1090 MHz Extended Squitter (ES). Meets requirements for Mode
Lincoln, MA 01773 (781) 402-7400 (800) AVIDYNE Fax: (781) 402-7599 Info @avidyne.com www.avidyne.com	C166b, ETSO 2C112b, ETSO C166b	240	6.3 x 1.57 x 9.4	10 -33 VDC	S elementary surveillance transponders. Slide-in replacement for existing KT76A/ KT78A transponders. Designed to upgrade existing Mode A/C equipment to Mode S, while adding additional functionality such as direct-entry numeric keypad, pressure altitude and GPS Lat/Long readout, Flight ID entry, one-touch VFR code entry, a stop- watch timer/flight timer, and altitude alerter. Supports the latest Version 2 1090 MHz Automatic Dependent Surveillance Broadcast (ADS-B) Extended Squitter (ADS-B out).
	AXP322 ETSOC112b, ETSO C166a, FAA TSO C112c, C166b	Remote-mounted Mode A/C/S with extended squitter; TIS & ADS-B OUT 250	1/0.97 2.68 x 1.90 x 6.30	\$2,999 10-33 VDC	Remote-mounted Class 1 Mode S Level 2 data-link transponder, with 1090 MHz Extended Squitter (ES). Meets requirements for Mode S elementary surveillance transpon- ders and supports legacy Traffic Information Service (TIS). Supports the latest Version 2 1090 MHz Automatic Dependent Surveillance Broadcast (ADS-B) Extended Squitter (ADS-B out). Designed to work with Avidyne's panel- mounted IFD540 & IFD440 panel-mounted FMS/GPS/NAV/COMs for display and control.
Garmin International 1200 E. 151st St. Olathe, KS 66062	GTX 327	A, C	1/3.1	\$2,036	IFR-certified panel-mounted Mode C
(800) 800-1020 (913) 397-8200	C74c Class 1A	200	6.25 x 1.65 x 8.73	11-33 VDC	
Fax: (913) 397-8282 www.garmin.com	GTX 330 ES	A, C, S, ES	1/4.2	\$8,636	IFR-certified panel-mounted Mode C
	C112d Level 2ens Class 1, C 166b Class B1S	250	6.25 x 1.65 x 11.25	11-22 VDC	transponder with ADS-B compliant ES capability.
	GTX 3000	A, C, S, ES	1/5.2	\$24,226	The GTX 3000 Mode S ES remote transpon- der features ADS-B OUT transmission and TCAS II/ACAS II compatibility.
	C112d Level 2adens, Class 1c, C166b Class B1	250 minimum, 300 nominal	2.58 x 6.47 x 10.94	14-28 VDC	

	Model	Modes	Units/Weight (lb.)	Price	
Manufacturer	TSO	Power Output (W)	Size or Form Factor	Power Required	Remarks
Garmin International 1200 E. 151st St.	GTX 335	A, C, S, ES	1/2.83	\$2,995	IFR-certified 250W panel-mounted Mode
Olathe, KS 66062 (800) 800-1020 (913) 397-8200	TSO-C88b, TSO- C112e Level 2es Class 1	250	6.25 x 1.65 x 10.07	9-33V	S transponder with ADS-B compliant ES capability.
Fax: (913) 397-8282 www.garmin.com	GTX 335 w/GPS	A, C, S, ES	1/2.94	\$3,795	IFR-certified 250W panel-mounted Mode S transponder with ADS-B compliant ES capability and built-in WAAS.
	TSO-C88b, TSO- C112e Level 2es Class 1, TSO-C145d Class B2	250	6.25x 1.65 x 10.07	9-33V	
	GTX 335R	A, C, S, ES	1/2.5	\$2,995	IFR-certified 250W remote-mounted Mode
	TSO-C88b, TSO- C112e Level 2es Class 1	250	6.25 x 1.65 x 10.07	9-33V	S transponder with ADS-B compliant ES capability.
	GTX 335R w/GPS	A, C, S, ES	1/2.6	\$3,795	IFR-certified 250W remote-mounted Mode
	TSO-C88b, TSO- C112e Level 2es Class 1, TSO-C145d Class B2	250	6.25 x 1.65 x 10.07	9-33V	S transponder with ADS-B compliant ES capability and built-in WAAS.
	GTX 345	A, C, S, ES	1/3.09	\$4,995	
	TSO-C88b, TSO- C112e Level 2es Class 1, TSO-C154c Class A15, TSO- C157a Class 1, TSO-C166b Class A15/B15, TSO-C195a Class C1, C2, C3, C4	250	6.25 x 1.65 x 10.07	9-33V	IFR-certified 250W panel-mounted Mode S transponder with ADS-B compliant ES capa- bility and ADS-B In benefits.

Manufacture	Model	Modes	Units/Weight (lb.)	Price	Demostra
Manufacturer	TSO	Power Output (W)	Size or Form Factor	Power Required	Remarks
Garmin International 1200 E. 151st St. Olathe, KS 66062 (800) 800-1020 (913) 397-8200 Fax: (913) 397-8282 www.garmin.com	GTX 345 w/GPS	A, C, S, ES	1/3.20	\$5,795	IFR-certified 250W panel-mounted Mode S transponder with ADS-B compliant ES capa- bility, built-in WAAS and ADS-B in benefits.
	TSO-C88b, TSO- C112e Level 2es Class 1, TSO-C145d Class 82, TSO- C154c Class A1S, TSO-C157a Class 1, TSO-C166b Class A1S/B1S, TSO-C195a Class C1, C2, C3, C4	250	6.25 x 1.65 x 10.07	9-33V	
	GTX 345R	A, C, S, ES	1/2.8	\$4,995	IFR-certified 250W remote-mounted Mode S transponder with ADS-B compliant ES capability and ADS-B In benefits.
	TSO-C88b, TSO- C112e Level 2es Class 1, TSO-C154c Class 415, TSO- C157a Class 1, TSO-C166b Class A15/B15, TSO-C195a Class C1, C2, C3, C4	250	6.25 x 1.65 x 10.07	9-33V	
	GTX 345R w/GPS	A, C, S, ES	1/2.9	\$5,795	IFR-certified 250W remote-mounted Mode S transponder with ADS-B compliant ES capability, built-in WAAS, and ADS-B In benefits.
	TSO-C88b, TSO- C112e Level 2es Class 1, TSO-C145d Class 82, TSO- C154c Class A1S, TSO-C157a Class 1, TSO-C166b Class A1S/B1S, TSO-C195a Class C1, C2, C3, C4	250	6.25 x 1.65 x 10.07	9-33V	
Honeywell Aerospace BendixKing Avionics 9201 San Mateo Blvd. NE Albuquerque, NM 87113 (855) 250-7027 www.bendixking.com	BendixKing KT 74	NA	1/2.8	\$3,278	Mode S ADS-B capable.
	ETSO C112d, ETSO C166b, TSO C112d and TSO C166b	240 W nominal; 125 W minimum	1.7 x 6.30 x 10.7	11 or 33 VDC	
	BendixKing KT 76A	A, C	1/3.1	\$2,265	Automatic reply-light dimmer; system test; remote ident capability adapter available.
	C47c; Class 1B	250	6.25 x 1.6 x 10.0	14 or 28 VDC	
	BendixKing KT 76C	A, C	1/3.1	\$3,278	Slide-in replacement for KT 76A. Program- mable VFR code; remote ident capability;
	C47c	250	6.25 x 1.63 x 10.73	11-33 VDC	code entry.
	BendixKing KT 73	A, C, S, TIS	1/3.6	\$6,719	Mode S data link with TIS. Meets European Elementary Surveillance mandate (non-diversity).
	C112	200	6.25 x 1.63 x 10.82	10-32 VDC	
	BendixKing MST 67A	A, C, S	2/8.5	\$46,143	Mada 6 divaraity transporter
	C37c/C38c C74c; Class 3A	250-625	14.0 x 3.0 x 8.9	115 VAC; 400 Hz	INIQUE S UIVERSILY TRANSPONDER.

Manufactures	Model	Modes	Units/Weight (lb.)	Price	Damada
Manufacturer	TSO	Power Output (W)	Size or Form Factor	Power Required	Remarks
L-3 Aviation Products 5353 52nd St. SE Grand Rapids, MI 49512 (616) 949-6600 Fax: (616) 977-6898 www.L-3Lynx.com	Lynx NGT-9000	A, C, S. ADS-B	1/2.96	\$5,495	Touchscreen ADS-B transponder and MFD display. 1090ES (Mode s ES) ADS-B Out, 1090 MHz and 978 MHz (UAT) ADS-B In. ADS-B traffic (1090 and 978 ADS-B, ADS-R and TIS-B) and 978 FIS-B input. WiFi interface module available for connectivity to PED (iPad). Embedded rule compliant position source (WAAS GPS). Embedded options include Class B TAWS and ADS-B aural traffic alerting for the verbal positioning of traffic conflicts.
	C112d, C113a, C145c, C147, C154c, C157a, C166b, C195a	25W minimum/250W maximum	1.8 x x 6.25	14 or 28 VDC	
	Lynx NGT-9000+	A, C, S. ADS-B	1/2.96	\$9,555	Same features as the NGT-9000, but with the added feature of the L-3 NextGen Active Traffic. Active traffic is embedded into the same LRU, requiring no separate boxes. Current SkyWatch owners can re-use existing antenna.
	C112d, C113a, C145c, C147, C154c, C157a, C166b, C195a	25W minimum/250W maximum	1.8 x x 6.25	14 or 28 VDC	
	Lynx NGT-9000D	A, C, S, ADS-B	1/2.96	\$6,915	Same features as NGT-9000 but with Antenna Diversity for the top and bottom of the aircraft.
	C112d, C113a, C145c, C147, C154c, C157a, C166b, C195a	25W minimum/250W maximum	1.8 x 6,25	14 or 28 VDC	
	Lynx NGT-9000D+	A, C, S, ADS-B	1/2.96	\$10,935	Same features as NGT-9000, but with the added feature of the L-3 NextGen Active Traffic and Antenna Diversity.
	C112d, C113a, C145c, C147, C154c, C157a, C166b, C195a	25W minimum/250W maximum	1.8 x 6.25	14 or 28 VDC	
	Lynx NGT-9000R	A, C, S, ADS-B	1/2.96	\$5,445	Remote version of the NGT-9000 that integrates and is conrolled by newer aircraft outfitted with glass panels.
	C112d, C113a, C145c, C147, C154c, C157a, C166b, C195a	25W minimum/250W maximum	1.8 x 6.25		
Rockwell Collins	TDR-94D	S	2/8.5	\$56,656	Mode S transponder; D0-260B ADS-B Out
Cedar Rapids, IA 52498 (319) 295-4085 Fax: (319) 295-2297 www.rockwellcollins.com	C112; Class 3A	250-625	4.9 x 3.3 x 12.5	28 VDC	hanced Surveillance compliant. Compatible with TCAS II Change 6.04, Change 7.0 and Change 7.1 systems. TDR-94 transponder also available for non-diversity applications. Flight ID capable CTL-92E controller available

WEATHER RADAR

	Model	Ranges	Dich Size	Scan	Stablztn.	Display	Scope	Units/	Price	
Manufacturer	TSO	Dewor	& Beam	Pulse Width	Stabl. Sig.	Interface	(dia./in.)	Weight		Remarks
	Circuits	Output Output (Peak KW)	(in./deg.)	Looks/Min.	Ant. Tilt	Colors	Indicator Size	RT. Size	Power	
Garmin International 1200 E. 151st St. Olathe, KS 66062 (800) 800-1020 (913) 397-8200 Fax: (913) 397-8282 www.garmin.com	GWX 70	Select- able: 2.5, 5, 10, 20, 40, 60, 80, 100, 120, 160, 240, and 320 nm		20, 40, 60, 90, or 120 (HSDB interface); 20 or 120 (ARINC interface)	±30°	HSDB		10 in 9.3 12-in -	in .3	Designed for use in a variety of aircraft.
	C63D Class B and C	(HSDB interface) 2.5, 5, 10, 20, 40, 80, 160 and 320 nm (ARINC interface)	10/9.0 12/7.8 18/5.6	1.6, 3.2, 6.4 or 13.6	_	ARINC 429/453	NA	9.5 18-in 11.0	\$21,995	include 10, 12, and 18 in. Circuits: horizontal and vertical scan; tile, bearing, sector san, gain, stabilization, ex attenuated color high- light, alt. compensated
	See remarks	40 W nominal		12 (Range 20 nm or below) 9 (Range 20 nm or above)	±15°	4	NA	8.0 x 9.69 x 7.08 (10 in. & 12 in.) 8.78 x 10.06	14-28 VDC 2.5A @28V	tile and ground clutter suppression control; turbulence detection.
	GWX 70R	Select- able: 2.5, 5, 10, 20, 40, 60, 80, 100, 120, 160, 240, and 320 nm		20, 40, 60, 90, or 120 (HSDB interface); 20 or 120 (ARINC interface)	±30°	HSDB		10 in 9.3 12 in -	\$21,995	Designed for use in a variety of aircraft. Available antenna sizes include 10, 12, and 18 in. Circuits: horizontal and vertical scan; tile, bearing, sector san, gain, stabiliza- tion, ex attenuated color highlight, alt. compensated tile and ground clutter suppression control; turbulence detec- tion.
	C63D Class B and C	(HSDB interface) 2.5, 5, 10, 20, 40, 80, 160 and 320 nm (ARINC interface)	10/9.0 12/7.8 18/5.6	1.6, 3.2, 6.4 or 13.6	_	429/453	NA	9.5 18-in 11.0		
	See remarks	40 Wnominal		12 (Range 20 nm or below) 9 (Range 20 nm or above)	±15°	4	NA	8.0 x 9.69 x 7.08 (10 in. & 12 in.)	4-28 VDC 2.5A @28V	
	GWX 70H	Select- able: 2.5, 5, 10, 20, 40, 60, 80, 100, 120, 160, 240, and 320 nm		20, 40, 60, 90, or 120 (HSDB interface); 20 or 120 (ARINC interface)	±30°	HSDB ARINC 429/453		10 in 9.3	\$31,995	The helicopter- optimized GWX 70H combines range and adjustable scanning profiles with preci- sion target definition
	C63D Class B and C	(HSDB interface) 2.5, 5, 10, 20, 40, 80, 160 and 320 nm (ARINC interface)	10/9.0 12/7.8 18/5.6	1.6, 3.2, 6.4 or 13.6	_		NA	9.5 18-in 11.0		for real-time weather analysis in the cockpit. The GWX 70H offers horizontal scan angles of up to 120 deg. and pilot- adjustable sector scapping from
	See remarks	40 W nominal		2 (Range 20 nm or below) 9 (Range 20 nm or above)	±15°		NA	8.0 x 9.69 x 7.08 (10 in. (10 in.) 4-28 VDC in.) 2.5A @28V 8.78 x 10.06 x 9.93 (18 in.)	 sector scanning from 20 deg.to 120 deg. Circuits: ; tile, bearing, sector san, gain, stabilization, ex attenuated color highlight, alt. compensated tile and ground clutter suppression control; turbulence detection. 	

WEATHER RADAR

	Model	Ranges	Dich Cizo	Scan	Stablztn.	Display	Scope	Units/	Price		
Manufacturer	TSO	-	& Beam	Pulse Width	Stabl. Sig.	Interface	(dia./in.)	Weight		Remarks	
	Circuits	Power Output (Peak KW)	(in./deg.)	Looks/Min.	Ant. Tilt	Colors	Indicator Size	RT. Size	Power		
Honeywell Aerospace BendixKing Avionics	BendixKing ADR 2000	10, 20, 40, 80, 160		90° or 100°	30°	KMD 850 EFIS	N/A	1/9.9	\$20,799	Vertical profile feature:	
Albuquerque, NM 87113 (855) 250-7027	C63c		10/10 12/8 15	4	20-220 mv/det.				28 \/DC+	scans horizontally or vertically on track line selected by pilot. Alpha-	
www.benaixking.com	Vertical profile, ext. STC, tgt., wx alert, atten., comp., variable gain-map mode	4		ARINC 429	±15°	4	N/A	10.28 dia.	10, 26, 115 VAC, 400 Hz	numeric display of range, function and tilt angle.	
	BendixKing RDR 2100	5, 10, 20, 40, 80, 160, 240,		90°, 100°, 120°	±30° pitch & roll	KMD 850 EFIS	N/A	1/9.9	\$21,181		
	C63c	320	12/8 10/10	range dependent	20/220 mv/deg. ARINC 429					Vertical profile feature: scans horizontally or vertically on track line selected by pilot; Alpha-	
	Vertical profile; extended STC; wx attenuation com- pensation; variable gain in map mode; wx alert; autotilt	6.0		15	±15°	5	N/A	10.28 dia.	28 VDC; 10, 26, 115 VAC, 400 Hz	numeric display of range, function and tilt angle. KMD 850 MFD, \$13,440.	
Honeywell Aerospace	Primus 660	2.5, 5, 10, 25, 50,		60° or 120°	±30°	ARINC 453/708 checklist,	4	2/15.9	\$99,531		
1944 East Sky Harbor Circle Phoenix, AZ 85034 (800) 601-3099 Fax: (602) 365-3343 www.honeywell.com	C83c	100, 200, 300	12/7.9 18/5.6	2	50 or 200 mv/deg. or ARINC 429	data nav, EFIS, MFD, LSZ-860	4	2/15.8		Single receiver/transmit- ter/antenna pedestal.	
	REACT; GMAP target alert, preset & variable gain	10	-	12/24	±15°	4	4.81 x 6.25 x 12.24	5.0 x 7.6 x 15.0	28 VDC		
	Primus 880	2.5, 5, 10, 25, 50,	12/7.9 18/5.6	60° or 120°	±30°	ARINC 453/708 checklist,	5	0/15.9	\$139,101		
	C63c	100, 200, 300		12/7.9 18/5.6	12/7.9 18/5.6 24/4	2	50 or 200 mv/deg. or ARINC 429	data nav, EFIS, MFD, LSZ-860	5	2/15.8	
	Doppler turb. detec., compensated, tilt, REACT, GMAP target alert, preset & vari- able gain	10	24/4	12/24	±15°	5	4.8 x 6.25 x 12.24	5.0 x 7.6 x 15.0	28 VDC		
	Primus 700A	¹ ⁄2, 1, 2.5,		60° or 120°	±30	ARINC 429/708 checklist,			\$110,702	Short-range and high- resolution system for	
	C63c	5, 10, 25, 50, 100, 200, 300	10/10, 12/7.9, 10 x 14/ 10 x 7.1, 18/5.6,	6	50 or 200 mv/deg. or ARINC 429	data nav, EFIS, lightning sensor LSZ-860	5	4/37	\$119,705	special search and surveillance missions, displayed menus. Mini- mum detect range at 450 ft. Allows full dual-mode	
	REACT; ground & sea clutter red.; turb. detect-preset & variable gain	10	24/4	12/24	±15°	5	4.81 x 6.25 x 12.24	5.0 x 7.6 x 15.0	28 VDC; 400 Hz	operation for pilot and copilot. Price reflects receiver/transmitter and pedestal.	
	Primus 701A	¹ / ₂ , 1, 2.5,		60° or 120°	±30°	ARINC 429/709 checklist.				Short-range and high-	
	C63c, C102	5, 10, 25, 50, 100, 200, 300	10/10, 12/7.9, 10 x 14/	6	50 or 200 mv/deg. or ARINC 429	data nav, EFIS, lightning sensor	5	4/39	\$125,254	special search and surveillance missions, displayed menus and AC 90-80A specified clear	
	REACT; ground clutter reduction; turbulence detect-preset & vari- able gain	10	18/5.6, 24/4	12/24	±15°	6	4.81 x 6.25 x 12.24	5.0 x 7.6 x 15.0	28 VDC; 115 VAC, 400 Hz	zones. Allows full dual- mode operation for pilot and copilot. Price reflects receiver/transmitter and pedestal.	

WEATHER RADAR

	Model	Ranges	Dich Cine	Scan	Stablztn.	Display	Scope	Units/	Price	
Manufacturer	TSO	Dower	& Beam	Pulse Width	Stabl. Sig.	Interface	(dia./in.)	Weight		Remarks
	Circuits	Output (Peak KW)	(in./deg.)	Looks/Min.	Ant. Tilt	Colors	Indicator Size	RT. Size	Power	
Honeywell Aerospace 1944 East Sky Harbor Circle Phoenix, AZ 85034 (800) 601-3099	IntuVue 3D weather radar RDR 4000	5 to 320 nm		Up to 90 deg/sec 1		ARINC 453		RP-1 - 10.5 TR-1 - 5.1 An- tenna+	Call dealer	IntuVue radar with volumetric buffer processing, automatic ground return elimination,
Fax: (602) 365-3343 www.honeywell.com	TSO C63c		30/3.2; 24/4.2; 18/5.6	2; uncom- 2; pressed 1 6 to 12 com- pressed	ARINC 429		N/A	Drive: 30 in. - 32.0 24 in.	28.VDC.or	automatic weather mode, altitude-based manual wx mode,
		40W Deels		(Non-Linear FM)		4		24.0 18 in. 20.8	115 VAC 400 Hz	REACT, predictive windshear, turbu-
	See remarks	40W Peak				4		RP Size: 3 MCU	on part number)	lence and optional hail and lightning prediction.
Rockwell Collins 400 Collins Rd. NE Cedar Rapids, IA 52498 (319) 295-4085 Fax: (319) 295-2297 www.rockwellcollins.com	RTA-4112	5-320 nm	-320 nm 12/7.4	±60°	stabilized to ±30 degrees in Pitch/Roll, Tilt adjust- ment is ±15° tilt	ARINC 708A	N/A 1/	1/15.1	\$180,976	
	C63c		12/7.4	3.44- 55usec	—					Price BCA estimate.
	_	38-75 W			_	4	_	12 in.	Typical 60 W; Max 80 W	
	RTS-4114	5-320 nm	14/6.7	±60°	stabilized to ±30 degrees in Pitch/Roll, Tilt adjust- ment is ±15° tilt	ARINC 708A	N/A	1/15.4	\$184,150	Price BCA estimate.
	C63c			3.44- 55usec	_					
	_	38-75 W		13		4	varies by indicator	14.0	80 W avg.	
	RTA-4118	5-320 nm		±60°	stabilized to ±30 degrees in	ARINC 708A			\$188,000	
	C63c	38-75 W	18/5.2	3.43- 55usec	Pitch/Roll, Tilt adjust- ment is ±15° tilt	_	N/A	1/17.0	Typical 60 Watts Max 80 W	Price BCA estimate
	RTA-4218	5-320 nm		±60° WX modes and ±45° in PWS	stabilized to ±30 degrees in Pitch/Roll, Tilt adjust- ment is	ARINC 708A			\$198,000	
	C63d	38-75 W	18/5.2	3.43- 55usec in WX modes and 2.11- 3.43usec in PWS	±15° tilt	4	_	18 in	Typical 60 W Max 80 W	Price BCA estimate

RADAR ALTIMETERS

Manufacture	Model	Alt. Range		Disates	Units/	Price	Barrata
Manufacturer	TS0	Pitch/Roll Limits	Accuracy	Display	Weight (lb.)	Power Required	Remarks
FreeFlight Systems 3700 Interstate 35 S. Waco, TX 76706 (254) 662-0000 Fax: (254) 662-9450	RA-4000 and RA-4500	-20-2,500 ft.	0 to 100 ft. ±3%.	RAD-40 Radar Altimeter		N/A	RA-4000 provides RS 485/422 and RS 232C outputs; RA- 4500 provides ARINC 429, RS 485/422 and RS 232 outputs. Two-year warranty. Optional night
www.freeflightsystems.com	C87	±20°/±30°	100 to 500 ft. ±3% 500 to 2,000 ft. ±5%	12,000 ft. ±3% 12,000 ft. ±5% RA-4000 and RA-4500		28 VDC	vision goggle (NVG) compatible display and round faceplate adapter for display. Optional 1/2 ATI (TSO'd) RAD-40 indicator, \$3,055; when purchased with RA-4000, \$11,190. RAD-40/RA 4500 w/installation kit, \$12,699.
	FRA-5500	-20-2,500 ft.	0 to 100 ft. ±3%.	RAD-40 Radar Altimeter		N/A	Provides compliance for FAR Part 29 operators who need to satisfy the Feb. 21, 2014 Final Rule (RIN 2120-AJ53) requiring installation of radar altimeters. Inegrates with electronic flight information systems (EFIC, flight director(s), and integrated flight decks via available BS-232 or
	C87	0 - 2,500 ft.	100 to 2,000 ft. ±5%	ible with the RA-4000 and RA-4500	1/2.37	28 VDC	ARINC 429. Options include the night-vision compatible RAD-40 panel-mounted diisplay for altitude pre-select and altimeter readout, and FTG-410 Tone Generator audio alert calls for flight crew attention to critical altitudes and other aircraft conditions.
Garmin International 1200 E. 151st St. Olathe, KS 66062 (800) 800-1020 (913) 397-8200 Fax: (913) 397-8282 www.garmin.com	GRA 55	-20 - 2,550 ft. AGL	±1.5 ft. (3 - 100 ft. AGL) ±2% (>100 - 2,500 Ft. AGL)	GI 205 and/or GIFD	1/3.5	\$6,300	All-digital design. Developed for use in hleicopters and general aviation aircraft.
	C87a Funcational Class A ETSO-2C87 Func- tional Cat- egory B/L/C (A1)/A	±20°/±30°	_	_	3.02 x 3.99 x 11.62	14-28 VDC 13.75 W max	
	GRA 5500	-20 - 2,550 ft. AGL	±1.5 ft. (3 - 100 ft. AGL) ±2% (>100 - 2,500 Ft. AGL)	GI 205 and/or GIFD	1/3.5	\$13,3000	All-digital design. Developed for helicopter, business jet, transport category and general aviation applications. Can inte- grate with Class A TAWS, TCAS II or CAT II ILS avionics.
	C87a Funcational Class A ETSO-2C87 Func- tional Cat- egory B/L/C (A1)/A	±20°/±30°	_	_	3.02 x 3.99 x 11.62	14-28 VDC 13.75 W max	

RADAR ALTIMETERS

Manufastura	Model	Alt. Range		Disates	Units/	Price	Bernarder
manuracturer	TS0	Pitch/Roll Limits	Accuracy	Display	Weight (lb.)	Power Required	Remarks
Honeywell Aerospace 1944 East Sky Harbor Circle Phoenix, AZ 85034 (800) 601-3099	AA-300 Radio Altimeter System	0-2,500 ft.				N/A	
Fax: (602) 365-3343 www.honeywell.com	RT300 C-87, RTCA DO-160A	N/A	N/A	N/A	4.56 x 4.09 x 11.07	21.32 VDC, 0.7 amp max	Pilot Activated Self Test (PAST) input available to verify system operation.
	Radar Altimeter KRA 405B	0-2,500 ft.				N/A	Internal 2,500 ft. capability for use with ground proximity sys- tems. Used with KNI-415 or KNI-
	C87/ETSO- 2C87	N/A	±2 ft (0.61 m) below 100 ft., ±3% at 100 ft. to 500 ft., and ±5% at 500 to 2,500 ft.	KNI 415 / 416 5V or 28V Black or Gray 28V Black Night Vision	3.0 x 3.5 x 11.0	27.5 VDC	416 indicators. Used with two KA- 54 or KA-54A antennas. Provides analog and ARINC 429 outputs for increased interface capability including GPWS, TCAS, autopilot. Option available with ARINC 552 auxiliary output (-0202 version) Option available that can accept DH input from ERS or KNI-415 or KNI-416 indicator to generate audio signal (-0202 version).
Rockwell Collins 400 Collins Rd. NE Cedar Rapids, IA 52498 (319) 295-4085 Fax: (319) 295-2297 www.rockwellcollins.com	ALT-1000 C87	0-2,500 ft ±40°/±50°	±2 ft or 2%	Analog only outputs*	2/6.8	\$17,988 28 VDC	*Requires separate converter for use with ARINC 429 sys- tems.
	ALT-4000 C87	0-2,500 ft. ±40°/±50°	±2 ft or 2%	EFIS (analog version available)	2/6.8	\$30,728 28 VDC	Interfaces to EFIS high-intensity monitor for Cat II/III certification. Includes two ANT-52 antennas.

THUNDERSTORM DETECTION SYSTEMS

	Model	Search Arc	Information	Display Size	Price	Remarks	
manuracturer	TS0	Max Range	Display	Units/Weight	Power Required	Remarks	
Avidyne Corp. 55 Old Bedford Rd.	TWX670	N/A	lightning strikes are displayed with	see Avidyne MHD300, EX600, EX5000, R9,	\$11,990	Third-generation lightning detection sys- tem with digital signal processing and	
(781) 402-7400 Fax: (781) 402-7599 www.avidyne.com info@avidyne.com	C110	200 nm	intensity (color). TWXCell mode highlights the most intense regions of thunderstorm activity, presenting a visually contoured color display with dynamic sectors.	IFD540, IFD440 specifications. TXW670 has 7 RS 323 ports and is compatible for monochrome strikes on many legacy RS232- capable lightning displays.	16-35 VDC	nm - 200 nm including critical 0 nm - 25 nm range for addded tactical benefit. Eliminates radial spread asso- ciated with older technology systems. Exclusive TWXCell display provides a dynamic map of the lightning discharge rate and density.	
L3 Aviation Products 5353 52nd St. SE Grand Rapids, MI 45912 (616) 949-6600 Fax: (616) 285-4224 www.L3aviationproducts.com	WX-500	pilot-selectable 120° & 360°		see remarks 2.5	\$6,656*	Remote-mount sensor interfaces with MFDs for graphical depiction of real-time lightning information Features	
	C110a	200 nm	graphical depiction of real-time lightning information in cell or strike modes	Processor: 5.6 x 2.2 12.0	11-32 VDC	ranges of 25-200 nm, input for heading stabilization and options for cell or strike mode data selection. Interfaces to MFDs via RS-422. A separate radar graphics computer (Model RGC-350) is needed for display on dedicated radar indicators. *Processor only.	
	WX-1000E (429 EFIS)	360°	_	depends on EFIS system	\$19,113*	Provides output on EFIS display or radar indicator when paired with RGC35C (sold separately); includes three levels of activity, bearing and distance; option-	
	C110a	200 nm		1/6.67	10-32 VDC	al displays for checklists. *Processor only. Price BCA estimate.	
	WX-1000E (429 Navaid)	360°		3 ATI	\$19,509	ARINC 429 interface allows simultane- ous display of thunderstorn info and course line to waypoints. Presentation	
	C110a	200 nm		2/10.95	10-32 VDC	of six user-selectable nav items. Course deviation indicator display. Consult manufacturer for approved interfaces.	

INTEGRATED FLIGHT CONTROL SYSTEMS

		Air Data	Autopilot		Weight	Price	
Manufacturer	Model	Attitude Sensors	Flight	Power	AP Only	AP Only	Remarks
Avidyne Corp. 55 Old Bedford Rd. Lincoln, MA 01773 (781) 402-7400 (800) AVIDYNE Fax: (781) 402-7599 www.avidyne.com info@avidyne.com	DFC 90	digital ADHARS from Avidyne En- tegra PFD or Aspen EFD1000 Pro	Director	28 VDC	1/2.02 NA	See remarks	Attitude-based digital autopilot interfac- es with Entegra PFD or Aspen EFD Pro. Is slide-in replacement for STEC55X, using existing servos. STEC30/50/60- 2/65 series autopilots may also be replaced by a DFC90. Currently certified in Cirrus, Beech Bonanza Series and Cessna 182 series. Price is \$9,995 for piston singles and \$14,995 for twins and turbine-powered aircraft.
	DFC 100	digital ADAHRS from Avidyne Entegra Release 9 PDF	combined	28 VDC	1/2.02 NA	\$9,995 piston singles \$14,995 twins and turbine-pow- ered aircraft	Attitude-based digital autopilot includes Straight & Level button, Envelope Protection, and full-time Envelope Alert- ing. DFC100 interfaces with Entegra Release 9 Integrated Flight Deck as a slide-in replacement for STEC55X, using existing servos. Certified in Cirrus SR20/22 and Piper Matrix & Mirage with Entegra R9. Price is \$9,995 for piston singles and \$14,995 for twins and turbine-powered aircraft.
Genesys Aerosystems One S-TEC Way, Municipal Airport Mineral Wells, TX 76067 (817) 215-7600 genesys-aerosystems.com Formerly Cobham (S-TEC)	IntelliFlight DFCS	digital ADAHARS	Magic EFIS N/A	14 or 28 VDC	14.5 N/A	See remarks	Designed for piston twins, turbine twins, and light jets. Features include Indicated air speed (IAS) hold; control wheel steering (CWS); GPS steering; heading preselect and hold PFD integration; altitude preselect and hold w/autotrim; digital vertical speed command.
	S-TEC 5000 Digital Autopilot	N/A	N/A	28	2.6 16 /01/4	NA/NA	An RVSM-compatible system offers full capabilities of a top-tier DFCS designed for high-performance jets and turbo- props. Straight and level button provides fast recovery from unusual attitudes with an annunciated alert. GPS Steering (GPSS) mode integrates the autopilot with the aircraft's GPS NAV receiver during precision approaches/missed approaches. Features include heading
	C198, ETSO- C9c, ETSO- C52b, RTCA DO-160G, RTCA DO-178B Level A	N/A	N/A 28 2.6	2.0 IU. / N/A	INA/ NA	preselect and hold; PFD integration; altitude preselect and hold w/autotrim; digital vertical speed command; course intercept capability; dual mode - HDG/ NAV and HDG/APR, VOR/LOC/GS/REV/ GPS course; NAV flag warnings; control wheel steering; GPS steering (GPSS); envelope protection and alerting; autopi- lot mode annunciations; voice annuncia- tions; all-axis trim control and more.	
Garmin International 1200 E. 151st St. Olathe, KS 66062		GDC 74 (B) DADC	combined		varies by installation	varies by installation	Digital, dual-channel fail-passive system for Cessna Mustang, Caravan, C-172, -182, -206, -350, -400, CJ525, C680 and C250, Cirrue 5020 and 5020
(800) 800-1020 (913) 397-8200 Fax: (913) 397-8282 www.garmin.com	GFC 700	GRS 77, GRS 7800		28 VDC			Diamond DA40 and DA42; Embraer Phenom 100 and 300; HBC G36 and G58; Learjet 40/45 and 70/75; Mooney M20R and M20S; Piper Seminole, Seneca, Matrix, Mirage and Meridian; Socata TBM 850; HondaJet.
Honeywell Aerospace		micro DADC	IC-600		varies by installation	varies by version	Digital fail-passive system. CAT II-
1944 East Sky Harbor Circle Phoenix, AZ 85034 (800) 601-3099 Fax: (602) 365-3343 www.honeywell.com	Primus 1000 (in remarks)	digital AHRS or IRS	combined	28 VDC	varies by installation	varies by version	capable; ARINC 429 interfaces, two-, three-, four- or five-tube, 8 in. x 7 in. EFIS. Bombardier Learjet 40, 45 and 45XR; Embraer ERJ-135, 140 and 145; Cessna Bravo, Encore, Excel and Ultra.

INTEGRATED FLIGHT CONTROL SYSTEMS

		Air Data	Autopilot		Weight	Price	
Manufacturer	Model	Attitude Sensors	Flight Director	Power	AP Only	AP Only	Remarks
Honeywell Aerospace 1944 East Sky		micro DADC	IC-600		varies by installation	varies by version	Digital fail-passive system. CAT II- canable: ARINC 429 interfaces, two-
Harbor Circle Phoenix, AZ 85034 (800) 601-3099 Fax: (602) 365-3343 www.honeywell.com	Primus 1000 (in remarks)	digital AHRS or IRS	combined	28 VDC	varies by installation	varies by version	three, four or five-tube, 8 in. x 7 in. EFIS. Bombardier Learjet 40, 45 and 45XR; Embraer ERJ-135, 140 and 145; Cessna Bravo, Encore, Excel and Ultra.
	Primus 1000	micro DADC	IC-615	28	varies by installation	varies by installation	Digital fail-passive system. CAT Il-capable; ARINC 429 interfaces;
	CDS	digital AHARS	combined	VDC	varies by installation	varies by installation	two- to five-tube 10 in. x 8 in. LCD EFIS. Cessna Citation XLS.
	Primus 2000	micro DADC	IC-800	28	varies by installation	varies by version	Digital, dual-channel fail-passive system. CAT II-capable w/optional auto-throttle, dual-sensor monitoring; five- or six-tube
		digital AHRS or IRS	combined	VDC	varies by installation	varies by version	8 in. x 7 in. CRT EFIS. Global Express and Global 5000; Cessna Citation X; Dassault Falcon 900EX/C.
	Primus Epic	micro DADC	FZ-800	28	varies by installation	varies by installation	Digital fail-passive system. CAT II- capable, ARINC 429 interfaces. Two-,
	CDS	digital AHRS or IRS	combined	VDC	varies by installation	varies by installation	three-, four- or five-tube 10 in. x 8 in. EFIS. SyberJet SJ30-2.
	Primus Epic	air data module and micro IRS	integrated modular avionics unit	28	varies by installation	varies by installation	Integrates all traditional avionics into modular avionics unit. Digital, dual- channel; fail operational system. CAT II-capable w/optional auto-throttle and envelope protection. Includes two- to
		air data module and micro IRS		VDC	varies by installation	varies by installation	Ne-tube 10 in. X 8 in. LCD EFIS or four 13 in. LCDs. Agusta/Bell AB139; Cita- tion Sovereign; Dassault Falcon 900EX, 2000EX and 7X; Embraer 170, 175, 190 and 195; Gulfstream G350, G450, G500 and G550; Hawker 4000.
Rockwell Collins 400 Collins Rd. NE	APS 4000	ADC-3000/3010 AHC-3000/4000	integrated	28	varies by installation	varies by installation	Available only as part of integrated Pro Line 21 system. Built-in diagnostics,
Cedar Rapids, IA 52498 (319) 295-4085	AI 0 4000	_	_	VDC	varies by installation	see remarks	dual channel, fail-passive, digital CAT-II certificated autopilot and flight director.
www.iockweitcomis.com		AHC-3000	APS-65	28	varies by installation	varies by installation	Built-in diagnostics; digital Cat II certificated autopilot. Optional EFIS
	APS-65	remote vertical gyro or dual AHRS	EFIS-84 (two tube)	VDC	50.6	*	and AHRS. STC kit installer fabricated. Compatible with EFIS-84. *Typical con- figuration, \$155,976.
		ADS-86	APS-85	20	varies by installation	varies by installation	Available only as full, dual-channel, fail-passive, digital system; digital Cat
	APS-85	dual AHRS AHC-3000	EFIS-85 (three tube)	VDC, 115 VAC, 400 Hz	varies by installation	*	II autopilot, 4- or 5-tube EFIS optional; ARINC 429 IRS interface available; includes yaw damper; extensive built-in diagnostics. STC kit installer fabricated. Compatible with EFIS-84. *Typical con- figuration, \$290,388.

Manufacturor	Model Processor Size		Prico	Remarks		
Manuracturer	TSO		Weight (lb.)	FILE	Reinai KS	
ACSS an L3 & Thales Company 19810 N. 7th Ave.	TAWS+	MFD, EFIS,	2 MCU	\$149.593	Terrain Advisory Line and Avoid Terrain features. With GPS version alerts based on	
Phoenix, AZ 85023 (623) 445-7000 Fax: (623) 445-7001	C151B Class A, C129b2	weather wadar wisplay	7.5		aircraft climb capability.	
www.acss.com	TCAS 2000 RT-950/951	MFD, EFIS,	4 MCU - 14.7 6 MCU - 15.8		Change 7.1 compliant. Standard positions on many regional and business jets. Bombardier, Cessna, Dassault Embraer, Gulfstream	
	C119b	VSI/TCAS display	_	\$245,367	Hawker Beechcraft. (SFE selectable on all Airbus and Boeing aircraft.)	
	TCAS 3000SP	MFD, EFIS,	4 MCU - 13.85 6 MCU - 16.08	\$254.271	Change 7.1 compliant. Flexible to add certified	
	C119b	weather radar display	_		ADS-B in applications combined with ICAS.	
	T ² CAS	MFD, EFIS, weather radar display.	6 MCU - 15.8	\$343.058	Combined TCAS and TAWS in one box. Change 7.1 compliant. ADS-B IN/Out capable. Certi-	
	C119b, C151b Class A, C129b2	VSI/TCAS display	_		fied on Airbus A320 family.	
Avidyne Corp. 55 Old Bedford Rd. Lincoln, MA 01773 (800) 284-3963	TAS600		7.25 x 11.67 x 3.10		Detects and interrogates other aircraft transponders within range, displaying the surrounding traffic on a host of compatible display systems and provides audible and visual alerts in the event of a potential traffic	
Fax: (614) 885-8307 www.avidyne.com	C-147	MFD, EFIS, weather displays, GPS map displays	8.71 (includes proces- sor, dual antennas and coupler)	\$9,749	conflict. Provides 30-second decision time at a closure rate of up to 1,200 kt. Head-Up Audible Position Alerting verbally indicates the conflicting aircraft's bearing, range and relative altitude for rapid visual acquisition of traffic. Includes Patented directional top and bottom antennas. Recommended for entry- level, single-engine piston aircraft. Features a 7-nm range, 3,500 ft. vertical separation maximum and 18,500-ft. service ceiling.	
	TAS605A		7.25 x 11.67 x 3.10		Recommended for mid-performance aircraft and helicopters. Features 13-nm range, 5,500-ft. vertical separation maximum and a	
	C-147	MFD, EFIS, weather displays, GPS map displays	8.71 (includes proces- sor, dual antennas and coupler)	\$10,999	55,000-ft. service ceiling. Accepts ARINC 429 heading input, permitting rapid respositioning of targets during high-rate turns. VeriTAS correlates active-surveillance targets along with 1090 MHz ADS-B IN targets and provides ADS-B collision avoidance logic.	
	TAS615A		7.25 x 11.67 x 3.10		Recommended for high-performance aircraft and helicopters, the TAS615 features 17-nm range, 10,000-ft. vertical separation maxi- mum and 55 000-ft service ceiling Accents	
	C-147	MFD, EFIS, weather displays, GPS map displays	8.71 (includes processor, dual antennas and coupler)	\$14,990	ARINC 429 heading, permitting rapid reposi- tions of targets during high-rate turns. VeriTAS correlates active-surveillance targets along with 1090 MHz ADS-B IN targets and provides ADS-B collision avoidance logic.	
	TAS620A	MFD, EFIS, weather displays,	7.25 x 11.67 x 3.10	\$20,990	Features 21-nm range, a 10,000-ft. vertical separation maximum and a 55,000-ft. service ceiling. Accepts ARINC 429 heading inut, permitting rapid repositioning of targets during high-rate turns.VeriTAS correlates active-surveillance targets along with 1090 MHz ADS-B IN targets and provides ADS-B collision avoidance logic.	
	C-147	uro map uispiays	8.71 (includes processor, dual antennas and coupler)			

Manufacturer	Model	Display Interface Ontions	Processor Size	Prico	Pemarks
manuracturer	TSO	Display interface options	Weight (lb.)	THCC	itelliains
Garmin 1200 E. 151st. St. Olathe, KS 66062	TAWS-B	GNS 400(W) series, 500(W) series GTN 600 series,	_	varies with	
(800)800-1020 (800)357-8200 Fax: (913) 397-8282 www.garmin.com	C151 ETSO-C151	GTN 700 series G600, G900X, G950, G1000, G1000 NXi G2000, G3000, G5000	_	Instanation	
	TAWS-A		—		
	C151 ETSO-C151	GTN 600/700 series, G900X, G950, G1000, G2000, G3000, G5000	-	varies with installation	
	HTAWS		N/A		Available as an option on GTN series
	C194 ETSO-C194	GNS 400 (W) series, GNS 500 (W) series, GTN 600/700 series, G1000H, G5000H	N/A	varies with installation	touchscreen avionics, as well as legacy GNS 430W/530W navigators, HTAWS (Helicopter Terrain Awareness and Warning System) offers "forward looking" terrain and obstacle avoidance (FLTA) capability to alert in advance where potential hazards may exist.
	GTS 800	GNS 400(W) series,	2.66 x 6.25 x 14.78		TAS traffic surveillance system able to track up to 45 targets up to a 22-nm interrogation range
	C147 Class A ETSO C147 Class A C166b	GTN 600 series, GTN 700 series, GTN 700 series, GNS 480, GMX 200 G500, G600 G900X, G950, G1000(H), G1000 NXi G2000, G3000, G5000(H) Third-party controller and display	1/8.92	\$9,995	
	GTS 825	GNS 400(W) series, 500(W)	6.2 x 3.0 x 12.1		
	C147 Class A ETSO C147 Class A C166b ETSO C166b	GTN 600 series, GTN 700 series GNS 480, GMX 200 G500, G600 G900X, G950, G1000(H), G1000 NXi G2000, G3000, G5000(H) Third-party controller and display	1/11.3	\$19,995	Affordable TAS Traffic survelliance system able to track up to 75 targets up to a 40-nm interrogation range.
	GTS 855	GNS 400(W) series, 500(W)	3.42 x 6.25 x 14.78		
-	C118 ETSO C118 C166b ETSO 166b	series GTN 600 series, GTN 700 series GNS 480, GMX 200 G500, G600 G900X, G950, G1000(H), G1000 NXi G2000, G3000, G5000(H) Third-party controller and display	1/11.3	\$\$24,995	TCAS I collision avoidance system able to track up to 75 targets within an 80-nm for- ward interrogation range
	GTS 8000	GNS 400(W) series	3.42 x 6.25 x 14.78		TCAS II Change 7.1 system, includes GTS
	C119c ETSO C119c C116b ETSO C166b	500(W) series GTN 600 series, GTN 700 series G900X, G950, G1000(H), G1000 NXi G2000, G3000, G5000(H) Third-party controller and display	1/11.3	\$89,995	8000 TCAS processor and two GTX 3000 TCAS transponder.

Manufacturer	Model	Display Interface Ontions	Processor Size	Price	Remarks	
Manufacturer	TSO	Display interface options	Weight (lb.)	THUE		
Honeywell Aerospace BendixKing Avionics	BendixKing KGP 560	KMD 550 MFD,	2.2 x 4.15 x 6.25	\$12 865	EGPWS exceeds Class B requirements. Pro-	
9201 San Mateo Blvd. NE Albuquerque, NM 87113	C151 Class B	and most MFDs	1.5	ΨI2,803	worldwide database by region.	
(855) 250-7027 www.bendixking.com	BendixKing KGP 860	KMD 550 MFD,	2.2 x 4.15 x 6.25	\$15,615	EGPWS exceeds Class B requirements. Pro- vides aural and visual warnings; internal GPS;	
	C151 Class B	and most MFDs	1.5	\$13,013	worldwide database by region. EFIS displays additional warning modes.	
	Mark XXI	KMD 550 MFD,	4.5 x 7.0 x N/A	\$10.011	Heliconter ECDWS	
	C118 Class B	and most MFDs	1.5	\$19,011		
	BendixKing KTA 870	KMD 550 MFD,	4.5 x 7.0 x 13.8	\$27.092	Traffic Advisory System (TAS) is an active	
	C147	and most MFDs	8.75	Ψ21,982	Single or dual directional antennas.	
	KTA 970	dual-color, flat-panel LCD combined IVSI/TA display, KMD 550	4.5 x 7.0 x 13.8	\$36.767	TCAS L system.	
	C118	EFIS, KMD 850 or weather radar	8.75	\$00,101		
	BendixKing KMH 880	KMD 550 MFD, KMD 850 MFD	4.5 x 7.0 x 13.8	¢42,720	Traffic Advisory System (TAS) and EGPWS in one box. Active traffic system providing aural and visual adviories. Single or dual directional antennas.	
	C147, C151, Class B	and most MFDs	8.75			
	BendixKing KMH 980	KMD 550 MFD, KMD 850 MFD	4.5 x 7.0 x 13.8	\$56.723	TCAS I and GA-EGPWS.	
	C118, C151 Class B	and most MFDs	9.68	+00,120		
	CAS 66A System	dual-color, flat-panel LCD combined IVSI/TA/RA	1/2 ATR-S (4 MCU)	\$136,934	TCAS I system. Includes processor, control panel, directional antenna and IVSI/TA	
	C118	EFIS or weather radar	17.0		display. Does not include installation kits. Upgradable to TCAS II.	
	CAS 67A System	CAS-67A systems includes one TPU-67A ; TCAS An-	1/2 ATR-S (4 MCU)	\$231 799		
	C118	tenna; Mode S Transponder; TA/RA/VSI IVA 81D	NA	+201,100		
	CAS 67B System	CAS-67A systems includes one TPU-67A ; TCAS An- tenna; Mode S Transponder;	1/2 ATR-S (4 MCU)	\$225,203		
		TA/RA/VSI IVA 81D				
	CAS-100 System C119c	Dual-color, flat-panel LCD combined IVSI/TA/RA dis- play (included in price show).	1/2 ATR (4 MCU)	\$219 179*	CAS-100 system includes one TPA-100B with Change 7.1; one ANT-81A; one IVA-81D VSI	
	C119c	Also can interface with KMD 550 MFD, KMD 850 MFD, EFIS or weather radar	1/13.5	¥210,110	display; one CTA-100A control panel. *BCA estimate.	
	EGPWS MK V-A	EFIS, MFD and radar indicators	7.9 x 2.4 x 12.8	\$115,858 (without inter-	MK V-A is for turbofan aircraft equipped with	
			1/6.5	nal GPS)		
	EGPWS MK XXI	See remarks	3.95 x 2.20 x 3.25		Helicopter EGPWS enhanced features: detailed terrain database, obstacle database, airports and heliports, look-ahead algorithms, terrain alerting, obstacle alerting, en route	
					terrain display (peaks), pop-up feature, auto ranging feature, geometric altitude, enhanced envelope modulation, speed expansion, internal GPS card.	

Monufacturor	Model	lodel Processor Size		Drico	Domosko	
Manufacturer	TSO	Display interface Options	Weight (lb.)	Price	Reliairs	
Honeywell Aerospace BendixKing Avionics	KGX 150T ADS-B UAT Transceiver	Mode A/C and Mode S	5 x 5.75 x 1.7		ADS-B receiver and UAT transmitter with optional Wi-Fi, best optimized for the those	
9201 San Mateo Bivd. NE Albuquerque, NM 87113 (855) 250-7027 www.bendixking.com	TSO-C157A (FISB) TSO-C195A (TIS-B) TSO-C154C (UAT) TSO-C154C (for GNSS) DO-160G DO-178B level C DO-254 Level C STC Approved in accordance with AC20- 165A	ARINC 429; 1 RS 485 and 4 discrete inputs 1 ARINC 429; 4 RS 232/422 and 2 discrete outputs 10-40 VDC input voltage .02 A @ 12 VDC Input current 6.5 VDC output voltage 350 mA output		\$2,849	who hy below 12,000 ft. Also includes an integrated ADS-B OUT Compliant WAAS GPS. The KGX 150T provides the ADS-B traffic and weather services to non-certified wireless tablet or certified compatible panel display. An optional control head is available for additional ADS-B required information and annunciations.	
	KGX 150R ADS-B UAT Receiver with Integrated WAAS	Mode A/C and Mode S transponder interface 2	5 x 5.75 x 1.7		ADS-B receiver with optional Wi-Fi, best op- timized for those who fly above and below 18,000 ft, or want to replace their existing	
	TSO-C157A (FISB) TSO-C157A (FISB) TSO-C154C (UAT) TSO-C145C (for GNSS) DO-160G DO-178B level C DO-254 Level C STC Approved in accordance with AC20- 165A			\$2,648	transponder with the KT 74 1090 extended squitter transponder. Also includes an integrated ADS-BOUT Compliant WAAS GPS. The KGX 150R provides the ADS-B traffic and weather services to non-certified wireless tablet or certified compatible panel display. No external controller is needed.	
	KGX 130R ADS-B UAT Receiver	Mode A/C and Mode S transponder interface 2	5 x 5.75 x 1.7		ADS-B receiver with optional Wi-Fi, best optimized for those who fly above and below 18,000 ft. and want to replace their existing transponder with the KT 74 1090 extended squitter transponder. The KGX 130R uses your existing WAAS Garmin GNS 430W/530W GPS and provides the ADS-B traffic and weather services to non-certified wireless tablet or certified compatible panel display.	
	TSO-C157A (FISB) TSO-C195A (TIS-B) TSO-C154C (UAT) DO-160G DO-178B level C DO-254 Level C STC Approved in accordance with AC20- 165A	ARING 429; 1 R5 485 and 4 discrete inputs 1 ARING 429; 4 RS 232/422 and 2 discrete outputs 10-40 VDC input voltage .02 A @ 12 VDC input current 6.5 VDC output voltage 350 mA output		\$1,699		
L3 Aviation Products 5353 52nd St. S.W.	LandMark TAWS 8000	TAWS compatible Arinc 453 EFIS, Arinc 453 weather	7.0 x 2.25 x 9.0		Remote processor that offers predictive warn- ing functions using position data from a GPS	
Grand Rapids, MI 49512 (616) 949-6600 Fax: (616) 285-4224 www.L3aviationprod- ucts.com	C151a Class B	ible MFDs. Display on non- Arinc 453 radar indicators requires the RGC 350 Radar Graphics Computer (sold separately)	3.35	\$14,120	terrain and obstacle database. Both aural and visual warnings are issued whenever CFIT situations arise. LandMark is designed to meet or exceed Class B requirements of TSO C151a. Baro-corrected altitude input required.	
	LandMark TAWS 8100	AWS compatible Arinc 453 EFIS, Arinc 453 weather radar indicators and compat- ible MFDs. Display on non- Arinc 453 radar indicators	7.0 x 2.25 x 9.0	\$15,230*	Features a WAAS GPS Sensor. With its ac- curate positioning information, the LandMark 8100 eliminates the need for multiple inputs from other aircraft sensors, simplifying the installation. The 8100 provides the bidbert	
	C151b Class B	Graphics Computer (sold separately)	3.40		integrity terrain data without complicated GPS, ADC or OAT inputs. 320 nm range. * <i>BCA</i> estimate.	

84 6	Model	Disular Interface Ontinue	Processor Size	Duine	Demonto	
Manutacturer	TSO	Display interface Options	Weight (lb.)	Price	Remarks	
L3 Aviation Products 5353 52nd St. SW Crond Papida, MI 40512	Lynx NGT-9000+		6.25 x 1.8 x 10.75		Panel-mounted touchscreen transponder that also displays traffic information onto compat- ible flight displays and iPad and Android apps. Can be configured to view ADS-B and active traffic on the same screen without the need for additional boxes. Aural traffic alerting is an available option.	
Grand Rapids, MI 49512 (616) 949-6600 Fax: (616) 285-4224 www.L3aviataionprod- ucts.com		see remarks	5.2	\$9,555		
Rockwell Collins 400 Collins Rd. NE	TCAS 4000	Collins FEIS MED	4 MCU		TCAS II system. European ACAS compatible Mode S Level III. AC/DC in one part number	
Cedar Rapids, IA 52498 (319) 295-1000 Fax: (319) 295-2297 www.rockwellcollins. com	pids, IA 5-1000) 295-2297 kwellcollins. Collins ErIS, MFD TCAS compatible VSI (RA) Collins TVI-920 (RA, TA) 17.0	\$422,064* (typical installation)	Includes control panel and two TRE antennas. Displays range/alt. separation from traffic. Max range 3 mi. Two surveillance volumes and MSL of traffic. Top/bottom antennas to optimize coverage. Upgrades to 8800 Gold. *BCA estimate.			
Sandel Avionics 2401 Dogwood Way Vista, CA 92081 (877) 726-3357 ((760) 727-4900	ST3400H HeliTAWS	Integrated rear projection	3 ATI panel-mount	\$18,950	 3-ATI helicopter TAWS with integrated display. Can replace existing radar altimeter indicator. Sunlight readable LED backlit display with 180 deg. viewing angle and over 10,000-hr. MTBF. NVIS compatible version \$22,200. 3-ATI Class A or Class B TAWS with integrated display. Sunlight readable LED backlit display with 180-deg. viewing angle and over 10,000 hr. MTBF. Optional interface for traffic, \$980. Class A version, \$38,600. 	
	C87, C113, C151b, C194	LCD with LED backlighting	2.9			
Fax: (760) 727-4899 www.sandel.com	ST3400 TAWS	Integrated rear	3 ATI panel-mount	\$04.050		
	C113, C151b	backlighting	2.9	\$24,230		
Universal Avionics Systems Corp.	TAWS A TAWS B	Universal Avionics EFI-890R, MFD-640, UNS FMS (5-in. display)	2 MCU LRU			
3260 E. Universal Way Tucson, AZ 85756 (520) 295-2300 (800) 321-5253 Fax: (520) 295-2395 www.uasc.com	C151b, C92c	Honeywell numerous weather radar, MFD and EFIS displays <i>Rockwell Collins</i> numerous weather radar, MFD and EFIS displays <i>Smiths</i> BAE ATP EFIS additional display options available	9.6	TAWS A \$40,700 TAWS B \$26,200	Worldwide terrain database with 480+ MB data. High-resolution analog video views; 3-D perspective view; profile view; map view. Map view of terrain can be output using ARINC 708 or WXPF formats for interface with various existing weather radars. Both version include obstacle database.	

COCKPIT VOICE RECORDERS (CVR)/FLIGHT DATA RECORDERS (FDR)

	Type Recording Medi		Size	Price		
Manufacturer	Model				Remarks	
	TS0	Duration	Weight (lb.)	Power Required		
Honeywell Aerospace	Business Aviation	solid-state	7.45 x 5.92 x 4.0	N/A		
Phoenix, AZ 85034	LW-CVR (429)				A fully compliant recorder developed for business aviation. *BCA estimate.	
Fax: (602) 365-3343	980-6044-002	120 min.	<7.0 with AR for factor mounting adapter	28 VDC		
·	Air Tranport	solid-state	¹ ⁄ ₂ ATR Short	N/A		
	HFR5-CVR	120 min	86	28 VDC	A fully compliant recorder developed air business aviation.	
	980-6032-003	120 mm.	8.0	115 AC		
	Business Aviation	solid-state	7.45 x 5,92 x 4.0	N/A	A fully compliant recorder developed air	
	LW-FDR (717)		<7.0 with AR form	28 VDC	business aviation.	
	980-4131-002	25 hr. @ 512 wps	foactor mounting adapter			
	Air Transport	solid-state	¹ ⁄ ₂ ATR Long or Short	N/A	A fully compliant recorder developed air	
	HFR5-FDR			445.140	business aviation.	
	1/2 ATR Long only	25 hr. @ 1,024 wps	10.0	28 VDC		
	Business Aviation	solid-state	7.45 x 5. 92 x 4.0	N/A		
	LWCVR/FDR				A fully compliant recorder developed air business aviation.	
	980-6050-042 (429 input) 980-6050-072 (717 imput)	120 min. CVR 25 hr. @512 wps FDR	<7.0 with AR form fac- tor mounting adapter	28 VDC		
	—	colid stato, digital	00	N/A		
	CVR AR 120 CVR	solid state, digital	0.0	IVA	Non-ARINC size with underwater locator	
	980-6023-002 ED 56A, C123a	120-min. CVR	9.5 x 5.88 x 5.75	28 VDC	beacon; control panel and mounting tray not required. ARINC 557 and ARINC 757.	
	_					
	AR FDR	solid-state, digital	8.8	N/A	Non-ARINC FDR, ARINC 717, 429. Mount-	
	980-4710-00X ED 55, C124e	25 hr. @ 64, 128, 256 wps	9.5 x 5.88 x 5.75	28 VDC	ing tray not required.	

COCKPIT VOICE RECORDERS (CVR)/FLIGHT DATA RECORDERS (FDR)

	Туре	Recording Medium Size		Price		
Manufacturer	Model	Duration	Weight (lb.)	Power Required	Remarks	
Honeywell Aerospace	CVR			N//A		
1944 East Sky Harbor Circle	AB 120 CVB	solid-state, digital	9.5 x 5.88 x 5.75	N/A	Non-ARINC size with underwater locator	
(800) 601-3099	980-6023-002	100 min			beacon; control panel and mounting tray not required. ARINC 557 and ARINC 757.	
Fax: (602) 365-3343	ED 56A, C123a	120 min.	8.8	28 VDC		
	CVR	solid-state, digital	¹ ⁄ ₂ ATR Short	N/A		
	SSCVR 980-6022-011	120 min	11.5	28 VDC	Solid-state CVR with underwater locator beacon. ARINC 557 and ARINC 757.	
	ED 56A, C123	120 mm.	11.5	115 AC		
	DVDR/FDR	solid-state, digital	¹ ⁄ ₂ ATR Short	N/A		
	AR Combi				Combination CVR/FDR; ARINC Form Factor. Mounting tray not required. Data	
	50-6021-06A	120-min. voice; 25-hr. data @ 64,	11.5	28 VDC	download through front access PCMCIA.	
	ED 56A, C123a	128, 256 wps				
100 Cattlemen Rd.	CVR/FDR	solid-state	1/2 ATR Short CVR; 1/2 ATR Short or Long FDR	\$32,719, CVR \$39,261 FDR	Includes underwater locator beacon, mounting tray required. ARINC 757 con-	
Sarasota, FL 34232 (941) 371-0811	FA2100	2-hr. min. CVR; 25-hr. min. FDR	CVR/FDR Short: 12.6 x 5.0 x 5.5		nector CVR, ARINC 747 connector FDR, GMT or FSK time-signaling source for	
www.L3aviationproducts.com	C123b, C124b, EUROCAE ED-112		10.0 10.0	115 VAC 400 Hz or 28 VDC	CVR. Separate RIPS module available for CVR, rotor-speed input for CVR for helicopter applications; CPDLC data link recording for CVR; minimum 25-hr. 64 wps up to 1024 wps recording rate for FDR; ramp (portable) and shop (bench) GSE hardward and software diagnostics and readout tools optional.	
	CVDR	solid-state	¹ ⁄ ₂ ATR Short 12.6 x 5.0 x 5.5	\$54,575	Includes underwater locator beacon, mounting tray required. ARINC 757 con- nector, GMT or FSK time-signaling source	
	FA2100		10.0	115 VAC 400 Hz or 28 VDC		
	C123b, C124b, EUROCAE ED-112	2-hr. min. CVR; 25-hr. min. FDR			for CVR. separate kirs inductie available for CVR, rotor-speed input for CVR for helicopter applications; CPDLC data link recording for CVR, OMS output for CVR, minimum 2-hr. 4-channel high-quality audio recording for CVR, minimum 25-hr., 128 wps up to 1024 wps recording rate for FDR; rap (portable) and shop (bench) GSE hardware and software diagnostics and readout tools available.	
	CVR/FDR	solid-state	¹ ⁄ ₂ ATR Short 12.6 x 4.8 x 6.5	\$50,703	Includes underwater locator beacon,	
	CVDR Model FA5000				nector, GMT or FSK time-signaling source	
	C123b — CVR C124b — FDR EUROCAE ED-112 — CVR and FDR	2-hr. min. CVR; 25-hr. min. FDR	7.9	115 VAC 400 Hz or 28 VDC	for CVR, rotor-speed input for CVR for helicopter applications; CPDLC data link recording for CVR, OMS output for CVR, minimum 2-hr. 4-channel high-quality audio recording for CVR, minimum 25-hr., 128 wps up to 1024 wps recording rate for FDR; rap (portable) and shop (bench) GSE hardware and software diagnostics and readout tools available. Ethernet data output.	

COCKPIT VOICE RECORDERS (CVR)/FLIGHT DATA RECORDERS (FDR)

	Туре	Recording Medium	Size	Price		
Manufacturer	Model	Duration	Weight (lb.)	Power Required	Remarks	
10 Aviation Decembers	150					
100 Cattlemen Rd.	CVDR/SRVIVR	solid-state	6.55 x 5.55 x 3.25	\$40,675		
Sarasota, FL 34232	_				Same as FA5000	
www.L3aviationproducts.com	C123b, C124b EUROCAE Ed-112	2 hr. CVR 25 hr. FDR	6.75	28 VDC		
	Lightweight Data Recorder	solid-state	8.0 x 3.9 x 4.9	\$21,370	No mounting tray required; 2-hr. 2-channel	
	LDR	2 br CVP			ing; 5-hr. ARINC 717 data recording; 2-hr. analog video recording at 5 fps. Ethernet	
	C197, EUROCAE ED-155	25 hr. FDR 2-hr. video	5.0	28 VDC	data output.	
	Micro Quick Access Recorder	Minimum 2 GB compact flash memory	2.7 x 2.2 x 1.8	\$8,513	ARINC 573/717/747 compatible; data rates 64 wps up to 1024 wps; USB or	
	_			115 VAC	Ethernet data output. Fixed or removable flash card media. Data download software	
	_	—	6 oz.	400 Hz or 28 VDC	utility optional.	
Universal Avionics Systems Corp. 3260 E. Universal Way Tucson, AZ 85756 (520) 295-2300	Combi CVR/FDR	solid-state flash memory	6.0 x 4.9 x 8.0	\$19,500	No internal batteries. No periodic main- tenance. Four channels of cockpit audio data, UTC from ARINC 429 bus, UTC fron a Frequency Shift Keying (FSK) simpling	
Fax: (520) 295-2395 www.uasc.com	CVFDR-145 C123b, C124b, C177, C123a, C124a, EUROCAE ED-112	120-min. voice & ambient audio +25 hr. (min.) Flight data +120 minute data link messaging	7.0	28 VDC	anlog/digital sensor signals via FDAU, ARINC 717 Flight Data Recording, analog/digital sensor signals via FDAU, ARINC 758 data link information. PC- based ramp testing/diagonstics.	
	Combi CVR/FDR w/ embedded Re- dorded Independedt Power Supply (RIPS)	solid-state flash memory	6.0 x 4.9 x 8.0	\$27,500	Embedded RIPS. Solid state memory. No internal batteries. No periodic mainte- nance. Four channels of cockpit audio data, UTC from ARINC 429 bus, UTC from a Frequency. Shift Keying (FSK) signaling	
	C123b, C124b, C123b, C124b, C155, C177, C123a, C124a, EUROCAE ED-112	45R 120-min. voice & ambient audio +25 hr. (min.) 8.68 28 VD0 124b, C123a, COCAE Flight data +120 minute data link 8.68 28 VD0 2 messaging 100 minute data link 100 minute data link		28 VDC	source, Rotor Speed for helicopter appli- cation. ARINC 717 Flight Data Recording, analog/digital sensor signals via FDAU, ARINC 758 data link information. PC- based ramp testing/diagonstics.	
	CVR	solid-state flash memory	6.0 x 4.9 x 8.0	\$16,500	No internal batteries. No periodic main- tenance. Four channels of cockpit audio data, UTC from ARINC 429 bus. UTC from	
	CVR-120A C123b, C177, C123a, EUROCAE ED-112	120-min. voice & ambient audio	7.9	28 VDC	a Frequency Shift Keying (FSK) signaling source, rotor Speed for helicopter applica- tions, ARINC 758 data link information. PC-based ramp testing/diagonstics.	

COCKPIT VOICE RECORDERS (CVR)/FLIGHT DATA RECORDERS (FDR)

	Туре	Recording Medium	ecording Medium Size			
Manufacturer	Model				Remarks	
	TS0	Duration	Weight (lb.)	Power Required		
Universal Avionics Systems Corp. 3260 E. Universal Way Tucson, AZ 85756 (520) 295-2300	CVR w/embedded Recorded Indepen- dent Power Supply (RIPS)	solid-state flash memory	6.0 x 4.9 x 8.0	\$24,500	Embedded RIPS. Solid-state memory. No internal batteries. No periodic mainte- nance. Four channels of cockpit audio	
Fax: (520) 295-2395	01/0 4000				data, UTC from ARINC 429 bus, UTC from a Frequency Shift Keying (FSK) signaling source, rotor speed for helicopter applica- tions. ARINC 758 data link information. PC-based ramp testing/diagonistics. No internal batteries. No periodic mainte- nance. ARINC 717 Flight Data Recording. Additional data storage beyond 25 hr., analog/digital sensor signals via FDAU.	
www.uasc.com	CVR-120R		8.68	28 VDC		
	C123b, C155, C177, C123a, EUROCAE ED-112	120 min. voice & ambient audio				
	FDR	solid-state flash memory	6.0 x 4.9 x 8.0	\$16,500		
	500 05					
	FDR-25	25 hr. (min) Flight				
	C124b, C124a, EUROCAE ED-112	data + 120 min. data link messaging	7.9	28 VDC	PC-based ramp testing/diagonstics.	

HEAD-UP DISPLAYS

Manufacturer	Model	Units/Weight (I		Price	Remarks	
	INIOUCI		Size or Form Factor	Power Required		
Elbit Systems of America- Fort Worth Operations 4700 Marine Creek Pkwy. Fort Worth TX 76179	Advanced Technology HUD	ARINC 429, ARINC 615 descrets, Enchaced Vision	3/35.0	\$356,000*	Fully digital EFVS video ready LCD HUD that is compact and	
www.elbitsystems-us.com	(AT-HŬD)	(EVS) Video, Synthetic Vision (SVS) video	14.0 x 6.0 x 5.0	28 VDC	*Contact manufacturer for specific pricing.	
Rockwell Collins (Head Up Guidance Systems)		ARINC 429 various	48.0 - 55.0	\$409,405*	Provides Cat III landing and Low	
400 Collins Rd. NE Cedar Rapids, IA 52498 www.rockwellcollins.com	HGS-4000	discretes, enhanced vision, synthetic vision	3 LRUs	N/A	Visibility Takeoff capability. *BCA estimate	
		ARINC 429, various	varies by configuration/less than 15 lb.		Compact wave-guide Head-Up Dis-	
	HG5-3500	discretes, ennanced vision, synthetic vision	3 LRUs	Price not provided	play developed for light to midsize business aircraft applications.	
		ARINC 429, various	48.0 - 53.0		First-generation digital Head-Up	
	HGS-5000	discretes, enhanced vision, synthetic vision	3 LRUs	Price not provided	commercial and business aircraft platforms.	
		ARINC 429, various	40.0 - 46.0		Second-generation digital Head-Up	
	HGS-6000	discretes, enhanced vision, synthetic vision	3 LRUs	Price not provided	commercial and business aircraft platforms.	

AIRCRAFT SITUATION DISPLAYS

	Model	Display			Units/Weight (lb.)	Price					
Manufacturer	TSO	Display Size	Inputs	Outputs	Size or Form Factor	Power Required	Remarks				
Aspen Avionics 5001 Indian School Rd. NE Albuquerque, NM 87110 (505) 856-5034 Fax: (505) 314-5440	1000 MFD	TFT AMLCD (400 x 760)	ARINC 429 (5) RS-232 (5)	ARINC 429	display: 2.6 lb w/mounting bracket remote sensor: 0.2 lb	\$8,995*	Includes integral ADAHRS backup battery and emergency GPS, integral altitude alterter/preselect, GPS				
www.aspenavionics.com	C2d, C3d, C4c, C6d, C8d, C10b, C106, C113	6.0-in diag.	Pitot/static quick connect	(1) RS-232 (3)	display: 3.50 x 7.0 x 4.15 depth: 6.35 in. remote sensor: 2.65 x 4.40 x 1.0 in.	14-28 VDC (provided by PFD)	flight plan map views: 360° and arc, slaved directional gyro with heading bug.				
	1000C3 Pro	TFT AMLCD (400 x 760)	ARINC 429 (5) RS-232 (5)	ARINC 429	display: 2.6 lb w/mounting bracket remote sensor: 0.2 lb	\$8,995	Same as EFD 1000, plus full EHSI with dual bearing pointers; dual GPS, dual VHF nav support; auto- pilot and flight director integration;				
	C2d, C3d, C4c, C6d, C8d, C10b, C106, C113	6.0-in. diag.	Pitot/static quick connect	(1) RS-232 (3)	display: 3.50 x 7.0 x 4.15 depth: 6.35 in. remote sensor: 2.65 x 4.40 x 1.0 in.	14-28 VDC (provided by PFD	integral GPS steering; base map with curved flight paths; (optional) traffic, weather overlays.				
	EFD1000 Pro Primary Flight Display	TFT AMLCD (400 x 760)	ARINC 429 (5) RS-232		Display: 2.6 lbs w/ mount- ing bracket remote sensor: 2.65.4.40 x 1.0	\$10,995	Economical full-feature glass primary flight display for GA retrofit;				
	C2D, C3D, C4C, C6D, C8D, C10B, C106, C113	6.0-in. diag.	(5) Pitot/ static quick connect		Display: 3.50 x 7.0 x 4.15, depth: 6.35-in. remote sensor: 2.65 x 4.40 x 1.0 in.	8 to 32 VDC	EFIS six-pack replacement; Compat- ible with many avionics.				
	EFD1000 Multifunction Display	TFT AMLCD (400 x 760)	ARINC 429 (5) RS-232 (5) Pitot/		Display: 2.6 lb. w/ mounting bracket remote sensor: 2.65. x 4.40 x 1.0 in.	\$8,995	Duplicate sensor set providing full PFD redundancy; may eliminate requirement for backup instru- ments; sectional-style moving maps with hazard awareness overlays:				
	C2D, C3D, C4C, C6D, C8D, C10B, C106, C113	6.0-in. diag.	static quick connect	t	Display: 3.50 x 7.0 x 4.15, depth: 6.35-in. remote sensor: 2.65 x 4.40 x 1.0 in.	8 to 32 VDC	charts and geo-referenced airport diagrams; customizable screen layouts; built-in back-up battery and emergency GPS.				
	EFD1000 Pro Plus Primary Flight Display	TFT AMLCD (400 x 760)	ARINC 429 (5) RS-232 (5) Pitot/ static quick connect		Display: 2.6 lb. w/ mounting bracket remote sensor: 2.65.4.40 x 1.0 in.	\$13,995	EFD1000 PFD with Evolution Synthetic Vision and angle of attack indicator; Lowest price full-featured glass panels for GA retrofit; ad- vanced FFIS six-pack replacement				
		6.0-in. diag.		static quick connect	static quick connect	static quíck connect	static quick connect	static quick connect		Display: 3.50 x 7.0 x 4.15, depth: 6.35-in. remote sensor: 2.65 x 4.40 x 1.0 in.	8 to 32 VDC
	EFD500 Multifunction Display	TFT AMLCD (400 x 760)	ARINC 429 (5) RS-232 (5) Pitot/ static quick connect	Display: 2.6 lb. w/ mounting bracket remote sensor: 2.65.4.40 x 1.0 in.	\$5,495	Sectional-style moving maps with hazard awareness overlays; custom- izable screen layouts; dharts and					
	C113	6.0-in. diag.		static quick connect		Display: 3.50 x 7.0 x 4.15, depth: 6.35-in. remote sensor: 2.65 x 4.40 x 1.0 in.	8 to 32 VDC	geo-referenced airport diagrams; built-in backup battery; broadest autopilot/flight director support.			
	EFD1000 VFR Primary Flight Display	TFT AMLCD (400 x 760)	ARINC 429 (5) RS-232		Display: 2.6 lbs w/ mount- ing bracket Remote Sensor: 2.65.4.40 x 1.0 in.	\$4,995	Consolidates traditional six-pack instrument information plus CDI into a single display with a back battery and emergency GPS; Lowest price, full-featured PFD for GA aircraft; works with your panel's existing				
	C2D, C3D, C4C, C6D, C8D, C10B, C106, C113	6.0-in. diag.	(5) Pitot/ static quick connect		Display: 3.50 x 7.0 x 4.15, depth: 6.35-in. Remote Sensor: 2.65 x 4.40 x 1.0-in.	8 to 32 VDC	avionics; unique PFD design slides into existing panel cutouts; Options include autopilot interface, (GPS steering); weather and traffic; Afford- able upgrades include HSI; bearing pointer and IFR features with easy software upgrade.				
	Model	Display			Units/Weight (lb.)	Price					
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Manufacturer	TSO	Display Size	Inputs	Outputs	Size or Form Factor	Power Required	Remarks				
Aspen Avionics 5001 Indian School Rd. NE Albuquerque, NM 87110 (505) 856-5034	EFD1000H Helicopter Primary Flight Display	TFT AMLCD (400 x 760)			Display: 2.6 lb. w/ mounting bracket Remote Sensor: 2.65.4.40 x 1.0 in.	\$15,195	Special vibration mount meets D0- 160F helicopter vibration standards; airspeed and altitude tapes, with al- titude alerter; built-in GPS steering; full cleatronic MSL with dual hearing				
Fax: (505) 314-5440 www.aspenavionics.com	C2D, C3D, C4C, C6D, C8D, C10B, C106, C11	6.0-in. diag.	ARINC 429 (5) RS-232 (5) Pitot/ static quick connect		Display: 3.50 x 7.0 x 4.15, depth: 6.35-in. Remote Sensor: 2.65 x 4.40 x 1.0-in.	8 to 32 VDC	binters; base map with flight plan legs and waypoints; integral air data computer and attitude heading reference system; built-in back-up battery; optional evolution hazard awareness provides traffic and weather displays; lowest price, full- featured glass panels; works with your panel's existing avionics.				
Avidyne Corp. 55 Old Bedford Rd. Lincoln, MA 01773 (800) 284-3963 Fax: (614) 885-8307	EX600	AMLCD			4.75	\$8,995 without radar; starting at \$12,990 w/radar	Full overlay of GPS flight plan along with traffic, wx radar, data-linked wx, and special-use airspace. Features include full vector-based moving map and interfaces for traffic and				
www.avidyne.com	C63c, C110a, C113, C118, C147, C157, C43c, C106	5.7-inch diago- nal 640 x 480 pixels (VGA)	RS-232 ARINC 429 ARINC568 DME	ARINC 429	6.25 x 4.93 x 11.0	28VDC	lightning, CMax approach charts and airport diagrams, plus 20 different radar models. Many optional radar interfaces and acts as a display replacement for many older CRT radar displays.Dedicated knobs for radar control of tilt and bearing, plus second set of context-sensitive knobs for range and other functions. Features map panning keys and allows pilot to toggle between the present position and a panned-to po- sition — such as destination airport — with a single button push. Many options available.				
Garmin International 1200 E. 151st St. Olathe, KS 66062 (800) 800-1020 (800) 357-8200 Fax: (913) 397-8282	TAWS -B	GNS 400 (W) series, 500 (W) series	Not	N/A	N/A	Varies					
www.garmin.com	C151 ETSO-C151	_	manufacturer	IN/A	N/A	N/A					
	TAWS-A	GTN 600 se- ries, GTN 700 series, G900X, G950	Not provided by	N/A	N/A	Varies					
	C151 ETSO-C151	_	manufacturer	,	N/A	N/A					

	Model	Display			Units/Weight (lb.)	Price	
Manufacturer	TS0	Display Size	Inputs	Outputs	Size or Form Factor	Power Required	Remarks
Garmin International 1200 E. 151st St. Olathe, KS 66062 (800) 800-1020 (800) 357-8200 Fax: (913) 397-8282	GTS 800	GNS 400 (W) series, 500 (W) series, GTN 600 series, GTN 700 series, GNS 480, GMX	Not provided by	_	1/8.92	\$9,995	TAS traffic surveillance system able to track up to 45 targets up to a 22-
www.garmin.com	C147 Class A ETSO 147 Class A C166B	_	manufacturer		2.66 x 6.25 X 14.78	2.6A @ 14 VDC 1.5A @ 28 VDC	nm interrogation range.
	GTS 825	GNS 400 (W) series, 500 (W) series, GTN 600 series, GTN 700 se- ries, GNS 480, GMS 200	Not provided by manufacturer	. –	1/11.3	\$19,995	TAS traffic surveillance system able to track up to 75 targets up to a 40- nm interrogation range.
	C147 Class A ETSO 147 Class A C166B ETSO C166b	_			3.42 x 6.25 x 14.78	3.5A @ 14 VDC 1.7A @ 28 VDC	
	GTS 855	GNS 400 (W) series, 500 (W) series, GTN 600 series, GTN 700 series, GNS 480, GMX	Not provided by manufacturer		1/11.3	\$24,995	High-performance TCAS I collision
	C118 ETSO C118 C116b ETSO C166b	_		_	3.42 x 6.25 x 14.78	3.5A @ 14 VDC 1.7A @ 28 VDC	up to 75 targets within an 80-nm forward interrogation range.
	GTS 8000	GNS 400 (W) series, 500 (W) series, GTN 600 series, GTN 700 series, G900X, G950	Not provided by manufac- turer	_	1/11.3	\$89,995	TCAS II Change 7.1 syste, includes GTS 8000 TCAS processor and two GTX 3000 TCAS transponders.
	C119c ETSO C119c C166b ETSO C166b	_			3.42 x 6.25 x 14.78	3.5A @ 14 VDC 1.7A @ 28 VDC	

	Model	Display			Units/Weight (lb.)	Price	
Manufacturer	TS0	Display Size	Inputs	Outputs	Size or Form Factor	Power Required	Remarks
Honeywell Aerospace BendixKing 9201 San Mateo Blvd. NE Albuquerque, NM 87113	BendixKing KDR 610 XM Weather Receiver	see remarks	ee arks weather displays via		1/1.5	\$6,888	Part of an MFD system; data link weather receiver provides high- speed textual and graphical weather to the cockpit. Available
(855) 250-7027 www.bendixking.com	C157	see remarks	weather displays via XM satellite	XM satellite interfaced to Bendix/King KMD 250, KMD 550 and KMD 850 MFDs	see remarks	10-32 VDC	weather products include composite NEXRAD radar, graphical METARs, AIRMETS and SIGMETS. The active flight plan can be overlaid on all graphical weather images. System enables user to pan, zoom and interrogate areas of interest via joystick.
	BendixKing KSN 770/765 Integrated Navigator WAAS/GPS/ NAV (KSN 770 only)/COMM (KSN 770 only)/MFD	Active Matrix LCD, 40 x 480 pixels (Full VGA) / 5.7 in. diagonal	ARINC 429 input; 10 ARINC 429 output; 2 RS-232 input; 4 RS- 232 output;		1/(770/765) 9.9/8.1/6.25" x 5.25" x 10"	List Price Starting at \$10,995	KSN 770 combines GPS navigation, terrain mapping, charting and safety sensor displays. It can also display XM Datalink Weather, radar-based weather, traffic and terrain. Offers many ways to interface with informa- tion via a combination of hard buttons, cursor control and touch-
	N/A	see remarks	232 output; 4 RS-222 input; 3 RS- 222 output; 28 discrete input; 20 discrete output		N/A	11-33 VDC	screen. The KSN 770 features Wide Area Augmentation System (WAAS) and Localizer Performance with Vertical Guidance (LPV). Can also display safety systems including on-board weather radar, Enhanced Ground Proximity Warning System (EGPWS), XM Datalink Weather. Numerous options.
Honeywell Aerospace 1944 East Sky Harbor Circle	Honeywell MFRD	LCD	50.000		1/7.5	\$64,525	
Phoenix, AZ 85034 (800) 601-3099 Fax: (602 365-3343 www.honeywell.com	C63c, C110a, C113, C196	6 in. diag.	ARINC 429, radar, datalink, EGPWS, traffic, NTSC video	display	6.24 (w) x 4.82 (h) x 8.38 (panel depth)	28 VDC or 115 VAC 400 Hz	Multi-function display of weather radar, traffic, terrain, navigation maps, checklists.
Innovative Solutions & Support (IS&S) 720 Pennsylvania Dr.	Integrated Standy Unit (ISU)	10-, 15-, 17-, 20-in. flat panel displays	RS 422/232: 3 channels Input/Output ARINC 429:	2 outputs, high speed/	N/A	N/A	Calculates and displays altitude,
Exton, PA 19341 (610) 646-9800 www.innovative-ss.com	NA	N/A	(configurable for VOR, ILS, DME, FMS, GPS)	low speed (softward configurable)	3 ATI clamp mount, optional panel mount	28 VDC 9.8 W	attitude, airspeed, slip/skid and navigation display information.
L3 Aviation Products 5353 52nd St. SW Grand Rapids, MI 49512 (616) 949-6600 Fax: (616) 285-4224	Trilogy ESI-1000	AMLCD; optional NVG compatibility			1/2.22	\$14,995	Electronic standby instrument designed to level "A" software and hardware compliances, the Trilogy ESI replaces traditional standby instruments and combines attitude, attitude and airspeed information
www.L3aviationproducts. com	C2d, C3e, C4c, C6e, C10b, C46a, C113, C179	4.0 x 3.0	N/A	N/A	3-ATI chassis 4.0 x. 3.35 x 7.66	14-28 VDC	into a compact 3.8-in. diagonal display while maintaining a 3-ATI chassis design. Heading is available when coupled with the optional magnetometer. For fixed-wing and helicopter applications.

	Model	Display			Units/Weight (lb.)	Price	
Manufacturer	TSO	Display Size	Inputs	Outputs	Size or Form Factor	Power Required	Remarks
L3 Aviation Products 5353 52nd St. SW Grand Rapids, MI 49512 (516) 949-6600	Trilogy ESI-2000	AMLCD; optional NVG compatibility			1/2.56	\$15,700	Electronic standby instrument incor- porates an internal battery to meet the requirements for independent, dedicated back un power for eigenet
Fax: (616) 285-4224 www.L3aviationproducts. com	C2d, C3e, C4c, C6e, C10b, C46a, C113, C179	4.0 x 3.0	NA	NA	3-ATI chassis 4.0 x. 3.00 x 6.7	14-28 VDC	without dual electrical system. The lithium ion battery is integrated into the ESI-2000 hardware with a triple redundant safetydesign and pro- vides a minimum of 1 hr. and up to 4 hr. of standby power. Heading is available when coupled with the optional magnetometer. For fixed- wing and helicopter applications.
	GH-3900 ESIS	Active matrix LCD			1/3.0	\$38,000	Features a lighter and shorter chas- sis than previous models and allows the installer to define multiple I/O
	C2d, C3e, C4c, C6e, C8e, C10b, C34e, C35d, C36e, C40c, C46a, C66c, C95a, C106, C113, C115b, C145c	3 ATI	ARINC 429, R5-232, discrete and analog	ARINC 429, RS-232, discrete and analog	8.33 x 3.19 x 3.19	Dual 28 VDC inputs (18 VDC emergency power)	interfaces., SSEC and VMO values. An Aircraft configuration PC Soft- ware Tool simplifies the setup of the unit, allowing installers to define and customize the presentation of col- ors, flight cues and navigation data. Designed for FAR Part 25, Part 23 (Class III & IV). Part 27 and Part 29. Variety of air data and heading input options as well as built-in accelerom- eters. Classified as Non-ITAR.
	GH-39RSU ESIS	DU-42 Display Acitve Matrix LCD	DU-42 Display: 3 ARINC 429; 1 USB Serial		DU-42 Display: 1.5 Remote Sensor Unit: 3.0	N/A	Features a 4.2-in. diagonal igh-
	DU-42 Display: C2d, C3e, C4c, C6e, C8e, C10b, C34e, C35d, C36e, C40c, C46a, C66c, C95a, C106, C113a Remote Sensor Unit: C2d, C3e, C4c, C6e, C8e, C10b, C46a, C95a, C106	1.50 (l) x 5.25 (w) x 3.0 (h) null	Serial Bus; 1 RS-232 Serial Bus; 12C Serial Bus; 1 Analog Remote Sen- sor Unit: 7 ARINC 429; 1 RS-232 Serial Bus; 6 Discrete Pneumatic pressure ports	DU-42 Display: 1 ARINC 429 Remote Sen- sor Unit: 3 ARINC 429; 2 Discrete; 2 Analog	_	+28 VDC nominal	resolution display (DU-42) and a separate Remote Sensor Unitt (RSU). 1.5-indeep display allows installation in aircraft with limited space behind the panel. Configu- rable I/O interfaces and SSEC and VMO values, as well as display pa- rameters. Designed for FAR Part 25 and Part 23 (Class III & IV aircraft, and Part 27 and Part 29 helicopters.
	ESI-500	24-bit color LCD; optional NVG compatibility	Inputs: - discrete pneumatic pressure		_	\$5,600	Standby system designed for piston and turboprop aircraft and helicop-
	C2d (Type B) C8e (Type B) C10b (Type 1, Range: -1,500 to +35,000 ft.) C34e C35d C36e C40c C46a (Range: 20 to 300 kt.) C106 C113a C179a C201 C2d (Type B) C8e (Type B) C8e (Type B) C10b (Type 1, Range: -1,500 To +35,000 ft.)	3.0 x 3.0	ports — ARINC 429 GPS or VLOC input or both (navigation) - MAG-500 magnetom- eter or an ARINC-429 (heading)- GPS (aircraft track) — OAT to compen- sate baro- corrected altitudes for temperature		2.75 3.25 x 3.25 in. bezel; 3.0 x 3.0 display	14-28 VDC	ters. Comes standard with altitude, attitude, slip/skid, vertical speed and aircraft track. Options available for display of navigation information and synthetic vision inputs, includ- ing terrain and obstacles. Magnetic heading optional when coupled with MAG-500 magnetometer. ESI-500 is compatible with existing NAV radios and GPS hardware. An internal lithium-ion battery pack automatically powers the system without interruption upon loss of main input power.

	Model	Display			Units/Weight (lb.)	Price		
Manufacturer	TS0	Display Size	Inputs	Outputs	Size or Form Factor	Power Required	Remarks	
Rogerson Aircraft 2201 Alton Pkwy. Irvine, CA 92606 (949) 660-0666 www.rogersonaircraft.com	5 ATI EFIS C3d, C4c, C5e, C6d, C8d, C9c, C34e, C35d, C36e, C40c, C41d, C52b, C63c, C66c, C67, C87, C92c, C113, C117a, C118, C119b, C129a, C147, C161a	AMLCD flat panel 5 ATI or 6.4-in. diagonal	analog synchro (XYZ, Sin/Cos) vari- able AC/DC discretes & digital ARINC 429, 419, 453, 735, RS-232	analog synchro (XYZ, Sin/Cos) vari- able AC/DC discretes & digital ARINC 5 ATI or 429, 419, 453, 735, RS-232		\$42,000* 28 VDC 44 W max	One, two or four programmable, self- contained flat-panel AMLCD EADI and EHSIs. Radio altimeter functions such as DH, expanded scale for landing helicopter operations, TCAS I and II, and EGPWS display capability, in addition to standard ADI, HSI, bearing pointers, CDI, autopilot annunciation, flight director cross bars or 'V' bars. Upgrade packages available. *BCA estimate.	
Sandel Avionics 2401 Dogwood Way Vista, CA 92081	SA4550 Primary Atti- tude Display	rear projection LCD w/LED backlighting	analog: attitude glideslope,		1/3.4 \$20,950*		Designed to upgrade legacy ADIs. Incorporates flight director command bars, glideslope/localizer deviation	
(877) 726-3357 (760) 727-4900 Fax: (760) 727-4899 www.sandel.com	C113, C3d, C4c, C34e, C36e, C52b	4 ATI	localizer, flight director command inputs, radar altimeter mode an- nunciators	NA	4 ATI	28 VDC 40 W	scale, fast/ slow indicator and mode annunciations. Selectable single-cue/split-cue display option. Sunlight readable LED backlit dis play with 180-degree viewing angle and over 10,000-hour MTBF. *High- vibration version, \$23,800. NVIS compatible version, \$27,050	
Universal Avionics Systems Corp. 3260 E. Universal Way Tucson, AZ 85756 (520) 295-2300 Fax: (520) 295-2395 www.uasc.com	EFI-890R	active matrix color LCD	Analog: 6 - ARINC 429 5 - CSDB 2 - ARINC 708 3 -Manches- ter bus ports 2 VGA or 1 - Manches- 1 - RDR-1E/F & 1 - VGS, 2 - RSS-170 0 rg2 - NTSC 5 - GND/OPN		1/ 12.0	\$62,000*	Horizontal viewing angle +60°/-60°, vertical viewing angle +45°/-10°; resolution: 780 x 780 pixels: 124 5 color groups per inch	
	C2d, C3d, C4c, C52, C6d, C8d, C10b, C34e, C35d, C36c, C40c,C41d, C52,b, C63c, C66c, C87, C92c, C95, C105, C113, C115b, C118, C119a, C129a, C151a	6.3 c 63 (8.0-om. dia.)	comp. or 18:1 1 - RS-232 (maint.) Digital: 28 GND/OPN discretes 14 - 28 VDC/ opn 4- ARINC 407 with 2 ref. inputs 15- analog DC	discretes 3 - 28 VDC/ OPN dis- cretes 6 - analog resolvers 2 - DC dif- ferential 2 - DC single ended	Bezel: 7.84 h x 7.42 w Depth: 9.79 (back of bezel to read of connector)	28 VDC	(CGPI); sunlight readability with greater than 10,000/1 dimming range. *Depending on configuration.	

ELECTRONIC FLIGHT BAGS

Monufacturar	Model	Display	Inputs and Outputs	Units/Weight (lb.)	Price	Domostvo
Manufacturer	Class	Display Size	inputs and Outputs	Size	Power Required	Rellidiks
Esterline CMC Electronics 600 Dr. Frederik Phillips Blvd. Montreal, Quebec, Canada 4HM2S9	PilotView CMA-1100 (8.4 in.) or CMA-1410 (10.4 in.) or CMA-1612 (12.1 in.)	touchscreen XGA AMLCD 8.4-in. or 10.4- in. diagonal	Ethernet, ARINC 429, discrete, RS422/232, USB 2.0, ARINC 717, ABINC 615	EDU 8.4 in.: 3.5 EDU 10.4 in.: 4.0 EDU 12.1 in.: 5.1 EEMU: 2.0	\$20,000- \$25,000	CMC's Aircraft Information Server acts as an integrated aircraft information management server and aircraft interface device, enabling a wide range of applications and interfaces with
(514) 748-3184 Fax: (514) 748-3100 www.cmcelectronics.ca	Class 2 and Portable	8.4-in. or 10.4- in. diagonal	ARINC 619		N/A	any display or tablet solution.
Garmin International 1200 E. 151st St.	aera 796	NA		1/26.4 oz.		Portable GPS with EFB, charting, terrain, mov-
Olathe, KS 66062 (913) 397-6200 Fax: (913) 397-8282 www.garmin.com			RS 232, USB,		¢1.800	ing map, weather, XM and other capabilities. New 3-D vision technology shows a virtual 3-D behind-the-aircraft perspective of surround- ing terrain derived from GPS and the onboard terrain database. With 2 serial ports, aera 796 allows for simultaneous connectivity with
	Class 1 or Class 2	7-in. diagonal	Bluetooth		\$T'922	other hardware. With optional GTX 330 Mode S transponser interface, can access Traffic Information Service (TIS) alerts, where avail- able, right on the device while also sending frequencies to a GTR 225 comm radio or GNC 255 nav/comm. Can also relay position reports to other devices.
	aera 660	F" Diagonal	RS-232, USB,	4/8.04.00	\$700	Portable GPS/EFB with charting, terrain,
	Class 1 or Class 2	Jiagoffal	Bluetooth, Wi-Fi	1/0.04 02.	\$199	wireless database updating and more.

ENHANCED/SYNTHETIC VISION SYSTEMS

Manufacture	Model	Display	to set a	Outrasta	Units/Weight (lb.)	Demoster
Manutacturer	TSO		Inputs	Outputs	Size	Kemarks
Esterline CMC Electronics 600 Dr. Frederik Phillips Blvd. Montreal, Quebec, Canada H4M2S9	CMA-2600 SureSight I-series EVS-IR Sensor	HUD/HDD	single, dual- band sensor operating in the short to medium wavelengths,	2-ANSI/SMPTE 170M ARINC 429 RS 422	1/LRU 21.0	Certified as part of an EFVS which provides operational landing credits as well as enhanced situational awareness to pilots in low-visibility conditions
Fax: (514) 748-3100 www.cmcelectronics.ca	—		1-5 microns	discretes	NA	
	CMA-2700 SureSight I-Series EVS-IR sensor	HUD/HDD	single, dual- band sensor operating in the short to medium wavelengths, 1-5 microns	2-ANSI/SMPTE 170M ARINC 429 RS 422 descretes 2 ARINC 818	1/LRU 21.0	Certified as part of an EFVS for operational landing credits by three leading Airworthi- ness Authorities (EASA, TCCA, and the FAA). Fully compliant for FAR 91.176
Elbit Systems of America Fort Worth Operations 4700 Marine Creek Pkwy. Fort Worth, TX 76179 www.elbitsystems-us.com	EVS II	HUD/HDD	1-5 micron infrared sensor	RS-170/SMPTE 170M; SMPTE 259; RS 232/RS 422; ARINC 429 descrets	3/22.0	EFVS certified for FAR 91.175 (I) and (m) operational credit. EFVS certified for Part 91, 135 and 121 operations on fixed- and rotary-wing applications. Contact manufacturer for specific application pricing
	—	—			1/2 ATR	
	GAVIS	any RS-170/ SMPTE, 170M analog video capable display	8-14 micron in- frared sdensor	RS-170/SMPTE 170M, analog video	1/3.5	EVS certified for situational awareness in all weather conditions. Certified for fixed- and rotary-wing aircraft. Contact manufacturer for specific application pricing
	see remarks				3.0 x 6.0 x 11.0	
L3 Aviation Products 5353 52nd St. SW Grand Rapids, MI 49512	IRIS A100	any RS-10 compatible displays	7 -14 micron		1/1.7	Uses uncooled BST technology, IRIS provides enhanced visibulity of almost any object. day or night, by measuring variations
(616) 949-6600 Fax: (616) 285-4224 www.L3aviationproducts. com	see remarks		uncooled ferroelectric sensor	RS-170, NTSC compatible video or PAL	5.4 x 5.4 x 3.4	in heat signatures. A real-time, black and white image of people, animals, aircraft and terrain is displayed on any compatible RS- 170 cockpit display. King Air, Bell 206 and Twin Commander STC kits additional.

ENHANCED/SYNTHETIC VISION SYSTEMS

Monufacturor	Model	Display	Inpute	Outpute	Units/Weight (lb.)	Pomorka
Manufacturer	TS0		inputs	Outputs	Size	Reliidiks
Astronics/MAX-VIZ, Inc. 11241 SE Hwy 212 Clackamas, OR 97015 (503)968-3036	Max-Viz 1500	MFD or EFB	long-wave uncooled	RS-170 video	sensor 2 lb.; PWS module 2.5 lb.	Multiple STCs for fixed- and rotory-wing air- craft. Turbine helicopter, high-performance
sales@mv.com	_		320 x 240	FOV discreet	3.75 x 5.0 x 2.25	turboprop and jet fixed-wing aircraft.
	Max-Viz 1400	MFD or EFB	long-wave	RS-170 video	1.2 lb.	The Max-Viz 1400 is a general aviation enhanced vision sensor using a 640 x 480
	—		640 x 480	polarity select	3.07 x 6.16 x 2.09	pixel resolution long-wave infrared thermal imager with electronic zoom.
	Max-Viz 1200	MFD or EFB	long-wave uncooled	RS-170 video	1.2 lb.	Developed for general aviation piston air- craft heliconters, and slower single-engine
	_		320 x 240		3.07 x 6.16 x 2.09	turboprop fixed-wing aircraft.
	Max-Viz 600	MFD or EFB	long-wave uncooled CMOS blended	RS-170 video	1.2 lb.	Developed for general aviation piston air- craft, helicopters, and slower single-engine
	—		with IR 320 x 240		3.77 x 8.69 x 2.21	turboprop fixed-wing aircraft.
Lexavia 4020 52nd Ave Ct. NW Gig Harbor, WA 98335 (850) 343-1147	LFS-3500 Long-Wave Infrared Sensor	any NTSC RS-170 or PAL compat- ible display	12: 28VDC input power,	PAL video output; serial control in- terface — RS-232	1.0 lb.	Price: \$29,250 (640 x 512 resolution), \$22,933 (336 x 256)/28 VDC. High-perfor- mance rugged sensor design provides an increased level of situational awareness
www.Lexavia.com		PND, MFD or dedicated display)	NTSC-RS-170	RS-422, RS-485	2.5 x 2.8 x 6.3	for improved safety of operations. Optional controller and stowable video displays also available.
	LFS-6000 Long-Wave Infrared Sensor	any NTSC RS-170 or PAL compat- ible display	12: 28VDC	PAL video output Serial Control In-	0.4 lb.	Price: \$39,495 (640 x 512 resolution), \$31,913 (336 x 256)/28 VDC. Compact, lightweight and aerodynamically shaped EVS sensor provides an increased level of
		device (PFD, PND, MFD or dedicated display)	NTSC-RS-170	terface — RS-232, RS-422, RS-485	2.42 x 2.32 x 5.31	situational awareness for improved safety of operations. Optional controller and stow- able video displays also available.
	LFX-2010 Long-Wave Infrared Sensor	any NTSC RS-170 or PAL compat- ible display device (PED	12: 28VDC input power,	PAL video output Serial Control In- terface — RS-232.	1.4 lb	Price: \$33,424 (640 x 512 resolution) 28 VDC. High-performance ruggedized sensor designed for special operations (hoist, fast rope and external operations) to provide an increased level of situational awarenees for
	_	PND, MFD or dedicated display)	NISC-RS-170	RS-422, RS-485	2.5 x 2.58 x 5.1	mission critical applications and improved safety of operations. Optional controller and stowable video displays also available.
Rockwell Collins 400 Collins Rd. NE Cedar Rapids, IA 52498	EVS-3000	HUD/HDD	uncooled multi-spectral		9.2 lb.	Provides situational awareness at night and in low-visibility conditions. When displayed head up, operational approval for landing
(319) 295-1000 Fax: (319) 295-2297 www.rockwellcollins.com	_		ARINC 429		1 LRU	able. Contact OEM for specific application pricing.

FLIGHT MANAGEMENT SYSTEMS

	Model	CDU Type	# Available ARINC 429	Vertical	Performance Management		Specific	Weight (lb.)	
Manufacturer	TSOs		(In/Out)	Nav moues	Remote Radio	Air Data In	Interfaces	CDU	Price / Remarks
		Disnlay Type	TSO'd	# Available ARINC 429	Turning	(# types)	ARINC	Dimensions	
	RNP Certification		Nav Sensors	Procedure Legs	Radar (In)		429 (Out)	Power	
Esterline CMC Electronics	CMA-9000	full alpha			Yes			8.0	Price varies by
Phillips Blvd.		keyborad	24/8	fully coupled	Yes	ARINC 429/ ARINC 575	429 DME std.; 429 VOR std.	2.10	installation. Cou-
H4M2S9 (514) 748-3184 Fax: (514) 748-3100	C129		21/0	VNAV				6.75 x 5.75 x 7.15	optimized and advisory VNAV for climb, cruise, de-
www.cmcelectronics.ca		AMLCD.color							Performance table based. FANS-1
	RNP 0.3, -10, BRNAV, PRNAV		GPS, WAAS, VOR, DME, INS, IRS, TACAN		No			_	capable LPV Ap- proach capabile. Optional NVG display.

FLIGHT MANAGEMENT SYSTEMS

	Model	СDU Туре	# Available ARINC 429	Vertical Nav Modes	Performance Management	A:	Specific	Weight (lb.)	
Manufacturer	TSOs		(In/Out)	# Available	Remote Radio Tuning	Data In (# types)	ARINC	CDU Dimensions	Price / Remarks
	RNP Certification	Display lype	Nav Sensors	ARINC 429 Procedure Legs	ARINC Radar (In)		429 (Out)	Power	
FreeFlight Systems 3700 Interstate 35 S.	2101 Approach Plus	Dzus	4/0	Advisory	No	ARINC 565, ARINC		3.65	\$7,245. Price
(254) 662-0000 Fax: (254) 662-9450	_				No	Coarse/ Fine A407 Synchro	ARINC 429	3.0 x 5.75 x 7.68	includes receiver, data card, instal- lation kit (with an-
in the ongregorithe of the original sector of	BRNAV	LED	GPS, WAAS	4	None	ARINC 545, TAS, ARINC 429 ADC, RS-232 ADC	GPS RS-232	10- 40 VDC	tena), installation manual and pilot guide, Unit also available with NVG capability.
	2101 I/O Ap- proach Plus	Dzus	4/0	Advisory	No	ARINC 565, ARINC		3.65	\$11,500. Price
	_				No	575; Coarse/ Fine	ARINC	3.0 x 5.75 x	includes receiver, datacard, instal-
						A407 Synchro, ARINC	429	7.68	lation kit (with an- tena), installation
	BRNAV	LED	GPS, WAAS	4	None	545, TAS, ARINC 429 ADC, RS-232 ADC	GPS RS-232	10- 40 VDC	guide. Sole means oceanic approval; interfaces with EGPWS.
Rockwell Collins	FMS 3000/5000	full alpha			Yes			80	LPV approach
Cedar Rapids, IA 52498 (319) 295-4085		keyboard	4/3	multi-waypoint					legs are available on some aircraft
Fax: (319) 295-2297 www.rockwellcollins.com	C129 GPS, C146 WAAS-B1, -C1	color LCD	GPS, WAAS, DME, INC, Loran C		Yes	see remarks	see remarks	6.375 x 5.75 x 6.33	types. +FMS //O provided by four redundant concen- trators. Remote computer dimen- sions 1.7 x 8.84 x 6.06 in.; FMS 5000 requires radio tuning unit; FMS 3000 radio tuning is internal.
	RNP 0.3, -10, BRNAV			S, 23	see remarks			20- 40 VDC	
	FMS 4200/6000	full alpha keyboard		multi-waypoint	Yes			4	LPV approach
	C129 GPS, C146 WAAS-B1, -C1		4/3		Yes			4	legs are available on some aircraft
	RNP 0.3, -10 BRNAV Color LCD GPS, WA DME, IN Loran					See	see	6.375 x 5.75 x 6.33	types. *FMS I/O provided by four redundant concentrators. See
		GPS, WAAS, DME, INC, Loran C	23	see remarks	remarks	remarks	20- 40 VDC	for remote com- puter; FMS4200 has advisory VNAV but not FMS-to-ILS auto transfer; Coupled VNAV available on FMS6000	
	FMS 6100	full alpha	1/2	multi-waypoint	Yes			4	ENO 1/0
-	C129 GPS, C146 WAAS-B1, -C1		-/ 3		Yes	see	see	6.375 x 5.75 x 6.33	FMS I/O provided by four redundant concentrators. See FMS 3000
	RNP 0.3, -10 BRNAV	color LCD	VOR< GPS, WAAS, DME, INS, Loran C	23	see remarks	remarks	remarks	20- 40 VDC	remarks for remote computer. WAAS/ SBAS capable.

FLIGHT MANAGEMENT SYSTEMS

	Model	CDU Type	# Available ARINC 429	Vertical	Performance Management		Specific	Weight (lb.)	
Manufacturer	TSOs		(In/Out)	# Available	Remote Radio Tuning	Air Data In (# types)	Interfaces ARINC	CDU Dimensions	Price / Remarks
	RNP Certification	Display Type	Nav Sensors	ARINC 429 Procedure Legs	ARINC Radar (In)	(429 (Out)	Power	
Universal Avionics				22		ARINC 575.	ARINC	2.9	\$54.500. Air data
3260 E. Universal Way Tucson, AZ 85756 (520) 295-2300	UNS-1Lw	Full alpha keyboard	8/5		opt.	ARINC 429 ADC std.; ARINC 565.	429 GPS, S422A CSDB DME, Arinc 429	4.5 x 5.75 x 6.33; remote	converter unit available; 3-D coupled approach
(800) 321-5253 Fax: (520) 295-2395 www.uasc.com	C129 GPS, C146B Gamma		GPS, WAAS, Optional: multi-w	multi-waypoint	opt.	Course/ Fine A407 Snchro.	DME, Bendix 429 VOR,	MCU, 7.7 lb.	for remote/ oceanic ops.; Uni-
	RNP 0.3, -5, -10	color LCD	VOR, DME, INS, IRS, Lo- ran, TACAN		std.	ARINC 545 TAS opt. See remarks	429 VOR, ARINC 429 INS	20- 40 VDC	ible; WAAS/SBAS capable.
	UNS-1LEw	full alpha keyboard	8/5	23	opt.	ARINC 575,	ARINC	7.86	
	C129 GPS,	-	8/5		opt.	ARINC 429 ADC std.; ARINC	429 GPS, S422A CSDB	6.38 x 5.75 x 8.96	\$69,000. 3-D coupled approach mode; PC program
		color LCD	GPS, WAAS, Optional: VOR, DME, INS, IRS, Lo- ran, TACAN	multi-waypoint	std.	565, Course/ Fine A407 Snchro, ARINC 545 TAS opt. See remarks	Arinc 429 DME, Bendix 429 VOR, ARINC 429 VOR, ARINC 429 INS	20- 40 VDC	for remote/ oceanic ops.; Uni- Link text compat- ible; WAAS/SBAS capable.
	UNS-1Espw	full alpha	8/5	8/5 23	opt.	ARINC 575, AF	ARINC	7.25	\$68,000. 3-D coupled approach mode; PC program
	C129 GPS, C146B	keyboard			opt.	ARINC 429 ADC std.;	429 GPS, S422A CSDB DMF	6.38 x 5.75 x 7.62	
	RNP 0.3, 5, 10	color LCD	GPS, WAAS, Optional: VOR, DME, INS, IRS, Lo- ran, TACAN	multi-waypoint	std.	565, Course/ Fine A407 Snchro, ARINC 545 TAS opt. See remarks	Arinc 429 DME, Bendix 429 VOR, ARINC 429 VOR, ARINC 429 INS	20- 40 VDC	for remote/ oceanic ops.; Uni- Link text compat- ible; WAAS/SBAS capable.
	UNS-1Fw	full alpha	9/5		opt.	ARINC 575,	ARINC	4.1	\$81.500. 3-D
	C129 GPS, C146B Gamma	keyboard	8/5	23	opt.	ARINC 429 ADC std.; ARINC 565,	429 GPS, S422A CSDB DME, Arinc 429	6.38 x 7.5 x 3.5; remote computer: 2.0 lb.	coupled approach mode; PC program for remote/ oceanic ops : Uni-
	RNP 0.3, 5, 10	color LCD	GPS, WAAS, Optional: VOR, DME, INS, IRS, Lo- ran, TACAN	multi-waypoint	std.	565, - Course/ Fine A407 Snchro, ARINC 545 TAS opt.	Arinc 429 DME, Bendix 429 VOR, ARINC 429 VOR, ARINC 429 INS	20- 40 VDC	Link text compat- ible; WAAS/SBAS capable.

				CDU Type		Weight (lb.)	
Manufacturer	Model	Inputs	Outputs		Operational Capabilities	Dimensions	Price/Remarks
				Dimensions		Power Required	
Avidyne Corp. 55 Old Bedford Rd.						18.75	Integrates primary flight informa- tion, navigation, terrain, weather,
Lincoln, MA 01773 (781) 402-7400 www.avidyne.com	Entegra	see remarks	see remarks	see remarks	FMS, PFD/ MFD, AP/ IFCS, EFIS,	two 10.4 in. diagonal, color active matris displays	traffic on two or three large- formate displays. Selectable IAS and V-speed ranges to suit aircraft installations. Dual-PFD version features CCS Cross Compare Sys-
	Release 8				TAWS, RMU, SVS, CAS/ TAWS	28 VDC	tem that monitors cross-side PDF and ADAHARS signals 30 times per second. Works with DFC90 or STEC 55 X autopilot and 3rd party GPS/NAV/Coms for position information.
					FMS, PFD/ MFD, AP/	18.75	Cirrus starting at \$90,000; Piper Matrix starting at \$105,800. Inte- grates primary flight information, navigation, weather and traffic on 2 or 3 large-format displays.
	Entegra Release9	see remarks	see remarks	see remarks	IFCS, EFIS, TAWS, RMU, SVS, CAS/ TAWS	two 10.4 in. diagonal, color active matris displays	Includes dual VHF nav/com, dual WAAS, GPS, dual FMS 900w dual ADAHARS, remote transponder tuning. ACD 215 alpha-numeric EMS learned with direnter Worke
						28 VDC	with DFC100 digital autopilot. Optional SVS.
Genesys Aerosystems One S-TEC Way		WX500, ADF,		color LCD	EMS PDE/	two screen: 2.0 four screen: 50.0	
Mineral Wells, TX 76067	Chelton Flight	ADS-B, TIS-B, radar altimeter,	ARINC 429, RS-232, RS-422,		MFD, AP/ IFCS, EFIS,	—	Two screens: \$95,000;
Formerly: Cobham Com- mercial Systems	EFIS	ARNC 429, RS- 232, RS-422, 10 discretes	autopilot	6.25 x 5.5 in. NVG compatible	TAWS, SVVS, CAS/TAWS	10-32 VDC	Four screens, \$150,000.
Garmin International 1200 E. 151st St. Olathe, KS 66062-3426 (913) 397-8200 Fax: (913) 397-8282 www.garmin.com	61000 N¥i	TCAS i/II, RS 232, RS-422, RS-485; ARINC	ARINC 429; HSDB, CD/HIS, RMI, air data. RS-232.	Varies by installation	See remarks	N/A	Price varies by installation. An all-glass avionics suite designed for OEM or custom retrofit instal- lation on a wide range of aircraft. Integrates primary flight informa- tion, navigation, communication, weather, terrain and traffic data on two or three large format displays. Tailored to specific OEM requirements. Features include 3-axis, all-digital flight control system: Synthetic Vision Pathway
		429; HSDB, CD/ HIS, RMI, air data	ARINC 429; HSDB, CD/HIS, RMI, air data	Varies by installation			navigation; dual AHRS; dual radio modules with WAAS certified IFR Oceanic-approved GPS, VHF Nav with ILS and VHF Com; dual RVSM compliant DADC; EICAS; ADS-B In and Out Transponder(s); Class B TAWS; Digital weather radar. Optional Bluetooth connectivity to select mobile devices. Retrofit system also available for King Air 300/350 and 200.
	G1000H (helicopter version)	TCAS i/II, RS 232, RS-422, RS-485; ARINC 429; HSDB, CD/ HIS, RMI, air data	ARINC 429; HSDB, CD/HIS, RMI, air data, RS-232, RS 422, RS-485; ARINC 429; HSDB, CD/HIS, RMI, air data	Varies by installation Varies by installation	see remarks	N/A	Price varies by installation. An all-glass avionics suite designed for OEM or custom retrofit instal- lation on a wide range of aircraft. Integrates primary flight informa- tion, navigation, communication, weather, terrain and traffic data on two or three large format displays. Tailored to specific DEM requirements. Features include 3-axis, all-digital flight control system; Synthetic Vision Pathway navigation; dual AHRS; dual radio modules with WAAS certified IFR Oceanic-approved GPS, VHF Nav with ILS and VHF Com; dual RVSM compliant DADC; EICAS; ADS-B In & Out Transponder(s); Class B TAWS; Digital weather radar. Optional Bluetooth connectivity to select mobile devices.

	Madal	Innuts	Outputs	CDU Type	Operational	Weight (lb.)	
Manufacturer	Model	Inputs	Outputs	Dimensions	Capabilities	Dimensions	Price/Remarks
Garmin International						Power Required	Price varies by installation.
1200 E. 151st St. Olathe, KS 66062-3426 (913) 397-8200 Fax: (913) 397-8282 www.gamin.com				12- or 14-in. backlit LED		N/A	Integrates primary flight informa- tion, navigation, communication, weather, terrain and traffic data on large format displays. Tailored to specific OFM requirements.
	G2000 (piston en- gine aircraft version)	TCAS I/II, RS- 232, RS-422, RS-485; ARINC 429; HSDB, CD/ HIS, RMI, air data	TCAS I/II, RS- 232, RS-422, RS -485; ARINC 429; HSDB, CD/HIS, RMI, air data		See remarks	N/A	Features include three-axis, all-digital automatic flight control system; Synthetic Vision Pathway navigation; dual solid-state AHRS; dual integrated radio modules with
				See remarks		N/A	WAAS certified IFR Oceanic-ap- proved GPS, VHF Nav with ILS and VHF Com with 16-W transceivers and 8.33-kHz spacing; dual RVSM compliant DADC; EICAS; Class B TAWS; digital weather radar; Garmin FliteCharts; and Garmin SafeTaxi.
		G3000 (light tur- bine aircraft version) TCAS I/II, RS- 232, RS-422, RS-485, ARINC 429; HSDB, CD/ HIS, RMI, air data	TCAS I/II, RS-232, RS-422, RS-485; ARINC 429; HSDB, CD/HIS, RMI, air data	14.1-in. diagonal WXGA		N/A	Price varies by installation. Integrates primary flight informa- tion, navigation, communication, weather, terrain and traffic data on large format displays. Tailored to
	G3000 (light tur-				See remarks	N/A	specific OEM requirements. Fea- tures include three-axis, all-digital automatic flight control system; Synthetic Vision Pathway naviga- tion; dual solid state AHRS; dual
	bine aircraft version)					N/A	integrated radio modules with WAAS certified IFR oceanic-approved GPS, VHF navigation with ILS and VHF communication with 16-watt transceivers and 8.33-kHz channel spacing; dual RVSM-compliant digital air-data computer; EICAS; Class B TAWS; XM Wx and/or digital weather radar; Garmin FliteCharts; and Garmin SafeTax
	G5000H (helicopter version)	TCAS I/II, RS- 232, RS-485; AZINC 429; HSDB, CD/ HIS, RMI, air data	RS 232, RS 422, RS 485, ARINC 429; HSDB, CD/ HIS, RMI, dis- cretes, air data	varies by installation	varies by installation	N/A	Price varies by installation. Advanced flight deck designed for OEM installation on medium-lift turbine helicopters. Bright high- resolution displays with Helicopter Synthetic Vision Technology (HSVT)
				See remarks	See remarks	N/A	conditions. Displays divide into 2 pages to help display multiple systems and sensors. Intuitive touchscreen interface with shallow menus and audible feedback.
						N/A	Graphical synoptics. Weather, charts, traffic, terrain and Global connectivity options. TOLD, per- formance planning and paperless cockpit support. Digital document display for electronic charts, flight manual data and more.

				CDU Type Operational	Weight (lb.)			
Manufacturer	Model	Inputs	Outputs	Dimonsions	Capabilities	Dimensions	Price/Remarks	
				Dimensions		Power Required		
Garmin International 1200 E. 151st St. Olathe, KS 66062-3426 (913) 397-8200 Fax: (913) 397-8282 www.garmin.com	G5000	TCAS i/II, RS 232, RS 422, RS 485; ARINC 429; HSDB, CD/HIS, RMI, air data		four backlit LED XGA 1280 X 800		NA	Price varies by installation. Intended for use aboard a broad range of professionally flown air transport category aircraft, ranging from light jets to large-cabin, transoceanic aircraft luborator orimony flight	
			RS 232, RS 422, RS 485; ARINC 429; HSDB, CD/ HIS, RMI, air data	touch-screen displays	see remarks	NA	information, navigation, communi- cation, weather, terrain and traffic data on large-format displays. Fea- tures include three-axis, all-digital automatic flight control system; Svnthetic Vision Pathway naviga-	
				see remaks		NA	tion; dual solid state AHRS; dual integrated radio modules with WAAS certified IFR oceanic approved GPS, VHF navigation with ILS and VHF communication with 16-watt	
	G500	TCAS i/II, RS 232, RS 422, RS 485; ARINC 429; HSDB, CD/HIS, RMI, air data	RS 232, RS 422, RS 485; ARINC 429; HSDB, CD, HIS, RMI, air data	dual 6.5-in. VGA LCDs		NA	\$15,995. Includes CDU, digital AHRS, ADC, magnetometer, tem- perature probe. Also certified to C2d, C10b and C34c. Replaces standard six-pack instruments.	
					see remarks	NA	AHRS. SVT is standard with G600 and optional for G500. Optional TAWS-B for G600 only. GWX70	
				_			radar sold separately. Includes CDU (dual 6.5-in. VGA LCD), digital AHRS, ADC, magnetometer,	
						NA	temperature probe. Enhanced au- topilot interface capabilities using the optional GAD 43	
		TCAS i/II, RS 232, RS 422, RS 485; ARINC 429; HSDB, CD/HIS, RMI, air data	RS 232, RS 422, RS 485; ARINC 429; HSDB, CD/ HIS, RMI, air data	dual 6.5-in. VGA LCDs		NA	\$29,995. Includes CDU, digital AHRS, ADC, magnetometer, temperature probe. Also certified to C2d, C10b and C34c. Replaces standard six-pack instruments.	
	G600				see remarks	NA	Features 6.5-in. PFD and MFD plus AHRS. SVT is standard with G600 and optional for G500.	
				_			Optional TAWS-B for G600 only. GWX70 radar sold separately. Includes CDU (dual 6.5-in. VGA	
						NA	netometer, temperature probe. Enhanced autopilot interface capabilities using the GAD 43.	

		Inputs		CDU Type		Weight (lb.)	
Manufacturer	Model		Outputs		Operational Capabilities	Dimensions	Price/Remarks
				Dimensions		Power Required	
Honeywell Aerospace BendixKing 9201 San Mateo Blvd. NF				12.0-in. color	FMS with Flight Director; Dual	see remarks	Integrated flight deck with three or four 12" LCDs, depending upon the aircraft insallation.
Albuquerque, NM 87113 (855) 250-7027	Primus Anov				ADAHRS; Graphi- cal flight planning; SmartView SVS: Digital	see remarks	features: digital autopilot capable of coupled VNAV, Smart- ViewTM Synthetic Vision System, Interactive Navigation (INAVTM) for graphical flight planning, and both a Cursor Control Device (CCD) and Multifunction Control-
www.bendixking.com	AeroVue (C106 in progress) C115b,	TCASI/II RS-232, RS-422, ARINC 429, ARINC 453, ethernet air data video dis	ARINC 429, ARINC 453, RS-232, RS- 422, discretes,		3-axis autopi- lot; Electronic checklist; XM Weather;		
	Class A1, B, C	cretes, analogs	analogs	Vertical nav profile; Video inputs; Dual WAAS GPS receivers; Integrated EIS; Mode S transponder; Dual audio panels with Bluetooth	vontaal nu profile; Video inputs; Dual WAAS GPS receivers; Integrated Els; Mode S transponder; Dual audio panels with Bluetooth	see remarks	ler for a more ergonomic user ex- perience. Weather radar, TCAS I, EGPWS, and radar altimeter also available. Price and weight are dependent upon installation. Announced programs include the Beechcraft King Air C90, 200, B200, and Cessna Citation V, Ultra, and Encore.

				CDU Type		Weight (lb.)	
Manufacturer	Model	Inputs	Outputs		Operational Capabilities	Dimensions	Price/Remarks
				Dimensions		Power Required	
Honeywell Aerospace BendixKing Avionics 9201 San Mateo Blvd. NE Albuquerque, NM 87113 (855) 250-7027 www.bendixking.com	Bendix King KSN 770/765 Integrated Navigator WAAS/ GPS/ NAV (KSN 770 only)/ COMM	RS-422 Inter- face; Weather Radar; Traffic; Terrain; EGPWS; XM Weather; Air data/ Heading Interface; Fuel Flow Air Data	AARINC 429 input; 10 ARINC 429 output; 2 R5-232 input; 4 R5-232 output; 4 R5-222 input; 3 R5-222 output; 28 discrete input; 20 discrete output	5.7 in.	WAAS, LPV. Can displays safety sys- tems informa- tion including On-board weather ra- dar, Enhanced Ground Proximity Warning Sys- tem (EGPWS), XM Datalink Weather, Terrain awarenese	8.5 lb.	Combines GPS navigation, Nav/ Com, terrain mapping, charting and safety sensor displays. Also displays XM Datalink Weather, radar-based weather, traffic and terrain. The KSN offers many ways of interfacing with your
	(KSN 770 only)/ MFD	Computer and others.		Active Matrix LCD	and warning System (TAWS) and Traffic Colli- sion Avoid- ance System (TCAS). Split-screen capabilities.	N/A	Information with a combination of hard buttons, cursor control and touchscreen.
Innovative Solutions & Support (IS&S) 70 Pennsylvania Dr.				AMLCD		7.0	Price varies by installation. Designed to replace existing instru- ments, including the EADI and
Exton, PA 19341 (610) 646-9800 Fax: (610) 646-0149	Cessna Citation AdViz Flat Panel Display	ARINC 429, A453/708, Eth- ernet, Descretes, Analog, Synchro, RS-422, CSDB, USB	ARINC 429, A453/708, Eth- ernet, Descretes, Analog, Synchro, RS-422, CSDB, USB		See remarks	NA	and vertical speed indicators. Retrofitting existing aircraft re- quires minimal changes to existing
				10.4 in.		NA	aircraft wiring while reducing power consumption and weight. Options include satellite weather, e-charts, video and remote radio control.
		ARINC 429, ARNC 453, RS 232, RS 42, Byteflite, Ether- net, discretes	ARINC 429, ARNC 453, RS 232, RS 42, Byteflite, USB, Eth- ernet, discretes	AMLCD		PFD: 8.5 MFD: 12.5	Price varies by installation. FMS options include either integrated WAAS-based FMS, exterior WAAS- based FMS errors WAAS-
	Eclipse Avio					PFD: 10.4 in. (2) MFD: 15.4 in.	FMS; system provides PFD/ND with MFD functions and engine in- struments; system interfaces with
	NG			PDF: 10.4 in. (2) MFD: 15.4 in.		PFD: 50 W MFD: 75 W	new or existing AP/FD/IFCS; TAWS display provided and connects directly with TAWS; remote tuned radios optional; e-charts, moving maps, radar display, satellite weather, TCAS-I, fuel management and aircraft systems pages.
				AMLCD		15 in. IPFD, 14 Ib., 70 W;	Price varies by installation. FMS options include either WAAS- based FMS, exterror WAAS-based
	Pilatus PC-12 FPDS	Contact OEM for	Contact OEM for		See remarks	10 in. IFPD, 8 lb., 35 W;	rws or non-wAAS-based FMS; systems provides PFD/ND with MFD functions; coupled WAAS LPV approach; system interfaces with
	System			10.4; 15.0		DCP, 3.0 lb., 8 W	new or existing AP/FD/IFCS; RVSM certified, options isnclude RS 170 or DVI video input on 5.15-in. IPFD; TAWS terrain display provided and connects directly with TAWS; e-charts certified.

Manufacturer	Model	Inputs	Outputs	CDU Type	Operational Capabilities	Weight (lb.) Dimensions	Price/Remarks
Innovative Solutions				Dimensions		Power Required	Drice verice by installation, FMC
As Support (IS&S) 70 Pennsylvania Dr. Exton, PA 19341 (610) 646-9800 Fax: (610) 646-0149 www.innovative-ss.com	Vantage Cockpit/IP Flat Panel Display System	Vantage Cockpit/IP Flat Panel Display System Contact OEM for details Contact OEM for details 10.4		See remarks	FPD: 6.0 lb., 30 watts RNCU: 9.75 lb., 25 watts; ECSU: 25 watts	Price varies by instaliation. FMS options include either WAAS-based FMS, exterror WAAS-based FMS or non-WAAS-based FMS; systems provides PFD/ND with MFD func- tions and engine instruments; sys- tem interfaces with new or existing AP/FD/IFCS; EVS input can be input fromEVS camera or other video camera via RS-170; TAWS terrain display provided and connect directly with TAWS; remote tuned radios optional. e-charts, moving maps (worldwide terrain 3-arc/second, radar display, satellite weather, TCAS-I/II, fuel management exceedance record- ing and video.	
Rockwell Collins 400 Collins Rd. NE Cedar Rapids, IA 52498 (319) 295-4085 Fax: (319) 295-2297 www.rockwellcollins. com	Pro Line	See Demorke	Sac Demoria	Color LCD	FMS, PFD/ Adapts to 3, 4 or 5 LCD graphic display	Various, depend-	Features include dual comm/nav, single, dual or triple FMS, GPS WAAS, single or dual integrated Flight Information system (IFIS), weather radar with turbulence de- tection, data link communication, onboard maintenance system, information management system, surface management, surveil- lance video, enhanced vision, sythethic vision, head-up guidance and functionality to meet Next Gen airspeace requirments. Display systems available with touch screen capability. Customized to OEM requirements. Price varies by installation.
	Fusion			15.1-in. color LCD SXGA: 14.1- in. color LCD WXGA	congiguration integrating PFD/MFD flight informa- tion	ing of instantion	
				Color LCD		Various, depend- ing on installation	Price varies by installation. The typical Pro Line 21 major retrofit package includes three-four 8 x 10 in. LCDs with advanced graphics, all digital CNS radios with dual comm/navs, dual transponders with enhanced surveillance dual
	Pro Line 21	Numerous	Numerous	6.375 (h) x 5.75 (w) x 6.33 (l)	FMS, PDF/ MFD, EFIS, TAWS, RMU, EVS, SVS pending		DME, single or dual FMS GPS WAAAS, Digital Flight Control System (DFCS) with coupled VNAV single or dual Integrated Flight Information Systems (IFIS), dual channel radar altimeter, dual solid-state Attitude Heading Refer ence Systems (AHRS), dual air data systems (RVSM compliant), solid-state radar with turbulence detection, Engine indications on PFD or MFD, 2nd or 3rd FMS, 3rd FMS, 3rd AHRS, 3rd VHF-4000, 2nd ALT-4000, TCAS 4000, ADS-F transponders, single or dual HF- 9000 radio, Satcom, CMU-4000 data link system, XM weather, maintenance diagnostics system, DBU-5000 data loader and all-ne wiring and connectors.

				CDU Type		Weight (lb.)		
Manufacturer	Model	Inputs	Outputs		Operational Capabilities	Dimensions	Price/Remarks	
				Dimensions	Capazintico	Power Required		
Rogerson Aircraft 2201 Alton Pkwy. Irvine, CA 92606 (949) 660-0666 www.rogersonaircraft. com	Series 700 Integrated Avionics System for Bell 412 and Bell 429; TC on Bell 429 and STC on Bell 412 using 6 x 8 ALMD displays	ARINC 429, Syn- chro, Discretes,	ARINC 429, vari- able DC, Discretes	Course Head- ing Select Panel (CHSP)	PFD, MFD, EICAS Mission func- tions: FLIR, RS-170 video, fuel and hydraulics	Each display unit:	Prices based on quantity; depen-	
		kGB, NISC, PAL video capability		6 x 8 ALMD displays			uent on engine type.	
	Series 600 Integrated Avionics	Series 600 Integrated Avionics ARINC 429, Syn-	ARINC 429, Syn-	Course Head- ing Select Panel (CHSP)	PFD, MFD Mission func- tions: FLIR.	Each display unit:		
	using 6 x 8 AMLCD displays	RGB, NTSC, PAL video capability	RGB, NTSC, PAL video capability	6 x 8 ALMD displays	RS-170 video, fuel and hydraulics	13.5	Prices based on quantity.	
Sandel Avionics 2401 Dogwood Way Vista, CA 92081 (877) 726-3357 (760) 727-4900 Fax: (760) 727-4899	Avilon	see capabilities	see capabilities	6 touchscreen displays*	ADS-B, ADC, AHRS, autopilot, audio, engine instruments, flight director, FMS, GPS, Mode S tran-	weight savings of 100-150 lb.	\$175,000 installed price. De- livered as a prewired assembly allowing for a five-day installation time. Initial STC for King Air 200 with additional models to follow. Designed for performance-based navigation with an emphasis on safety. *Existing panel is removed and replaced with Avilon.	
www.sandel.com				—	sponder, Nav, Com, TAWS, weather radar display			

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2017 Business Airplanes

Business jet operators are flying more than **4.3 million missions per year**, the highest since 2009 and even more than in 2008 prior to the Great Recession.

BY FRED GEORGE fred.george@penton.com

he U.S. economy has shown steady improvement as indicated by the 0.2%, 0.5% and 0.6% increases for November and December 2016 and January 2017, respectively, in the Conference Board Leading Economic Index, a composite measure of manufacturing activity, consumer and business demand for goods and services, stock prices and new building permits, among other factors. But you'd never know there was any improvement from looking at the general aviation market.

New aircraft sales revenues plunged by nearly \$5 billion in 2016 from one year earlier, according to the General Aviation Manufacturers Association (GAMA). Business jet deliveries fell from 718 units in 2015 to 661 units in 2016, the industry's lowest figure since 2004. Activity was strongest in North America and Europe, but a prolonged and pronounced slump in Latin America, Asia-Pacific, the Middle East and Africa dragged down total sales. North America and Europe accounted for more than 80% of turbofan deliveries and more than two-thirds of the turboprop deliveries.

GAMA reports that turboprops

fared slightly better than in the previous year, with a slight uptick in deliveries from 557 units in 2015 to 582 deliveries in 2016. North America, Asia-Pacific and Europe saw slight increases, while Latin America witnessed a minor decline. Overall, turbine aircraft deliveries have remained flat since 2009 and actually declined since 2013. More telling, turbine aircraft sales revenues fell nearly 15% in 2016 compared to the previous year.

Piston aircraft deliveries also fell by nearly 5% in 2016, although North America had a slight increase, accounting for nearly 70% of the sales.

Yet, the size of the world's turbofan and turboprop fleet increased slightly to 36,674 aircraft, according to GAMA citing data published by Jetnet LLC.

Sales and deliveries of new aircraft historically have tracked with global economic activity. But that's no longer the case in the business aircraft industry, says Rolland Vincent of his eponymous Aviation Consulting firm in Plano, Texas. His firm surveys 500 business aircraft owners and operators every 90 days.

In collaboration with Utica, New York-based Jetnet, Vincent publishes

quarterly history and forecast reports used for planning purposes by the business aircraft industry.

The Jetnet IQ report for first quarter 2017, for instance, says that 80% of North American respondents believe the economy there will grow faster in the next 12 months than in the previous year. More than 80% of North Americans believe the Donald Trump administration will be beneficial to aviation during the next year. And business jet operators are flying more than 4.3 million missions per year, the highest since 2009 and even more than in 2008 prior to the Great Recession.

Robert Stallard of Vertical Research Partners also notes that business aircraft operations grew at 2.9% in early 2017 year-over-year. For early 2016, year-over-year growth was only 1.1% versus 2015.

The economies of China and India should continue to expand, but the average GDP growth of 18 other nations, including the U.S., will hover near 2.0% in 2017, according to Vincent. These 20 nations account for most of the world's business aircraft.

Still, potential buyers are not rushing to new aircraft sales offices and asking for demo flights. In fact, Vincent projects that new turbofan aircraft deliveries will drop again this year to 640 units, accompanied by a slight decline in sales revenues. And he forecasts another 5.5% decrease to 605 units in 2018.

The reason? Oversupply. Book-to-bill ratios for Bombardier, Dassault, Embraer, Gulfstream and Textron all are below 1:1, meaning that the manufacturers are taking fewer orders for new equipment than the number of units they ship from their plants. Dassault, for example, had a book-to-bill ratio of less than 0.5 to 1 in 2015 and 2016.

Asking prices for turbofan aircraft are soft in 2017. Compare list prices in *BCA*'s *May 2016 Handbook* with prices this year. Most turbofans are priced the same as last year, though a few Falcon and Gulfstream models show modest increases. To increase competitiveness, Embraer dropped the Legacy 600 in favor of the new Legacy 650E that is priced \$5.7 million less than last year's Legacy 650. And Gulfstream dropped the G150 from its lineup due to low demand.

There also is a widening gap between list prices and sale prices. For instance, Vincent says Bombardier is selling some models at a 33% discount, forcing other manufacturers to sacrifice profit margins or lose sales. While the Canadian manufacturer garnered the largest number of business aircraft deliveries in 2016 among business jet makers, any such discounting would likely result in razor-thin margins.

Textron Aviation is faring better than most others. CEO Scott Ernest's capacity discipline resulted in the best book-to-bill ratio of any of the five jet makers from 2013 through 2016. But last year it still was hovering at slightly less than 1:1, according to Vincent, hardly a banner year for business jets.

This year, the FAA revised its general aviation fleet forecast, lowering growth of the general aviation fleet to 0.1% per year for the next two decades, with new turbine aircraft deliveries offsetting a projected contraction of the piston aircraft fleet, according to its Aerospace Forecast Report Fiscal Years 2017 to 2037. GAMA also notes that the general aviation pilot population is shrinking, although there was a slight uptick in student starts in 2015. While the general aviation fleet growth is lackluster, the FAA estimates that business jet operations will increase 3.0% from 2017 to 2037 in its latest forecast.

The report also says "there is uncertainty regarding the impact of the new U.S. administration's policies on economic growth." And with both U.S. Rep. Bill Shuster (R-Penn.), chairman of the House Transportation and Infrastructure Committee, and President Trump pushing to spin off FAA ATC into to a private corporation with a board of directors dominated by the airlines, business aircraft operators potentially could face substantial airspace and airport user fees.

On a more positive note, the FAA believes that the price for turbine fuel will increase only modestly in 2017 because the price of crude oil should stabilize at about \$47 per barrel, up from \$39 per barrel in 2016. Crude oil shouldn't again reach its 2013 price of \$100 per barrel until 2026, according to the FAA Forecast.

Regardless of the price of fuel or user fees, the FAA estimates that piston

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aircraft deliveries will continue to decline. In 2016, piston-engine aircraft deliveries from U.S. manufacturers were down 4.2% from 2015, according to GAMA. The FAA estimates that the piston aircraft fleet will atrophy at 0.8% per year from 2017 to 2036, due to "unfavorable pilot demographics, overall increased cost of aircraft ownership" and "new aircraft deliveries not keeping pace with retirements of the aging fleet."

Nonetheless, most piston aircraft manufacturers are hiking prices this

HTF7700L turbofans, features Garmin G5000 avionics and offers double-club seating for eight passengers.

Vincent foresees a sweet spot in the business jet market for 3,000-nm to 4,000-nm super-mids, such as the Longitude. Textron's new model could spur Bombardier, Embraer and Gulfstream to look at derivatives or new models in this segment. He's also bullish on the Falcon 5X because of its cabin size, range and fuel efficiency. But ongoing problems with its Snecma Silvercrest turbofans have slowed Dassault's de-



year. That includes Cirrus Aircraft, Piper and Textron Aviation, but Mooney, whose future seems uncertain, is holding 2016 pricing for its M20 models. Notably, GAMA reports Mooney delivered just seven aircraft in 2016, and there is very little activity at the factory in Kerrville, Texas. However, the M20U Ovation Ultra and M20V Acclaim Ultra, models featuring left- and right-side doors, received certification in March, and development of the diesel-powered models was still pending as we closed this issue.

Not all the news for 2017 is bad, however. This year, Textron Aviation's 3,500-nm range super-midsize Cessna CE-700 Citation Longitude makes its debut in the Purchase Planning Handbook. Due for certification late this year, the Longitude's evolutionary design combines a stretched and strengthened Citation Latitude fuselage mated to proven wing and empennage structures that were modified and adapted for the mission. The aircraft is powered by well-proven Honeywell velopment program by several years.

Gulfstream's 6,200-nm range, Mach 0.85 cruise G600 also is making its debut in this year's Handbook. A longer cabin, wider wingspan and longer-range derivative of the G500, it features active side-sticks, fly-by-wire (FBW) flight controls and Gulfstream's signature Symmetry flight deck. It's slated for certification late next year.

Bombardier's Global 7000 was due to make its debut in this year's *Handbook*. But the manufacturer declined to release performance details despite having two aircraft in flight testing. A third test aircraft, slated for first flight later this year, should be fully production conforming, Vincent believes. Look for the Global 7000 to appear in the 2018 *Handbook*.

The single-engine turboprop sector also remains stable to strong. Epic, Piper, Mahindra, Quest and Textron held prices unchanged or close to 2016 levels. Epic Aircraft is making changes to the E1000 to ensure it complies with upcoming certification requirements. Daher is replacing the TBM 900 with the TBM 910, a derivative upgraded with Garmin G1000 NXi avionics and other modifications. Both Daher and Pilatus increased prices in response to strong order books.

While most new piston and turbofan aircraft deliveries remain stubbornly stagnant, several developments are buoying spirits in the business aircraft industry. The European Aviation Safety Agency (EASA) issued final regulations permitting commercial singleengine turbine aircraft operations in instrument meteorological conditions (IMC). Notably, Europe is the last large business aircraft market that, with few exceptions, did not permit commercial single-engine operations in IMC.

After seven years, the 36-state International Civil Aviation Organization (ICAO) council adopted uniform CO₂ emission standards for aircraft. Such standardization facilitates creation of market-based measures to move toward carbon-neutral growth of aircraft operations by 2020. Reduction in CO₂ will be made possible by more-efficient air traffic management, use of sustainable alternative fuels, replanting rain forests and developing more-fuel-efficient aircraft.

The FAA also continues to progress through Phase II of its Piston Aviation Fuels Initiative by developing a drop-in replacement unleaded avgas by 2018. Shell Oil and Swift Fuels have been selected to partner with the FAA to develop ASTM standards for unleaded avgas that will have the least technical and financial impact on general aviation aircraft operators and establish a fuel distribution infrastructure. However, it's still not clear how much the price of that fuel will change from the cost per gallon of 100LL gasoline.

So, in the short term, look for singleengine and multiengine turboprops to be solid sellers. The piston-engine market is in for a rough ride because of aging pilot demographics, increasing direct operating costs and tougher local airport authority rules, regulations and restrictions, particularly in California. The turbofan aircraft market will remain relatively flat because of oversupply in almost all segments. But a new generation of roomy, fuel-efficient and fast U.S. transcontinental-range and transatlantic-range super-midsize to large-cabin aircraft hold the promise to lift the turbofan sector out of its doldrums. BCA

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Purchase Planning Handbook

How to Use the Airplane Charts



or an aircraft to be listed in the Purchase Planning Handbook, a production conforming article must have flown by May 1 of this year. The dimensions, weights and performance characteristics of each model listed are representative of the current production aircraft being built or for which a type certificate application has been filed. The basic operating weights we publish should be representative of actual production turboprop and turbofan aircraft because we ask manufacturers to supply us with the average weights of the last 10 commercial aircraft that have been delivered. However, spot checks of some manufacturers' BOW numbers reveal anomalies. We reserve the right to make adjustments to weights, dimensions and performance data. These data adjustments will be noted in the Remarks section for specific models as "BCA Estimated Data."

The takeoff field length distances are based on maximum takeoff weight for maximum range missions.

Please note that "all data preliminary" in the Remarks section indicates that actual aircraft weight, dimension and performance numbers may vary considerably after the model is certified and delivery of completed aircraft begins. ***All data for these aircraft is highlighted with a blue tint.***

Manufacturer, Model and Type Designation

In some cases, the airplane manufacturer's name is abbreviated. The model name and the type designation also are included in this group.

BCA Equipped Price

Price estimates are first quarter, current year dollars for the next available delivery. Some aircraft have long lead times, thus the actual price will be higher than our published price because of block point changes and inflation adjustments. Note well, manufacturers may change prices without notification.
Piston-powered airplanes – Computed retail price with at least the level of equipment specified in the "BCA Required Equipment List."

► Turbine-powered airplanes — Computed retail price with at least the level of equipment specified in the "BCA Required Equipment List," if available. Some manufacturers decline to provide us with actual prices of delivered aircraft, so we may estimate them. The aircraft serial numbers aren't necessarily consecutive because of variations in completion time and because some aircraft may be configured for non-commercial, special missions.

Characteristics

► Seating: Crew + Typical Executive Seating/High-Density Seating/Max Certification Seating — For example. 2+8/13/19 indicates that the aircraft requires two pilots, there are eight seats in the typical executive configuration, 13 seats with optional high-density seating and up to 19 passenger seats based upon FAA and/or EASA certification limits. A four-place single-engine aircraft is shown as 1+3/3, indicating that one pilot is required and there are three other seats available for passengers. We require two pilots for all turbofan airplanes, except for single-pilot certified aircraft such as the Cirrus Vision SF-50, Eclipse 550, Cessna Citation CJ series, HondaJet and Syberiet SJ30-2, which have, or will have, a large percentage of single-pilot operators. Four crewmembers are specified for ultra-long-range aircraft — three pilots and one flight attendant. However, Dassault only provides data with three crewmembers aboard for its ultra-long-range aircraft, thus the notations for the Falcon 8X.

Each occupant of a turbine-powered airplane is assumed to weigh 200 lb., thereby allowing for stowed luggage and carry-on items. In the case of pistonengine airplanes, we assume each occupant weighs 170 lb. There is no luggage allowance for piston-engine airplanes. Wing Loading – MTOW divided by to-

tal wing area.

▶ Power Loading – MTOW divided by total rated takeoff horsepower or total rated takeoff thrust.

► FAR Part 36 Certified Noise Levels – Flyover noise in A-weighted decibels (dBA) for small and turboprop aircraft. For turbofan-powered aircraft, we provide Part 36 EPNdB (effective perceived noise levels) for Lateral, Flyover and Approach.

Dimensions

External Length, Height and Span dimensions are provided for use in determining hangar and/or tie-down space requirements.

Internal Length, Height and Width are based on a completed interior, including insulation, upholstery, carpet, carpet padding and fixtures. Note well: These dimensions are not intended to be



based upon green aircraft dimensions. They must reflect the actual net dimensions with all soft goods installed. Some manufacturers provide optimistic measurements, thus prospective buyers are advised to measure aircraft themselves.

As shown in the Cabin Interior Dimensions illustration, for small airplanes other than "cabin-class" models, the length is measured from the forward bulkhead ahead of the rudder pedals to the back of the rear-most passenger seat in its normal, upright position. The upright position of the aft seat backs allows room for luggage in the cabin.

For so-called cabin-class and larger aircraft, we show two or three dimensions, depending on aircraft class. The first is the overall length of the passenger cabin, measured from the aft side of the forward cockpit/cabin divider to the aft-most bulkhead of the cabin. The aft-most point is defined by the rear side of a baggage compartment that is accessible to passengers in flight or the aft pressure bulkhead. The overall length is reduced by the length of any permanent mounted system or structure that is installed in the fuselage ahead of the aft bulkhead. For example, some aircraft have full fuselage cross-section fuel tanks mounted ahead of the aft pressure bulkhead.

The second length number is the net length of the cabin that routinely is occupied by passengers. It's measured from the aft side of the forward cockpit/ cabin divider to an aft point defined by the rear of the cabin floor capable of supporting passenger seats, the rear wall of an aft galley or lavatory, an auxiliary pressure bulkhead or the front wall of the pressurized baggage compartment. Some aircraft have the same net and overall interior length because the manufacturer offers at least one interior configuration with the aft-most passenger seat located next to the front wall of the aft luggage compartment.

The third length dimension is the main seating area of the cabin, including all passenger seats in the standard aircraft configuration that are certified for full-time occupancy. Some manufacturers may fit their aircraft with forward, side-facing divans, ahead of areas with individual fore-aft facing chairs. The main seating length dimension may include such forward cabin side-facing divans at the discretion of the manufacturer. The length of the lavatory, even though it may have a seat certified for full-time occupancy, may not be included in the main seating length dimension.

Interior height is measured at the center of the cabin cross-section. If the aircraft has a dropped aisle, the maximum depth below the adjacent cabin floor is shown. Some aircraft have dropped aisles of varying depths, resulting in less available interior net height in certain sections of the cabin.

Two width dimensions are shown for multiengine turbine airplanes — one at the widest part of the cabin and the other at floor level. The dimensions, however, are not completely indicative of the usable space in a specific aircraft because of individual variances in interior furnishings.

Power

Number of engines, if greater than one, and the abbreviated name of the manufacturer: GE — General Electric; GE/ Honda — General Electric and Honda; Honeywell; CFMI — CFM International; IAE — International Aero Engines; Lyc — Textron Lycoming; P&WC — Pratt

Purchase Planning Handbook

& Whitney Canada; RR — Rolls-Royce; Snecma; TCM — Teledyne Continental; and Wms — Williams International.

▶ **Output** – Takeoff rated horsepower for propeller-driven aircraft or pounds thrust for turbofan aircraft. If an engine is flat rated, enabling it to produce takeoff rated output at a higher than ISA (standard day) ambient temperature, the flat rating limit is shown as ISA+XXC. Highly flat-rated engines, i.e. engines that can produce takeoff rated thrust at a much higher than standard ambient temperature, typically provide substantially improved high density altitude, climb and high-altitude cruise performance.

▶ Inspection Interval is the longest scheduled hourly major maintenance interval for the engine, either "t" for TBO or "c" for compressor zone inspection. In some fuel required to fly 1.5 hr. at high-speed cruise.

▶ Max ramp, max takeoff and max landing weights may be the same for light aircraft that may only have a certified max takeoff weight.

▶ EOW/BOW – Empty Operating Weight is shown for piston-powered airplanes. EOW is based on the factory standard weight, plus items specified in the "BCA Required Equipment List," less fuel, loose equipment and cabin stores.

Basic Operating Weight is shown for turbine-powered airplanes. BOW is based on the average EOW weight of the last 10 commercial deliveries, plus 200 lb. for each required crewmember. Three flight crewmembers and one cabin crewmember are required for ultra-long-range aircraft, unless otherwise noted.



AIRBUS CORPORATE JETS

cases, we show a second number if the engine manufacturer has obtained an extended maintenance interval, provided that the engines are enrolled in the manufacturer's service program. OC is shown only for engines that have "on condition" repair or replace parts maintenance.

Weights (lb.)

Weight categories are listed as appropriate to each class of aircraft.

Max Ramp – Maximum ramp weight for taxi.

► Max Takeoff - Maximum takeoff weight as determined by structural limits.

► Max Landing — Maximum landing weight as determined by structural limits.

► **Zero Fuel** – Maximum zero fuel weiht, shown by "c," indicating the certified MZFW or "b," a *BCA*-computed weight based on MTOW minus the weight of While there is no requirement to add in the weight of cabin stores, some manufacturers choose to include galley stores and passenger supplies as part of the BOW build-up. Life vests, life rafts and appropriate deep-water survival equipment are included in the weight buildup of the 80,000+ lb., ultra-longrange aircraft.

▶ Max Payload – Zero Fuel weight minus EOW or BOW, as appropriate. For piston-engine airplanes, Max Payload frequently is a computed value because it is based on the *BCA* ("b") computed maximum ZFW.

▶ Max Fuel – Usable fuel weight based on 6.0 lb. per U.S. gallon for avgas or 6.7 lb. per U.S. gallon for jet fuel. Fuel quantity is based upon the largest capacity tanks that are available as standard equipment.

► Available Payload With Max Fuel – Max Ramp weight minus the tanks-full weight, not to exceed Zero Fuel weight minus EOW or BOW. ► Available Fuel With Max Payload – Max Ramp weight minus Zero Fuel weight, not to exceed maximum fuel capacity.

Limits

BCA lists V speeds and other limits as appropriate to the class of airplane. These are the abbreviations used on the charts:

► VNE - Never exceed speed (redline for piston-engine airplanes).

VNO – Normal operating speed (top of the green arc for piston-engine airplanes).

VMO – Maximum operating speed (redline for turbine-powered airplanes).

MM0 – Maximum operating Mach number (redline for turbofan-powered airplanes and a few turboprop airplanes).

FL/VMO – Transition altitude at which VMO equals MMO (large turboprop and turbofan aircraft).

► VA — Maneuvering speed (except for certain large turboprop and all turbofan aircraft).

▶ **VDEC** – Accelerate/stop decision speed (multiengine piston and light multiengine turboprop airplanes).

VMCA – Minimum control airspeed, airborne (multiengine piston and light multiengine turboprop airplanes).

Vso – Maximum stalling speed, landing configuration (single-engine airplanes).

Vx – Best angle-of-climb speed (single-engine airplanes).

▶ VXSE – Best angle-of-climb speed, oneengine inoperative (multiengine piston and multiengine turboprop airplanes under 12,500 lb.).

► VY - Best rate-of-climb speed (singleengine airplanes).

VYSE – Best rate-of-climb speed, oneengine inoperative (multiengine piston and multiengine turboprop airplanes under 12,500 lb.).

► V2 — Takeoff safety speed (large turboprops and turbofan airplanes).

► VREF – Reference landing approach speed (large turboprops and turbofan airplanes, four passengers, NBAA IFR reserves; eight passengers for ultralong-range aircraft).

PSI – Cabin pressure differential (all pressurized airplanes).

Airport Performance

Airplane Flight Manual takeoff runway performance is shown for sea level, standard day and for 5,000-ft. elevation/25C day density altitude. All-engine takeoff distance (TO) is shown for single-engine and multiengine piston, and turboprop airplanes with an MTOW of less than 12,500 lb. Takeoff distances and speeds assume MTOW, unless otherwise noted.

► Accelerate/Stop distance (A/S) is shown for small multiengine piston and small turboprop airplanes.

► Takeoff Field Length (TOFL), the greater of the one-engine inoperative (OEI) takeoff distance or the accelerate/stop distance, is shown for FAR Part 23 Commuter Category and FAR Part 25 airplanes. If the accelerate/stop and accelerate/stop distances are equal, the TOFL is the balanced field length.

► Landing distance (LD) is shown for FAR Part 23 Commuter Category and FAR Part 25 Transport Category airplanes. The landing weight is BOW plus four passengers and NBAA IFR fuel reserves. We assume that 80,000+ lb. ultra-long-range aircraft will have eight passengers on board.

▶ V2 and VREF speeds are useful for reference when comparing the TOFL and LD numbers because they provide an indication of potential minimum-length runway performance when low RCR or runway gradient is a factor.

BCA lists two additional warm day airport performance numbers for large turboprop- and turbofan-powered airplanes. First, we publish the Mission Weight, which is the maximum allowable takeoff weight when departing a 5,000-ft. elevation/ISA+20C airport with at least four passengers aboard.

Mission Weight, when departing from a 5,000-ft./ISA+20C airport, may be less than the MTOW at sea level on a standard day because of FAR Part 25 second-segment, one-engine-inoperative, climb performance requirements. If maximum allowable mission weight at takeoff is restricted under said conditions, it's flagged with a "p." Aircraft with highly flat-rated engines are less likely to have a performance limited mission weight when departing under said warm day conditions.

Second, we publish the NBAA IFR range for said warm day conditions, assuming a transition into standardday, ISA flight conditions after takeoff. For purposes of computing NBAA IFR range, the aircraft is flown at the long-range cruise speed shown in the "Cruise" block or at the same speed as shown in the "Range" block. Notably, some aircraft may actually have slightly better range performance when departing from said warm day airport because



they have a 5,000-ft. head start on the climb to cruise altitude.

Climb

The all-engine time to climb provides an indication of overall climb performance, especially if the aircraft has an all-engine service ceiling well above our sample time-to-climb altitudes. We provide the all-engine time to climb to one of three specific altitudes, based on type of aircraft departing at MTOW from a sea-level, standard-day airport: (1) FL 100 (10,000 ft.) for normally aspirated single-engine and multiengine piston aircraft, plus pressurized singleengine piston aircraft and unpressurized turboprop aircraft; (2) FL 250 for pressurized single-engine and multiengine turboprop aircraft; or (3) FL 370 for turbofan-powered aircraft. These data are published as time-to-climb in minutes/climb altitude. For example, if a non-pressurized twin-engine piston aircraft can depart from a sea-level airport at MTOW and climb to 10,000 ft. in 8 min., the time to climb is expressed as 8/FL 100.

We also publish the initial all-engine climb feet per nautical mile gradient, plus initial engine-out climb rate and gradient, for single-engine and multiengine pistons and turboprops with MTOWs of 12,500 lb. or less.

The one-engine-inoperative (OEI) climb rate for multiengine aircraft at MTOW is derived from the Airplane Flight Manual. OEI climb rate and gradient are based on landing gear retracted and wing flaps in the takeoff configuration used to compute the published takeoff distance. The climb gradient for such airplanes is obtained by dividing the product of the climb rate (fpm) in the Airplane Flight Manual times 60 by the VY or VYSE climb speed, as appropriate.

The OEI climb gradients we show for FAR Part 23 Commuter Category and FAR Part 25 Transport Category aircraft are the second-segment net climb performance numbers published in the AFMs. Please note: The AFM net second-segment climb performance numbers are adjusted downward by 0.8% to compensate for variations in pilot technique and ambient conditions.

The OEI climb gradient is computed at the same flap configuration used to calculate the takeoff field length.

Ceilings (ft.)

► Maximum Certificated Altitude – Maximum allowable operating altitude



FAR Part 25 and Part 23 Commuter Category OEI Climb Performance

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determined by airworthiness authorities. ► All-Engine Service Ceiling – For turbofan aircraft: maximum altitude at which at least a 300-fpm rate of climb can be attained, assuming the aircraft departed a sea-level, standard-day airport at MTOW and climbed directly to altitude. For piston and turboprop aircraft: 100 fpm rate of climb.

▶ OEI (Engine Out) Service Ceiling

► Sea-Level Cabin (SLC) Altitude — Maximum cruise altitude at which a 14.7-psia, sea-level cabin altitude can be maintained in a pressurized airplane.

Cruise

Cruise performance is computed using EOW with four occupants or BOW with four passengers and one-half fuel load. Ultra-long-range aircraft carry eight passengers for purposes of computing cruise performance.

Assume 170 lb. for each occupant of a piston-engine airplane and 200 lb. for each occupant of a turbine-powered aircraft.

▶ Long Range — True air speed (TAS), fuel flow in pounds/hour, flight level (FL) cruise altitude and specific range for longrange cruise by the manufacturer.

▶ **Recommended (Piston-Engine Airplanes)** – TAS, fuel flow in pounds/hour, FL cruise altitude and specific range for normal cruise performance specified by the manufacturer.

► High Speed - TAS, fuel flow in pounds/hour, FL cruise altitude and specific range for short-range, high-speed performance specified by the air-craft manufacturer.

Speed, fuel flow, specific range and altitude in each category are based on one mid-weight cruise point and these data reflect standard-day conditions. They are not an average for the overall mission and they are not representative of the above standard-day temperatures at cruise altitudes commonly encountered in everyday operations.

BCA imposes a 12,000-ft. maximum cabin altitude requirement on CAR3/ FAR Part 23 normally aspirated aircraft. Non-pressurized turbocharged piston-engine airplanes are limited to FL 250, providing they are fitted with supplemental oxygen systems having sufficient capacity for all occupants for the entire duration of the mission. Pressurized CAR3/FAR Part 23 aircraft are limited to a maximum cabin altitude of 10,000 ft. For FAR Part 25 aircraft, the maximum cabin altitude for computing cruise performance is 8,000 ft.

To conserve space, we use flight levels (FL) for all cruise altitudes, which is appropriate considering that we assume standard-day ambient temperature and pressure conditions. Cruise performance is subject to *BCA*'s verification.

Range

BCA shows various paper missions for each aircraft that illustrate range versus payload tradeoffs, runway and cruise performance, plus fuel efficiency. Similar to the cruise profile calculations, *BCA* limits the maximum altitude to 12,000 ft. for normally aspirated, non-pressurized CAR3/FAR Part 23 aircraft, 25,000 ft. for turbocharged non-pressurized airplanes with supplemental oxygen, 10,000 ft. cabin altitude for pressurized CAR 3/FAR Part 23 airplanes and 8,000 ft. cabin altitude for FAR Part 23 Commuter Category or FAR Part 25 aircraft.

► Seats-Full Range (Single-Engine Piston Airplanes) — Based on typical executive configuration with all seats filled with 170-lb. occupants, with maximum available fuel less 45-min. IFR fuel reserves. We use the lower of seats full or maximum payload.

► Tanks-Full Range (Single-Engine Piston Airplanes) — Based on one 170-lb. pilot, full fuel less 45-min. IFR fuel reserves.

► Max Fuel With Available Payload (Single-Engine Turboprops) — Based on BOW, plus full fuel and the maximum available payload up to maximum ramp weight. Range is based on arriving at destination with NBAA IFR fuel reserves, but only a 100-mi. alternate is required.

▶ Ferry (Multiengine Piston Airplanes and Single-Engine Turboprops) — Based on one 170-lb. pilot, maximum fuel less 45-min. IFR fuel reserves.

Please note: None of the missions for piston-engine aircraft includes fuel for diverting to an alternate. However, single-engine turboprops are required to have NBAA IFR fuel reserves, but only a 100-mi. alternate is required.

NBAA IFR range format cruise profiles, having a 200-mi. alternate, are used for turbine-powered aircraft with MTOWs equal to, or greater than, 22,000 lb. Turbine aircraft having MTOWs less than 22,000 lb. only need a 100-mi. NBAA alternate. The difference in alternate requirements should be kept in mind when comparing range performance of various classes of aircraft.

► Available Fuel With Max Payload (Multiengine Turbine Airplanes) — Based on aircraft loaded to maximum zero fuel weight with maximum available fuel up to maximum ramp weight, less NBAA IFR fuel reserves at destination.

► Available Payload With Max Fuel (Multiengine Turbine Airplanes) — Based on BOW plus full fuel and maximum available payload up to maximum ramp weight. Range based on NBAA IFR reserves at destination.

► Full/Max Fuel With Four Passengers (Multiengine Turbine Airplanes) – Based on BOW plus four 200-lb. passengers and the lesser of full fuel or maximum available fuel up to maximum ramp

NBAA IFR RANGE PROFILE



weight. Ultra-long-range aircraft must have eight passengers on board.

► Ferry (Multiengine Turbine Airplanes) — Based on BOW, required crew and full fuel, arriving at destination with NBAA IFR fuel reserves.

We allow 2,000-ft. increment step climbs above the initial cruise altitude to improve specific range performance, even though current air traffic rules in North America provide for 4,000-ft. altitude semicircular directional traffic separation above FL 290. The altitude shown in the range section is the highest cruise altitude for the trip — not the initial cruise or mid-mission altitude.

The range profiles are in nautical miles, and the average speed is computed by dividing that distance by the total flight time or weight-off-wheels time en route. The Fuel Used or Trip Fuel includes the fuel consumed for start, taxi, takeoff, cruise, descent and landing approach but not after-landing taxi or reserves.

The Specific Range is obtained by



dividing the distance flown by the total fuel burn. The Altitude is the highest cruise altitude achieved on the specific mission profile shown.

Missions

Various paper missions are computed to illustrate the runway requirements, speeds, fuel burns and specific range, plus cruise altitudes. The mission ranges are chosen to be representative for the airplane category. All fixeddistance missions are flown with four passengers on board, except for ultra-long-range airplanes, which have eight passengers on board. The pilot is counted as a passenger on board piston-engine airplanes. If an airplane cannot complete a specific fixed distance mission with the appropriate payload, *BCA* shows a reduction of payload in the remarks section or marks the fields NP (Not Possible) at our option.



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POWERPLANT SYSTEMS										
Engine synchronization						•	•	•	•	•
Fire detection, each engine						٠		٠	٠	٠
Fire extinguishing, each engine			_		_		•	-	•	•
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Thrust reversers										
AVIONICS										
Altitude alerter						•	•	•	•	•
Altitude encoder	•	•	٠	•	٠	•	٠	•	•	•
Audio control panel	•	-		-	•		•		•	
Automatic flight guidance, 3-axis, alt hold		-		-				•		
Digital air data computer										
DME or approved GPS distance indication	•		•		•	-	•	•	•	
ELT	•		•		•	•	•	÷	•	•
FMS (TSO C115) or GPS (TSO C129/145/146)	•	٠	•	٠	٠	٠	۲	•	۲	•
Marker beacon receiver	•	•	•	•	•	-	•	•	•	-
RVSM certification			_			•	•	•	•	•
Satcom, Iridium, or Inmarsat									٠	٠
TAS or TCAS I					_	-	•	-		
TCAS I/II						•	•	•	•	•
Transponder, Mode S 1090ES	•	•	•	•	•	•	•			
VHF comm transceiver, 25-KHz spacing	•	•	•	•	•	•	•	•		
VOR/ILS									Ŏ	•
Weather data link										
GENERAL										
Air conditioning, vapor cycle (not required with APU)				٠		٠	٠	٠	٠	٠
Anti-skid brakes (not required MTOW <10,000 lb.)					_			•	•	-
Cabin/cockpit bulkhead divider	-						-	•	•	÷
Corrosion-proofing	•	•	•	•	•	•	•	•	•	•
Exterior paint, tinted windows	•		•	•	•				•	
Fire extinguisher, cockpit		٠				•	•	•	•	•
Fuel tanks, long-range	•	•	•	۲	•	~		~		~
Ground power jack										
Lavatory		-		-		-		•	•	•
Lights, external — nav/beacon/strobe/landing/taxi	•	•	•	•	•	•	•	•	•	•
Lights, internally illuminated instrument/cockpit flood	•		•	•	•		•			
Refreshment center		-				-	•	•	•	•
Seats, crew, articulating	•	•	•	•	•	•	•	•	•	•
Seats, passenger, reclining Shoulder harness, all seats/crew with inertial reel										
Tables, cabin work		-						•	•	ě
ICE AND RAIN PROTECTION										
Alternate static pressure source (not required with dual DADC)								•		•
Ice protection plates						_	•	•		_
Pitot heat			•				•	•	•	
Windshield rain removal, mechanical/pneumatic/hygroscopic										
Angle-of-attack stall margin indicator									٠	٠
EGT	•	•	•	•	•	-		-		-
OAT										
Primary flight instruments	•	•	•	•	•	•	•	•	•	•
Dequired										

Runway performance is obtained from the Approved Airplane Flight Manual. Takeoff distance is listed for single-engine airplanes; accelerate/ stop distance is listed for piston twins and light turboprops; and takeoff field length, which often corresponds to balanced field length, is used for FAR Part 23 Commuter Category and FAR Part 25 large Transport Category airplanes.

Flight Time (takeoff to touchdown, or weight-off-wheels, time) is shown for turbine airplanes. Some piston-engine manufacturers also include taxi time, resulting in a chock-to-chock, Block Time measurement. Fuel Used, though, is the actual block fuel burn for each type of aircraft, but it does not include fuel reserves. The cruise altitude shown is that which is specified by the manufacturer for fixed-distance missions.

▶ 200 nm – (Piston-engine airplanes).

500 nm – (Piston-engine airplanes).

▶ 300 nm — (Turbine-engine airplanes, except ultra-long-range).

▶ 600 nm – (Turbine-engine airplanes, except ultra-long-range).

▶ 1,000 nm - (All turbine-engine airplanes).

▶ **3,000 nm** – (Ultra-long-range turbine-engine airplanes).

▶ 6,000 nm – (Ultra-long-range turbine-engine airplanes).

Remarks

In this section, *BCA* generally includes the base price, if it is available or applicable; the certification basis and year; and any notes about estimations, limitations or qualifications regarding specifications, performance or price. All prices are in 2017 dollars, FOB at a U.S. delivery point, unless otherwise noted. The certification basis includes the regulation under which the airplane was originally type certified, the year in which it was originally certified and, if applicable, subsequent years during which the airplane was re-certified. "BCA Estimated Data" indicates that we made adjustments to data provided by manufacturers.

General

The following abbreviations are used throughout the tables: **"NA**" means not available; **"—**" indicates the information is not applicable and **"NP**" signifies that specific performance is not possible. **BCA**

Single-Engine Pistons normally aspirated

Manufacture	r		Cirrus Design	Piper	Textron Aviation	Cirrus Design
Model			SR20	Arrow	Cessna Skylane	SR22
BCA Equipped	d Price		\$389 900	\$466.880	\$480,000	\$539,900
DUA LQUIPPEC		Seating	1+3/4	1+3/3	1+3/3	1+3/4
Character		Wing Loading	21.7	16.2	17.8	23.5
istics		Power Loading	14.65	13.75	13.48	11.61
13003		Noice (dPA)	92.4	77.7	77.7	927
External		longth	26.0	24.7	29.0	26.0
Dimensions		Heidht	8.9	7.9	9.3	8.9
(ft)		neight Soon	28.2	25.4	3.3	29.2
Internel		Length	8.0	77	7.2	8.0
Dimensions		Heidht	4.1	3.7	1.2	4.1
(ft)		Width	4.1	3.5	3.5	4.1
(10.)						Cont
_		Engine	IO-390-C3B6	IO-360-C1C6	IO-540-AB1A5	10-550-N
Power		Output (hp)	215	200	230	310
		Inspection Interval	2,000t	2,000t	2,000t	2,000t
		Max Ramp	3,160	2,758	3,110	3,610
		Max Takeoff	3,150	2,750	3,100	3,600
		Max Landing	3,150	2,750	2,950	3,600
		Zero Fuel	3,043b	2,636b	2,976b	3,400c
		EOW	2,120	1,798	1,965	2,260
Weights (lb.)		Max Payload	923	838	1,011	1,140
		Useful Load	1,040	960	1,145	1,350
		Max Baggage	130	200	200	130
		Max Fuel	336	432	522	552
	Ava	ailable Payload w/Max Fuel	704	528	623	798
	Ava	ailable Fuel w/Max Payload	117	122	135	210
		VNE	201	183	175	205
Limits		VNO	164	146	140	176
		VA	133	118	110	140
	TO (SL elev./ISA temp.)		2,530	1,600	1,514	1,756
Airport		TO (5,000-ft. elev.@25C)	4,305	3,250	2,708	3,016
Perfor-		Vso	62	55	49	64
mance	mance		81	78	65	88
		Vy	88	90	80	108
Climb	Ti	me to Climb (min.)/Altitude	20/FL 100	16/FL 100	15/FL 100	11/FL 100
		Initial Gradient (ft./nm)	540	560	694	775
Ceiling (ft.)		Service	17,500	16,200	18,100	17,500
		TAS	135	124	125	160
	Long Range	Fuel Flow	53	51	61	68
		Altitude	FL 080	FL 100	FL 100	FL 080
		Specific Range	2.547	2.431	2.049	2.353
		IAS	145	130	135	1/1
Cruise	Recommended	Fuel Flow	61	68	69	92
		Altitude	FL 080	FL 090	FL 100	FL 080
		Specific Range	2.309	1.912	1.957	1.859
		IA3 Eucl Flow	71	76	22	107
	High Speed	Altitudo	FL 080	FL 060	55 EL 060	EL 080
		Specific Pande	2 129	1 803	1 735	1.682
		Nautical Miles	672	537	795	1 118
		Average Sneed	135	121	131	162
	Seats Full	Fuel Lised	275	156	414	492
		Specific Range/Altitude	2 444/FL 080	3 442/FL 070	1 920/FL 120	2 272/FL 080
Ranges		Nautical Miles	672	926	912	1.118
		Average Speed	135	121	131	162
	Tanks Full	Fuel Used	275	408	471	492
		Specific Range/Altitude	2.444/FL 080	2.270/FL 070	1.936/FL 120	2.272/FL 080
		Runway	1,685	1,600	1,216	1,303
		Block Time	1+26	1+29	1+37	1+09
	200 nm	Fuel Used	112	125	123	127
Missions		Specific Range/Altitude	1.786/FL 080	1.600/FL 070	1.626/FL 120	1.575/FL 080
(4 occupants)		Runway	1,685	1,600	1,369	1,519
	500	Block Time	3+30	3+50	3+52	2+49
	500 nm	Fuel Used	245	278	269	305
		Specific Range/Altitude	2.041/FL 080	1.799/FL 090	1.859/FL 120	1.639/FL 080
		Suggested Base Price	\$389,900	\$466,880	\$480,000	\$539,900
Remarks		Certification Basis	FAR 23, 1999/2017 Includes Garmin Perspective+ avionics.	CAR 3, 1976/2001 Garmin G500 standard.	FAR 23, 1996/2001 A 23-6 Garmin G1000 NXi with GFC 700 autopilot.	FAR 23, 2000 Includes Garmin Perspective+ avionics.

Single-Engine Pistons normally aspirated

Model Ovation Ultra M2OU Ovation Ultra M2OU BCA Equipped Price \$689,000 Seating 1+3/4 Character- Wing Loading 19.3	Airvan GA-8 \$726,960 1+6/7 20.7 13.33 84.0	Beechcraft Bonanza G36 G36 \$815,000 1+4/5 20.2
BCA Equipped Price \$689,000 Seating 1+3/4 Character- Wing Loading 19.3	\$726,960 1+6/7 20.7 13.33	\$815,000 1+4/5 20.2
Seating 1+3/4 Character- Wing Loading 19.3	1+6/7 20.7 13.33	1+4/5 20.2
Character-Wing Loading 19.3	20.7 13.33	20.2
	13.33	
ISLICS Power Loading 10.86	84.0	12.17
Noise (dBA) NA	04.9	76.7
External Length 26.9	29.3	27.5
Dimensions Height 8.3	12.8	8.6
(ft.) Span 36.1	40.7	33.5
Internal Length 8.1	11.6	12.6
Dimensions Height 3.7	3.7	4.2
(ft.) Width 3.6	4.2	3.5
Engine Cont	Lyc 10-540-K145	Cont
Power Output (hp) 310	300	300
Inspection Interval 2,200t	2.000t	1,900t
Max Ramp 3,374	4,014	3,663
Max Takeoff 3,368	4,000	3,650
Max Landing 3,200	4,000	3,650
Zero Fuel 3,197b	3,849b	3,509b
EOW 2,260	2,241	2,600
Weights (lb.) Max Payload 937	1,608	909
Useful Load 1,114	1,773	1,063
Max Baggage 120	180	670
Max Fuel 600	540	444
Available Payload W/ Max Fuel 514	1,233	154
	185	203
	143	165
Va 127	121	139
T0 (SL elev./ISA temp.) 2,300	1,860	1,913
Airport T0 (5,000-ft. elev.@25C) 3,400	3,670	3,450
Perfor- Vso 59	57	59
mance Vx 75	70	84
W 105	86	100
Climb Time to Climb (min.)/Altitude 10/FL 100	15/FL 100	14/FL 100
Initial Gradient (tt./nm) NA	/8/	/30
	20,000	160
Fuel Flow 50	78	71
Long Range Altitude FL 120	FL 120	FL 080
Specific Range 3.260	1.628	2.254
TAS 186	135	167
Fuel Flow 84	88	86
Altitude FL 121	FL 080	FL 080
Specific Range 2.214	1.534	1.942
TAS 196	142	174
High Speed Altitude Floor	101	94
Allulue FL 080	1 406	1 851
Nautical Miles 1.075	487	217
Average Speed 161	124	153
Seats Full Fuel Used 438	339	115
Ranges Specific Range/Altitude 2.454/FL 121	1.437/FL 120	1.887/FL 040
Nautical Miles 1,465	690	859
Average Speed 173	125	159
Fuel Used 558	464	403
Specific Range/Altitude 2.625/FL 121	1.487/FL 120	2.132/FL 080
Kunway 1,230	1,860	1,664
200 nm Eval Leed 115	157	130
Missions Snecific Range/Altitude 1 739/FL 050	1 274/FL 120	1.538/FL 060
(4 occupants) Runway 1,290	1,860	1,870
Block Time 2+58	3+55	2+54
Fuel Used 221	339	304
Specific Range/Altitude 2.262/FL 100	1.475/FL 120	1.645/FL 060
Suggested Base Price \$689,000	\$726,960	\$815,000
Remarks Certification Basis Certification Basis CAR 3/FAR 23, 1955/94; STC SA02483CH Includes Garmin G1000; composite fuselage shell with left and right doors.	FAR 23 A 54 Includes Garmin G500. All data preliminary. 2016 data.	CAR 3, 1956/69/83/2005 A/C system standard; Garmin G1000 NXi.

Single-Engine Pistons turbocharged

Manufacturer			Cirrus	Textron Aviation	Textron Aviation	GippsAero	Mooney
Model			SR22T	Cessna Turbo Stationair HD	Cessna TTx	GA8 Airvan TC	Acclaim Ultra
BC4 Equipped	Price		\$639,900	\$665.000	\$715,000	\$761.030	\$769.000
Dun Equipped	1100	Seating	1+3/4	1+5/5	1+3/3	1+6/7	1+3/3
Character-		Wing Loading	23.5	21.8	25.5	20.7	19.2
istics		Power Loading	11.43	12.22	11.61	13.13	12.03
		Noise (dBA)	80.3	82.6	81.4	85.4	78.0
External		Length	26.0	28.3	25.2	28.3	26.9
Dimensions		Height	8.9	9.3	9.0	9.3	8.3
(ft.)		Span	38.3	36.0	36.0	36.0	36.4
Internal		Length	8.0	9.3	7.9	11.6	8.1
Dimensions		Height	4.1	4.1	4.1	3.7	3.7
(ft.)		Width	4.1	3.7	4.0	4.2	3.6
		Engine	Cont	Lyc TIO-540-4114	Cont	Lyc TIO-540-4H14	Cont
Power		Output (hp)	315	310	310	320	280
		Inspection Interval	2.000t	2.000t	2.000t	1.800t	2.200t
		Max Ramp	3.610	3.806	3.600	4.214	3.374
		MaxTakeoff	3.600	3.789	3.600	4,200	3.368
		Max Landing	3,600	3,600	3,420	4,000	3,200
		Zero Fuel	3,400c	3,618b	3,300c	4,053b	3,173b
		EOW	2,342	2,336	2,535	2,349	2,378
Weights (lb.)		Max Payload	1,058	1,282	765	1,704	795
		Useful Load	1,268	1,470	1,065	1,865	996
		Max Baggage	130	180	120	180	120
		Max Fuel	552	522	612	540	612
	Ava	ilable Payload w/Max Fuel	716	948	453	1,325	384
Av		ilable Fuel w/Max Payload	210	188	300	161	201
		VNE	205	182	230	185	195
Limits		VNO	176	149	181	143	174
		VA	140	125	158	121	127
T0 (SL elev./ISA Ter Airport T0 (5,000-ft. elev.@2		TO (SL elev./ISA Temp.)	1,517	1,970	1,900	1,840	1,900
		T0 (5,000-ft. elev.@25C)	2,268	2,845	2,460	2,788	3,300
Perfor-		Vso	64	59	61	61	60
mance		VX VA	102	70	82	71 01	105
	Tin	ne to Climb (min)/Altitude	7/FL 100	00 12/EL 100	7/EL 100	01 13/EL 100	7/FL 100
Climb	111	Initial Gradient (ft /nm)	782	724	701	825	770
		Certificated	25.000	25 000	25.000	20.000	25,000
Ceilings (ft.)		Service	25,000	27,000	25,000	20,000	25.000
		TAS	171	137	208	125	215
		Fuel Flow	76	85	78	68	99
	Long Range	Altitude	FL 250	FL 240	FL 250	FL 200	FL 250
		Specific Range	2.250	1.612	2.667	1.838	2.172
		TAS	201	155	227	130	227
Cruise	Recommended	Fuel Flow	98	99	130	78	128
ordise	Recommended	Altitude	FL 250	FL 240	FL 250	FL 200	FL 180
		Specific Range	2.051	1.566	1.746	1.667	1.773
		TAS	213	164	235	135	242
	High Speed	Fuel Flow	110	114	152	98	130
		Altitude Specific Dec	FL 250	FL 200	1 E46	FL 200	1 860
		Nautical Miles	1.930	512	1.540	2.370	500
		Average Sneed	171	137	202	125	178
	Seats Full	Fuel Used	486	387	345	220	259
Ranges		Specific Range/Altitude	2.101/FL 250	1.323/FL 200	1.930/FL 250	1.059/FL 200	1.931/FL 160
		Nautical Miles	1,021	655	1,270	618	1,122
	Tanka Fall	Average Speed	171	138	204	125	200
	Tanks Full	Fuel Used	486	459	572	459	539
		Specific Range/Altitude	2.101/FL 250	1.427/FL 240	2.220/FL 250	1.346/FL 200	2.082/FL 250
		Runway	1,405	1,396	1,730	1,743	1,300
	200 nm	Block Time	1+08	1+23	1+03	1+35	1+05
		Fuel Used	197	163	159	125	139
Missions		Specific Range/Altitude	1.015/FL 100	1.227/FL 150	1.258/FL 150	1.600/FL 120	1.439/FL 120
(4 occupants)		Runway	1,699	1,597	1,880	1,743	1,380
	500 nm	Block Time	2+28	3+22	2+24	3+30	2+54
		Fuel Used	300	385	338	3/3	259
		Suggested Rose Driss	1.309/FL 180	1.299/FL 240	1.4/9/FL 250	1.340/FL 200 \$507.500	\$760.000
		Suggesteu Dase MICe	4039,900	4000,000	\$110,000	φ397,300	\$103,000
Remarks		Certification Basis	FAR 23, 2010 Includes Garmin Perspective+ Global avionics.	FAR 23, 1998 Utility version w/2,183-lb. EOW, \$658,650; Garmin G1000 NXi w/GFC 700 a/p; new interior.	FAR 23 Includes Garmin G2000, SVT, AP, TAWS, TAS, ESP, A/C, Ti LE, leather.	FAR 23, 1998 Garmin G500; KC 225 All data preliminary. 2016 data.	CAR 3, 1955/89/2006 Includes Garmin G1000; new composite fuselage shell with left and right doors.

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Single-Engine Pistons pressurized

Manufacturer			Piper Aircraft Piper Aircraft			
Model			Matrix	M350		
BCA Equipped	Price		\$916.680	\$1 178 610		
BCA LQUIPPEU	FILE	Seating	1+4/5	1+4/5		
Character-		Wing Loading	24.8	24.8		
istics		Power Loading	12.40	12.40		
		Noise (dBA)	81.0	81.0		
External		Length	28.9	28.9		
Dimensions		Height	11.3	11.3		
(TL.)		Span	43.0	43.0		
Dimensions		Height	3.9	3.9		
(ft.)		Width	4.2	4.2		
		Engine	Lyc	Lyc		
Power		Light	TIO-540-AE2A	TIO-540-AE2A		
		Output (hp)	350	350		
		Max Ramp	4.358	4,358		
		Max Takeoff	4,340	4,340		
		Max Landing	4,123	4,123		
		Zero Fuel	4,123c	4,123c		
		EOW	2,969	3,146		
Weights (lb.)		Max Payload	1,154	977		
		Useful Load	1,389	1,212		
		Max Baggage	200	200		
	Avai	Max Fuel	669	120		
	Avai	lable Fuel w/Max Pavload	235	235		
		VNE	198	198		
1.1		VNO	168	168		
LIMITS		VA	133	133		
		PSI	5.5	5.5		
		TO (SL elev./ISA temp.)	2,090	2,090		
Airport		T0 (5,000-ft. elev.@25C)	2,977	2,977		
Performance		VSO	58	58		
		VX Vy	110	110		
<u></u>	Tim	e to Climb (min.)/Altitude	8/FL 100	8/FL 100		
Climb		Initial Gradient (ft./nm)	703	703		
		Certificated	25,000	25,000		
Ceilings (ft.)		Service	25,000	25,000		
		Sea-Level Cabin	-	12,300		
		IAS Fuel Flow	66	66		
	Long Range	Altitude	FL 250	FL 250		
		Specific Range	2.364	2.364		
		TAS	203	203		
Cruise	Recommended	Fuel Flow	108	108		
oraioo		Altitude	FL 250	FL 250		
		Specific Range	1.880	1.880		
		TAS Fuel Flow	120	120		
	High Speed	Altitude	FL 250	FL 250		
		Specific Range	1.775	1.775		
		Nautical Miles	867	535		
	Seats Full	Average Speed	151	138		
		Fuel Used	457	312		
Ranges		Specific Range/Altitude	1.897/FL 200	1./15/FL 120		
		Average Speed	158	159		
	Tanks Full	Fuel Used	658	670		
		Specific Range/Altitude	2.041/FL 250	2.004/FL 250		
		Runway	2,090	2,090		
	200 nm	Block Time	1+07	1+06		
Minoir		Fuel Used	168	167		
MISSIONS		Specific Range/Altitude	1.190/FL 120	1.198/FL 200		
(+ occupants)		Runway	2,090	2,090		
	500 nm	Fuel Used	350	350		
		Specific Range/Altitude	1.429/FL 250	1.429/FL 250		
		Suggested Base Price	\$916,680	\$1,178,610		
Remarks		Certification Basis	FAR 23, 1983/88 Garmin G1000; FIKI optional.	FAR 23, 1983/88 Garmin G1000; FIKI optional.		

Multiengine Pistons normally aspirated

Model P - 68C	Manufacturer	_	-	Vulcanair SpA Vulcanair SpA			
Number P 68C P 68C P 68C BAC Equipped Price \$530.300.0 \$584.200.0 \$548.45.200.0 Character. Noise (dB) 74.7 78.8 External 11.49 11.37 31.3 Dimensions Heigh 31.3 31.3 Internal Sam 39.4 39.4 Dimensions Heigh 3.0.5 39.4 Internal Used (dB) 3.3 3.3 (ft.) Sam 3.9.4 39.4 Dimensions Heigh 3.0.5 3.9 (ft.) Sam Sam 3.8 Power Optor (in each) 2.00 2.000 Vertice Ligs 0.3.3 3.9 Vertice Sam 4.157 4.374b Weights (lb.) Max Paydio 1.014 1.177 Max Paydio 1.014 1.177 1.331 Limits In 0.62 1.043 1.043 Max Paydio 1.014 1.177 1.312 1.226	Madal			P.68C	Victor		
BdA Equipped Price \$830,800 \$8484,200 Character Wing Loading 11-5/6 14-5/6 isitics Wing Loading 22.9 22.7 Isitics Nisse (dBA) 74.7 78.8 External Units 31.3 31.3 Dimensions Heigh 11.2 11.2 (ft) Sam 39.4 39.4 Dimensions Heigh 3.6 10.6 (ft) Wide 3.8 3.8 Power Output (hp seid) 2000 2000 Mas Takeff 4.532 4.321 Verights (lb.) Mas Takeff 4.533 3.137 Mas Takeff 4.543 4.321 1.260 Mas Takeff 4.543 1.331 1.312 Mas Takeff 4.543 1.417 1.335 Mas Takeff 4.543 1.417 1.335 Mas Takeff 4.543 1.4157 1.4157 Mas Takeff 4.543 1.4157 1.4157	IVIOUEI			P 68C	P 68R		
Character- istics 1+5/6 1+5/6 Searing 1-15/6 1-15/6 Power Lading 11.49 11.37 Obmensions 31.3 31.3 Obmensions 39.4 39.4 Internal 10.6 10.6 Dimensions Height 3.9 3.9 (ft) Span 39.4 39.4 Power 0.060/A1B6 0.0300/A1B6 0.0300/A1B6 Opticitive exit 2.100 2.200 2.200 Mark and divide 4.504 4.548 Mark and divide 4.504 4.548 Mark and divide 4.505 4.321 Zero Hai 4.167 4.349 Mark and divide 1.014 1.177 Used Mark and divide 1.014 1.177 Used Mark and divide 1.0163 1.063 Mark and divide 1.0163 1.0163 Mark and divide 1.014 1.177 Limits No 1.014 1.177 <td< th=""><th>BCA Equipped</th><th>Price</th><th></th><th>\$830,800</th><th>\$848,200</th></td<>	BCA Equipped	Price		\$830,800	\$848,200		
Characteries mig.takaning Poert.caning 22.3 (1.49) 22.1 (1.49) 21.12 11.2	Character		Seating Wing Loading	1+5/6	1+5/6		
Situ:Source 11.00 11.00 11.00 11.00 External Dimensions (ft.) Iargent (ft.) 31.3 31.3 31.3 Unitarial Dimensions (ft.) Sam 39.4 39.4 39.4 Unitarial Dimensions (ft.) Sam 39.4 39.4 39.4 Dimensions (ft.) Sam 39.4 39.4 39.4 Dimensions (ft.) Sam 3.8 3.8 3.8 Power Orago Albe (ft.) 0.060 Albe (ft.) 0.060 Albe (ft.) 0.060 Albe (ft.) Weights (ft.) Max Faleeff 4.594 4.548 4.548 Max Faleeff 4.594 4.548 4.548 4.548 Max Faleeff 4.563 4.374 7.6 7.6 Max Faleeff 4.563 4.374 7.4 7.6	inting		Wing Loading Power Loading	22.9	11 37		
External Dimensions 13.3 31.3 31.3 Dimensions Heigh Heigh 11.2 11.2 11.2 Internal Dimensions Heigh Heigh 13.9 3.9.4 38.4 38.4 Othersions Heigh Heigh 3.9 3.9.9 3.9.9 3.9.9 Power Different Dispection Interval Long Ramp 10.6 10.6 10.66 Weights (Ib.) Fights 0.21 kp (2.000 t 2.000 t 2.000 t Max landing Long to the standing Long to the	15005		Noise (dBA)	74.7	78.8		
Dimensions (ft.) High Span 11.2 11.2 11.2 Internal Dimensions (ft.) Span 39.4 39.4 39.4 Power Umber of the special internal Dimensions 3.9 3.9 3.9 Power Eigene Ubgection internal Dimensions 2.1yc (D.360A1B6 0.360A1B6 0.360A1B6 Power Dimensions 2.000t 2.000t 2.000t Weights (b.) Max Ramp 4.630 4.548 Max Bacter 4.954 4.548 4.548 Weights (b.) Max Payota 1.014 1.17 Max Payota 1.014 1.17 1.351 Max Payota 1.0163 1.063 1.063 Max Payota 1.014 1.07 1.351 Max Payota 1.014 1.17 1.351 Limits Wei 1.14 1.17 1.12 Limits Wei 1.14 1.17 1.12 Limits Wei 1.14 1.17 1.12 Limits Na	External		Length	31.3	31.3		
(ft.) Span 39.4 39.4 Internal Dimensions Length 10.6 10.6 (ft.) Weth 3.9 3.9 (ft.) Weth 3.9 3.9 (ft.) Span (2.1)cs (0.300.4186 (ft.) 0.300 2.000t 2.000t (ft.) Markamin 4.630 4.548 Markamin 4.630 4.548 Markamin 4.365 4.321 Zoo Toil 4.167c 4.374b Markamin 4.167c 4.374b Markamin 1.063 1.063 Markamin 1.063 <	Dimensions		Height	11.2	11.2		
Internal Dimensions 10.6 10.6 Power 0.6 10.6 0 3.9 3.9 Power 0.10 2.1/2 0.2 0.03.60.A1B6 0.200.01 2.0001 0.03.60.A1B6 0.200.01 2.0001 Weights (b.) Mar.Bamp 4.630 4.548 Mar.Bamp 4.630 4.548 Mar.Bamp 4.630 4.548 Mar.Bamp 4.630 4.531 Mar.Bamp 4.630 4.531 Mar.Bamp 4.635 4.321 Mar.Bamp 1.014 1.177 Mar.Bamp 1.014 1.177 Mar.Bamp 1.063 1.063 Mar.Bamp 1.014 1.177 Mar.Bamp 1.010 1.020 </th <th>(ft.)</th> <th></th> <th>Span</th> <th>39.4</th> <th>39.4</th>	(ft.)		Span	39.4	39.4		
Dimensions (ft.) 3.9 3.9 3.9 Weights (b.) Weights (b.) Weights (b.) Weights (b.) Up (b) each base definition (b) each base definitent (b) each base definition (b) each base definition (b) each	Internal		Length	10.6	10.6		
(f.) Weth 3.8 3.8 Power -2 kg 0.2 kg 0.2 kg Power -0.360 A1B6 10.360 A1B6 0.0360 A1B6 Weights (b) Max Ramp 4.630 4.548 Max Landing 2.000t 2.000t 2.000t Weights (b) Max Ramp 4.630 4.548 Max Landing 7.807 kg 4.167c 4.374b Zero Pat 4.167c 4.374b 4.374b Max Rayton 1.014 1.177 1.351 Max Rayton 1.312 1.2600 1.961 Max Rayton 1.312 1.2600 1.961 Max Rayton 1.312 1.2600 1.961 Max Rayton 1.961 4.633 1.971 Max Rayton 1.961 4.630 1.971 Max Rayton 1	Dimensions		Height	3.9	3.9		
Power 0 2 Lyc Uptu (the sect) Inspection Intered 0 2 0.00 2 0.00 Weights (b,) Max Ram Max lawling 4,630 4,548 Weights (b,) Max Ram Max lawling 4,630 4,548 Weights (b,) Max Ram Max lawling 4,167c 4,374b Max Fard 4,167c 4,374b 3,197 Max Fard 1,063 1,063 1,063 Max Fard 1,063 1,063 <	(ft.)		Width	3.8	3.8		
Power Dutput (hp end) Inspection Interval 2.000c Double (h) = 200 2.000t Double (h) = 200 2.000t Weights (h). Image: horizon (h) = 200 (h) = 200 2.000t 2.000t 2.000t Weights (h). Max latert Max later Max latert Max latert Max			Engines	2 Lyc	2 Lyc		
Impection Interval Max Ramp Max Islanding Zer Pist Max Landing Zer Pist Max Paylead Max Landing A:155 2.000t 2.000t Weights (b). Max Ramp Max Islanding A:554 4.548 Weights (b). EW 3.153 3.3197 Weights (b). Max Paylead Max Islanding Available Payload w/Max Paylead 1.014 1.177 Max Paylead 1.063 1.0633 1.0633 Max Isla 1.063 1.063 1.063 Max Isla 1.060 2.000 4.000 Max Isla 1.32 1.260 1.010 Max Isla 1.27E 0.00 4.000 Max Isla 1.410 1.27E 1.020 Max Isla 1.020 1.020	Power		Output (hp each)	200	200		
Weights (b). Mex Ramp Max Takedt Max Lakedt Max			Inspection Interval	2,000t	2,000t		
Weights (b). Max latert Max latert Max latert Max latert Max latert Max latert E0W 4,565 4,521 Weights (b). Max latert Max Parter Max Parter Max Parter Max Parter Max Parter Max File 4,167C 4,374b Weights (b). Max File 1,047 1,351 Max File 1,043 1,063 Max File 1,043 1,063 Max File 1,045 1,063 Max File 1,053 1,063 Max File 415 289 Max lable Fuel W/Max Paytest 403 1,147 Max Stref 1,12 1,260 Max Stref 4,157 1,312 Max Stref 1,312 1,260 Max Stref 4,150 2,150 Max Stref 1,312 1,400 Max Stref 2,950 2,370 Max Stref 1,00 92 Max Engine Out Gradient (ft/rm) 1,100 12/FL 100 Max Engine Out Gradient (ft/rm) 1,100 92 Max Engine Out Gradient (ft/rm) 1,100 1,315			Max Ramp	4,630	4,548		
Max Landing 4,365 4,321 Zer Fiel 4,1676 4,3740 Line 200 3,153 3,197 Weights (lb.) Max Parlead 1,014 1,177 Max Fiel 1,063 1,063 Mariable Payload w(Max Fuel 415 289 Available Payload w(Max Fuel 415 289 Max Berle 194 197 Limits 105,000 fuelew/SA Payload 463 Limits 105,000 fuelew/SA Payload 463 Airport 1,312 1,260 Vis 132 127 Vis 660 60 Vis 660 60 Vis 82 88 Base 88 88 Climb Initial Engine-Out Genetic ft(1, /m) 1,100 920 Initial Engine-Out Cardient ft, /m 1,100 920 1,532 Climb Initial Engine-Out Cardient ft, /m 1,100 920 Initial Engine-Out Cardient ft, /m 1,100 920			Max Takeoff	4,594	4,548		
Weights (b) Image: Constraint of the second se			Max Landing	4,365	4,321		
Weights (lb.) How Ar Payload Max Payload Lissful Load 1.014 1.014 1.177 Max Ibel Max Ibel M			Zero Fuel	4,167c	4,374b		
Image: Constant and the second seco	Weights (lb.)		EOW Max Dayland	3,153	3,197		
Max Fiel 1.063 1.063 Maxibile 1.063 1.063 Available Payload WMax Fuel 445 174 Limits Ve 194 197 Limits Ve 194 197 Limits Ve 132 127 Ve 132 127 1.312 Available Fuel w/Max Payload 4633 1.410 Auron 1.312 1.260 Ve 132 1.27 Ve 1.32 1.260 Ve 60 60 Ve 88 88 Performance Vec 88 88 Time to Climb (min)/Altuide 12/FL 100 12/FL 100 Initial Engine-Out Gradient (fL/rm) 1.407 247 Initial Engine-Out Gradient (fL/rm) 1.417 144 Cruise Recommended Fuel Row 1.00 20,000 Specific Range 1.532 1.532 1.532 Cruise Recommended Aurona Vecentin 1.			Max Payload	1,014	1 351		
Available Payload w/Max Fue Available Fuel w/Max Payload 415 289 Limits Vea 445 174 Limits Vea 194 197 Limits Vea 154 157 Vea 154 157 Vea Airport 10 (SL elev,/ISA Temp.) 1312 1276 Airport A/S (SL elev,/ISA 2.150 1.410 A/S (SL elev,/ISA 2.150 2.370 Performance Wea 60 60 Vea 88 98 88 Climb Initial Engine-Out Braite (Imn 1.100 1.2/FL 100 Initial Engine-Out Service 50.000 5.650 50 Climbs Initial Engine-Out Service 1.600 FL 140 <td< th=""><th></th><th></th><th>Max Fuel</th><th>1.063</th><th>1.063</th></td<>			Max Fuel	1.063	1.063		
Available Fuel w/Max Payload 463 174 Limits Vie 194 197 Vie 132 127 Vie 132 127 Airport 10 (Stolev/ISA Temp) 1,312 1,260 Performance Vie 1,312 1,260 Vie 2,950 2,370 Performance Vie 70 70 Vie 82 88 88 Climb Initial Engine-Out Rate (trym) 2,17 2,17 Initial Air-Engine Gradient (trym) 1,100 12/FL 100 12/FL 100 Initial Engine-Out Rate (trym) 1,417 144 Ceilings (tt.) All-Engine Serice 18,000 20,000 Engine-Out Service 5,000 5,650 1.55 Tris 144 144 144 Long Range Tris 1462 162 Iong Range Tris 1.62 1.632 Yiei Plow 108 108 108 Ranges		Avai	lable Payload w/Max Fuel	415	289		
Limits Vec 194 197 Visit 154 157 Visit 132 127 Visit 4000 4,000 Alitport 4,000 4,000 Alitport 4,000 4,000 Alitport 4,000 4,000 Alitport 4,000 4,000 Visit 82 32 Visit 82 88 Bite 12,7 217 Visit 82 88 Visit 82 88 Visit 82 88 Climb Initial Engine-Out Rate (pm) 217 217 Initial Alitengine Gradent (tr,/m) 147 147 Ceilings (tt) Long Range 18,000 20,000 Egeecinte Range		Avai	lable Fuel w/Max Payload	463	174		
Limits Viol 154 157 Air 20 132 127 Airport 10 (SL elev,/ISA Temp.) 1,312 1,260 Airport A/S (SL elev,/ISA 2,150 1,410 Alrport A/S (SL elev,/ISA 2,150 1,410 Alrport A/S (SL elev,/ISA 2,150 1,410 Alrport A/S (SL elev,/ISA 2,150 1,410 Visa 80 60 60 Vec 70 70 70 Vec 82 82 82 Vec 70 70 7217 7217 Initial Engine-Out Rate (pm) 1,100 920 1614 144 Initial Engine-Out Cardient (ft,/m) 1,410 144 144 Cruise Ing Range Ing 155 155 Ing Range Ing Ing Ing 155 155 Ing Range Ing Ing Ing 162 162 Cruise Recommended Fuel Flow 162 </th <th></th> <th></th> <th>VNE</th> <th>194</th> <th>197</th>			VNE	194	197		
Ki 132 127 10 (Stelew()StTemp) 1.312 1.260 Airport	Limits		VNO	154	157		
Airport 10 (SL eler,/ISA lemp.) 1,312 1,260 No (S. 00-L. eler,25C) 2,950 2,370 Performance Vec 60 60 Vec 70 70 Vec 80 60 60 Vec 70 70 Vec 82 82 Vec 82 88 Bital Engine-Out Rate (pm) 217 217 Vec 83 88 Cilimb Initial Engine-Out Rate (pm) 217 217 Initial All-Engine Gradient (tr/m) 1,100 920 1,147 Ceilings (ft.) All-Engine Service 18,000 20,000 Initial Engine-Out Service 5,000 5,650 144 144 Cruise Recommended Fuel Row 94 94 94 Iong Range TAS 144 144 144 144 Iong Range Fuel Row 162 162 162 Ranges Tas 144 144 <th></th> <th></th> <th>VA</th> <th>132</th> <th>127</th>			VA	132	127		
Airport Airo (Subol-ft, elev/8250) 4,000 4,000 A/S (S. lev/N3) 2.150 1,410 Performance Wax 60 60 Vec 70 70 Vec 70 70 Vec 70 70 Vec 82 82 Climb Intre to Climb (min.)/Altitude 12/FL 100 12/FL 100 Intial Engine-Out Rate (pm) 217 217 Intial Engine-Out Rate (pm) 117 217 Intial Engine-Out Rate (pm) 147 147 Ceilings (ft.) Engine-Out Service 5000 5,650 Intial Engine-Out Service 5,000 5,650 5,650 Cruise Recommended Fuel Row 144 144 Ital 1155 1155 155 TAS 155 1155 155 Fuel Row 108 108 60 Specific Range 1.337 1.337 1.337 Ranges Specific Range Altitude			TO (SL elev./ISA Temp.)	1,312	1,260		
Airport Performance Image: hysic lew(#50) 2,150 1,443 Performance A/S (5,000-ft, lew(#250) 2,950 2,370 Visc 60 60 60 Visc 82 88 88 Climb Inter to Climb (min/Altitude 12/FL 100 12/FL 100 12/FL 100 Initial Engine-Out Rate (fm) 2.1 7 217 117 Initial Engine-Out Gradient (ft/m) 1.100 920 147 147 Ceilings (ft.) Certificated			IU (5,000-Tt. elev.@25C)	4,000	4,000		
Performance No. 199 (1000) No. 199 (1000) Performance Performance No. 2000 Performance No. 2000 Performance Performance <thperformance< th=""> Performance</thperformance<>	Airport		A/S (5 000-ft_elev@25C)	2,150	2,370		
Vec 70 70 Visi 82 82 Visi 83 88 Climb Initial Engine-Out Rate (fpm) 217 217 Initial All-Engine Gradient (ft,/m) 1,100 920 12/FL 100 Initial All-Engine Gradient (ft,/m) 1,100 920 147 Ceillings (ft.) Certificated - - - Ceillings (ft.) Engine-Out Service 18,000 20,000 5,650 Kass Specific Range 1.532 1.532 1.532 Cruise Recommended Fuel Flow 94 94 Kass 1.532 1.532 1.532 Recommended Fuel Flow 108 108 Max Payload Specific Range 1.435 1.435 High Speed Fuel Flow 116 116 High Speed Specific Range 1.397 1.397 Max Payload Naucical Miles 3.00 300 Missions Specific Range/Altitude 1.026/F	Performance	·'	VMCA	60	60		
Viss 82 82 82 Climb Time to Climb (min.)/Altitude 12/FL 1.00 12/FL 1.00 Initial Engine-Out Braid (min.)/Altitude 12/FL 1.00 12/FL 1.00 Initial Engine-Out Gradient (ft./m) 1.10 920 Initial Engine-Out Gradient (ft./m) 1.47 1.47 Ceilings (ft.) Certification			VDEC	70	70		
Vsc 88 88 Climb Time to Climb (min)/Altitude 12/FL 100 12/FL 100 Initial Fegine-Out Brad (fpm) 217 217 Initial Engine-Out Gradient (fL/m) 1,100 920 Initial Engine-Out Gradient (fL/m) 147 147 Ceillings (ft.) Engine-Out Gradient (fL/m) 147 144 Ceillings (ft.) Engine-Out Service 5,000 5,650 Engine-Out Service 5,000 5,650 5,650 Fuel Flow 94 94 94 Fuel Flow 94 94 94 Fuel Flow 116 116 116 TMS 155 155 155 Fuel Flow 108 1008 1008 Fuel Flow 116 116 116 High Speed Fuel Flow 116 116 Fuel Flow 116 116 116 Fuel Flow 116 116 116 Fuel Flow 116 116 145			Vxse	82	82		
Climb Time to Climb (min)/Mittude 12/FL 100 12/FL 100 Initial Engine-Out Rate (from) 217 217 Initial All-Engine-Out Gradient (fr/m) 1.100 920 Initial Engine-Out Gradient (fr/m) 1.47 147 Ceillings (ft.) Certificated All-Engine-Out Service 18,000 20,000 5,650 Engine-Out Service 5,000 5,650 5,000 5,650 Cruise Fas 144 144 144 Fuel Flow 94 94 94 Autude FL 080 FL 080 FL 080 Specific Range 1.532 1.532 1.532 Cruise Recommended Autude FL 080 FL 080 Specific Range 1.435 1.435 1.435 High Speed Autude FL 080 FL 080 Nautical Miles 300 300 300 Aurage Speed 1.40 140 140 Ranges Specific Range/Altitude <t< th=""><th></th><th></th><th>Vyse</th><th>88</th><th>88</th></t<>			Vyse	88	88		
Climb Initial Engine-Out Rate (pm) 217 217 Initial All-Engine Gradient (ft,/nm) 1,100 920 Initial Engine-Out Gradient (ft,/nm) 1,47 1,47 Ceilings (ft.) Ceilings (ft.) Certificated Certificated Certificated Certificated Certificated Fuel Flow All-Engine Service 18,000 20,000 Engine-Out Service 5,000 5,650 TAS 144 144 IAS 1.532 1.532 IAS 1.532 1.532 IAS 1.632 1.632 IAS 1.612 1.62 1.62 IAS 1.62 1.62 1.62 IAS 1.632 1.632 IAS 1.62 1.62		Time to Climb (min.)/Altitude			12/FL 100		
And Areingine Gradient (LT, Im) 1,100 920 Initial Engine-Out Gradient (LT, Im) 147 147 Ceillings (ft.)	Climb	Init Initial All	al Engine-Out Rate (fpm)	217	217		
Ceilings (ft.) Initial cigato Got Gloud (ft) fm) Initial cigato Got Gloud (ft) fm) Initial cigato Got Gloud (ft) fm) Ceilings (ft.) All-Engine Service 18,000 20,000 Initial cigato Got Gloud (ft) fm) 144 144 Long Range TAS 144 144 Fuel Flow 94 94 Attitude FL 080 FL 080 Recommended Specific Range 1.532 1.532 TAS 155 155 Fuel Flow 108 108 Attitude FL 080 FL 080 Fuel Flow 116 116 High Speed Attitude FL 080 FL 080 Max Payload Specific Range 1.397 1.397 Max Payload Average Speed 140 140 Ferry Nautical Miles 300 300 Average Speed 145 145 145 Fuel Flow 116 1.026/FL 080 1.026/FL 080 Auerage Speed 140 1.420 1.435 </th <th></th> <th>Initial Eng</th> <th>sine_Out Gradient (ft /nm)</th> <th>1,100</th> <th>920</th>		Initial Eng	sine_Out Gradient (ft /nm)	1,100	920		
Ceilings (ft.) All-Engine Service Engine-Out Service 18,000 20,000 Fighe-Out Service 5,000 5,650 Idag TAS 144 144 Fuel Flow 94 94 94 Alt-Engine-Out Service 5,000 5,650 5,650 Cruise Fuel Flow 94 94 94 Fuel Flow 94 94 94 94 Fuel Flow 94 94 94 Fuel Flow 1.532 1.532 1.532 Fuel Flow 108 108 108 108 Fuel Flow Altitude FL 0.800 FL 080 FL 080 High Speed TAS 162 162 162 Altitude FL 0.800 FL 080 FL 080 130 Altitude FL 080 FL 080 140 140 Altitude FL 080 FL 080 145 145 Altitude Specific Range/Altitude 1.0000 1,000 1,000			Certificated				
Cruise Engine-Out Service 5,000 5,650 Cruise Long Range TAS 144 144 Fuel Flow 94 94 94 Attitude FL 080 FL 080 FL 080 Cruise Recommended Specific Range 1.532 1.532 High Speed Fuel Flow 108 108 Attitude FL 080 FL 080 FL 080 High Speed Specific Range 1.435 1.435 High Speed TAS 162 162 Fuel Flow 116 116 116 Attitude FL 080 FL 080 Specific Range 1.397 Max Payload Specific Range 1.397 1.397 Max Payload Specific Range/Attitude 0.952/FL 080 0.952/FL 080 Nautical Miles 1.0000 1.0000 1.0000 Average Speed 145 145 145 Tip Fuel 975 975 975 Specific Range/Attitude 1.026/	Ceilings (ft.)		All-Engine Service	18,000	20,000		
Cruise Long Range TAS Fuel Flow Altitude 144 144 Fuel Flow Altitude 94 94 94 Cruise Recommended Specific Range Fluel Flow Altitude FL 080 FL 080 Recommended TAS Fluel Flow Altitude 1.532 1.532 High Speed Altitude FL 080 FL 080 Specific Range Fluel Flow Altitude 1.435 1.435 1.435 High Speed TAS Fuel Flow Altitude 1.62 1.62 1.62 Max Payload Specific Range Specific Range/Altitude 1.0307 1.1397 Max Payload Tip Fuel Tip Fuel 315 3.15 3.15 Ranges Specific Range/Altitude 0.952/FL 080 0.952/FL 080 Maxeage Speed Tip Fuel Specific Range/Altitude 1.026/FL 080 1.026/FL 080 1.450 Missions Specific Range/Altitude 1.450 1.450 Missions Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Missions Specific Range/Altitude 1.450 1.450 Missions Spec			Engine-Out Service	5,000	5,650		
Long Range Fuel Flow Altitude 94 94 Cruise Iong Range Altitude FL 080 FL 080 Recommended ISS 155 155 Fuel Flow Altitude 1532 1.532 1.532 Prescription 108 108 108 Altitude FL 080 FL 080 FL 080 Specific Range 1.435 1.435 1.435 High Speed Attitude FL 080 FL 080 FL 080 High Speed Attitude FL 080 FL 080 1.080 Specific Range 1.397 1.397 1.397 Max Payload Average Speed 140 140 Tip Fuel 315 315 315 Specific Range/Altitude 0.952/FL 080 0.952/FL 080 0.952/FL 080 Nautical Miles 1.000 1.0000 1.0000 Nautical Miles 1.000 1.0000 1.0000 Average Speed 145 145 145 Grecrifi Range/Altitude <th></th> <th></th> <th>TAS</th> <th>144</th> <th>144</th>			TAS	144	144		
Cruise Attude FL 080 FL 080 Recommended Specific Range 1.532 1.532 Fuel Flow 108 108 108 Attude FL 080 FL 080 FL 080 Fuel Flow 108 108 108 Attude FL 080 FL 080 FL 080 High Speed Attude FL 080 FL 080 High Speed 1435 1.435 1.435 High Speed Attude FL 080 FL 080 Max Payload Specific Range 1.397 1.397 Nautical Miles 300 300 300 Average Speed 140 140 140 Ranges Specific Range/Attude 0.952/FL 080 0.952/FL 080 Nautical Miles 1,000 1,000 1,000 Average Speed 145 145 Trip Fuel 975 975 Specific Range/Attude 1.026/FL 080 1.026/FL 080 Misssions Specific Range/Attude 14		Long Range	Fuel Flow	94	94		
Cruise Recommended TAS 155 155 Fuel Flow 108 108 108 High Speed Attitude FL 080 FL 080 FL 080 High Speed TK 162 162 162 High Speed Fuel Flow 116 116 116 Max Payload Specific Range 1.397 1.397 Max Payload Nautical Miles 300 300 Average Speed 140 140 140 Trip Fuel 315 315 315 Specific Range/Altitude 0.952/FL 080 0.952/FL 080 0.952/FL 080 Nautical Miles 1.000 1.000 1.000 Average Speed 145 145 Trip Fuel 975 975 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Missions Specific Range/Altitude 1.420 1.420 Go onm Specific Range/Altitude 1.420 1.420 Specific Range/Altitude 1.420 1		Long Nange	Altitude	FL 080	FL 080		
Cruise Recommended Fuel Flow Altitude 100 108 Fuel Flow Altitude 1.435 1.435 FL 080 FL 080 FL 080 Fuel Flow High Speed TKS 162 162 High Speed TKS 162 162 High Speed Altitude FL 080 FL 080 Max Payload Specific Range 1.397 1.397 Nautical Miles 300 300 300 Average Speed 140 140 140 Average Speed 145 145 145 Specific Range/Altitude 0.952/FL 080 0.952/FL 080 0.952/FL 080 Nautical Miles 1,000 1,000 1,000 1,000 Average Speed 145 145 145 Ferry Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Missions Specific Range/Altitude 1.420 1.420 Groupantsi Specific Range/Altitude 1.420 1.420 Solon m Specific Range/Altitude<			Specific Range	1.532	1.532		
Cruise Recommended Altitude FL 080 FL 080 High Speed Altitude FL 080 FL 080 FL 080 High Speed TAS 162 162 High Speed Fuel Flow 116 116 Max Payload Specific Range 1.397 1.397 Nautical Miles 300 300 300 Average Speed 140 140 140 Tip Fuel 315 315 315 Specific Range/Altitude 0.952/FL 080 0.952/FL 080 0.952/FL 080 Nautical Miles 1,000 1,000 1,000 Average Speed 145 145 145 Ferry Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Missions Specific Range/Altitude 1.425 1.420 Go nm Specific Range/Altitude 1.420 1.420 Specific Range/Altitude 1.420 1.420 1.420 Go nm Specific Range/Altitude 1.420 1.420			Fuel Flow	108	108		
Migh Speed Specific Range Fuel Flow 1.435 1.435 High Speed TAS 162 162 Attitude Fluel Flow 116 116 Attitude FL080 FL080 FL080 Specific Range 1.397 1.397 1.397 Max Payload Average Speed 140 140 Trip Fuel 315 315 315 Specific Range/Attitude 0.952/FL 080 0.952/FL 080 0.952/FL 080 Nautical Miles 1,000 1,000 1,000 Average Speed 145 145 Trip Fuel 975 975 Specific Range/Attitude 1.026/FL 080 1.425/FL 080 Missions 1.4450 1.4450 1.4450 Geordic Range/Attitude 1.420 1.4450 Specific Range/Attitude 1.420/FL 080 1.429/FL 080 Karway 1.450 1.429/FL 080 1.429/FL 080 Missions Specific Range/Attitude 1.420 1.420 Soon nm Spe	Cruise	Recommended	Altitude	FL 080	FL 080		
High Speed TAS Fuel Flow Altitude 162 162 High Speed Fuel Flow Altitude 116 116 Specific Range 1.397 1.397 Max Payload Nautical Miles Specific Range/Altitude 300 300 Max Payload Average Speed Tip Fuel 315 315 Specific Range/Altitude 0.952/FL 080 0.952/FL 080 0.952/FL 080 Nautical Miles 1.000 1.000 1.000 Average Speed 145 145 145 Ferry Nautical Miles 1.020 1.026/FL 080 1.026/FL 080 Missions Specific Range/Altitude 1.026/FL 080 1.026/FL 080 1.450 Missions Runway 1.450 1.450 1.450 Missions Specific Range/Altitude 1.429/FL 080 1.429/FL 080 1.429/FL 080 Missions Specific Range/Altitude 1.429/FL 080 1.429/FL 080 1.429/FL 080 Missions Suggested Base Price 375 375 375 Specific Range/Altitude 1.333/FL 0			Specific Range	1.435	1.435		
High Speed Fuel Flow Altitude 116 116 Ranges Altitude FL 080 FL 080 Specific Range 1.397 1.397 Max Payload Specific Range 300 300 300 300 Max Payload Average Speed 140 140 140 Trip Fuel 315 315 315 Specific Range/Altitude 0.952/FL 080 0.952/FL 080 0.952/FL 080 Nautical Miles 1.0000 1.0000 1.0000 Average Speed 145 145 Trip Fuel 975 975 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Missions Specific Range/Altitude 1.450 1.4450 High Specific Range/Altitude 1.450 1.4450 Koccupants) Rumway 1.450 1.429/FL 080 Missions Specific Range/Altitude 1.429/FL 080 1.429/FL 080 Koccupants) Specific Range/Altitude 1.333/FL 080 1.333/FL 080 Suggested Base Price \$830,800 <th></th> <th></th> <th>TAS</th> <th>162</th> <th>162</th>			TAS	162	162		
Max Payload Atitude Specific Range FL 080 FL 080 Nautical Miles 300 300 300 Nautical Miles 300 300 300 Average Speed 140 140 Trip Fuel 315 315 Specific Range/Altitude 0.952/FL 080 0.952/FL 080 Nautical Miles 1.000 1.0000 Nautical Miles 1.026/FL 080 1.026/FL 080 Missions Rumway 1.450 1.4450 Missions Specific Range/Altitude 1.429/FL 080 1.429/FL 080 Missions Specific Range/Altitude 1.429/FL 080 1.429/FL 080 Sourceit Range/Altitude 1.333/FL 080 1.333/FL 080 1.333/FL 080 Sugg		High Speed	Fuel Flow	116	116		
Autor Autor Autor Autor Name 300 300 300 Natical Miles 300 300 300 Natical Miles 300 300 300 Ranges Specific Range/Altitude 140 140 Specific Range/Altitude 0.952/FL 080 0.952/FL 080 0.952/FL 080 Ferry Nautical Miles 1,000 1,000 Average Speed 145 145 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Block Time 1+28 1+28 Participant Fuel Used 140 140 (4 occupants) Specific Range/Altitude 1.420/FL 080 1.429/FL 080 Suggested Base Price \$830,800 \$848,200 \$848,200 Remarks Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data.		0 11	Altitude	FL 080	FL 080		
Max Payload Max Payload 300 300 300 Average Speed 440 140 140 Ranges Specific Range/Altitude 0.952/FL 080 0.952/FL 080 Nautical Miles 1,000 1,000 Nautical Miles 1,000 1,000 Nautical Miles 1.026/FL 080 0.952/FL 080 Ferry Provide Marage Speed 145 145 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Block Time 1+28 1+28 Hards 140 140 Advances Specific Range/Altitude 1.420/FL 080 1.429/FL 080 (4 occupants) Specific Range/Altitude 1.323/FL 080 1.333/FL 080 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 Specific Range/Altitude 1.333/FL 080 1.333/FL 080			Specific Range Nautical Miles	1.397	1.397		
Max Payload Tip Fuel Specific Range/Altitude 315 315 Ranges Nautical Miles 1,000 1,000 Nautical Miles 1,026/FL 080 1.026/FL 080 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Block Time 1+28 1+28 H450 1.450 Missions Q00 nm Runway 1,450 1.429/FL 080 Specific Range/Altitude 1429/FL 080 1.429/FL 080 (40 occupants) Specific Range/Altitude 375 375 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 1.333/FL 080 S8488,200 \$848,200 \$843,080 <t< th=""><th></th><th></th><th>Average Speed</th><th>140</th><th>140</th></t<>			Average Speed	140	140		
Ranges Specific Range/Altitude 0.952/FL 080 0.952/FL 080 Nautical Miles Perry Perry Rumway Provide Specific Range/Altitude Specific Rang		Max Payload	Trip Fuel	315	315		
Ferry Nautical Miles Average Speed Trip Fuel 1,000 1,000 975 145 145 975 975 975 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Missions (4 occupants) Runway 1,450 1.450 500 nm Specific Range/Altitude 1420/FL 080 1.429/FL 080 8 Runway 1.420/FL 080 1.429/FL 080 1.429/FL 080 9 Specific Range/Altitude 1.429/FL 080 1.429/FL 080 1.429/FL 080 9 Block Time 3+25 3+25 3+25 9 Fuel Used 375 375 375 9 Suggested Base Price \$830,800 \$848,200 8 Remarks Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data. SEAS 23, 2009	Ranges		Specific Range/Altitude	0.952/FL 080	0.952/FL 080		
Ferry Average Speed Trip Fuel 145 145 975 975 975 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Rumay 1.450 1.450 Missions Rumay 1.425 1.450 Missions Specific Range/Altitude 1.026/FL 080 1.4250 Missions Specific Range/Altitude 1.420 1.420 Missions Specific Range/Altitude 1.429/FL 080 1.429/FL 080 Specific Range/Altitude 1.429/FL 080 1.429/FL 080 1.500 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 1.333/FL 080 Suggested Base Price \$830,800 \$848,200 \$848,200 Remarks Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data. SEAS 23, 2009 SBCA estimated data.			Nautical Miles	1,000	1,000		
Tirp Fuel 975 975 Specific Range/Altitude 1.026/FL 080 1.026/FL 080 Rumway 1.450 1.4450 Missions 1420 1.420 Year Fuel Used 140 Fuel Used 140 140 Specific Range/Altitude 1.429/FL 080 1.429/FL 080 Specific Range/Altitude 1.429/FL 080 1.429/FL 080 Soon m Block Time 3+25 3+25 Block Time 3+25 3+25 3+25 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 1.333/FL 080 Suggested Base Price \$\$830,800 \$\$848,200 \$\$848,200 Garmin G950; STEC 55X DFCS. BCA estimated data. \$\$BCA estimated data. \$\$BCA estimated data.		Ferry	Average Speed	145	145		
Specific Range/Attude 1.020/FL 080 1.020/FL 080 8 Rumay 1.450 1.450 Missions Block Time 1+28 1+28 Missions Fuel Used 140 140 6 Specific Range/Attitude 1.429/FL 080 1.429/FL 080 8 Rumay 1.500 1.500 8 Block Time 3+25 3+25 9 Block Time 3+25 3+25 9 Block Time 3+25 3+25 9 Suggested Base Price \$830,800 \$848,200 8 Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. SEA estimated data.			Trip Fuel	975	975		
Numey 1,450 1,450 200 nm Block Time 1+28 1+28 Missions (4 occupants) Specific Range/Altitude 1.429/FL 080 1.429/FL 080 Specific Range/Altitude 1.429/FL 080 1.429/FL 080 Sugested Base Price 375 375 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 Suggested Base Price \$830,800 \$848,200 Remarks Certification Basis FAR 23,1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data.			Specific Range/Altitude	1.026/FL 080	1.026/FL 080		
200 nm 200 nm 2120			Block Time	1+28	1+28		
Missions (4 occupants) Specific Range/Altitude 1.429/FL 080 1.429/FL 080 800k Time Fuel Used 1,500 1,500 1,500 900 nm Block Time Fuel Used 375 375 900 nm Specific Range/Altitude 1.333/FL 080 1.333/FL 080 900 nm Suggested Base Price \$830,800 \$848,200 900 nm Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data. EASA 23, 2009 Garmin G950; STEC 55X DFCS. BCA estimated data. BCA estimated data.		200 nm	Fuel Used	140	140		
(4 occupants) Rumway 1,500 1,500 500 nm Block Time Fuel Used 3+25 3+25 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 Suggested Base Price \$\$830,800 \$\$848,200 Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data. EASA 23, 2009 Garmin G950; STEC 55X DFCS. BCA estimated data.	Missions		Specific Range/Altitude	1.429/FL 080	1.429/FL 080		
Block Time Fuel Used 3+25 3+25 Specific Range/Altitude 375 375 Suggested Base Price \$830,800 \$8488,200 Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data. EASA 23, 2009 Garmin G950; STEC 55X DFCS. BCA estimated data.	(4 occupants)		Runway	1,500	1,500		
Fuel Used 375 375 Specific Range/Altitude 1.333/FL 080 1.333/FL 080 Suggested Base Price \$830,800 \$848,200 Remarks Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data. EASA 23, 2009 Garmin G950; STEC 55X DFCS. BCA estimated data.		500 nm	Block Time	3+25	3+25		
Specific kange/Altitude 1.333/FL 080 1.333/FL 080 Suggested Base Price \$830,800 \$848,200 Remarks Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data. EASA 23, 2009 Garmin G950; STEC 55X DFCS. BCA estimated data.			Fuel Used	375	375		
Suggested base Proce \$830,800 \$848,200 Remarks Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. EASA 23, 2009 Garmin G950; STEC 55X DFCS. BCA estimated data. BCA estimated data.			Specific Range/Altitude	1.333/FL 080	1.333/FL 080		
Remarks Certification Basis FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data. BCA estimated data.			Suggested Base Price	\$830,800	\$848,200		
	Remarks		Certification Basis	FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated data.	EASA 23, 2009 Garmin G950; STEC 55X DFCS. BCA estimated data.		

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Multiengine Pistons turbocharged

Mandal			P 68C-TC	Seneca V	
Model			P 68C-TC	PA-34-220T	
BCA Equipped Pr	ice		\$877,500	\$999,900	
		Seating	1+5/5	1+4/5	
Character-		Wing Loading	20.7	22.8	
istics		Power Loading	10.94	10.80	
		Noise (dBA)	74.7	75.6	
External		Length	31.3	28.6	
Dimensions		Height	11.2	9.9	
(ft.)		Span	39.4	38.9	
Internal		Length	10.6	10.4	
Dimensions		Height	3.9	3.6	
(ft.)		Width	3.8	4.1	
		Endinos	2 Lyc	2 Cont	
Power		Lingines	TIO-360-C1A6D	TSI0-360-RB	
1 000001		Output (hp each)	210	220	
		Inspection Interval	2,000t	1,800t	
		Max Ramp	4,630	4,773	
		Max Takeoff	4,594	4,750	
		Max Landing	4,365	4,513	
		Zero Fuel	4,140b	4,479c	
Weights (lh)		EOW	3,197	3,491	
101B1100 (101)		Max Payload	943	988	
		Useful Load	1,433	1,282	
		Max Fuel	1,062	732	
	Av	ailable Payload w/Max Fuel	371	550	
	Av	ailable Fuel w/Max Payload	490	294	
		VNE	194	204	
Limits		VNO	154	164	
		VA	132	139	
		TO (SL elev./ISA temp.)	1,260	1,707	
		T0 (5,000-ft. elev.@25C)	2,200	2,435	
Airport		A/S (SL elev./ISA)	1,800	2,510	
Airport		A/S (5,000-ft. elev.@25C)	2,400	3,117	
Perior-	A/S (5,000-tt. elev.@25C) Vmca Vdec Vxse		66	66	
mance			NA	73	
			78	83	
		Vyse	88	88	
	Ti	me to Climb (min.)/Altitude	10/FL 100	7/FL 100	
Olivert	Ir	itial Engine-Out Rate (fpm)	240	253	
Climb	Initial A	All-Engine Gradient (ft./nm)	1,400	996	
	Initial Engine-Out Gradient (ft./nm)		NA	173	
		Certificated	20.000	25.000	
Ceilings (ft.)		All-Engine Service	20.000	25.000	
		Engine-Out Service	10.000	16,500	
		TAS	144	167	
		Fuel Flow	104	108	
	Long Range	Altitude	FL 080	FL 230	
		Specific Range	1.385	1.546	
		TAS	155	196	
0		Fuel Flow	125	144	
Cruise	Recommended	Altitude	FL 080	FL 250	
		Specific Range	1.240	1.361	
		TAS	162	200	
		Fuel Flow	150	156	
	High Speed	Altitude	FL 080	FL 230	
		Specific Range	1.080	1.282	
		Nautical Miles	1,100	866	
		Average Speed	145	160	
Range	Ferry	Trip Fuel	960	648	
		Specific Range/Altitude	1.146/FL 080	1.336/FL 180	
		Runway	NA	1.520	
		Block Time	1+28	1+10	
	200 nm	Fuel Used	260	213	
Missions		Specific Range/Altitude	0.769/FL 080	0.939/FL 120	
(4 occupants)		Runway	NA	1.610	
()		Block Time	3+25	2+41	
	500 nm	Fuel Used	485	476	
	Specific Range/Altitude		1.031/FL 080	1.050/FL 200	
			\$877.500	\$999 900	
Remarks		Certification Basis	FAR 23, 1982 Garmin G950 glass cockpit; STEC 55X DFGS. BC4 estimated data	FAR 23, 1971/80/9 Garmin G1000 standard.	

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Manufacturer			Textron Aviation
Model			Beechcraft Baron G58
	Duine		G58
BCA Equipped	Price	Contine	\$1,400,000
Character-		Wing Loading	27.6
istics		Power Loading	9.17
100100		Noise (dBA)	77.6
External		Length	29.8
Dimensions		Height	9.8
(ft.)		37.8	
Internal		12.6	
Dimensions		4.2	
(IT.)		3.5 2 Cont	
_		I0-550-C	
Power		300	
		Inspection Interval	1,900t
		Max Ramp	5,524
		Max lakeott	5,500
		Max Landing Zero Fuel	5,400 5,215b
		FOW	3 970
Weights (lb.)		Max Payload	1,245
		Useful Load	1,554
		1,164	
	А	vailable Payload w/Max Fuel	390
	A	309	
Limite		223	
Linnes		VNO Va	165
		TO (SL elev./ISA Temp.)	2,345
		T0 (5,000-ft. elev.@25C)	4,144
		A/S (SL elev./ISA)	3,009
Airport		A/S (5,000-ft. elev.@25C)	4,335
Performance		Vmca	84
		100	
		VXSE	100
		10/FL 100	
Climb		390	
CIIIID	Initial	988	
	Initial	232	
Opiling (ft.)			
Cellings (IL.)		All-Eligine Service	20,688
	Engine-Out Service		185
		Fuel Flow	144
	Long Range Recommended High Speed	Altitude	FL 080
		Specific Range	1.285
		TAS	192
Cruise		Fuel Flow	174
		Alulude Specific Pande	FL 080
		TAS	200
		Fuel Flow	190
		Altitude	FL 080
		Specific Range	1.053
		Nautical Miles	333
	Max Payload	Average Speed	178
Ranges		ITIP FUEL Specific Range/Altitude	293 1 137/FL 040
Nunges		Nautical Miles	1.480
	F	Average Speed	180
	Ferry	Trip Fuel	1,081
		Specific Range/Altitude	1.369/FL 120
		Runway	2,862
	200 nm	Block Time	21+02
Missions		Specific Range/Altitude	0.885/FL 060
(4 occupants)		Runway	2,941
(,,,,,,, .	E00 mm	Block Time	2+31
	500 nm	Fuel Used	531
	Specific Range/Altitude		0.942/FL 060
		Suggested Base Price	\$1,400,000
Remarks		CAR 3, 1957/69/83/2005 A/C system standard; Garmin G1000.	

Single-Engine Turboprops

Manufacturer		Mahindra Aerospace	Textron Aviation	Piper Aircraft	Quest Aircraft	Textron Aviation	
Model		Airvan 10	Cessna Caravan	M500	Kodiak Kodiak 100	Cessna Grand Caravan EX	
BCA Equipped	d Price		\$999 500*	\$1,950,000	\$1 999 900	\$2.454.725	\$2 527 900
DUA LQUIPPEC		Seating	1+9/	1+9/13*	1+4/5	1+6/9	1+9/13*
Character-	Wing Loading Power Loading		28.6	28.6	27.8	30.2	31.3
istics			10.56	11.85	10.18	9.67	10.16
		Noise (dBA)	79.0	79.0	76.8	84.4	84.1
External		Length	33.5	37.6	29.6	33.8	41.6
Dimensions		Height	12.7	14.9	11.3	15.3	14.8
(ft.)		Span	40.6	52.1	43.0	45.0	52.1
Internal		Length	16.1	12.7	12.3	15.8	16.7
Dimensions		Height	3.8	4.5	3.9	4.8	4.5
(ft.)	width		4.2	5.3	4.1	4.5	5.3
		Engine	M250 B-17F/2	P&WC PT6A-114A	P&WC PT6A-42A	P&WC PT6A-34	P&WC PT6A-140
Power		Output (shp)/Flat Rating	450/ISA+31C	675/ISA+31C	500/ISA+55C	750/ISA+7C	867/ISA+24C
	Inspection Interval		3,500t	3,600t	3,600t	4,000t	3,600t
	Max Ramp Max Takeoff Max Landing Zero Fuel		4,775	8,035	5,134	7,305	8,842
			4,750	8,000	5,092	7,255	8,807
			4,750	7,800	4,850	7,255	8,500
			4,182b	7,432b	4,850c	6,490c	8,152b
Weights (lb.)		BOW Nex Devload	2,475	4,930	3,634	4,417	5,510
		Ividx Pdylodu	2 300	2,302	1,210	2,075	3 332
		Max Fuel	1 025	2 224	1 160	2,000	2 246
	Available Pavload w/Max Fuel		1,275	881	340	744	1,086
	Ava	ailable Fuel w/Max Payload	594	604	284	815	691
		Vмо	175	175	188	180	175
Limits		VA	150	150	127	143	148
		PSI	—	—	5.6	_	—
A time is us		TO (SL elev./ISA temp.)	1,600	2,055	2,438	1,468	2,160
Airport		10 (5,000-tt. elev.@25C)	2,973	2,973	3,691	2,396	3,661
Perfor-		VSO	61	61	69	60	61
mance	¥X. Me		107	107	125	101	108
<u></u>	Tir	me to Climb (min.)/Altitude	9/FL 100	9/FL 100	19/FL 250	9/FL 100	9/FL 100
Climb		Initial Gradient (ft./nm)	771	771	753	915	816
		Certificated	20,000	25,000	30,000	25,000	25,000
Ceilings (ft.)	Service		25,000	25,000	30,000	25,000	25,000
		Sea-Level Cabin	_	_	12,600	_	
		IAS Evel Flave	157	157	179	164	156
	Long Range	Fuel Flow Altitude	281 FL 100	281 FL 100	135 FL 280	201	328 FL 100
Cruise	High Speed	Specific Range	0.559	0.559	1.326	0.653	0.476
		TAS	186	186	258	175	185
		Fuel Flow	379	379	242	335	437
		Altitude	FL 100	FL 100	FL 280	FL 120	FL 100
		Specific Range	0.491	0.491	1.066	0.522	0.423
	Full Fuel	Nautical Miles	965	288	834	1,005	291
	(with available	Average Speed	156	153	1/1	1/5	155
NDAA IFR	payload)	Inp Fuel Specific Pande /Altitude	1,795 0.538/FL 100	0.496/FL 100	1 115/EL 280	2,130	0.430/EL 100
(100-nm		Nautical Miles	970	970	834	1.236	816
alternate)	_	Average Speed	156	156	171	164	156
	Ferry	Trip Fuel	1,800	1,800	748	2,130	1,772
		Specific Range/Altitude	0.539/FL 100	0.539/FL 100	1.115/FL 280	0.580/FL 200	0.460/FL 100
		Runway	1,468	1,468	1,550	1,468	1,428
	300 nm	Flight Time	1+40	1+40	1+22	1+47	1+41
		Fuel Used	0.463/EL 100	0.463/51.400	3/9	58/ 0.511/EL 120	
		Specific Ralige/Altitude	1.675	1 675	0.792/FL 280	0.511/FL 120	1 792
Missions		Flight Time	3+17	3+17	2+32	3+30	3+19
(4 passen-	600 nm	Fuel Used	1,260	1,260	660	1,140	1,462
gers)		Specific Range/Altitude	0.476/FL 100	0.476/FL 100	0.909/FL 280	0.526/FL 120	0.410/FL 100
	1,000 nm	Runway	NP	NP	1,700	1,467	NP
		Flight Time	NP	NP	4+18	5+47	NP
		Fuel Used	NP	NP	985	1,878	NP
		Specific Range/Altitude	NP	NP	1.015/FL 280	0.532/FL 120	NP
		Suggested base Price	INA	INA	\$1,999,900	\$2,075,000	INA
Remarks		Certification Basis	FAR 23, 1984/98 *BCA estimated price. Garmin G1000 with GFC700 autopilot. 2016 data.	FAR 23, 1984/98 *Export only. Garmin G1000 with GFC700 autopilot.	FAR 23 A 52 *1,000 nm, 3 passengers. Garmin G1000 with SVS.	FAR 23, 2007 Normal category Includes Garmin G1000; GFC700 with coupled GA; Summit interior option.	FAR 23, 1986/2012 *Export only. Includes cargo pod, Garmin G1000 with GFC700 autopilot.

Single-Engine Turboprops

Manufacturer		Piper Aircraft	Epic Aircraft	Daher	Daher	Pilatus	
Model			M600	Epic	TBM 910	TBM 930	PC-12 NG
PO4 Equipped Dripp		PA-46-600TP	E1000	TBM 700 N	TBM 700 N	PC-12/47E	
BCA Equipped	a Price	Contine	\$2,899,000	\$2,995,000	\$3,683,260	\$3,979,750	\$4,923,000
Character		Seaung Wind Loading	28.7	26.9	1+5/0	1+5/0 38.2	37.6
ietice		PowerLoading	10.00	6.25	8 70	8 70	8 71
101100	Noise (dBA)		76.8	76.0	76.2	76.2	77.0
External		Length	29.6	35.8	35.2	35.2	47.3
Dimensions		Height	11.3	12.5	14.3	14.3	14.0
(ft.)		Span	43.2	43.0	42.1	42.1	53.3
Internal		Length	12.3	10.5	15.0	15.0	16.9
Dimensions		Height	3.9	4.9	4.1	4.1	4.8
(ft.)		Width	4.1	4.6	4.0	4.0	5.0
(***)			P&WC	P&WC	P&WC	P&WC	P&WC
Power		Engine	PT6A-42A	PT6A-67A	PT6A-66D	PT6A-66D	PT6A-67P
		Output (shp)/Flat Rating	600/ISA+55C	1,200/ISA+35C	850/ISA+49C	850/ISA+49C	1,200/ISA+35C
	Inspection Interval		3,600t	3,500t	3,500t	3,500t	3,500t
	Max Ramp		6,050	7,500	7,430	7,430	10,495
		Max Takeoff	6,000	7,500	7,394	7,394	10,450
		Max Landing	5,800	7,500	7,024	7,024	9,921
		Zero Fuel	4,8500	5,4000	6,0320	6,0320	9,0390
Weights (lb.)		BUW Max Payload	3,850	4,000	4,029	4,029	0,762
		WidX Pdyludu	2,000	2 900	2 601	2 601	2,257
		May Fuel	1 7/12	1,876	2,001	2,001	2 704
	Available Payload w/Max Fuel Available Euel w/Max Payload		458	1 024	584	584	1,009
			1 200	2 100	1.398	1.398	1 456
		VM0	250	280	266	266	240
Limits		VA	151	170	160	160	163
		PSI	5.6	6.7	6.2	6.2	5.8
		TO (SL elev./ISA temp.)	2,635	1,600	2,380	2,380	2,600
Airport		TO (5,000-ft. elev.@25C)	3,998	NA	3,475	3,475	4,270
Perfor-	Vso		62	65	65	65	67
mance		Vx	95	124	100	100	120
		VY	122	144	124	124	130
Climb	Tir	ne to Climb (min.)/Altitude	21/FL 250	10/FL 250	13/FL 250	13/FL 250	20/FL 250
	Initial Gradient (ft./nm)		785	1,500	1,000	1,000	860
0 - 11: (6+)	Certificated		30,000	34,000	31,000	31,000	30,000
Cellings (ft.)		Service	30,000	34,000	31,000	31,000	30,000
		Sea-Level Cabin	12,600	18,000	14,390	14,390	13,100
		IAS Fuel Flow	184	200	252	252	225
	Long Range	Altitudo	EL 290	Z00	241 EL 210	241 EL 210	200 EL 200
	High Speed	Specific Pande	1 1 1 8 7	0.989	1.046	1.046	0.840
Cruise		Specific Range	274	330	330	330	285
		Fuel Flow	324	402	412	412	497
		Altitude	FL 280	FL 280	FL 260	FL 260	FL 200
		Specific Range	0.846	0.821	0.801	0.801	0.573
		Nautical Miles	1,406	1,650	1,514	1,514	1,608
	Full Fuel	Average Speed	179	265	252	252	261
NBAA IFR	(with available payload)	Trip Fuel	1,324	1,599	1,599	1,599	2,282
Ranges	pujiouuj	Specific Range/Altitude	1.062/FL 280	1.032/FL 310	0.947/FL 310	0.947/FL 310	0.705/FL 300
(100-nm		Nautical Miles	1,406	1,594	1,594	1,594	1,650
alternate)	Ferry	Average Speed	179	252	252	252	264
	,	Trip Fuel	1,324	1,598	1,598	1,598	2,294
		Specific Range/Altitude	1.062/FL 280	0.997/FL 310	0.997/FL 310	0.997/FL 310	0.719/FL 300
		Runway	1,593	1,765	1,/65	1,/65	1,563
	300 nm	Flight Lime	1421	1+00	140 140	140	1+10
		Fuel Used Spacific Pango/Altitudo	429 0.600/EL 280	0.692/EL 290	440	0.692/EL 290	0.546/EL 260
Missions		Specific Ralige/Altitude	1 687	2 005	2 005	2 005	1 753
		Flight Time	2+31	1+55	1+55	1+55	2+16
(4 passen-	600 nm	Fuel Used	735	830	830	830	975
gers)		Specific Range/Altitude	0.816/FL 280	0.723/FL 280	0.723/FL 280	0.723/FL 280	0.615/FL 270
	1,000 nm	Runway	1,812	2,380	2,380	2,380	2,026
		Flight Time	4+06	3+10	3+10	3+10	3+46
		Fuel Used	1,142	1,320	1,320	1,320	1,520
		Specific Range/Altitude	0.876/FL 280	0.758/FL 290	0.758/FL 290	0.758/FL 290	0.658/FL 280
		Suggested Base Price	\$2,899,000	NA	\$3,658,336	\$3,899,887	\$4,095,000
Remarks		Certification Basis	FAR 23 A 62, 2016 Garmin G3000 with SVS and enhanced AFCS.	FAR 23 pending Garmin G1000 NXI.	FAR 23, 1990/2006/07/14 Pilot door standard; 5-blade propeller; G1000 NXi; AA-ESP-USP; satcom; weather; 5-year system warranty.	FAR 23, 1990/2006/07/14 All features of TBM 900 plus advanced interior; Garmin G3000; 5-year system warranty.	FAR 23, 1996/2005/08 Honeywell APEX avionics: SmartView; ADS-B Out; BMW executive interior; Hartzell 5-blade propeller.
Multiengine Turboprops ≤12,500-LB. MTOW

Manufacturer			Vulcanair SpA	Nextant Aerospace	Evektor	Textron Aviation
Model			Viator	G90XT	Outback	Beechcraft King Air C90GTx
BC4 Equipped Price			\$2 485 900	\$2,750,000	\$3,000,000	\$3 595 000
		Seating	1+7/10	1+7/10	1+9/14	1+7/8
0.		Wing Loading	33.0	34.4	37.4	34.4
Characteristics		Power Loading	10.08	9.55	9.46	9.53
		Noise (dBA)	71.7	71.7	NA	74.8
External		Length	37.0	35.5	46.6	35.5
Dimensions (ft.)		Height	20.4	14.3	10.8	14.3
		Length: OA/Net	11.9/17.2	12.4/12.4	16.5/20.0	12.4/12.4
Internal		Height	4.1	4.8	4.5	4.8
Dimensions (ft.)		Width: Max/Floor	3.7/3.7	4.5/4.1	5.3/4.7	4.5/4.1
		Engines	2 RR	2 GE Czech	2 P&WC	2 P&WC
Power			250 B1/C	H75-100	P16A-21	P16A-135A
	0	utput (snp each)/ Flat Rating	328/ISA+250 2 500t	4 000t	2 600t	2 600t
		Max Ramn	6 669	10 560	10 207	10 545
		Max Takeoff	6,613	10,500	10,141	10,485
		Max Landing	6,283	9,700	10,141	9,832
		Zero Fuel	5,621c	9,650c	9,810c	9,378c
Weights (lb.)		BOW	3,850	7,200	5,965	7,265
110.B.100 (101)		Max Payload	1,771	2,450	3,845	2,113
		Usetul Load Max Fuol	2,819	3,360	4,242	3,280
	Α	vailable Pavload w/Max Fuel	1 332	787	829	707
	A	vailable Fuel w/Max Payload	1,048	910	397	1,167
		Vmo	200	208	205	226
Limits		VA	141	169	140	169
		PSI	-	5.0	-	5.0
		IU (SL elev./ISA temp.)	2,034	2,100	1,378	1,984
		A/S (SL elev /ISA temp)	2,950	2,800	1,837	3,375
Airport Performance		A/S (5,000-ft. elev.@25C)	2,953	5,100	2,395	5,855
		VMCA	77	92	66	80
		VDEC	85	97	79	97
		Vxse	90	101	92	100
		Vyse	105	111	95	108
		lime to Climb (min.)/Altitude	//FL 100	18/FL 250	6/FL 010	18/FL 250
Climb	Initial	All-Engine Gradient (ft /nm)	1 500	1 900	1 107	1 900
	Initial I	Engine-Out Gradient (ft./nm)	180	260	219	260
		Certificated	25,000	30,000	24,000	30,000
Ceilings (ft)		All-Engine Service	25,000	30,000	24,000	30,000
		Engine-Out Service	8,050	22,000	15,420	19,230
		Sea-Level Cabin		11,065		11,065
		Fuel Flow	261	213	432	332
	Long Range	Altitude	FL 100	FL 280	FL 010	FL 260
Oruioo		Specific Range	0.648	0.729	0.417	0.627
Cruise		TAS	214	283	220	270
	High Speed	Fuel Flow	375	578	610	612
	• •	Altitude Specific Bando	FL 100	FL 240	FL 200	FL 200
		Nautical Miles	543	324	0.361	260
	Max Pavload	Average Speed	180	203	NP	229
	(with available fuel)	Trip Fuel	781	600	NP	620
		Specific Range/Altitude	0.695/FL 100	0.540/FL 220	NP/-	0.419/FL 270
	Max Fuel	Nautical Miles	837	1,300	1,046	1,026
	(with available	Average Speed	179	207	217	252
	payload)	Trip Fuel	1,220	1,782	3,008	2,044
(100 pm alternata)		Specific Kange/Altitude	0.080/FL 100	0.730/FL 280	0.348/FL 100	0.502/FL 2/0
(100-nin alternate)	Full Fuel	Average Sneed	179	207	217	252
	(with 4 passsengers)	Trip Fuel	1,220	1,769	3,008	1,949
	,	Specific Range/Altitude	0.686/FL 100	0.729/FL 280	0.348/FL 100	0.500/FL 270
		Nautical Miles	837	1,369	1,051	1,045
	Ferry	Average Speed	179	203	218	255
	,	Trip Fuel	1,220	1,850	3,008	2,053
		Specific Kange/Altitude	0.080/FL 100	0.740/FL 280	0.349/FL 100 2 162	0.509/FL 2/0
		Flight Time	1+35	3,010	1+26	3,004
	300 nm	Fuel Used	419	584	943	748
		Specific Range/Altitude	0.716/FL 100	0.514/FL 220	0.318/FL 100	0.401/FL 210
		Runway	1,558	3,350	1,289	3,347
Missions	600 nm	Flight Time	3+18	2+12	2+22	2+22
(4 passengers)		Fuel Used	866	1,162	1,773	1,353
		Specific Range/Altitude	0.095/ FL 100	3 500	0.336/FL 100 1 565	3.690
		Flight Time	NP	3+39	4+ 36	3+57
	1,000 nm	Fuel Used	NP	1,938	2,881	1,990
		Specific Range/Altitude	NP/NP	0.516/FL 280	0.347/FL 100	0.503/FL 270
		Suggested Base Price	\$2,485,900	NA	NA	NA
Remarks	Suggested Base Price narks Certification Basis		FAR 23, 1986 Garmin G950; STEC 2100 autopilot. BCA estimated data.	ST01902CH; SA3593NM; SA4010NM; SA3593NM; SA01902CH; SA01456WI-D; SA02133SE.	EASA/FAR 23 pending 2016 data.	CAR 3, 1959/2007 Pro Line Fusion standard.; STC SA10747SC weight increase; SA02054SE winglets; SA3593NM swept props; SA4010NM dual aft strakes; 1,000-mm mission 755-lb ald

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$Multiengine \ Turboprops \ \leq 12,500 \text{-LB. MTOW}$

Manufacturer		Textron Aviation	Viking Air	Piaggio Aero Industries	
Model			Beechcraft King Air 250	400 Series	Avanti Evo
WIUUEI			B200GT	DHC-6-400	P180
BCA Equipped Price		0	\$5,995,000	\$6,500,000	\$7,695,000
		Seating	1+8/10	1+11/19	1+7/9
Characteristics		Wing Loading	40.3	29.8	70.3
		Power Loading Noise (dBA)	7.35 TRD	10.08	7.12
		Length	43.8	51.8	47.3
External		Height	14.8	19.5	13.0
Dimensions (ft.)		Span	57.9	65.0	46.0
Internal		Length: OA/Net	16.7/16.7	18.4/24.5	17.5/17.5
Dimensione (ft.)		Height	4.8	4.9	5.8
Dimensions (ft.)		Width: Max/Floor	4.5/4.1	5.4/4.4	6.1/3.5
		Engines	2 P&WC	2 P&WC	2 P&WC
Power		Engines	PT6A-52	PT6A-34	PT6A-66B
	C	utput (shp each)/Flat Rating	850/ISA+37C	620/ISA+27C	850/ISA+28C
		Inspection Interval	3,600t	3,600t	3,600t
		Max Ramp	12,590	12,525	12,150
		Max Landing	12,500	12,300	11,500
		Zero Fuel	11 000c	11 655b	9 800c
		BOW	8 830	8 100	8.375
Weights (lb.)		Max Pavload	2 170	3 555	1 425
		Useful Load	3.760	4.425	3.775
		Max Fuel	3,645	3,549	2,802
	A	vailable Payload w/Max Fuel	115	876	973
	A	vailable Fuel w/Max Payload	1,590	870	2,350
		VMO	260	170	260
Limits		VA	182	136	202
		PSI	6.5		9.0
		TO (SL elev./ISA temp.)	2,111	1,490	3,262
		IU (5,000-ft. elev.@25C)	3,099	NA	4,700
Airport		A/S (SL elev./ISA temp.)	3,687	2,220	5,750
Dorformonoo		A/ 5 (0,000-11. elev.@200)	4,859	NA	1,400
Performance		VMUA	94	NA	106
		VXSE	115	NA	132
		Vyse	121	NA	140
		Time to Climb (min.)/Altitude	13/FL 250	NA/FL 100	10/FL 250
Olinah		Initial Engine-Out Rate (fpm)	682	340	670
CIIIID	Initia	All-Engine Gradient (ft./nm)	1,170	NA	1,106
	Initial	Engine-Out Gradient (ft./nm)	364	NA	287
Ceilings (ft.)		Certificated	35,000	25,000	41,000
		All-Engine Service	35,000	26,700	39,400
		Engine-Out Service	26,000	11,600	23,800
		Sea-Level Cabin	15,293		24,000
		IAS Eucl Flow	200	NA NA	318
	Long Range	Altitude	FI 350	FL 100	FL /10
		Snecific Range	0.595	NA	0.779
Cruise		TAS	310	180	400
		Fuel Flow	750	580	792
	nigii Speeu	Altitude	FL 260	FL 100	FL 310
		Specific Range	0.413	0.310	0.505
		Nautical Miles	321	NP	1,070
	Max Payload	Average Speed	267	NP	315
	(with available fuel)	Trip Fuel	870	NP	1,715
		Specific Range/Altitude	0.369/FL 330	NP	0.624/FL 390
	Max Fuel	Nautical Miles	1,403	NA	1,450
	(with available	Average Speed	291	NA NA	311
NBAA IFR Ranges	payload)	Specific Rande /Altitude	0.477/FL 330		0.669/FL /10
(100-nm alternate)		Nautical Miles	1 038	NA	1 510
	Full Fuel	Average Speed	288	NA	317
	(with 4 passsengers)	Trip Fuel	2,225	NA	2,167
		Specific Range/Altitude	0.467/FL 330	NA/FL 100	0.697/FL 410
		Nautical Miles	1,420	NA	1,530
	Form	Average Speed	293	NA	318
	relly	Trip Fuel	2,942	NA	2,167
		Specific Range/Altitude	0.483/FL 330	NA/FL 100	0.706/FL 410
		Runway	3,504	NA	2,350
	300 nm	Flight lime	1+03	NA	0+53
		Fuel Used	0.245/51.250	NA NA /FL 100	0.426/EL 240
		Specific Range/Altitude	2 527	NA/ FL 100	0.430/FL 310 2 550
Missions		Flight Time	2+03	NA	1+44
(4 passengers)	600 nm	Fuel Lised	1,494	NA	1,144
(Specific Range/Altitude	0.402/FL 290	NA/FL 100	0.524/FL 350
		Runwav	3,677	NA	2,700
	1 000 mm	Flight Time	3+28	NA	3+02
	1,000 IIM	Fuel Used	2,147	NA	1,603
		Specific Range/Altitude	0.466/FL 330	NA/FL 100	0.624/FL 390
		Suggested Base Price	NA	NA	\$7,395,000
Remarks		Certification Basis	FAR 23, 1973/80/2008/11 Rockwell Collins Pro Line Fusion	EASA/FAR 23 A 57, 2010	EASA 23, 2014; FAR 23, 2015 Includes Rockwell Collins Pro Line 21 avionics: TCAS I: Iridium satcom:
			Standard; WI-FI optional; STC SA02131SE.	2016 data.	RVSM approved; optional 390-lb. capacity internal tank: \$275,000.

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Multiengine Turboprops >12,500-LB. MTOW

Manufacture	r		Textron Aviation	Textron Aviation	Textron Aviation	Textron Aviation
Model			Beechcraft King Air 250 EP	Beechcraft King Air 350i	Beechcraft King Air 350HW	Beechcraft King Air 350iER
			B200GT	B300	B300	B300ER
BCA Equipped	d Price		\$6,231,025	\$6,995,000	\$7,329,055	\$8,445,625
Character		Seating	1+8/10	1+9/11	1+9/14	1+9/11
Character-		Wing Loading	43.3	48.4	53.2	53.2
istics		Power Loading	7.89	7.14	7.86	7.86
Enternal.		NOISE (dBA)	85.3	/2.1	81.5	81.5
External		Length	43.8	40.7	40.7	40.7
Dimensions		neigiit	14.0	14.5	14.5	14.5
(ft.)		Span	57.9	57.9	57.9	57.9
Internal		Length: OA/Net	16.7/16.7	19.5/19.5	19.5/19.5	19.5/19.5
Dimensions		Height	4.8	4.8	4.8	4.8
(ft.)		Width: Max/Floor	4.5/4.1	4.5/4.1	4.5/4.1	4.5/4.1
		Endinee	2 P&WC	2 P&WC	2 P&WC	2 P&WC
Power	Enginos		PT6A-52	PT6A-60A	PT6A-60A	PT6A-60A
i onci	()utput (shp each)/Flat Rating	850/ISA+37C	1,050/ISA+10C	1,050/ISA+10C	1,050/ISA+10C
		Inspection Interval	3,600t	3,600t	3,600t	3,600t
		Max Ramp	13,510	15,100	16,600	16,600
		Max landing	12,420	15,000	16,500	16,500
		VidX Edituting	11,000	12,5000	12,000	12,000
		Zelo Tuel ROW	8 865	9 955	9 290	10 215
Weights (lb.)		Max Pavload	2 135	2 545	3,710	2 785
		Useful Load	4 645	5 145	7 310	6.385
		Max Fuel	3.645	3.611	3.611	5,192
	A	vailable Payload w/Max Fuel	1,000	1,534	3,699	1,193
	Available Fuel w/Max Pavload		2,510	2,600	3,600	3,600
		Ммо	0.58	0.58	0.58	0.58
Limits		Trans. Alt. FL/VMO	FL 210/259	FL 210/263	FL 240/245	FL 240/245
Linito		VA	182	182	182	182
		PSI	6.5	6.6	6.6	6.5
		TO (SL elev./ISA temp.)	4,005	3,300	4,057	4,057
Airport		IUFL (5,000-ft. elev.@25C)	5,780	5,376	5,140	1,675
Dorfor		MISSION Weight	13,220	14,190	13,080	16,100
Perior-		INDAA IFK Kdiige	100	109	111	2,257
mance		V2	97	100	104	104
		Landing Distance	2.780	2,390	2,720	2.728
		Time to Climb (min.)/Altitude	15/FL 250	15/FL 250	23/FL 250	18/FL 250
Climb	*FAR 25 Initial Engine-Out Rate (fpm)		580	552	274	337
	FAR 25 Initial Engine-Out Gradient (ft./nm)		255	304	172	182
	Certificated		35,000	35,000	35,000	35,000
Ceilings (ft.)		All-Engine Service	35,000	35,000	35,000	35,000
6 6 m . B 6 (. u)		Engine-Out Service	24,400	21,500	17,100	17,100
		Sea-Level Cabin	15,293	15,293	15,293	15,293
		IA5 Evol Flow	233	235	232	238
	Long Range	Fuel Flow	309	302 EL 220	392	402 EL 220
		Aluuue Specific Rande	0.631	0.649	0.592	0.592
Cruise		TAS	308	312	303	303
		Fuel Flow	750	773	766	764
	Hign Speed	Altitude	FL 260	FL 240	FL 240	FL 240
		Specific Range	0.411	0.404	0.396	0.397
		Nautical Miles	802	896	1,254	1,316
	Max Payload (with available fuel)	Average Speed	275	273	258	261
		Trip Fuel	1,802	1,891	2,838	2,880
		Specific Range/Altitude	0.445/FL 330	0.474/FL 350	0.442/FL 350	0.457/FL 350
	Max Fuel	Nautical Miles	1,393	1,485	1,260	2,223
NBAA IER	(with available	Average Speed	283	280	258	269
Ranges	payload)	Irip Fuel	2,947	2,944	2,884	4,528
(100 pm		Specific Kange/Altitude	0.473/FL 330 1 414	0.504/FL 350 1 522	0.437/FL 350 1 427	0.491/FL 350
(100-IIII	Full Fuel	Averade Speed	285	2,555	276	2,211
alternate)	(with 4 passsengers)	Trin Fuel	2 950	200	2 930	4 533
	(inter i passoongols)	Specific Range/Altitude	0,479/FL 330	0.519/FL 350	0,490/FL 350	0.501/FL 350
		Nautical Miles	1,442	1,560	1,473	2,338
	Farme	Average Speed	289	289	282	276
	Ferry	Trip Fuel	2,956	2,958	2,942	4,543
		Specific Range/Altitude	0.488/FL 330	0.527/FL 350	0.501/FL 350	0.515/FL 350
		Runway	3,524	2,586	2,634	2,795
	300 nm	Flight Time	1+05	1+02	1+06	1+05
		Fuel Used	848	881	954	919
		Specific Range/Altitude	0.354/FL 250	0.341/FL 250	0.314/FL 250	0.326/FL 250
Missions		Runway Flight Time	3,011	2,702	2,/40	2,927
(4 passen-	600 nm	Fight Hille	1 472	1 470	1 561	1 529
gers)		Specific Range/Altitude	0.408/FI 290	0.408/FL 290	0.384/FL 290	0.392/FL 290
		Runway	3.702	2.827	2.883	3.048
	4	Flight Time	3+31	3+27	3+33	3+35
	1,000 nm	Fuel Used	2,123	2,102	2,227	2,195
		Specific Range/Altitude	0.471/FL 330	0.476/FL 330	0.449/FL 330	0.456/FL 330
		Suggested Base Price	NA	NA	NA	NA
Remarks		Certification Basis	FAR 23, 1973/80/2008/11 Commuter category Rockwell Collins Pro Line Fusion; Wi-Fi optional; STC SA11103SC for IGW; 14,000-ib. MTOW also available.	FAR 23, 1989 Commuter category Rockwell Collins Pro Line Fusion; Wi-Fi standard; RVSM approved.	FAR 23, 1989/2007 Commuter category 17,500-b. MTOW optional; Rockwell Collins Pro Line Fusion; Wi-Fi standard; factory-installed Slick interior available for special missions; BVSM approved	FAR 23, 1989/2007 Commuter category Rockwell Collins Pro Line Fusion; Wi-Fi standard; RVSM approved.

Manufacturer	r		Cirrus Design	Eclipse Aerospace
Model			Vision	Eclipse 550
BCA Equipped	l Price		\$1.960.000	\$2 995 000
DUA LQUIPPEC		Seating	1+4/6	1+4/5
Character-		Wing Loading	30.7	41.0
istics		Power Loading	1.67	3.33
Extornal	Noise (EPNdB)	: Lateral/Flyover/Approach	NA/NA/NA 20.7	69.2/78.9/81.9
Dimensions		Height	10.9	11.0
(ft)		Span	38.7	37.9
Internal		Length: OA/Net	11.5/9.8	12.3/10.0
Dimensions	н	leight/Dropped Aisle Depth	4.1/NA	4.2/NA
(ft.)	Width: Max/Floo		5.1/3.1	4.7/3.0
Porraro	Internal: Cu. ft./lb.		24/NA	16/260
Бавваве		External: Cu. ft./lb.	30/NA	NA/NA
		Engine(s)	1 Wms Intl	2 P&WC PW610F
Power	0)utput (lb. each)/Flat Rating	1.800/ISA+10C	900/ISA+10C
	Inspection Interval/	Manu. Service Plan Interval	3,500t/	3,500t/
		Max Ramp	6,040	6,034
		Max Takeoff	6,000	6,000
		Max Landing	5,550	5,600
		BOW	3,772	3,923
weights (lb.)		Max Payload	1,128	999
		Useful Load	2,268	2,111
	A	Max Fuel	2,000	1,680
	AV	raiidule rayload w/Max Fuel railable Fuel w/Max Pavload	<u>∠08</u> 1 140	431
	AV	Ммо	0.530	0.640
Limits		Trans. Alt. FL/VMO	FL 183/250	FL 200/285
		PSI	6.4	8.7
		TOFL (SL elev./ISA temp.)	2,036	2,394
Airport		Mission Weight	6,000	5 893
Perfor-		NBAA IFR Range	1,125	1,015
mance		V2	90	102*
		VREF	87	89
	Time to Climb/Altitude			2,340 25/EL 370
Climb	FAR 25 Engine-Out Rate (fpm)		NA	500
	FAR 25 Engine-Out Gradient (ft./nm)		NA	294
		Certificated	28,000	41,000
Ceilings (ft.)		All-Engine Service	28,000	25.000
	Sea-Level Cabin		NA	21,500
	TAS		256	334
	Long Range	Fuel Flow	358	321
		Altitude Specific Pande	FL 280	FL 410
Cruise		TAS	300	369
	High Spood	Fuel Flow	466	462
	nigii speeu	Altitude	FL 280	FL 350
		Specific Range	0.644	0.799
	May Payload	Average Speed	251	307
	(with available fuel)	Trip Fuel	845	677
		Specific Range/Altitude	0.651/FL 280	0.783/FL 410
	Max Fuel	Nautical Miles	1,167	1,125
NBAA IFR	(with available	Average Speed	248	319
Ranges	payload)	INP FUEL Specific Range/Altitude	0.728/EL 280	0.897/FL 410
(100-nm		Nautical Miles	796	825
alternate)	Four Passengers	Average Speed	250	317
	(with available fuel)	Trip Fuel	1,076	965
		Specific Range/Altitude	0.740/FL 280	0.855/FL 410
	_	Average Speed	218	312
	Ferry	Trip Fuel	1,680	1,263
		Specific Range/Altitude	0.726/FL 280	0.942/FL 410
		Runway	1,857	2,038
	300 nm	Flight Lime Fuel Lised	568	456
		Specific Range/Altitude	0.528/FL 280	0.658/FL 350
Missions		Runway	2,171	2,258
(4 passen-	600 nm	Flight Time	2+15	1+46
gers)		Fuel Used Specific Pange/Altitude	1,033 0.581/EL 280	0.717/EL 390
		Runway	2,437	2,318
	1 000 pm	Flight Time	3+36	3+04
	1,000 mill	Fuel Used	1,642	1,137
		Specific Range/Altitude	0.609/FL 280	0.880/FL 410
				FAR 23, 2006/15
Domorke		Contification D.	FAR 23, 2016	1,000-nm mission flown with
Remarks		Cerufication Basis	Some data preliminary.	*V50 used in lieu of V2.
				2016 data.

Manufacture	r		Textron Aviation	Embraer	Textron Aviation	Honda Aircraft Co.	Nextant Aerospace
Model			Cessna Citation Mustang	Phenom 100 EV	Cessna Citation M2	HondaJet	Nextant 400 XTi
	1.5.1		CE-510	EMB-500	CE-525	HA-420	BE 400A
BCA Equipped	a Price	Casting	\$3,350,000	\$4,495,000	\$4,500,000	\$4,850,000	\$5,304,500
Character-	1475	Seating	1+5/5/-	1+5/ // /	1+1/1/-	1+5/6/6	2+1/9/-
istics	Noise (EPNdB):	I steral /Flyover/Approach	41.2/2.90	70 / /81 / /86 1	44.0/2.12	85 //72 9/87 5	07.0/2.07 76.0/01.5/88.8
External	Hoise (El Hub).	Lateral/ Hyover/ Approach	40.6	42 1	42.6	42.6	48.4
Dimensions		Height	13.4	14.3	13.9	14.9	13.9
/f+)		Snan	43.2	40.4	47.3	39.8	43.5
(IL.)	Longth	Main Coating (Nat /Cross	6 7/0 8/0 8	0.0/11.0/11.0	9 9/11 0/11 0	10.1./10.1./NA	155/155/
Dimensione	Lengu	eight /Dronned Aisle Denth	4 5/0 3	4 9/0 3	4 8/0 4	12.1/12.1/NA 4.8/NA	4.8/flat floor
Dimensions		Width: May/Elear	4.6/0.0	F 1 /2 G	4.8/2.1	=.0/NA	4.0/4.0
(11.)		Internal On A //h	4.0/3.1	10/00	4.8/3.1	J.U/ INA	4.5/4.0
Baggage		External: Cu. IL/ID.	57/620	10/99	/	INA/INA	27/410
		External: Gu. IL/ID.	2 P&WC	2 P&WC	40/725 2 Wmc Intl	2 GE Hondo	20/430 2 Wmc Intl
_		Engines	PW615F	PW 617F-E	FJ44-1AP-21	HF-120-H1A	FJ44-3AP
Power	01	utput (lb. each)/Flat Rating	1.460/ISA+10C	1.730/ISA+8C	1.965/ISA+7C	2.037/ISA+10C	3.052/ISA+7C
	Inspection Interval/M	Manu. Service Plan Interval	3,500t/—	3,500t/—	3,500t/5,000	NA/—	5,000t/—
		Max Ramp	8,730	10,748	10,800	10,680	16,500
		Max Takeoff	8,645	10,703	10,700	10,600	16,300
		Max Landing	8,000	9,877	9,900	9,860	15,700
		Zero Fuel	6,750c	9,072c	8,400c	8,800c	13,000c
Weights (lh.)		BOW	5,600	7,298	6,990	7,279	10,950
Hoighto (ib.)		Max Payload	1,150	1,774	1,410	1,521	2,050
		Useful Load	3,130	3,450	3,810	3,401	5,550
		Max Fuel	2,580	2,804	3,296	2,845	4,912
	Ava A	allable Fuel w/Max Fuel	1 090	1.676	2 400	1 990	2 500
	AVa	anable Fuer w/ Max Payload	1,980 0.630	0.700	2,400	0.720	0.780
Limits		Trans Alt FL/Vwo	FL 271/250	280/275	FL 305/263	FL 302/270	FL 290/320
Linnes		PSI/Sea-Level Cabin	8.3/21.280	8,3/21.280	8.5/22.027	8.8/23.060	9.1/24.000
		TOFL (SL elev./ISA temp.)	3,110	3,199	3,210	3,934	3,821
	1	TOFL (5,000-ft. elev.@25C)	6,600	5,663	5,580	6,108	5,088
Airport		Mission Weight	8,645	10,703	10,700	10,600	14,500p
Perfor-		NBAA IFR Range	984	1,092	1,204	1,223	1,197
mance		V2	97	99	111	120	116
		VREF	88	95	101	105	105
	Landing Distance		2,137	2,473	2,340	2,795	2,960
	Time to Climb/Altitude FAR 25 Engine-Out Rate (fpm) FAR 25 Engine-Out Gradient (ft./nm)		20/FL 370	19/FL 370	18/FL 370	15/FL 370	16/FL 370
Climb			432	597	618	933	305
			267	316	334	400	158
Ceilings (ft.)	Certificated		41,000	41,000	41,000	43,000	45,000
		Findine=Out Service	26 900	24 045	26,800	27,000	27 500
	TAS/Fuel Flow (lb./hr.)		319/498	340/543	323/516	360/558	406/740
	Long Range	Altitude/Specific Range	FL 390/0.641	FL 410/0.626	FL 410/0.626	FL 430/0.645	FL 450/0.549
Cruise		TAS/Fuel Flow (lb./hr.)	339/609	406/955	401/920	420/972	447/968
	High Speed	Altitude/Specific Range	FL 350/0.557	FL 330/0.425	FL 350/0.436	FL 330/0.432	FL 430/0.462
		Nautical Miles	716	466	812	600	1.024
	Max Pavload	Average Speed	294	325	361	347	367
	(with available fuel) Max Fuel	Trip Fuel	1,300	1,036	1,706	1,230	2,411
		Specific Range/Altitude	0.551/FL 410	0.450/FL 410	0.476/FL 410	0.488/FL 430	0.425/FL 450
NBAA IFR		Nautical Miles	1,141	1,194	1,357	1,282	1,895
Ranges		Average Speed	304	333	372	361	384
(FAR Part 23,	(with available navload)	Trip Fuel	1,947	2,196	2,675	2,273	3,953
100-nm	payloady	Specific Range/Altitude	0.586/FL 410	0.544/FL 410	0.507/FL 410	0.564/FL 430	0.479/FL 450
alternate;		Nautical Miles	963	1,092	1,183	1,065	1,801
FAR Part 25,	Four Passengers	Average Speed	301	333	370	361	383
200-nm	(with available fuel)	Inp Fuel	1,664	2,038	2,352	1,976	3,706
alternate)		Nautical Miles	1 20/	1.050/ FL 410	0.505/FL 410 1 /00	1 358	1 081
		Averade Sneed	315	329	378	358	381
	Ferry	Trin Fuel	1.965	2.220	2.705	2.290	3.986
		Specific Range/Altitude	0.613/EL 410	0.565/FL 410	0.518/FL 410	0.593/FL 430	0.497/FL 450
		Runwav	2,498	2,909	2,625	3,564	3,015
		Flight Time	1+00	0+53	0+52	0+53	0+48
	300 nm	Fuel Used	670	753	804	676	786
		Specific Range/Altitude	0.448/FL 370	0.398/FL 390	0.373/FL 370	0.444/FL 430	0.382/FL 390
Missions		Runway	2,700	3,121	2,692	3,732	3,044
(4 nassen-	600 nm	Flight Time	1+56	1+45	1+38	1+38	1+30
(i passeni ders)		Fuel Used	1,135	1,236	1,362	1,179	1,323
8010)		Specific Range/Altitude	0.529/FL 390	0.485/FL 390	0.441/FL 390	0.509/FL 430	0.454/FL 430
		Runway	3,110	3,179	3,009	3,909	3,101
	1,000 nm	Flight Lime	3+19	2+54	2+42	2+40	2+28
		Fuel Useu Spacific Pando (Altitudo	0.570/EL 410	0.521/EL /10	2,018 0.496/EL.410	1,003	2,145
Remarks	Specific Range/Altitude Certification Basis		FAR 23, 2006 1,000-nm mission flown with 713-lb. payload.	FAR 23, 2008	FAR 23, 2013	FAR 23, 2015	FAR 25, 1981/85 STC 02371LA; STC 10959SC; STC 03960AT

Manufacturer		Textron Aviation	Syberjet	Pilatus Aircraft	Embraer	Textron Aviation	
Model			Cessna Citation CJ3+	SJ30i SJ30-2	SVJ PC-24	Phenom 300 EMB-505	Cessna Citation CJ4
BCA Equipped	d Price		\$7.995.000	\$8.306.452	\$8.900.000	\$8,995,000	\$8.995.000
Oberreter		Seating	1+8/9/	1+5/6/	1+8/11/NA	1+7/10/10	2+8/9/-
Character-	Wi	ng Loading/Power Loading	47.2/2.46	73.2/3.03	53.1/2.60	60.0/2.74	51.8/2.36
ISUCS	Noise (EPNdB)	Lateral/Flyover/Approach	88.7/74.0/88.6	78.5/86.2/91.8	NA/NA/NA	69.9/88.8/88.5	92.8/75.6/89.5
External		Length	51.2	46.8	55.2	51.2	53.3
Dimensions		Height	15.2	14.2	17.3	16.7	15.3
(ft.)	1	Span	53.3	42.3	55.8	52.2	50.8
Internal	Length	i: Main Seating/Net/Gross	12.3/15.//—	12.5/12.5/—	NA/NA/23.0 5.1/flat floor	14.8/17.2/17.2	12.9/17.3/17.3
Dimensions	11	Width: Max/Eloor	4.8/0.4	4.4/ NA	5.1/11/11/11/00	4.9/0.3 5.1/2.6	4.8/0.4
(TT.)		Internal: Cu. ft. /lb	4.0/3.1	6/100	0.0/0.0	10/77	4.0/3.3
Baggage		Fxternal: Cu. ft /lb	65/1 000	53/500	NA/NA	74/573	71/1 000
		External Odridy Ior	2 Wms Intl	2 Wms Intl	2 Wms Intl	2 P&WC	2 Wms Intl
Power	Ligites		FJ44-3A	FJ44-2A	FJ44-4A	PW 535E	FJ44-4A
1 OWCI	0	utput (lb. each)/Flat Rating	2,820/ISA+11C	2,300/ISA+8C	3,400/NA	3,360/ISA+15C	3,621/ISA+11C
	Inspection Interval/	Manu. Service Plan Interval Max Pamp	4,000t/5,000	3,500t/— 14.050	5,000t/NA	5,000t/—	5,000t/5,000
		Max Takeoff	13.870	13.950	17,650	18.387	17.110
		Max Landing	12,750	12,725	16,250	17,042	15,660
		Zero Fuel	10,510c	10,500c	NA	14,220c	12,500c
Weights (lb.)		BOW	8,540	8,917	NA	11,583	10,280
		Max Payload	1,970	1,583	2,500	2,637	2,220
		Useiui L0ad Max Fiiel	4,710	4,850	5.965	5,353	5.828
	Ava	ailable Payload w/Max Fuel	820	283	915	1,561	1,122
	Ava	ailable Fuel w/Max Payload	3,560	3,550	NA	4,277	4,730
		Ммо	0.737	0.830	NA	0.780	0.770
Limits		Trans. Alt. FL/VMO	FL 293/278	FL 295/320	NA/NA	FL 263/320	FL 279/305
		TOFL (SL elev /ISA tamp)	8.9/23,586 3.180	3 939	NA/23,500	9.4/25,560	3.0/24,005
		TOFL (5.000-ft, elev.@25C)	4,750	8.784	4,430	5.400	5.180
Airport		Mission Weight	13,870	13,125	17,750	18,387	16,788
Perfor-		NBAA IFR Range	1,827	1,915	NA	2,019	1,948
mance		V2	114	112	NA	113	117
		VREF	99	104	NA	104	2 291
	Time to Climb/Altitude FAR 25 Engine-Out Rate (fpm) FAR 25 Engine-Out Gradient (ft./nm)		2,422 15/FL 370	16/FL 370	NA/FL 370	15/FL 370	14/FL 370
Climb			808	312	NA	872	839
			425	167	NA	437	430
	Certificated		45,000	49,000	45,000	45,000	45,000
Ceilings (ft.)	All-Engine Service		45,000	44,000	45,000	45,000	45,000
	TAS/Fuel Flow (lb /br)		26,250	436/684	26,000 NA/NA	30,137	28,200
	Long Range	Altitude/Specific Range	FL 450/0.564	FL 450/0.637	NA/NA	FL 450/0.506	FL 450/0.464
Cruise	High Cased	TAS/Fuel Flow (lb./hr.)	415/1,197	475/1,188	NA/NA	444/1,312	442/1,470
	High Speed	Altitude/Specific Range	FL 350/0.347	FL 360/0.400	FL 300/NA	FL 350/0.338	FL 370/0.301
		Nautical Miles	1,172	1,635	NA	1,351	1,425
	Max Payload	Average Speed	368	402	NA	397	407
	(with available fuel)	Inp Fuel Specific Dends (Altitude	2,552	2,908	NA NA (NA	3,362	3,753
NBAA IFR		Nautical Miles	1 814	2 598	NA	1 883	1 913
Ranges	Max Fuel	Average Speed	377	410	NA	406	413
(FAR Part 23,	(with available navload)	Trip Fuel	3,846	4,241	NA	4,469	4,904
100-nm	payloady	Specific Range/Altitude	0.472/FL 450	0.613/FL 490	NA/NA	0.421/FL 450	0.390/FL 450
alternate;	Free Deservation	Nautical Miles	1,825	2,205	NA	1,936	1,927
FAR Part 25,	(with available fuel)	Average Speed Trin Fuel	3 767	3 71 3	NA	411	410
200-nm	(marataliable fact)	Specific Range/Altitude	0.484/FL 450	0.594/FL 490	NA/NA	0.429/FL 450	0.392/FL 450
allemale)		Nautical Miles	1,900	2,667	NA	1,985	1,955
	Ferry	Average Speed	383	411	NA	417	420
	,	Trip Fuel	3,872	4,246	NA	4,473	4,955
		Specific Kange/Altitude	0.491/FL 450 2 608	0.028/FL 490 2 822	NA/NA NA	0.444/FL 450 2 613	0.395/FL 450 2 429
		Flight Time	0+49	0+45	NA	0+47	0+46
	300 nm	Fuel Used	969	846	NA	1,058	1,087
		Specific Range/Altitude	0.310/FL 370	0.355/FL 410	NA/NA	0.284/FL 390	0.276/FL 390
Missions		Runway	2,609	3,025	NA	2,747	2,444
(4 passen-	600 nm	Flight Lime	1+35	1 313	NA	1+29	1 865
gers)		Specific Range/Altitude	0.382/FL 410	0.457/FL 450	NA/NA	0.346/FL 410	0.322/FL 410
		Runway	2,720	3,336	NA	2,808	2,490
	1.000 nm	Flight Time	2+36	2+21	NA	2+26	2+23
	2,000 mm	Fuel Used	2,315	1,980	NA	2,471	2,823
Remarks	Specific Range/Altitude		FAR 23, 2004/14 Commuter category Garmin G3000.	FAR 23 Commuter category	EASA CS 23, FAR 23 Commuter category pending Pricing in 2017 dollars; FJ44-4 with quiet power mode APU function.	FAR 23, 2009 Commuter category Performance-based upon optional increased weights.	FAR 23, 2010 Commuter category

Manufacture	r		Textron Aviation	Bombardier	Textron Aviation	Bombardier	Textron Aviation
Model			Cessna Citation X Elite	Learjet 70	Cessna Citation XLS+	Learjet 75	Cessna Citation Latitude
BCA Equipped	1 Price		\$6,500,000	\$11,300,000	\$12,750,000	\$13,800,000	\$16,350,000
DUA LQUIPPE		Seating	2+8/11/-	2+6/7/7	2+9/12/-	2+8/9/9	2+9/9/10
Character-	Wi	ing Loading/Power Loading	68.5/2.67	69.6/2.79	54.6/2.45	69.6/2.79	56.8/2.61
istics	Noise (EPNdB)	: Lateral/Flyover/Approach	83.8/71.2/90.3	87.4/74.3/93.4	86.8/72.2/92.8	87.4/74.3/93.4	87.7/73.5/87.7
External		Length	72.3	56.0	52.5	58.0	62.3
Dimensions	Height		19.3	14.0	17.2	14.0	20.9
(ft.)		Span	63.9	50.9	56.3	50.9	72.3
Internal	Length	h: Main Seating/Net/Gross	17.0/23.9/23.9	10.6/17.7/17.7	14.3/18.5/18.5	13.4/19.8/19.8	15.9/21.8/21.8
Dimensions	H	eight/Dropped Aisle Depth	5.7/0.7	4.9/flat floor	5.7/0.7	4.9/flat floor	6.0/flat floor
(ft.)		Width: Max/Floor	5.5/3.9	5.1/3.2	5.5/3.9	5.1/3.2	6.4/4.1
Raggage		Internal: Cu. ft./lb.	variable/variable	15/150	10/100	15/150	26/NA
Duggugu		External: Cu. ft./lb.	82/775	50/500	80/700	50/500	100/1,000
	Engines		2 RR	2 Hon	2 P&WC	2 Hon	2 P&WC
Power	0	utnut (lb. each) /Flat Pating	AE300701 6.764/ISA±150	3 850/ISA±230	PW5450 // 119/ISA±100	3 850/ISA+230	5 907/ISA±160
	Inspection Interval/	Manu Service Plan Interval	4 500t*/-	6 000t/-	5,000t/-	6,000t/-	6,000t/-
	inspection interval/1	Max Ramp	36.400	21,750	20.400	21.750	31.050
		Max Takeoff	36,100	21,500	20,200	21,500	30,800
		Max Landing	31,800	19,200	18,700	19,200	27,575
		Zero Fuel	24,400c	16,000c	15,100c	16,000c	21,200c
Woights (lb.)		BOW	22,100	13,900	12,860	14,050	18,656
weights (ib.)		Max Payload	2,300	2,100	2,240	1,950	2,544
		Useful Load	14,300	7,850	7,540	7,700	12,394
		Max Fuel	12,931	6,062	6,740	6,062	11,394
	Ava	allable Payload w/Max Fuel	1,369	1,/88	800	1,638	1,000
	Ava	anable Fuel w/ Max Payload	12,000	5,750	0,300	5,750	9,850
Limits		Trans Alt FL/Ving	0.920 FL 307/350	EL 270/330	EL 265/305	FL 270/330	FL 298/305
Linits		PSI/Sea-Level Cabin	9,3/25.230	9,4/25,700	9,3/25.230	9,4/25.700	9,7/25.400
		TOFL (SL elev./ISA temp.)	5,140	4,440	3,560	4,440	3,580
	1	TOFL (5,000-ft. elev.@25C)	7,350	5,191	5,430	5,272	5,070
Airport		Mission Weight	34,980p	20,632	20,200	20,782	30,675
Perfor-		NBAA IFR Range	2,980	2,045	1,740	2,026	2,700
mance		V2	137	125	118	125	115
	VREF		112	112	106	113	95
	Landing Distance		2,730	2,326	2,740	2,338	2,085
Climb	FAP 25 Engine_Out Pate (from)		18/FL 370	15/FL 370	15/FL 370	15/FL 370	15/FL 370
Cimb	EAD 25 En	C 25 Engine-Out Rate (IPIII)	480	430	280	430	240
	Certificated		51 000	51,000	45.000	51.000	45,000
Ceilings (ft.)	All-Engine Service		43.000	45,200	45,000	44,700	43,000
o o ini igo (i ti)		Engine-Out Service	26,000	28,400	28,600	27,900	26,260
Oruioo	Land Danda	TAS/Fuel Flow (lb./hr.)	470/1,529	437/970	353/865	437/977	368/1,114
	Long Kange	Altitude/Specific Range	FL 470/0.307	FL 470/0.451	FL 450/0.408	FL 470/0.447	FL 430/0.330
Cruise	High Speed	TAS/Fuel Flow (lb./hr.)	513/2,229	452/1,080	431/1,238	451/1,079	432/1,765
	ingii opeeu	Altitude/Specific Range	FL 410/0.230	FL 470/0.419	FL 410/0.348	470/0.418	FL 390/0.245
		Nautical Miles	2,703	1,728	1,150	1,728	2,135
	Max Payload	Average Speed	462	425	385	425	394
	(with available fuel)	Trip Fuel	9,973	4,575	3,663	4,575	7,901
NBAA IFR		Specific Range/Altitude	0.271/FL 470	0.378/FL 470	0.314/FL 450	0.378/FL 470	0.270/FL 450
Ranges	Max Fuel	Nautical Miles	3,070	1,881	1,719	1,881	2,645
(FAR Part 23.	(with available	Average Speed	402	420	390	420	401
100-nm	payload)	Specific Range/Altitude	0.278/FL 490	0.384/FL 470	0.328/FL 450	0 384/FL 470	9,580 0.276/EL 450
alternate:		Nautical Miles	3.125	2.045	1.719	2.026	2.678
FAR Part 25	Four Passengers	Average Speed	463	426	395	427	401
200_nm	(with available fuel)	Trip Fuel	11,078	5,064	5,168	5,058	9,594
alternate)		Specific Range/Altitude	0.282/FL 490	0.404/FL 470	0.333/FL 450	0.401/FL 470	0.279/FL 450
anomato)		Nautical Miles	3,221	2,150	1,785	2,129	2,731
	Ferry	Average Speed	463	427	403	427	405
	,	Trip Fuel	11,118	5,099	5,268	5,093	9,628
		Specific Range/Altitude	0.290/FL 490	0.422/FL 490	0.339/FL 450	0.418/FL 490	0.284/FL 450
		KUNWAY Elistet Timo	3,030 0±41	3,388	2,/34	3,398	2,760
	300 nm	Figure 1 lead	0⊤4⊥ 1 837	1 072	1 246	1 075	1 610
		Specific Range/Altitude	0.163/FL 370	0.280/FL 470	0.241/FL 390	0.279/FL 470	0.186/FL 390
Miccierc		Runwav	3,580	3,632	2,758	3,642	2,845
WISSIONS	600	Flight Time	1+16	1+24	1+29	1+23	1+29
(4 passen-	800 nm	Fuel Used	2,855	1,805	2,094	1,810	2,573
gers)		Specific Range/Altitude	0.210/FL 430	0.332/FL 470	0.287/FL 410	0.331/FL 470	0.233/FL 430
		Runway	3,672	3,691	3,028	3,701	2,951
	1,000 nm	Flight Time	2+03	2+18	2+26	2+18	2+25
	Fuel Used		4,469	2,/87	3,211	2,792	3,989
Remarks		Certification Basis	FAR 25, 1996/2002; JAR 25 1999/2002 *Engine flight hour inspection interval.	FAR/EASA CS 25	FAR 25, 2008	FAR/EASA CS 25	FAR 25, 2015 Garmin G5000.

Manufacturer		Embraer	Textron Aviation	Embraer	Textron Aviation	Textron Aviation	
Model			Legacy 450	Cessna Citation Sovereign+	Legacy 500	Cessna Citation X+	Cessna Citation Longitude
BCA Equippor	d Drico		EMB-545	CE-680	EMB-550	CE-750	CE-700
DCA Equipped		Seating	\$10,570,000 2±7/9/9	\$17,695,000 2±9/12/12	219,995,000	\$23,303,000 2±0/12/	¢23,995,000 2+8/12/12
Character-	Wi	ng Loading / Power Loading	74 0/2 73	56 7/2 60	79 4/2 73	69 4/2 60	2+0/12/12 NA/NA
istics	Noise (EPNdB)	: Lateral/Flyover/Approach	84.2/72.8/89.9	87.8/71.9/87.9	85.5/73.1/89.9	87.7/72.4/89.3	NA/NA/NA
External		Length	64.6	63.5	68.1	73.6	73.2
Dimensions		Height	21.1	20.3	21.2	19.2	19.4
(ft.)		Span	66.4	72.3	66.4	69.2	68.9
Internal	Length	n: Main Seating/Net/Gross	17.4/20.6/24.0	17.4/25.3/25.3	21.3/24.1/27.5	18.3/25.2/25.2	16.5/25.2/28.1
Dimensions	H	eight/Dropped Aisle Depth	6.0/flat floor	5.7/0.7	6.0/flat floor	5.7/0.7	6.0/flat floor
(ft.)		Width: Max/Floor	6.8/4.7	5.5/3.9	6.8/4.7	5.5/3.9	6.4/4.1
Portroro		Internal: Cu. ft./lb.	40/330	35/415	45/330	22/NA	112/1,115
Бавваве		External: Cu. ft./lb.	110/882	100/1,000	110/882	82/775	NA/NA
		Engines	2 Hon	2 P&WC	2 Hon	2 RR	2 Hon
Power	0	utput (lb. open) /Elat Pating	HIF/500E	PW306D	T 026/ISA+190	AE300702	HIF/700L 7.600//SA±100
	UI	Manu Service Plan Interval	0,540/15A+160	6,000t/	1,030/13A+160	1,034/13A+130 1,500t*/	00/
	mopeedon interval/	Max Ramp	35.891	31.025	38.537	36.900	NA
		Max Takeoff	35.759	30,775	38.360	36,600	NA
		Max Landing	32,518	27,575	34,524	32,000	NA
		Zero Fuel	25,904c	21,000c	26,499	24,978c	NA
Weights (lb.)		BOW	22,983	18,235	23,699	22,114	NA
weights (ib.)		Max Payload	2,921	2,765	2,800	2,864	2,725
		Useful Load	12,908	12,790	14,838	14,786	NA
		Max Fuel	12,108	11,390	13,058	12,931	NA
	Ava	aliable Payload w/ Max Fuel	800	1,400	1,780	1,855	1,600
	Ava	Manager Fuel W/ Max Payload	9,987	10,025	12,038	0.025	NA 0.840
Limits		Trans Alt FL /Vwo	0.030 FL 395/320	FL 298/305	FL 295/320	FL 307/350	NA/NA
Linito		PSI/Sea-Level Cabin	9,7/26.520	9,3/25,230	9,7/26.520	9,3/25.230	9,7/25.400
		TOFL (SL elev./ISA temp.)	3,907	3,530	4,084	5,250	4,900
	1	TOFL (5,000-ft. elev.@25C)	5,189	4,760	5,523	7,317	NA
Airport		Mission Weight	35,759	30,250	38,360	35,645	NA
Perfor-		NBAA IFR Range	2,919	3,093	3,131	3,396	3,520
mance	V2		117	117	120	139	NA
		VREF	101	96	102	116	NA
	Landing Distance Time to Climb/Altitude FAR 25 Engine-Out Rate (fpm)		2,090	2,144	2,114	2,727	NA 12 (5), 270
Olinala			14/FL 370	13/FL 370	14/FL 370	13/FL 370	13/FL 370
CIIIID			324	377	387	267	NA
	TAIL 20 LI	Certificated	45 000	47,000	45 000	51 000	45 000
Ceilings (ft.)	All-Engine Service Engine-Out Service		44,000	45.000	44.000	47.000	45.000
6 6 i i i Bo (i ci)			24,476	29,740	28,189	25,900	27,500
Oruioo	Long Dongo	TAS/Fuel Flow (lb./hr.)	438/1,404	368/1,059	440/1,441	470/1,470	457/1,591
	Long Kange	Altitude/Specific Range	FL 450/0.312	FL 450/0.347	FL 450/0.305	FL 470/0.320	FL 450/0.287
Cluise	High Sneed	TAS/Fuel Flow (lb./hr.)	462/1,621	448/1,756	467/1,741	520/2,453	476/1,933
	ingh opecu	Altitude/Specific Range	FL 430/0.285	FL 390/0.255	FL 430/0.268	FL 410/0.212	FL 430/0.246
		Nautical Miles	2,170	2,484	2,603	2,838	3,074
	Max Payload (with available fuel)	Average Speed	428	396	438	463	452
		Inp Fuel	8,084	8,170	9,908	9,952	11,600
NBAA IFR		Specific Range/Altitude	0.268/FL 450	0.304/FL 470	0.263/450	0.285/FL 490	0.265/FL 450
Ranges	Max Fuel	Nautical Miles	2,904	2,996	2,998	3,241	3,422
(FAR Part 23.	(with available payload)	Average Speed	431	9.658	11 151	11 108	403
100-nm		Specific Range/Altitude	0.282/FL 450	0.310/FL 470	0.269/FL 450	0.292/FL 490	0.268/FL 450
alternate:		Nautical Miles	2,904	3,069	3,125	3,372	3,500
FAR Part 25.	Four Passengers	Average Speed	431	402	433	465	454
200-nm	(with available fuel)	Trip Fuel	10,285	9,679	11,222	11,157	12,787
alternate)		Specific Range/Altitude	0.282/FL 450	0.317/FL 470	0.278/FL 450	0.302/FL 490	0.274/FL 450
		Nautical Miles	2,973	3,138	3,153	3,463	3,568
	Ferry	Average Speed	430	405	440	465	454
		Inp Fuel	10,313	9,708	11,250	11,195	12,810
		Specific Kange/Altitude	0.288/FL 450 3.674	0.323/FL 4/0 2 501	0.280/FL 450	0.309/FL 490 3.725	0.279/FL 450
		Flight Time	0+45	0+45	0+45	0+41	2,144
	300 nm	Fuel Lised	1,543	1,506	1.545	1.827	1,516
		Specific Range/Altitude	0.194/FL 450	0.199/FL 390	0.194/FL 450	0.164/FL 370	0.198/FL 450
Missions		Runway	2,696	2,600	2,817	3,775	2,880
Missions Missions	600 nm	Flight Time	1+26	1+26	1+26	1+16	1+23
(4 passell-	000 mm	Fuel Used	2,478	2,404	2,478	2,937	2,457
Beiol		Specific Range/Altitude	0.242/FL 450	0.250/FL 430	0.242/FL 450	0.204/FL 430	0.244/FL 450
		Runway	2,873	2,650	2,963	3,849	3,025
	1,000 nm	Flight lime	2+21	2+21	2+21	2+02	2+16
		Fuel Used Specific Pange / Altitude	3,710	3,750 0.267/EL 420	3,750 0.267/EL 450	4,080	3,740
Remarks		Certification Basis	RBAC/FAR/EASA CS 25, 2015	FAR 25, 2013 Garmin G5000.	RBAC/FAR/EASA CS 25, 2014	FAR 25, 2014 Garmin G5000. *Engine flight hour inspection interval.	FAR 25 pending Garmin G5000.

Manufacture	r		Gulfstream Aerospace	Embraer	Bombardier	Dassault	Bombardier
Model			Gulfstream 280	Legacy 650E	Challenger 350	Falcon 2000S	Challenger 650
PCA Equippor	d Prico		\$24,500,000	\$25,000,000	\$26.672.000	\$20 550 000	\$22,250,000
DCA Equipped		Section	\$24,500,000	2+12/14/10	\$20,073,000	\$29,550,000	\$32,330,000 2+12/12/10
Character-	Wi	ording /Power Loading	80.0/2.60	07 2/2 07	776/277	77 7/2 93	98.6/2.61
istics	Noise (EPNdB):	Lateral/Flyover/Approach	75.2/89.5/90.5	86.9/78.0/91.7	87.6/75.3/89.6	75.1/91.8/90.5	86.2/81.2/90.3
External		Length	66.8	86.4	68.7	66.3	68.4
Dimensions		Height	21.3	21.8	20.0	23.2	20.7
(ft.)		Span	63.0	69.5	69.0	70.2	64.3
Internal	Length	Main Seating/Net/Gross	17.7/25.8/32.3	30.3/42.4/49.1	16.6/25.2/28.6	17.1/26.2/31.0	15.4/25.6/28.3
Dimensions	He	eight/Dropped Aisle Depth	6.1/4.5	6.0/2.5	6.0/flat floor	6.2/flat floor	6.0/flat floor
(ft)		Width: Max/Floor	6.9/5.4	6.9/5.2	7.2/5.1	7.7/6.3	7.9/6.9
(10.)		Internal: Cu_ft /lb	154/1 980	286/1 441	106/750	131/1 600	112/900
Baggage		External: Cu. ft./lb.	_/_	_/_	_/_	8/92	_/_
			2 Hon	2 RR	2 Hon	2 P&WC	2 GE
Dowor	Engines		HTF7250G	AE 3007A2	HTF 7350	PW308C	CF34-3B
Power	Οι	utput (lb. each)/Flat Rating	7,624/ISA+17C	9,020/ISA+15C	7,323/ISA+15C	7,000/ISA+15C	9,220*/ISA+15C
	Inspection Interval/N	Ianu. Service Plan Interval	0C/—	0C/—	0C/—	7,000c/—	0C/—
		Max Ramp	39,750	53,727	40,750	41,200	48,300
		Max Takeoff	39,600	53,572	40,600	41,000	48,200
		Max Landing	32,700	44,092	34,150	39,300	38,000
		Zero Fuel	28,2000	21 217	28,2000	29,7000	32,0000
Weights (lb.)		Max Pavload	4 000	4 939	3 400	4 950	4 750
		Useful Load	15 550	22 510	15 950	16 450	21.050
		Max Fuel	14.600	20,600	14.045	14.600	19.852
	Ava	ilable Payload w/Max Fuel	950	1,910	1,905	1,850	1,198
	Ava	ilable Fuel w/Max Payload	11,550	17,571	12,550	11,500	16,300
		Ммо	0.850	0.800	0.830	0.862	0.850
Limits		Trans. Alt. FL/VMO	FL 280/340	FL 276/320	FL 290/320	FL 250/370	FL 222/348
		PSI/Sea-Level Cabin	9.2/25,000	8.4/21,650	8.8/23,338	9.3/25,300	8.8/23,000
		TOFL (SL elev./ISA temp.)	4,750	5,741	4,829	4,325	5,640
Airport		UFL (5,000-TT. elev.@25C)	7,320	7,979	6,451	6,055	9,233
Perfor-		NRAA IER Rande	39,600	3 953	39,495	39,950	47,802
Ferior-		NDAA II K Kalige Vo	137	144	133	123	147
mance	VEFE		115	115	111	106	117
		Landing Distance	2,373	2,346	2,302	2,295	2,365
	Time to Climb/Altitude		14/FL 370	21/FL 370	14/FL 370	16/FL 370	21/FL 370
Climb	FAR	25 Engine-Out Rate (fpm)	845	633	552	528	581
	FAR 25 En	gine-Out Gradient (ft./nm)	371	259	249	257	237
		Certificated	45,000	41,000	45,000	47,000	41,000
Ceilings (ft.)		All-Engine Service	45,000	41,000	44,000	43,265	38,250
Cruise		Engine-Out Service	27,500	23,128	27,800	22,187	20,000
	Long Range	IAS/FUELFIOW (ID./ III.)	409/1,488	425/1,901 EL 410/0.224	459/1,590 EL 450/0.280	437/1,400 EL 470/0.212	424/1,832 EL /10/0.221
		TAS/Fuel Flow (lb /br)	482/1 925	459/2 570	470/1 832	482/2 075	470/2 448
	High Speed	Altitude/Specific Range	FL 430/0 250	FL 370/0 179	FL 430/0 257	FL 410/0 232	FL 370/0 192
		Nautical Miles	2 577	3.076	2 719	2 450	3 011
	Max Pavload	Average Speed	448	417	447	426	417
	(with available fuel)	Trip Fuel	9,591	15,238	10,689	9,640	14,256
		Specific Range/Altitude	0.269/FL 450	0.202/FL 410	0.254/FL 450	0.254/FL 450	0.211/FL 410
NBAA IFR	Max Fuel	Nautical Miles	3,636	3,839	3,235	3,445	3,974
Ranges	(with available payload)	Average Speed	452	417	449	429	419
(FAR Part 23,		Trip Fuel	12,757	18,380	12,206	12,740	17,939
100-nm		Specific Range/Altitude	0.285/FL 450	0.209/FL 410	0.265/FL 450	0.270/FL 470	0.222/FL 410
alternate;		Nautical Miles	3,646	3,919	3,250	3,540	4,011
FAR Part 25,	Four Passengers	Average Speed	451	415	448	430	419
200-nm	(with available fuer)	Inp Fuel Specific Pande/Altitude	12,701 0.286/FL 450	18,422	12,212 0.266/EL.450	12,740 0.278/FL 470	17,953 0.223/EL /10
alternate)		Nautical Miles	3.724	3.980	3.307	3.615	4.085
	_	Average Speed	452	414	450	430	419
	Ferry	Trip Fuel	12,789	18,450	12,236	12,740	17,982
		Specific Range/Altitude	0.291/FL 450	0.216/FL 410	0.270/FL 450	0.284/FL 470	0.227/FL 410
		Runway	2,957	3,346	3,611	2,795	3,389
	300 nm	Flight Time	0+47	0+49	0+47	0+47	0+47
	000 1111	Fuel Used	1,505	1,773	1,583	1,525	1,595
		Specific Range/Altitude	0.199/FL 450	0.169/FL 410	0.190/FL 450	0.197/FL 470	0.188/FL 410
Missions		Ruriway Elight Timo	2,997	3,518	3,000	2,800	3,421
(4 passen-	600 nm	Fuel Lised	2 41 2	3 146	2 577	2 465	2.835
gers)		Specific Range/Altitude	0.249/FL 450	0 191/FL 410	0.233/FL 450	0.243/FL 470	0.212/FL 410
		Runway	3.136	3.573	3.718	2.920	3.483
	1 000	Flight Time	2+18	2+33	2+18	2+20	2+19
	1,000 nm	Fuel Used	3,645	4,815	3,925	3,755	4,532
		Specific Range/Altitude	0.274/FL 450	0.208/FL 410	0.255/FL 450	0.266/FL 470	0.221/FL 410
Remarks		Certification Basis	FAR 25, 2012; EASA CS 25, 2013	FAR 25, 2011 *Factory modification DCA 145-000- 00020/2008	FAR 25 A 98; JAR 25 Chg 15 Rockwell Collins Pro Line 21 Advanced.	FAR/EASA CS 25, 2013 EASy II flight deck; 2017 delivery price.	FAR 25, 1980/83/ 87/95/2006/15 Rockwell Collins Pro Line 21 Advanced. *9,220-lb. max takeoff; 8,729-lb. normal takeoff

Manufacturer		Dassault	Gulfstream Aerospace	Dassault	Gulfstream Aerospace	
Model			Falcon 2000LXS	Gulfstream 450	Falcon 900LX	Gulfstream 500
BCA Equipped	d Price		\$34,700.000	\$43,150,000	\$44.300.000	\$44.650.000
Character		Seating	2+8/10/19	2+14/16/19	2+12/12/19	2+13/19/19
ietice	Wi	ng Loading/Power Loading	81.2/3.06	78.4/2.69	92.9/3.27	80.9/2.54
ISUCS	Noise (EPNdB):	: Lateral/Flyover/Approach	76.4/91.7/90.5	76.2/89.5/92.3	78.2/90.3/92.1	NA/NA/NA
External		Length	66.3	89.3	66.3	91.2
Dimensions		Snon	70.2	77.9	24.8	23.3
(TL.)	Longth	Julia Spating (Nat (Cross	171/26 2/21 0	25 8/27 0/45 1	22 5 (22 2 (20 2	00.3
Dimonsions	Lengui	eight/Dropped Aisle Depth	6.2/flat floor	6.0/flat floor	6.2/flat floor	6.2/flat floor
(ft)		Width: Max/Floor	77/6.3	7 0/5 4	77/6.3	76/61
		Internal: Cu. ft./lb.	131/1.600	169/2.000	127/2.866	230/2.250
Baggage		External: Cu. ft./lb.	8/92			
		Engines	2 P&WC	2 RR	3 Hon	2 P&WC
Power	0.	utput (lb. aaab) (Elat Dating	PW308C	Tay Mk 611-8C	TFE731-60	PW814GA
	Inspection Interval/M	Manu Service Plan Interval	7,000/15A+15C	12 000t or 0C/	6 000c/	00/
		Max Ramp	43,000	75,000	49,200	77,250
		Max Takeoff	42,800	74,600	49,000	76,850
		Max Landing	39,300	66,000	44,500	64,350
		Zero Fuel	29,700c	49,000c	30,864c	52,100c
Weights (lb.)		Max Pavload	4 950	5 800	<u> </u>	5 500
	Useful Load		18,250	31,800	22,450	30,650
	Max Fuel		16,660	29,281	20,905	28,850
	Ava	ailable Payload w/Max Fuel	1,590	2,519	1,545	1,800
	Ava	allable Fuel w/Max Payload	13,300	26,000	18,336	25,150
Limits		Trans Alt FL/Vwo	EL 250/370	0.880 FL 280/340	U.870 EL 250/370	0.925 NA/NA
		PSI/Sea-Level Cabin	9.3/25,300	9.6/26,700	9.6/25,300	10.7/31,900
		TOFL (SL elev./ISA temp.)	4,675	5,600	5,360	5,200
A former and	1	TOFL (5,000-ft. elev.@25C)	6,840	8,200	7,615	7,930
Airport		Mission Weight	42,010	74,600	48,255	76,850
Perior-		NBAA IFK Kange	126	4,328	4,085	5,000 NA
mance	VREF		106	123	111	NA
		Landing Distance	2,295	2,663	2,455	NA
or 1	Time to Climb/Altitude FAR 25 Engine-Out Rate (fpm) FAR 25 Engine-Out Gradient (ft /nm)		17/FL 370	16/FL 370	19/FL 370	15/FL 370
Climb			463	/12	723	NA
	Certificated		47.000	45.000	51.000	51.000
Ceilings (ft.)		All-Engine Service	42,315	42,400	39,630	NA
	Engine-Out Service		21,010	25,000	24,980	NA
	Long Range	TAS/Fuel Flow (Ib./hr.)	437/1,485	459/2,585	431/1,665	488/2,440
Cruise		TAS/Fuel Flow (Ib /br)	FL 450/0.294 	476/3 055	430/0.259 474/2 225	FL 450/0.200
	High Speed	Altitude/Specific Range	FL 390/0.208	FL 410/0.156	FL 390/0.213	FL 410/0.149
		Nautical Miles	2,915	3,549	3,790	4,129
	Max Payload (with available fuel)	Average Speed	427	452	422	478
		Trip Fuel	11,438	22,622	16,340	22,365
NBAA IFR		Specific Range/Altitude	0.255/FL 450	0.157/FL 450	0.232/FL 430	0.185/FL 470
Ranges	Max Fuel		430	4,210	4,565	480
(FAR Part 23,	(with available	Trip Fuel	14,798	26,023	18,909	26,172
100-nm	payload)	Specific Range/Altitude	0.270/FL 470	0.162/FL 450	0.241/FL 430	0.191/FL 490
alternate;		Nautical Miles	4,065	4,328	4,650	5,075
FAR Part 25,	Four Passengers	Average Speed	430	452	420	480
200-nm	(with available fuer)	IND Fuel Specific Range/Altitude	0,275/FL 470	0,166/FL 450	0,246/FL 430	0,194/FL 490
alternate)		Nautical Miles	4,155	4,382	4,740	5,137
	Ferry	Average Speed	431	453	419	480
	iony	Trip Fuel	14,798	26,116	18,909	26,222
		Specific Range/Altitude	0.281/FL 470 2 795	0.168/FL 450	0.251/FL 430	0.196/FL 490
		Flight Time	0+47	0+46	0+47	0+45
	300 nm	Fuel Used	1,525	2,599	1,595	2,274
		Specific Range/Altitude	0.197/FL 470	0.115/FL 450	0.188/FL 470	0.132/FL 490
Missions		Runway	2,855	3,258	2,865	NA
(4 passen-	600 nm	Flight Lime	2.465	4 113	2.625	2 561
gers)		Specific Range/Altitude	0.243/FL 470	0.146/FL 450	0.229/FL 470	0.168/FL 490
		Runway	2,920	3,304	2,880	NA
	1.000 nm	Flight Time	2+20	2+18	2+20	2+12
	1	Fuel Used	3,755	6,176 0.162/EL 450	4,070	5,313
Remarks	Specific Range/Altitude		FAR/EASA CS 25, 2013 EASy II flight deck; 2017 delivery price.	FAR/EASA CS 25, 2004	FAR/EASA 25, 1979/2010 EASy II flight deck; 2017 delivery price.	FAR/EASA 25 pending

Manufacturer		Bombardier	Embraer	Dassault	Airbus	
Model			Global 5000	Lineage 1000E	Falcon 7X	A320 Prestige
			BD-700-1A11	ERJ 190-100 ECJ	Falcon 7X	A320-214
BCA Equipped	a Price	Casting	\$50,441,000	\$53,000,000	\$53,800,000	\$95,000,000
Character-	14/5	Seating	3+13/15/19	3+13/19/19	3+12/14/19	4+18/1/9/
istics	Wi Noise (EPNdB)	I ateral /Flyover/Approach	90.6/3.14	92 7/86 4/92 5	92.0/3.04	130.3/3.18
External	Noise (El Nub)	Lateral/ Hyover/Approach	96.8	118.9	76.7	123.3
Dimensions		Height	25.5	34.7	25.7	38.6
/f+)		Snan	94.0	94.2	86.0	111.8
(IL.)	Leasthe Main Conting (Not /Cont		27.2/40.7/45.7	67.2/76.6/84.2	26.2/30.1/46.5	00.3/00.3/
Dimensione	Lengu	eight /Dronned Aisle Denth	6 2/flat floor	6.6/flat floor	6 2/flat floor	7 4/flat floor
Dimensions	Width: May/Floo		70/65	8 8 / 8 0	77/6.2	10.1/11.7
(π.)		wiuui: WdX/ FIOU	1.9/0.5	0.0/0.0	1.1/0.3	
Baggage	External: Cu. IL/ID.		195/1,000	323/2,293	140/2,004	
		External: Cu. IL/ID.		2.65	/	2 CEMI
_		Engines	BR700-710A2-20	CF34-10E7-B	PW307A	CFM56-5B4/3*
Power	01	utput (lb. each)/Flat Rating	14,750/ISA+20C	18,500/ISA+15C	6,402/ISA+17C	27,000/ISA+29C
	Inspection Interval/	Manu. Service Plan Interval	0C/—	0C/—	7,200c/—	0C/—
		Max Ramp	92,750	120,593	70,200	172,850
		Max Takeoff	92,500	120,152	70,000	171,950
		Max Landing	78,600	100,972	62,400	145,500
		Zero Fuel	58,000c	80,469c	41,000c	137,800c
Weights (lb.)		BOW	50,861	70,548	36,600	109,000
		Max Payload	7,139	9,921	4,400	28,800
		USETUI LOAD	41,889 38 959	20,045 48,217	33,000	53,850
	Δue	ailable Payload w/Max Fuel	2 930	1 828	1 660	10 400
	Ave	ailable Fuel w/Max Pavload	34.750	40.124	29.200	35.050
		Ммо	0.890	0.820	0.900	0.820
Limits		Trans. Alt. FL/VMO	FL 303/340	FL 289/320	FL 270/370	FL 250/350
		PSI/Sea-Level Cabin	10.3/30,125	8.8/23,190	10.2/29,200	8.3/NA
		TOFL (SL elev./ISA temp.)	5,540	6,076	5,710	6,920
A.1	1	TOFL (5,000-ft. elev.@25C)	7,223	9,500	8,045	9,355
Airport		Mission Weight	90,370	112,038	69,140	171,950
Perfor-		NBAA IFR Range	5,475	3,965	5,795	4,300
mance		V2	133	140	133	NA
			2 190	2.038	2 1 2 0	2.400
	Time to Climb/Altitude		18/FL 370	2,038 29/EL 350	19/FL 370	23/FL 360
Climb	FAR	25 Engine-Out Rate (fpm)	704	NA	597	NA
0	FAR 25 En	igine-Out Gradient (ft./nm)	318	NA	269	NA
	Certificated		51,000	41,000	51,000	39,000
Ceilings (ft.)		All-Engine Service	44,600	35,000	40,215	NA
	Engine-Out Service		20,600	19,178	25,480	NA
	Long Range	TAS/Fuel Flow (lb./hr.)	470/2,856	454/4,184	459/2,260	451/4,730
Cruise	88-	Altitude/Specific Range	FL 450/0.165	FL 380/0.109	FL 430/0.203	FL 370/0.095
	High Speed	IAS/Fuel Flow (lb./hr.)	499/3,582	471/5,033	497/3,205	473/5,860
	• •	Altitude/Specific Range	FL 410/0.139	FL 350/0.094	FL 390/0.155	350/0.081
		Nautical Miles	4,920	3,493	5,000	2,100
	(with available fuel)	Average Speed	403	25 560	453	428
	(with available fact)	Specific Pando /Altitudo	0.147/EL 470	0.098/EL 400	0.186/EL 450	0.075/EL 250
NBAA IFR		Nautical Miles	5 486	4 532	5 670	3 852
Ranges	Max Fuel	Average Speed	464	446	454	438
(FAR Part 23,	(with available	Trip Fuel	35.723	43.962	29.560	46.930
100-nm	payload)	Specific Range/Altitude	0.154/FL 470	0.103/FL 410	0.192/FL 470	0.082/FL 390
alternate;		Nautical Miles	5,475	4,602	5,760	4,330
FAR Part 25,	Four Passengers	Average Speed	463	446	454	438
200-nm	(with available fuel)	Trip Fuel	35,719	44,240	29,560	48,057
alternate)		Specific Range/Altitude	0.153/FL 470	0.104/FL 410	0.195/FL 470	0.090/FL 390
		Nautical Miles	5,526	4,640	5,840	4,380
	Ferry	Average Speed	404	440	404	438
		Specific Pande /Altitude	0 155/FL /70	0 105/EL /10	0 108/FL /70	0 001 /FL 200
		Runway	2 487	3 002	2 500	3 670
		Flight Time	0+46	0+48	0+46	0+55
	300 nm	Fuel Used	2,773	3,426	2,075	4,265
		Specific Range/Altitude	0.108/FL 450	0.088/FL 390	0.145/FL 450	0.070/FL 350
Missions		Runway	2,575	3,133	2,515	3,700
(A nassen-	600 nm	Flight Time	1+23	1+26	1+25	1+34
(4 passell-	000 mm	Fuel Used	4,445	5,862	3,285	7,080
geis)		Specific Range/Altitude	0.135/FL 490	0.102/FL 410	0.183/FL 470	0.085/FL 390
		Runway	2,697	3,251	2,640	3,/60
	1,000 nm	Flight Lime	2+13	2+20	2+17	2+28
		Specific Pande/Altitude	0,732	0.110/EL./10	4,943	0.091/FL 390
Remarks	rks Certification Basis		FAR 25, 1998/2004; EASA 25, 2004 Global Vision flight deck	FAR/EASA 25, 2008	FAR/EASA 25, 2007 EASy II flight deck; DFCS; 2017 delivery price.	FAR 25, 1999 *Also available with 26,500- Ibf IAEV2527M-A5 engines; includes 2 additional center tanks and VIP cabin. BCA estimated data.

Ultra-Long-Range Jets

Model Market and Solution 200 Results and Solution 200 Results 200 <thresults 200<="" th=""> Resu</thresults>	Manufacturer			Gulfstream Aerospace	Dassault	Gulfstream Aerospace	Bombardier	Gulfstream Aerospace
Bits Construct Bits Alexanolog Bits Alexan	Model			Gulfstream 600	Falcon 8X	GV-SP	Global 6000	Gulfstream 650
Standborg Standborg <thstandborg< th=""> Standborg <th< th=""><th>BC4 Equipped</th><th>Price</th><th></th><th>\$56,200,000</th><th>\$58 400 000</th><th>\$61,500,000</th><th>\$62,310,000</th><th>\$67,400,000</th></th<></thstandborg<>	BC4 Equipped	Price		\$56,200,000	\$58 400 000	\$61,500,000	\$62,310,000	\$67,400,000
Only Billing Number of the second secon	Ohanastan		Seating	4+16/19/19	3+12/14/19	4+16/18/19	4+13/15/19	4+16/19/19
BACS Integlifted Lengthener (Lange) MA (MV/M) 61.598.9(-0.5) 79.3/0.2(-0.5) 98.7/0.5(-0.7) 77.5/0.8.9(-0.7) Integlifted Length MS bing (MV/M) 91.598.9(-0.5) 92.4 95.4	Character-	Win	ng Loading/Power Loading	78.9/2.92	95.9/3.62	80.1/2.96	97.5/3.37	77.6/2.95
Store Homework Homework <t< td=""><th>ISUCS</th><td>Noise (EPNdB):</td><td>Lateral/Flyover/Approach</td><td>NA/NA/NA</td><td>81.5/88.9/90.6</td><td>79.3/90.2/90.8</td><td>88.7/83.5/89.7</td><td>77.5/89.8/88.3</td></t<>	ISUCS	Noise (EPNdB):	Lateral/Flyover/Approach	NA/NA/NA	81.5/88.9/90.6	79.3/90.2/90.8	88.7/83.5/89.7	77.5/89.8/88.3
Unrealized Decision (1) Log (2) Cols (External		Length	96.1	80.2	96.4	99.4	99.8
Min Mark	Dimensions		Height	25.3	26.1	25.8	25.5	25.7
Montes Market for the set of the set	(ft.)		Span	94.1	86.3	93.5	94.0	99.6
Multisetter T26.8 T26.8 T27.8.3 T26.8.4 T26.8.4 <tht26.8.4< th=""> <th< td=""><th>Internal</th><td>Length:</td><td>int / Dropped Aisle Depth</td><td>30.2/45.2/51.3 6.2/flat floor</td><td>29.8/42.1/50.1 6.2/flat floor</td><td>30.3/42.6/50.1 6.0/flat floor</td><td>27.3/43.3/48.3 6.2/flat floor</td><td>32.1/40.8/53.0 6.3/flat floor</td></th<></tht26.8.4<>	Internal	Length:	int / Dropped Aisle Depth	30.2/45.2/51.3 6.2/flat floor	29.8/42.1/50.1 6.2/flat floor	30.3/42.6/50.1 6.0/flat floor	27.3/43.3/48.3 6.2/flat floor	32.1/40.8/53.0 6.3/flat floor
0.1 memis 1, 10, 226, 2280 140, 200, 228, 2300 140, 200, 228, 2300 232, 2300 230, 2300 <t< td=""><th>Dimensions</th><td>ne</td><td>Width: Max/Floor</td><td>76/61</td><td>77/63</td><td>70/54</td><td>79/65</td><td>8 2 / 6 7</td></t<>	Dimensions	ne	Width: Max/Floor	76/61	77/63	70/54	79/65	8 2 / 6 7
Bagggob Power Instruction Instruction <thinstruction< th=""> <thinstruction< th=""></thinstruction<></thinstruction<>	(11.)		Internal: Cu ft /lb	230/2 250	140/2 004	226/2 500	195/1.000	235/2 500
Point 2 Marc 3 Marc 9 Mar 9 Mar 9 Mar	Baggage		External: Cu. ft./lb.		_/	_/_	_/_	_/_
Protect Protect <t< td=""><th></th><td></td><td>Endinos</td><td>2 P&WC</td><td>3 P&WC</td><td>2 RR</td><td>2 RR</td><td>2 RR</td></t<>			Endinos	2 P&WC	3 P&WC	2 RR	2 RR	2 RR
Aussing Council and if a large of the second seco	Power		Lingines	PW815GA	PW307D	BR700-710C4-11	BR700-710A2-20	BR700-725A1-12
Implementation Model Process Documentation Documentation Documentation Documentation Weights (b). Mastering 76,800 62,000 75,300 76,800 62,000 Weights (b). Mastering 76,800 62,000 75,300 76,800 62,000 Weights (b). Mastering 76,800 62,000 75,300 76,800 62,000 Weights (b). Mastering 76,800 62,000 76,300 64,000 62,000 64,000 62,000 64,000 64,000 62,000 64,000	1 01101	Ou	tput (lb. each)/Flat Rating	15,680/ISA+15C	6,722/ISA+17C	15,385/ISA+15C	14,750/ISA+20C	16,900/ISA+15C
Meg/15 (b) Mathematication 92,000 23,000 94,000 94,000 94,000		Inspection Interval/ M	Max Ramp	92,000	7,2000/	91 400	99.750	100.000
Metal control 76,800 60,400 75,300 76,800 83,500 Weights (b) $21,400$ 32,400 44,000 52,800 54,000 52,800 54,000 52,800 54,000 52,800 54,000 52,800 54,500 55,500 <			Max Takeoff	91,600	73,000	91,000	99,500	99,600
Neights (b) Zerriel Str.4400 54,000c 55,000c 50,000c			Max Landing	76,800	62,400	75,300	78,600	83,500
Weights (b). Above to the second			Zero Fuel	57,440c	41,000c	54,500c	58,000c	60,500c
Martine Month <	Weights (lb.)		BOW	51,440	36,800	48,700	52,560	54,500
Marine Dependent Value (%) 35:141 40:394 47:16 47:26 47:26 Anishe Projek Value (%) 30:201 1,259 1,706 2,474 1,300 Umits Marine Fair (Xite Pyrat 34,660 32:200 36:600 41.700 33:500 Umits Marine Fair (Xite Pyrat 34,660 32:200 36:600 41.700 33:500 Apport Marine Fair (Xite Pyrat 34,660 32:200 36:600 42:71 33:500 Apport Marine Fair (Xite Pyrat 34,660 32:200 10:12,71:300 10:271:300 10:271:300 10:271:300 10:371:300 55:68 35:60 35:60 35:60 35:60 35:60 35:60 35:60 35:60 35:60 10:71:100 31:41 30:60 35:60	Ŭ, v		Max Payload	6,000	4,200	5,800	5,440	6,000
Anable bright with and Anable bright with and Anable bright with angel Anable bright with angel Bright and Carl Bright angel Bright angel Brigh			Max Fuel	38,760	35.141	40,994	44,716	44,200
Autober Tarl Vist. Synap 36,560 32,200 36,800 41,750 38,800 Limits M 6/262 0.900 0.855 0.800 0.025 Synap 10,272,390 10,270,300 10,272,390 10,270,300 10,270,125 Apport Fill Sink (visit was) 9,000 7,251 91,000 94,513 60,000 Performance MM 8,000 6,200 6,413 6,230 5,544 6,912 Performance MM 107,71 12 10,00 94,5130 69,600 10,112 Performance MM 10,71 12 10,00 94,5130 69,600 Cimb MM 10,71 12 10,00 94,5130 69,600 Cimb MM 10,71 12 10,00 94,5130 69,600 Cimb MM 2,720 10,471,370 12,71,370 12,71,370 12,71,370 12,71,370 12,71,370 12,71,370 12,71,370 12,71,370 12,71,370 <t< td=""><th></th><td>Avai</td><td>ilable Payload w/Max Fuel</td><td>1,800</td><td>1,259</td><td>1,706</td><td>2,474</td><td>1,300</td></t<>		Avai	ilable Payload w/Max Fuel	1,800	1,259	1,706	2,474	1,300
Limits Mass. Rt. Vis. 0.395 0.3900 0.3955 0.3900 0.9255 Limits Mass. Rt. Vis. Mar. And Vir. And		Avai	ilable Fuel w/Max Payload	34,560	32,200	36,900	41,750	39,500
Centres main as in the main and mai	Limite		Ммо	0.925	0.900	0.885	0.890	0.925
OPI,SI sen;G4 mp. 2,200 2,500 2,510 0,747 1,742 1,441 0,747 0,747 1,742 1,441 1,442 1,441 1,442 1,441 1,442 1,441 1,442 1,441 1,442 1,441 1,442 1,441 1,442 1,441 1,442 1,441 1,442 1,441	Limits		ITATIS. AIT. FL/ VM0 PSI/Sea-Level Cabin	10.7/31.900	10.4/30.300	FL 270/340 10.2/29.200	10.3/30 125	10,7/31,900
NIDL (5.00 + ten str25) MA 8.555 9.070 7.880 9.000 Performance NAM (8 step) 6.200 6.415 6.738 5.594 6.900 Performance V MA 138 1.17 1.42 1.46 Performance V MA 1.00 1.12 1.00 1.14 Climb Max 2.72.570 1.07.71 1.07.17 1.02 1.01 1.14 Climb Max 2.72.570 1.07.17 1.07.27.07 1.07.17 1.00 1.14 Climb Max 2.72.570 1.07.17 1.07.13 1.00 1.0			TOFL (SL elev./ISA temn.)	5,700	5,880	5,910	6,476	5,858
Airport Performance Masce Weight (No. 10, 10, 10, 10, 10, 10, 10, 10, 10, 10,		T	OFL (5,000-ft. elev.@25C)	NA	8,555	9,070	7,880	9,000
Performance vision Max Resp. vision 6,200 (min) 6,415 (min) 6,735 (min) 5,634 (min) 6,312 (min) Climb Vision Min 1,000 (min) 1,127 (min) 1,420 (min) 1,440 (min) 1,440 (min	Airport		Mission Weight	91,600	72,591	91,000	94,513p	99,600
MA 137 113	Performance		NBAA IFR Range	6,200	6,415	6,738	5,594	6,912
Ladiag Between NA 2.245 2.240 2.2433 2.560 Climb Fite 25 Engle Out Rate (bm) NA 774 594 474 NA Climb Fite 25 Engle Out Rate (bm) NA 774 594 474 NA Climb Fite 25 Engle Out Rate (bm) NA 774 594 474 NA Coling (ft) Fite 25 Engle Out Rate (bm) NA 733 2242 200 NA Coling (ft) Fite States (bm) State (bm) NA 737 594 4700 51,000 51,000 51,000 51,000 22,000 42,700 42,700 42,800 42,700 42,800 42,800 42,700 42,80			V2 VDEE	NA NA	138	147	142	146
Climb The to Dimkinute 12/FL 370 22/FL 370 13/FL 370 21/FL 370 Celling (ft) R2 35 Eggle-04 East (try) NA 339 242 200 NA Celling (ft) R2 35 Eggle-04 East (try) NA 339 242 200 NA Celling (ft) R2 35 Eggle-04 East(try) NA 339 242 200 NA Celling (ft) R2 35 Eggle-04 East(try) NA 339 242 200 NA Ciling (ft) Res 25 Eggle-04 East(try) A49 409 427.00 42.000 42.000 42.000 42.000 42.000 42.000 42.000 42.000 42.600 42.60 14.00 488 499 499 490 516		Landing Distance		NA	2,245	2,240	2,243	2,680
Climb PR25 Espine-06 are to final NA 774 1594 474 NA Celling (ft) R25 Espine-06 are to final construct (minutes) NA 339 242 200 NA Celling (ft) R25 Espine-06 are to final construct (minutes) NA 339 242 200 NA Colling (ft) R25 Espine-06 are to final construct (minutes) R27,000 40,000 12,000 12,000 22,000 12,000 12,000 12,000 12,000 12,000 22,000 13,000 12,000 13,000 12,000 13,000 12,000 13,000 12,000 13,000 14,000 12,000 <th></th> <td></td> <td>Time to Climb/Altitude</td> <td>17/FL 370</td> <td>20/FL 370</td> <td>18/FL 370</td> <td>21/FL 370</td> <td>19/FL 370</td>			Time to Climb/Altitude	17/FL 370	20/FL 370	18/FL 370	21/FL 370	19/FL 370
Celling (ft.) MA 339 242 200 NA Celling (ft.) All 25 tigge- 0.4 source 51,000 51,000 51,000 51,000 51,000 10,000 42,000 <	Climb	FAR	25 Engine-Out Rate (fpm)	NA	774	594	474	NA
Celling (L) Unitable bytec 0.1 (1) 0.1000 (1) 0.10000 (1) 0.10000 (1) 0.100000 (1) 0.100000000000000000000000000000000000		FAR 25 Engine-Out Gradient (ft./nm)		NA 51.000	339	242	200	NA 51.000
NBAA IFR Range Max Field (with available fuel) Description (Frunction for the second memory of the second second memory of the second memory of the second memory of the second second memory of the second second memory of the second memory of the second second memory of the second second memory of the second memory of the second second memory of the second memory of the second second memory of the second memory of the second second memory of the second second memory of the second memory of the second second memory of the second memory of the second second memory of the second memory of the second second memory of the second memory of the second second memory of the second memory of the second memory of the second memory of the second memory of the second memory	Ceiling (ft)			42 700	40.075	42 700	42 400	42 700
Cruise NS 488 459 459 470 488 Cruise Failbox 2,769 2,254 2,553 3,046 2,225 Cruise Failbox 2,769 2,2254 2,563 3,046 2,225 High Speed Attude FL 450 FL 450 FL 450 FL 450 FL 450 High Speed Specific Range 0,176 0,204 0,179 0,154 0,173 Missions Specific Range 0,133 0,191 0,151 0,131 0,165 Max Pryoad (with available fine) Top Fait 31,622 29,507 33,993 40,415 36,225 Ranges (200 mm Specific Range/Hutes 0,167/FL 450 0,138/FL 470 0,117/FL 490 0,146/FL 470 0,164/FL 470 NBAAL Frage 31,622 29,507 33,993 40,415 36,285 C200 mm Specific Range/Hutes 0,167/FL 450 0,138/FL 470 0,116/FL 470 0,164/FL 470 Ranges 220 0,107/FL 490	001111 <u>B</u> (11.)		Engine-Out Service	25,000	26,645	25,820	18,000	25,000
Lung Rung Fuel For Multion 2,769 2,254 2,563 3,046 2,225 Cruise			TAS	488	459	459	470	488
Cruise Attoo FL 450 FL 450 FL 450 FL 450 FL 450 Cruise Specific Regging, Mouth 126 0.204 0.2179 0.154 0.173 High Speed Attouts 3.891 2.508 3.228 3.766 3.136 High Speed Attouts FL 430 FL 430 FL 430 FL 430 FL 430 Name Specific Regging, Mouth FL 430 FL 430 FL 430 FL 430 FL 450 Name Specific Regging, Mouth Attouts FL 430 FL 430 FL 440 FL 430 FL 450 FL 450 <th></th> <td>Long Range</td> <td>Fuel Flow</td> <td>2,769</td> <td>2,254</td> <td>2,563</td> <td>3,046</td> <td>2,825</td>		Long Range	Fuel Flow	2,769	2,254	2,563	3,046	2,825
Cruise Output Output <thoutput< th=""> <thoutput< th=""> <thoutput< td="" th<=""><th></th><td></td><td>Altitude Specific Pange</td><td>FL 450</td><td>FL 430</td><td>FL 450</td><td>FL 450</td><td>FL 450</td></thoutput<></thoutput<></thoutput<>			Altitude Specific Pange	FL 450	FL 430	FL 450	FL 450	FL 450
High Speed Full Former Methods 3,891 2,508 3,228 3,726 3,136 Missions Specific Range (with mailable fuel) 0,133 0,191 0,151 0,131 0,465 NBAA IFR Ranges (200-mm) Max Peytod (with mailable fuel) Specific Range Munical Mise 0,220 2,9507 33,993 40,415 36,285 Ranges (200-mm) Specific Range/Mutual 0,167/L 450 0,188/L 470 0,170/L 430 0,145/L 470 0,146/FL 470 0,170/FL 510 0,139/FL 470 0,170/FL 510 0,139/FL 470 0,170/FL 510 0,139/FL 470 0,170/FL 510 0,149/FL 470 0,170/FL 510 0,149/FL 470 0,170/FL 510 0,149/FL 470 0,176/FL 400 0,169/FL 470 0,176/FL 400 0,169/FL 470 0,176/FL 400 0,169/FL 470 0,176/FL 400 0,169/FL 470 0,176/FL 400 0,149/FL 470 0,149/FL	Cruise		Specific Ralige TAS	516	480	488	499	516
NBAA IFR Ranges (with available fue) Mitude Specific Range/Mitude (with available fue) RL 410 Specific Range/Mitude (with available fue) FL 410 Specific Range/Mitu		Web Court	Fuel Flow	3,891	2,508	3,228	3,796	3,136
NBAA IFR (with available fue) Nuture in the Marrage Speed (with available fue) 0.131 Nuture in the Marrage Speed (with available fue) 0.131 Nuture in the Marrage Speed (with available fue) 0.131 Nuture in the Marrage Speed (Marrage Speed (Marrage Speed (Marrage Speed (Marrage Speed (Marrage Speed) 0.131 Association (Marrage Speed) 0.134 Association (Marrage Speed) 0.134 Association (Marrage Speed) 0.134 Association (Marrage Speed) 0.134 Association (Marrage Speed) 0.137/FL 450 Association (Marrage Speed) 0.137/FL 45		Hign Speed	Altitude	FL 410	FL 430	FL 430	FL 410	FL 450
Nax Payod (wfb available fiel) Nax Payod Nax Payod (wfb available fiel) Nature Nax Payod (wfb available fiel) S286 Specific Regr/Mitude 5,286 0.122 2,555 0.167 5,767 5,882 452 5,934 452 NBA IFR Ranges (wfb available fiel) Max Feld (wfb available payloal) The Fuel Specific Regr/Mitude 31,622 0.128/FL 470 0.1170/FL 490 0.146/FL 430 0.164/FL 490 NBA IFR Ranges (wfb available payloal) Specific Regr/Mitude 0.173/FL 490 0.194/FL 470 0.175/FL 490 0.149/FL 470 0.170/FL 510 (wfb available payloal) Specific Regr/Mitude 0.173/FL 490 0.194/FL 470 0.175/FL 490 0.149/FL 470 0.149/FL 470 0.176/FL 490 (wfb available fiel) Specific Regr/Mitude 0.173/FL 490 0.194/FL 470 0.149/FL 470 0			Specific Range	0.133	0.191	0.151	0.131	0.165
Math available fuel (with available fuel particul) Avail (with		Max Dayland	Nautical Miles	5,286	5,555	5,767	5,882	5,934
NBAA IFR Ranges (200-nm atemate) Specific Range/Altitude (with available pa)(od) 0.167/FL 450 0.188/FL 470 0.170/FL 490 0.146/FL 470 0.164/FL 470 NBAA IFR (with available pa)(od) Kattel (with available pa)(od) 6,200 6,325 6,698 6,200 6,981 Specific Range/Altitude (200-nm atemate) Specific Range/Altitude (with available feet) 0.173/FL 490 0.194/FL 470 0.175/FL 490 0.146/FL 470 0.170/FL 510 Bight Passengers (200-nm atemate) Natical Miles (with available feet) 0.173/FL 490 0.194/FL 470 0.175/FL 490 0.146/FL 470 0.170/FL 510 Ferry Fight Passengers (with available feet) Avarage Speed (451 453 453 464 481 Newsage Speed (451 0.173/FL 490 0.194/FL 470 0.176/FL 490 0.148/FL 470 0.169/FL 510 Natical Miles (with available feet) Specific Range/Altitude 0.173/FL 490 0.194/FL 470 0.176/FL 490 0.168/FL 470 0.168/FL 470 0.168/FL 510 Natical Miles (with available feet) 0.727/FL 490 0.194/FL 470 0.176/FL 490 0.168/FL 470 0.168/FL 470 0.168/FL 470		(with available fuel)	Trin Fuel	31.622	29 507	33 993	40.415	36 285
NBAA IFR Ranges Max Fuel (with available payload) Aminge Speed Arrage Speed (atta) 6.200 6.325 6.698 6.200 6.981 Ranges average Speed (atta) 481 453 454 464 482 Ranges Speedic Range/Attude 0.173/FL 490 0.194/FL 470 0.175/FL 490 0.194/FL 470 0.170/FL 510 Remarks Eight Passengers (with available fue) Average Speed 481 453 453 464 481 Speedic Range/Attude 0.173/FL 490 0.194/FL 470 0.176/FL 490 0.149/FL 470 0.148/FL 470 0.160/FL 510 Natical Miles 6.353 6.475 6.853 6.233 7.105 Natical Miles 6.953 6.475 6.853 6.233 7.105 Natical Miles 6.953 0.127/FL 490 0.198/FL 470 0.178/FL 510 0.150/FL 470 0.173/FL 510 Natical Miles 6.953 0.477 4.904 5.599 6.842 5.942 Specific Range/Attude 0.178/FL 490 0.200/FL 470 0.178/FL 510 0.150/FL 470<			Specific Range/Altitude	0.167/FL 450	0.188/FL 470	0.170/FL 490	0.146/FL 470	0.164/FL 490
NBAA IFR Ranges (200-nm alternate) Max Hold (mit) available payload) Average Speed (1) Foreign (200-nm alternate) 48:1 (1) Foreign (200-nm alternate) 44:1 (1) Foreign (200-nm alternate) 44:1 (1) Foreign (200-nm alternate) 44:1 (1) Foreign (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm alternate) 45:4 (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-nm alternate) 45:3 (200-nm (200-		Mary Fred	Nautical Miles	6,200	6,325	6,698	6,200	6,981
Normality psyload To Fuel 35,918 32,558 38,202 41,472 41,129 (200-nm alternate) Specific Range/hittude 0.173/FL 490 0.194/FL 470 0.175/FL 490 0.149/FL 470 0.170/FL 510 (200-nm alternate) Eight Passengers (with available fee) Average Speed 481 453 453 464 481 5eedic Range/hittude 0.173/FL 490 0.194/FL 470 0.176/FL 490 0.148/FL 470 0.169/FL 510 Specific Range/hittude 0.173/FL 490 0.194/FL 470 0.176/FL 490 0.148/FL 470 0.169/FL 510 Natical Miles 6,353 6,475 6,853 6,233 7,105 Average Speed 481 454 454 464 482 Specific Range/hittude 0.173/FL 490 0.198/FL 470 0.150/FL 470 0.173/FL 510 Na 2,6853 38,251 441,487 41,168 Specific Range/hittude 0.173/FL 490 0.128/FL 470 0.179/FL 510 0.150/FL 470 0.173/FL 510 NA 2,6854 3,436		(with available	Average Speed	481	453	454	464	482
Fail ges Specific Range/Attitude 0.174/FL 490 0.176/FL 490 0.176/		payload)	Trip Fuel	35,918	32,558	38,202	41,472	41,129
Histors Bight Passengers (with available fuel) Discussion (with available fuel)	(200-nm		Specific Range/Altitude Nautical Miles	0.173/FL 490 6 217	0.194/FL 4/0 6 235	0.1/5/FL 490 6 708	0.149/FL 470 6 124	0.1/0/FL 510 6 912
(with available fuel) Trp Fuel Specific Range/Altitude Average Specific Range/Altitude Specific Range/Altitude Average Specific Range/Altitude Average Specific Range/Altitude Average Specific Range/Altitude Average Specific Range/Altitude Specific Range/Altitude Average Specific Range/Altitude Average Specific Range/Altitude Specific Range/Altitude Average Specific Range/Altitude Average Specific Range/Altitude Specific Range/Altitude Average Specific Range/Altitude	alternate)	Eight Passengers	Average Speed	481	453	453	464	481
Specific Range/Altitude 0.173/FL 490 0.194/FL 470 0.176/FL 490 0.148/FL 470 0.169/FL 510 Natical Miles 6,353 6,475 6,853 6,233 7,105 Natical Miles 6,353 6,475 6,853 6,233 7,105 Nerrage Speed 481 454 454 464 482 Tip Fuel 35,966 32,653 38,251 41,487 41,168 Specific Range/Altitude 0.177/FL 490 0.198/FL 470 0.179/FL 510 0.150/FL 470 0.173/FL 510 Missions 8 Flight Time 2+12 2+17 2+20 2+13 2+10 Specific Range/Altitude 0.177/FL 490 0.200/FL 470 0.179/FL 490 0.466/FL 470 0.178/FL 510 Missions 3,000 nm Runway NA 3,540 3,559 6,842 5,942 Specific Range/Altitude 0.177/FL 490 0.200/FL 470 0.179/FL 490 0.164/FL 470 0.184/FL 510 Missions 8,000 nm Fuel Used 16,060 14,122		(with available fuel)	Trip Fuel	35,924	32,204	38,205	41,437	40,820
Nutled Miles 6,353 6,475 6,853 6,233 7,105 Average Speed To Firbiti Specific Range/Altitude 481 454 454 464 482 To Firbiti Specific Range/Altitude 0.177/FL 490 0.198/FL 470 0.179/FL 510 0.150/FL 470 0.173/FL 510 Numay NA 2,685 3,436 2,852 3,241 Filer Time 2+12 2+17 2+20 2+13 2+10 Filer Time 5,728 4,994 5,599 6,842 5,942 Specific Range/Altitude 0.175/FL 490 0.200/FL 470 0.179/FL 490 0.164/FL 470 0.168/FL 510 Rumay NA 3,540 3,599 3,858 3,591 Bigst Time 6+19 6+39 6+42 6+20 6+17 Bigst Time 6+19 0.137/FL 490 0.212/FL 470 0.146/FL 470 0.168/FL 510 Rumay NA 5,645 5,277 6,293 5,241 Hight Time 12+29 13+12 13+15 12+39			Specific Range/Altitude	0.173/FL 490	0.194/FL 470	0.176/FL 490	0.148/FL 470	0.169/FL 510
Ferry Average spect $4 \circ A$ $2 \circ A$			Nautical Miles	6,353	6,475	6,853	6,233	7,105
Ninstons Specific Range/Altitude 0.177/FL 90 0.188/FL 470 0.179/FL 510 0.150/FL 470 0.173/FL 510 Missions Runway NA 2,685 3,436 2,852 3,241 Hight Time 2+12 2+17 2+20 2+13 2+10 Specific Range/Altitude 0.175/FL 490 0.200/FL 470 0.179/FL 490 0.146/FL 470 0.168/FL 510 Missions Specific Range/Altitude 0.175/FL 490 0.200/FL 470 0.179/FL 490 0.146/FL 470 0.168/FL 510 Missions Runway NA 3,540 3,599 6,842 5,942 Runway NA 3,540 3,599 3,858 3,591 Runway NA 3,540 3,599 6,842 6,280 Bight Time 6,190 14,122 15,474 19,538 16,280 Gamma Fuel Used 0.187/FL 490 0.212/FL 470 0.194/FL 490 0.154/FL 470 0.184/FL 510 Remarks Certification Basis FAR, EASA CS 25 5,241 13,412 <th></th> <td>Ferry</td> <td>Average Speed Trin Fuel</td> <td>35,966</td> <td>32,653</td> <td>404 38,251</td> <td>404</td> <td>482 41,168</td>		Ferry	Average Speed Trin Fuel	35,966	32,653	404 38,251	404	482 41,168
Nissions Runway NA 2,685 3,436 2,852 3,241 1,000 nm Fight Time 2+12 2+17 2+20 2+13 2+10 Missions Specific Range/Altitude 5,728 4,994 5,599 6,842 5,942 Missions Bassengers) 3,000 nm Flight Time 6+19 6+39 6+42 6+20 6+17 6 passengers) 3,000 nm Flight Time 6+19 6+39 6+42 6+20 6+17 6,000 nm Flight Time 6+19 6+39 6+42 6,283 5,241 8,000 nm Flight Time 6+19 0.212/FL 470 0.194/FL 490 0.154/FL 470 0.184/FL 510 8,000 nm Flight Time 12+29 13+12 13+15 12+39 12+28 6,000 nm Flight Time 12+29 13+12 13+15 12+39 12+28 8,000 nm Flight Time 12+29 13+12 13+15 12+39 12+28 9,000 nm Flight Ti			Specific Range/Altitude	0.177/FL 490	0.198/FL 470	0.179/FL 510	0.150/FL 470	0.173/FL 510
Homon Flight Time Fuel Used Specific Range/Altitude 2+12 2+17 2+20 2+13 2+10 Missions Specific Range/Altitude 5,728 4,994 5,599 6,842 5,942 Missions Specific Range/Altitude 0.175/FL 490 0.200/FL 470 0.179/FL 490 0.146/FL 470 0.168/FL 510 Missions NA 3,540 3,599 3,858 3,591 Bassengers) Rumay NA 3,540 3,599 3,858 3,591 Bassengers) Flight Time 619 6+39 6+42 6+20 6+17 Fuel Used 16,060 14,122 15,474 19,538 16,280 Bassengers) Rumay NA 5,645 5,277 6,293 5,241 Rumay NA 5,645 5,277 6,293 5,241 12+29 Rumay NA 5,645 5,277 6,293 5,241 Fuel Used 34,432 30,729 33,428 41,053 34,622 <td< td=""><th></th><td></td><td>Runway</td><td>NA</td><td>2,685</td><td>3,436</td><td>2,852</td><td>3,241</td></td<>			Runway	NA	2,685	3,436	2,852	3,241
Missions (8 passengers) 3,000 nm Fuel Used Specific Range/Altitude (8 passengers) 5,728 (8 passengers) 4,994 (8 passengers) 5,599 (8 passengers) 6,842 (8 passengers) 5,942 (8 passengers) 3,000 nm Application 0.175/FL 490 (8 passengers) 0.200/FL 470 (8 passengers) 0.119/FL 490 (8 passengers) 0.146/FL 470 (8 passengers) 0.146/FL 470 (9 passengers) 0.146/FL 470 (9 passengers) 0.146/FL 470 (9 passengers) 0.146		1,000 nm	Flight Time	2+12	2+17	2+20	2+13	2+10
Missions (8 passengers) 3,000 nm Bight Time Fuel Used Specific Range/Altitude 0.179/TL 490 0.120/FL 470 0.119/FL 490 0.180/FL 470 0.173/FL 490 0.173/FL 490 0.173/FL 490 0.17		,	Fuel Used	5,728	4,994	5,599	6,842	5,942
Missions (8 passengers) 3,000 nm Flight Time Fuel Used Specific Range/Altitude 6+19 6+39 6+42 6+20 6+11 6,000 nm Fuel Used Specific Range/Altitude 16,060 14,122 15,474 19,538 16,280 6,000 nm Runway Flight Time Fuel Used Specific Range/Altitude 0.187/FL 490 0.212/FL 470 0.194/FL 490 0.154/FL 470 0.184/FL 510 6,000 nm Runway Flight Time Fuel Used Specific Range/Altitude NA 5,645 5,277 6,293 5,241 6,000 nm Flight Time Fuel Used Specific Range/Altitude 0.174/FL 490 0.195/FL 470 0.197/FL 490 0.146/FL 490 0.173/FL 510 Remarks Certification Basis FAR, EASA CS 25 pending FAR/EASA 25, 2016 EASy III flight deck; DFCS; 2017 delivery price. FAR 25, 1997/2003; EASA 25 CS, 2004 FAR 25, 1998/2003; JAR 25 BEVS and new Global Vision flight deck standard. FAR, EASA CS 25, 2012			Runwav	NA	3.540	3.599	3.858	3.591
(8 passengers) 5.000 nm Fuel Used Specific Range/Altitude 16,060 14,122 15,474 19,538 16,280 Remarks Remarks Rumay NA 5,645 5,277 6,293 5,241 Bassengers) NA 5,645 5,277 6,293 5,241 Bassengers) Flight Time 12+29 13+12 13+15 12+39 12+28 Bassengers) Specific Range/Altitude 0.174/FL 490 0.195/FL 470 0.179/FL 490 0.146/FL 490 0.173/FL 510 Remarks Certification Basis FAR, EASA CS 25 pending FAR/EASA 25, 2016 EASy III flight deck; DFCS; 2017 delivery price. FAR 25, 1997/2003; EASA 25 CS, 2004 FAR 25, 1998/2003; JAR 25 BEVS and new Global Vision flight deck standard. FAR, EASA CS 25, 2012	Missions	2 000	Flight Time	6+19	6+39	6+42	6+20	6+17
Specific Range/Altitude 0.187/FL 490 0.212/FL 470 0.194/FL 490 0.154/FL 470 0.184/FL 510 Rumway NA 5,645 5,277 6,293 5,241 Fight Time 12+29 13+12 13+15 12+39 12+28 Specific Range/Altitude 34,432 30,729 33,428 41,053 34,622 Specific Range/Altitude 0.174/FL 490 0.195/FL 470 0.179/FL 490 0.146/FL 490 0.173/FL 510 Remarks Certification Basis FAR, EASA CS 25 pending FAR/EASA 25, 2016 EASy III flight deck; DFCS; 2017 delivery price. FAR 25, 1997/2003; EASA 25 CS, 2004 FAR 25, 1998/2003; JAR 25 BEVS and new Global Vision flight deck standard. FAR, EASA CS 25, 2012	(8 passengers)	3,000 nm	Fuel Used	16,060	14,122	15,474	19,538	16,280
Kumway Filight Time Fuel big Fuel big Remarks NA 5,645 5,277 6,293 5,241 Remarks 6,000 nm 12+29 13+12 13+15 12+39 12+28 Flight Time Fuel big Specific Range/Altitude 12+29 13+12 13+15 12+39 12+28 Remarks Certification Basis FAR, EASA CS 25 pending FAR/EASA 25, 2016 EASy III flight deck; DFCS; 2017 delivery price. FAR 25, 1997/2003; EASA 25 CS, 2004 FAR 25, 1998/2003; BEVS and new Global Vision flight deck standard. FAR, EASA CS 25, 2012			Specific Range/Altitude	0.187/FL 490	0.212/FL 470	0.194/FL 490	0.154/FL 470	0.184/FL 510
6,000 nm right little Field statute 12429 13+12 13+15 12439 12428 Remarks Specific Range/Altitude 34,432 30,729 33,428 41,053 34,622 Remarks Certification Basis FAR, EASA CS 25 pending FAR/EASA 25, 2016 EASy III flight deck; DFCS; 2017 delivery price. FAR 25, 1997/2003; EASA 25 CS, 2004 FAR 25, 1998/2003; BEVS and new Global Vision flight deck standard. FAR, EASA CS 25, 2012			Runway	NA 12+20	5,645	5,277	6,293	5,241
Specific Range/Altitude 0.174/FL 490 0.195/FL 470 0.179/FL 490 0.146/FL 490 0.173/FL 510 Remarks Certification Basis FAR, EASA CS 25 pending FAR/EASA 25, 2016 EASy III flight deck; DFCS; 2017 delivery price. FAR 25, 1997/2003; EASA 25 CS, 2004 FAR 25, 1998/2003; JAR 25 BEVS and new Global Vision flight deck standard. FAR, EASA CS 25, 2012		6,000 nm	Filght Lime Fuel Lised	34,432	30,729	33,428	41,053	34,622
Remarks Certification Basis FAR, EASA CS 25 pending FAR/EASA 25, 2016 EASy III flight deck; DFCS; 2017 delivery price. FAR 25, 1997/2003; EASA 25 CS, 2004 FAR 25, 1998/2003; JAR 25 BEVS and new Global Vision flight deck standard.			Specific Range/Altitude	0.174/FL 490	0.195/FL 470	0.179/FL 490	0.146/FL 490	0.173/FL 510
	Remarks		Certification Basis	FAR, EASA CS 25 pending	FAR/EASA 25, 2016 EASy III flight deck; DFCS; 2017 delivery price.	FAR 25, 1997/2003; EASA 25 CS, 2004	FAR 25, 1998/2003; JAR 25 BEVS and new Global Vision flight deck standard.	FAR, EASA CS 25, 2012

Ultra-Long-Range Jets

Model Use of the second of the s	Manufacturer			Gulfstream Aerospace	Boeing	Airbus	Boeing	Boeing
Bit Accord From Head Coll 197,000,000	Model			Gulfstream 650ER	BBJ	ACJ319	BBJ MAX8	BBJ MAX9
Display Source (Longetty) Sou	POL Fanders al Dates			GVI \$60,400,000	\$70,000,000	A319-133	¢05 200 000	¢102 200 000
Charactor Witchangerowski 167/37 127/23/33 127/23/34	BCA Equipped	ped Price		\$69,400,000	\$79,000,000	\$87,000,000	\$95,300,000	\$103,300,000
Biolog New (PMU) New (Note) New (Note) New (Note) New (Note) Unrearieston 100 </td <td>Character-</td> <td>Win</td> <td>Seauring / Dowor Loading</td> <td>4+16/19/19</td> <td>4+19/00/149</td> <td>4+19/19/150</td> <td>125 1/2 24</td> <td>4+19/75/220</td>	Character-	Win	Seauring / Dowor Loading	4+16/19/19	4+19/00/149	4+19/19/150	125 1/2 24	4+19/75/220
Secural biology (b) 1/40 (1) 0.9.8 0.9.10.5 0.9.11.0 1.9.29.7 1.9.29.8 (1) 1.9.9 1.9.2 1.1.2 1.9.29.7 1.9.29 1.9.3 (1) 1.9.9 1.9.2 1.1.2	istics	Noise (FPNdR)	I ateral / Flyover/Approach	78 7/89 6/88 3	85 4/94 9/95 8	85 4/94 6/94 2	NA/NA/NA	NA /NA /NA
Spensions	External	10130 (E1 110B).	Length	99.8	110.3	111.0	129.7	138.3
more in the instruction of the instrume in the instrume in the instrume in the instrume instrume in the instrume instrume in the instrume in the instrume in the instrume in the instrume instrume instrume in the instrume instru	Dimensions	Height		25.7	41.2	38.6	40.3	40.3
Important International Internation	(ft)			99.6	117.4	111.8	117.8	117.8
Densition Image of the set	Internal	l ength:	Main Seating/Net/Gross	32 7/46 8/53 6	72 7/79 2/	78.0/78.0/	91 9/98 5/98 5	100 6/107 2/107 2
mm mm <thm< th=""> mm mm mm<td>Dimensions</td><td>He</td><td>ight/Dropped Aisle Depth</td><td>6.3/flat floor</td><td>79.3/flat floor</td><td>7.4/flat floor</td><td>7.1/flat floor</td><td>7.1/flat floor</td></thm<>	Dimensions	He	ight/Dropped Aisle Depth	6.3/flat floor	79.3/flat floor	7.4/flat floor	7.1/flat floor	7.1/flat floor
Ld			Width Max/Floor	8 2/6 7	11 6/10 7	12 2/11 6	11 6/10 7	11 6/10 7
Bagggoot Image: Chi Price 125/1/A NA/A 72/A 20/A/A Power margine interval functions in the chi Price Accord and a chi Price Accord and and a chi Price Accord and a chi Price Accord an	(11.)		Internal: Cu. ft /lb	235/2 500	NA /NA	160/NA	NA /NA	
Power Const. Const. <thcons.< th=""> <thcons.< th=""> Cons.</thcons.<></thcons.<>	Baggage		External: Cu. ft /lb		159/NA	NA/NA	71.3/NA	874/NA
Protect Protect PRODUCTSAL12 CPAR06-192/74 CPAR06-192/74 CPAR06-192/74 CPAR06-192/74 seminational function of the state 15000/16-4150 27000/16-4120 28000/15-4120 28000/15-4120 seminational function of the state 10000/16-4120 100		Engines Output (lb. each)/Flat Rating		2 RR	2 CFMI	2 CFMI	2 CFMI	2 CFMI
Huma monutation analytication 23,000/18-11/50 27,000/18-12/50 28,000/18-11	Davisa			BR700-725A1-12	CFM56-7B27E	CFM56-5B7/3*	LEAP-1B	LEAP-1B
Image on utricity/leng. Since Final Intervel 10.000//- 00//- 00//- 00//- 00//- 00//- 00//- 00//- 00//- Weights (h). Weights (h). Weights (h). Weights (h). 125.200 <td>Power</td> <td>16,900/ISA+15C</td> <td>27,300/ISA+15C</td> <td>27,000/ISA+29C</td> <td>28,000/ISA+15C</td> <td>28,000/ISA+15C</td>	Power			16,900/ISA+15C	27,300/ISA+15C	27,000/ISA+29C	28,000/ISA+15C	28,000/ISA+15C
Martine 104.003 171.500 183.50 184.700 193.200 Weg/15 (b)		Inspection Interval/Manu. Service Plan Interval		10,000t/—	0C/—	0C/—	0C/—	0C/—
Neights (b) Image: https://www.image: https://wwwwimage: https://wwwwwwwwwwwwwwwwwwwwwwwwwwwwwwwwwww		Max Ramp		104,000	171,500	169,530	181,700	195,200
Neights (b): Neights (b):<		Max Takeoff		103,600	1/1,000	168,650	181,200	194,700
Weights (b)		Wax Landing Zoro Evol		83,500	134,000	137,790	145,800	156 5000
Weights (b) More Peakson 0.000 222800 322501 34.400 38.420 Market Market 48,200 73.460 73.460 73.700 77.120 Market Market 48,200 73.737 77.500 68.814 73.337 Market Market 1000 1.723 500 1.888 3.708 Market Market 1000 1.723 500 1.889 0.890 <td></td> <td colspan="2" rowspan="2">BOW Max Pavload</td> <td>54 500</td> <td>98.040</td> <td>96.450**</td> <td>110,000</td> <td>118 080</td>		BOW Max Pavload		54 500	98.040	96.450**	110,000	118 080
Interface 19,200 73,840 73,080 72,700 77,220 Anable Pyriod vills in the 1,300 1,723 72,580 65,814 73,235 Umits Inter AR, Rive II, 300 1,723 520 1,888 3,795 Umits Inter AR, Rive II, 520,340 FL250,350 9,240,00 9,224,000 <	Weights (lb.)			6,000	27 960	32 520	.35 400	.38 420
Mailabe Project Vial Fail 48,200 72,737 72,260 08,814 72,335 Anallabe Project Vial Fail 1,300 1,723 500 1,886 3,735 Limits Mail Definition Vial Project 43,500 45,500 40,550 0,8370			Useful Load	49,500	73,460	73,080	71,700	77,120
Anabez Pojes uf Ma-mail 1.300 1.723 520 1.888 3.795 Limits Mateber Pojes uf Ma-mail 3.500 45.500 46.500 46.500 36.200 <td></td> <td>Max Fuel</td> <td>48,200</td> <td>71,737</td> <td>72,560</td> <td>69,814</td> <td>73,325</td>			Max Fuel	48,200	71,737	72,560	69,814	73,325
Matable fuel yllash yold 44,550 45,500 40,560 36,300 38,700 Limits Image fuel yllash yold Image fuel yllash		Avai	ilable Payload w/Max Fuel	1,300	1,723	520	1,886	3,795
Linits ms. R. IV. F. 250: 340 0.820 0.820 0.820 0.820 All post. Left 64 10.733.100 9.024.000 8.272.005 II. 260:340 III. 260:340 IIII. 260:340 IIIII. 260:340 IIIII. 260:340 IIIII. 260:340 IIIII. 260:340 IIIIII. 260:340 IIIIII. 260:340 IIIIIIIII. 260:340 IIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIII		Available Fuel w/Max Payload		43,500	45,500	40,560	36,300	38,700
Limits Item & H. Wu FL 250/340 FL 250/340 FL 260/340 FL 260/34			Ммо	0.925	0.820	0.820	0.820	0.820
Arport Performante ruy de surve Mais Rege 103,800 20/24,000 9.0/24,0	Limits		Trans. Alt. FL/VMO	FL 290/340	FL 260/340	FL 250/350	FL 260/340	FL 260/340
Arport Performance Units Bergissempi (R) (B) (B) (B) (B) (B) (B) (B) (B) (B) (B		PSI/Sea-Level Cabin		10.//31,900	9.0/24,000	8.3/22,000	9.0/24,000	9.0/24,000
Autorit Mark Book 11/2000 10/2000 50/2000 Mark Mark Performance			IUFL (SL EIEV./ ISA TEMP.)	0,299	0,085	0,1/0	0,030	8,200
Amport Performance NMAI IP may (1.417) 1.6.297 (2.97) 2.600 NM MA We 1.14 1.17 1.13 1.22 1.24 We 1.14 1.17 1.13 1.22 1.24 Climb The follow family fibration 2.680 2.2200 2.2440 2.2570 Climb Ref 5 (dinkAmbe 21/FL 370 252/FL 370 222/300 2.440 2.570 Climb Ref 25 (dipe of diseting fl.mm NA NA NA NA NA NA Celling (fL) Certificate Certificate 1.000 41.000 41.000 41.000 41.000 41.000 MA MA NA			Mission Weight	103 600	171 000	168 650	NA	NA
Performance v 148 141 137 MA MA MA Laming Distance 2,680 2,360 2,2360 2,240 2,570 Clinb Fite to Clinh Atta 117 111 122 124 Clinb Fite to Clinh Atta 117 111 123 24/1.350 26/71.300 Clinb Fite to Clinh Atta NA NA NA MA MA MA Colling (ft.) Ma Display Clinh Atta NA NA NA MA MA MA Colling (ft.) Ma Display Clinh Atta 11000 4100 4400	Airport		NBAA IFR Range	7.437	6.297	6.000	NA	NA
NBALIER No. 114 117 111 122 124 Imme to limit/hittidue 2.680 2.380 2.220 2.440 2.570 Clinb Max Stepschart NA NA NA NA NA Delta Stepschart Entrate Climit/hittidue 22/FL 370 22/300 22/410 22/FL 350 41.000 41.000 41.000 41.000 41.000 41.000 41.000 41.000 41.000 42.67 44.65 45.77 44.77 47.71	Performance	V2 Vref		148	141	137	NA	NA
Image: constraint of the second sec				114	117	111	122	124
Circle The to CompAthedia 21/FL 370 22/FL 370 22/APL 380 26/FL 330 26/FL 330 Circline FR2 35 Eigen- 0.0.8te (fm) NA NA <t< td=""><td colspan="2">Landing Distance</td><td>2,680</td><td>2,360</td><td>2,220</td><td>2,440</td><td>2,570</td></t<>		Landing Distance		2,680	2,360	2,220	2,440	2,570
Clinito R82 5 Egine-0.4 Ret (m) NA <	Climb	Time to Climb/Altitude		21/FL 370	25/FL 370	22/360	24/FL 350	26/FL 330
Child Stright-Out Cadent (Lym) MA		FAR 25 Engine-Out Rate (fpm)		NA	NA	NA	NA	NA
Ceiling (ft.) Lefthese 0.1000 41.000 <t< td=""><td></td><td colspan="2">FAR 25 Engine-Out Gradient (ft./nm)</td><td>NA 51.000</td><td>NA</td><td>NA</td><td>NA</td><td>NA</td></t<>		FAR 25 Engine-Out Gradient (ft./nm)		NA 51.000	NA	NA	NA	NA
Curring (L) The regular strate 12,000 NA 38,000 NA MA Integration of the regular strate 22,000 NA 18,000 NA MA MA Cruise Integration strate 22,800 NA 18,000 NA MA MA Cruise Integration strate 2,883 402 447 493 47 Cruise Integration strate 1,590 ND 0,700 NA NA High Speed Field Star 5,150 Field Star 5,150 NA NA NBA Fight National Mark Field Star 3,206 2,679 2,682 2,628 NBA Fight National Mark 6,459 3,306 0,080/Fi 370 NA/Fi 370 NA NA NBA Fight National Mark 7,507 6,285 6,134 6,521 6,300 NBA Fight National Mark 7,507 6,285 6,134 6,521 6,300 NA Mark Ma NA </td <td>Colling (ft)</td> <td colspan="2" rowspan="2">All-Engine Service</td> <td>51,000</td> <td>41,000</td> <td>41,000</td> <td>41,000</td> <td>41,000</td>	Colling (ft)	All-Engine Service		51,000	41,000	41,000	41,000	41,000
Cruise Log Ford Mark 486 482 447 455 457 Cruise Ing Range Fiel Row 2.833 4.679 4.695 MM MM Cruise Fiel Row 2.833 4.679 4.695 MM MM Cruise Fiel Row 2.830 FI.450 FI.390 FI.370 FI.380 FI.380 Nigh Speed Attack FI.450 FI.370 FI.370 FI.370 FI.360 FI.380 Na Attack FI.450 FI.370 FI.370 FI.360 FI.370 FI.370 FI.360	Cennig (IL.)			25,000	NA NA	18,000	NA NA	NA NA
Under Range Full Flow Mutual Mutual FL 450 FL 300 FL 300 FL 370 FL 370 FL 380 FL 380 Cruise Sectif Renge Mitig Speed 0.169 0.097 0.095 NA NA High Speed Flat Flow Mutual FL 450 FL 516 470 471 471 471 High Speed Flat Flow Mutual FL 450 FL 370 FL 370<	Cruise	TAS		488	452	447	455	457
Long Range Attude FL 450 FL 300 FL 300 FL 320 FL 320 FL 320 Cruise Seciel Reg. 0.169 0.097 0.0955 NA NA High Speed FL 350 FL 300		Long Range	Fuel Flow	2,883	4,679	4,695	NA	NA
Cruise Specific Range Fuel For 0.169 0.097 0.095 MA MA MA High Specific Range Human 15.16 470 470 471 471 471 High Specific Range Human 1.05 5.16 470 FL 350 FL 360 FL 380 Specific Range (with available fuel (with available fuel (%) available fuel (%) available fuel (%) available fuel (%) a			Altitude	FL 450	FL 390	FL 370	FL 380	FL 360
NBA IFR Ranges Form Task Attude 516 Full 470 Attude 471 Full 471 Attude 4			Specific Range	0.169	0.097	0.095	NA	NA
High Speet Full Flow Autor 3.136 5.550 5.830 NA NA Name Autor FL360 FL360 FL360 FL360 FL360 Name Name 0.165 0.085 0.081 NA NA Name Name 0.165 0.085 0.081 NA NA Name Name 0.165 0.085 0.081 NA NA Name Name 0.4649 3.306 2.679 2.692 2.628 Name Name Name NA NA NA NA Specific Range/Altrude 0.160/FL400 0.084/FL30 0.080/FL370 NA/EL370 NA/EL370 Ranges (200-m) Specific Range/Altrude 0.166/FL510 0.094/FL410 0.094/FL410 NA/EL370 NA/EL370 NA/EL370 Ranges (with available fue) Specific Range/Altrude 0.166/FL510 0.094/FL410 0.094/FL410 NA/FL30 NA/FL30 Natrel difue Specific Range/Altrude <		High Speed	TAS	516	470	470	471	471
NBAA IR Ranges (thit available fue) Nuture (thit fu			Fuel Flow	3,136	5,550	5,830	NA	NA
NBA IFR (with available field) Specific Range (with available field) 0.105 (Astroname 0.0085 (Astroname 0.0081 (Astroname NA NA NBA IFR Ranges (200-mm alternate) Max Feld (with available field) NA/FL 370 NA/FL 370 NA/FL 350 (200-mm alternate) Max Feld (With available field) Max Feld (With available field) Na/FL 150 0.094/FL 430 0.080/FL 430 NA NA Specific Range/Attinué (200-mm alternate) Eight Passengers (with available field) Navical Miss 7.437 6.6270 6.002 6.555 6.376 Ferry Maxical Miss 7.437 6.6273 NA NA NA Navical Miss 7.437 6.6274 0.0092/FL 410 NA/FL 390 NA/FL 390 NA/FL 390 Navical Miss 7.636 6.5134 6.5558 NA NA NA Navical Miss 7.636 6.348 6.200 6.619 6.441 Navavical Miss 7.638 6.6886			Altitude	FL 450	FL 370	FL 370	FL 360	FL 360
Max Payload (wfh waliable fuel) Max Payload Aereage Speed 0.10.9 (wfh waliable fuel) 0.10.9 (wfh waliable fuel) 0.10.07 (wfh waliable fuel) <t< td=""><td></td><td></td><td>Specific Range</td><td>0.105</td><td>2 206</td><td>0.081</td><td>NA</td><td>NA</td></t<>			Specific Range	0.105	2 206	0.081	NA	NA
NBAA IFR (with available feel) Integroup 40.285 39,508 33,677 MA MA NBAA IFR Ranges (200-nm alternate) Max Fuel (with available feel) Specific Range,Altitude 0.160/FL 420 0.084/FL 320 0.080/FL 370 NA/RL 370 NA/RL 370 NA/RL 370 NBAA IFR Ranges (200-nm alternate) Max Fuel (with available feel) Name (bit Participant) Name (bit Participant) 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/RL 390 NA/RL 390 Specific Range/Altitude 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 390 Bit Passengers (with available feel) National Miles 7,437 6,270 6,002 6,555 6,376 Bit Passengers (with available feel) Specific Range/Altitude 0.066/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 National Miles 7,636 6,723 65,558 NA NA NA National Miles 7,636 0.348 6,200 6,619 6,441 National Miles 7,536 0.348 6,207 NA NA	NBAA IFR Ranges	Max Pavload	Average Sneed	481	437	434	2,092 NA	2,028 NA
NBA IFR Ranges (200-nm alternate) Specific Range/Attuide Markied Miss (with available poload) 0.160/FL 490 0.084/FL 390 0.080/FL 370 NA/FL 370 NA/FL 350 NBA IFR Ranges (200-nm alternate) Markied (with available poload) Nature Markied Miss (Poload) 7.507 6.285 6.134 6.521 6.300 Bigh Passengers (200-nm alternate) Markied Miss (with available fue) 7.437 6.270 6.002 6.555 6.376 Bigh Passengers (with available fue) Natureal Miss (with available fue) Natureal Miss (Police Range/Attuide 5.9edific Range/Attuide 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 390 Bigh Passengers (with available fue) Marking Miss (Portice Range/Attuide 5.9edific Range/Attuide 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Missions (g passengers) 1.000 nm Runway 5.9edific Range/Attuide 5.942 0.168/FL 510 0.095/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Missions (g passengers) 3.000 nm Runway 5.9edific Range/Attuide 0.168/FL 510 0.095/FL 410 0.090/FL 410 NA/NA NA Missions (g passengers) 3.000 nm		(with available fuel)	Trin Fuel	40.285	39 508	33 677	NA	NA
Max Fuel (with available payload) Natical Miles Average Speed (all source) 7,507 6,285 6,134 6,521 6,300 (with available payload) (with available payload) Speedic Range/Attude 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 390 (with available fue) alternate) Eight Passengers (with available fue) Farge Speed (Max available fue) 7,637 6,6285 6,673 NA NA Barnate Eight Passengers (with available fue) Speedic Range/Attude Die fue 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Barnate Speedic Range/Attude Die fue 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Barnate Speedic Range/Attude Die fue 0.166/FL 510 0.099/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Maxical Miles 7,636 6,348 6,200 6,619 6,411 Marce Speedic Range/Attude 0.169/FL 510 0.099/FL 410 NA/FL 390 NA/FL 410 Missions 1,000 nm Rumay 3,241 3,485 4.075			Specific Range/Altitude	0.160/FL 490	0.084/FL 390	0.080/FL 370	NA/FL 370	NA/FL 350
MBAA IFR (with available payload) Average Speed (with available payload) Average Speed (with available payload) 482 443 442 NA NA Ranges (200-nm alternate) (with available payload) Specific Range/Altrude (with available fuel) 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 390 (200-nm alternate) Eight Passenger, (with available fuel) Specific Range/Altrude (0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 420 Specific Range/Altrude (with available fuel) Specific Range/Altrude (0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 420 NA NA Notacial Miles (with available fuel) Specific Range/Altrude (0.168/FL 510 0.095/FL 410 0.092/FL 410 NA/FL 420 NA NA Notacial Miles (0.168/FL 510 0.095/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Notacial Miles (1.000 nm Fuel Used (0.168/FL 510 0.095/FL 410 0.092/FL 410 NA/FL 420 Missions (8 passenger) 1.000 nm Fuel Used (0.168/FL 510 0.095/FL 410 0.0370 NA NA Missions (8 passenger) 0.000			Nautical Miles	7,507	6,285	6,134	6,521	6,300
NBAA IFR Ranges (200-m alternate) twice areasize payload) Tip Fuel Specific Range/Attude 45.129 66.854 66.673 NA NA Bight Passengers (with available fuel) Specific Range/Attude 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 390 NA/FL 390 Bight Passengers (with available fuel) Karege Speed 4482 443 4422 NA NA Perry Specific Range/Attude 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Naticial Miles 7,636 6,348 6,200 6,619 6,441 Naticial Miles 7,636 6,348 6,200 6,619 6,441 Naticial Miles 7,636 6,886 67,207 NA NA Naticial Miles 7,636 0.095/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Naticial Miles 7,636 0.427 2426 NA NA NA Missions 1,000 nm Full Wathing 0.168/FL 510 0.0095/FL 410 0.096/FL 410 <t< td=""><td>Max Fuel</td><td>Average Speed</td><td>482</td><td>443</td><td>442</td><td>NA</td><td>NA</td></t<>		Max Fuel	Average Speed	482	443	442	NA	NA
Ranges (200-nm) Specific Range/Altitude Natical Miles 0.066/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 390 (200-nm) Bight Passengers (with available fuel) Natical Miles 7.437 6.270 6.002 6.555 6.376 alternate) Eight Passengers (with available fuel) Natical Miles 7.437 6.270 6.002 6.555 6.376 Auroical Miles 7.636 6.348 442 NA NA Natical Miles 7.636 6.348 6.200 6.619 6.441 Natical Miles 7.636 6.348 6.200 6.619 6.441 Natical Miles 7.636 6.348 6.200 6.619 6.441 Natical Miles 7.636 0.095/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Natical Miles 7.636 0.442 442 A42 NA NA Natical Miles 7.636 0.095/FL 410 0.096/FL 410 NA/FL 390 NA/FL 410 Natical Miles 7.636 0.169/FL 510		(WITH available	Trip Fuel	45,129	66,854	66,673	NA	NA
High Passengers (with available turi) Nautical Miles Fight Passengers (with available turi) Nautical Miles Firs Field 7.437 6.270 6.002 6.555 6.376 Hight Passengers (with available turi) Wareage Speed 482 443 442 NA NA Fiery Specific Range/Altitude 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Average Speed 482 442 442 NA NA NA Average Speed 1.69/FL 510 0.095/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Fight Time 5.942 10.478 10.370 NA NA Hould Specific Range/Altitude 0.168/FL 510 0.095/FL 410 NA/NA NA/NA Specific Range/Altitude 0.164/FL 510 <t< td=""><td>payloady</td><td>Specific Range/Altitude</td><td>0.166/FL 510</td><td>0.094/FL 410</td><td>0.092/FL 410</td><td>NA/FL 390</td><td>NA/FL 390</td></t<>		payloady	Specific Range/Altitude	0.166/FL 510	0.094/FL 410	0.092/FL 410	NA/FL 390	NA/FL 390
Light Passengers (with available fue) Average Speed (with available fue) 482 443 442 NA NA Imp Fuel Tip Fuel 44,820 66,723 65,558 NA NA NA Ferry Specific Range/Altitude 0.166/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Natrical Miles 7,636 6,348 6,200 66,619 6,441 Natrical Miles 7,636 6,348 6,200 6,619 6,441 NA NA NA NA NA NA Missions Tip Fuel 45,168 66,886 67,207 NA NA Rumay 3,241 3,485 4,075 NA NA NA Page/Altitude 0.168/FL 510 0.0095/FL 410 0.092/FL 410 NA/NA NA Missions 3,000 nm Fuel Used 5,942 10,478 10,370 NA NA Karay 3,241 5,855 6,160 NA NA NA <	(200-nm	-	Nautical Miles	7,437	6,270	6,002	6,555	6,376
Missions 3,000 nm Fight Time 6-12 0.595 (FL 410) NA NA Missions 8,000 nm 6,000 nm 0.168/FL 510 0.094/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Missions 8,000 nm 10,000 nm <t< td=""><td>alternate)</td><td>Light Passengers</td><td>Average Speed</td><td>482</td><td>443</td><td>442</td><td>NA</td><td>NA</td></t<>	alternate)	Light Passengers	Average Speed	482	443	442	NA	NA
Ferry Specific Range/Attitude OLDGY/TE 310 OLDGY/TE 410 NA/FE 390 NA/FE 410 Natrical Miles 7.636 6.348 6.200 6.619 6.441 Average Speed 482 442 442 NA NA Specific Range/Attitude 0.169/FL 510 0.095/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Specific Range/Attitude 0.169/FL 510 0.095/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Missions Runway 3.241 3.485 4.075 NA NA Field Marge 5.942 10.478 10.370 NA NA Missions Specific Range/Attitude 0.168/FL 510 0.095/FL 410 0.096/FL 410 NA/NA NA Missions 8 3.000 nm FightTime 6+17 6+54 6+54 NA NA Runway 3.591 4.290 4.280 NA NA NA 8 passengers) 3,000 nm FilghtTime 6+17 6+54 NA <td< td=""><td></td><td>(with available fuel)</td><td>Irip Fuel Specific Panets (Altitude</td><td>44,820</td><td>0.004/EL.410</td><td>00,000 0 000/EL 440</td><td>NA NA /EL 200</td><td>NA NA/EL 410</td></td<>		(with available fuel)	Irip Fuel Specific Panets (Altitude	44,820	0.004/EL.410	00,000 0 000/EL 440	NA NA /EL 200	NA NA/EL 410
Ferry Number Network Notestimate Outbox			Nautical Miles	7.636	6.348	6 200	6 619	6 441
Ferry Integration Integration <thintegrater< th=""> <thintegration< th=""> <thint< td=""><td></td><td></td><td>Average Speed</td><td>482</td><td>442</td><td>442</td><td>0,015</td><td></td></thint<></thintegration<></thintegrater<>			Average Speed	482	442	442	0,015	
Specific Range/Altitude 0.169/FL 510 0.095/FL 410 0.092/FL 410 NA/FL 390 NA/FL 410 Homman Runway 3,241 3,485 4,075 NA NA J,000 nm Flight Time 2+10 2+27 2+26 NA NA Specific Range/Altitude 0.168/FL 510 0.095/FL 410 0.096/FL 410 NA/NA NA Missions Specific Range/Altitude 0.168/FL 510 0.095/FL 410 0.096/FL 410 NA/NA NA Missions Altitude 0.168/FL 510 0.095/FL 410 0.096/FL 410 NA/NA NA Bight Time Flight Time 6+17 6+54 6+54 NA NA Remarks Specific Range/Altitude 0.184/FL 510 0.102/FL 410 0.100/FL 410 NA/NA NA Bight Time Flight Time 12+28 13+34 13+35 NA NA Bight Time Flight Time 12+28 13+34 13+35 NA NA Bight Time 0.173/FL 510 0.095/FL 410		Ferry	Trip Fuel	45,168	66,886	67,207	NA	NA
Homman Rumway Flight Time Specific Range/Altitude 3,241 3,485 4,075 NA NA Missions (8 passengers) 1,000 nm Fight Time Fuel Used Specific Range/Altitude 2+10 2+27 2+26 NA NA NA Missions (8 passengers) 3,000 nm Fight Time Fuel Used Specific Range/Altitude 0.168/FL 510 0.095/FL 410 0.096/FL 410 NA/NA NA/NA Missions (8 passengers) 3,000 nm Flight Time Fuel Used 6+17 6+54 6+54 NA NA NA 6,000 nm Flight Time Fuel Used 0.184/FL 510 0.102/FL 410 0.100/FL 410 NA/NA NA 6,000 nm Flight Time Fuel Used 0.184/FL 510 0.102/FL 410 0.100/FL 410 NA/NA NA 6,000 nm Flight Time Fuel Used 0.173/FL 510 0.095/FL 410 0.092/FL 410 NA/NA NA Remarks Certification Basis FAR 25, 2014 FAR 25 A 77, 1967/98 Split scimitar winglets. 2016 data. FAR 25, 1999 *Also available with A5 engines; includes 6 additional center tanks 6 Additional center tanks 6 Additional center tanks 8 Add ast prelimin			Specific Range/Altitude	0.169/FL 510	0.095/FL 410	0.092/FL 410	NA/FL 390	NA/FL 410
I,000 nm Flight Time Fuel Used Specific Range/Altitude 2+10 2+27 2+26 NA NA Missions (8 passengers) Specific Range/Altitude 5,942 10,478 10,370 NA NA Missions (8 passengers) 3,000 nm Specific Range/Altitude 0.168/FL 510 0.095/FL 410 0.096/FL 410 NA/NA NA/NA Bassengers) 3,000 nm Fuel Used Specific Range/Altitude 6+17 6+54 6+54 NA NA Bassengers) Specific Range/Altitude 0.184/FL 510 0.102/FL 410 0.100/FL 410 NA/NA NA Bassengers) Rumway 5,241 5,855 6,160 NA NA Bassengers) Rumway 5,241 5,855 6,160 NA NA Bassengers) Rumway 5,241 5,855 6,160 NA NA Bassengers/Bitute 0.134/FL 510 0.095/FL 410 0.092/FL 410 NA/NA NA Bassengers/Bitute 0.173/FL 510 0.095/FL 410 0.092/FL 410 NA/NA NA<		1,000 nm	Runway	3,241	3,485	4,075	NA	NA
Remarks Fuel Used 5,942 10,478 10,370 NA NA Specific Range/Altitude 0.168/FL 510 0.095/FL 410 0.096/FL 410 NA/NA NA/NA Missions (8 passengers) 3,000 nm Flight Time Fuel Used 6+17 6+54 6+54 NA NA Bassengers) 7 6+17 6+54 6+54 NA NA Bassengers) 8 7 6+17 6+54 6+54 NA NA Bassengers) 8 5 6+17 6+54 6+54 NA NA Bassengers) 8 5 6+10 0.100/FL 410 NA/NA NA Bassengers) 8 6,000 nm Flight Time 12+28 13+34 13+35 NA NA BassengerAltitude 0.173/FL 510 0.095/FL 410 0.092/FL 410 NA/NA NA Specific Range/Altitude 0.173/FL 510 0.095/FL 410 0.092/FL 410 NA/NA NA/NA Missingeral Specific Range/Altitude			Flight Time	2+10	2+27	2+26	NA	NA
Specific Range/Altitude 0.168/FL 510 0.095/FL 410 0.096/FL 410 NA/NA NA/NA Missions (8 passengers) 3,000 nm Rumay Flight Time Fuel Used 6+17 6+54 6+54 NA NA 6,000 nm Flight Time Fuel Used 16,280 29,534 30,070 NA NA 7 6+54 0.102/FL 410 0.100/FL 410 NA/NA NA 8 Specific Range/Altitude 0.184/FL 510 0.102/FL 410 0.100/FL 410 NA/NA 8 Rumay 6,000 nm Flight Time Flight Time 12+28 13+34 13+35 NA NA 8 Certification Basis FAR 25, 2014 FAR 25 63,311 65,528 NA NA/NA 8 Certification Basis FAR 25, 2014 FAR 25 A 77, 1967/98 Split scimitar winglets. 2016 data. FAR 25, 1999 *Also available with blus VIP cabin. **Spec weight. <i>Refuel estimited data</i> FAR 25 A TBD All data preliminary. 2016 data. All data preliminary. 2016 data. 2016 data.		2,000 min	Fuel Used	5,942	10,478	10,370	NA	NA
Missions (8 passengers) 3,000 nm Hummay Flight Time Fuel Use Specific Range/Altitude 3,991 (6,280 4,290 (6+54 4,280 (6+54 NA NA 6,000 nm Fight Time Fuel Use Rummay 6+17 (5,280 0.102/FL 410 0.100/FL 410 NA NA 6,000 nm Flight Time Fuel Use Specific Range/Altitude 0.184/FL 510 0.102/FL 410 0.100/FL 410 NA/NA NA 6,000 nm Flight Time Fuel Use Specific Range/Altitude 0.173/FL 510 0.095/FL 410 0.092/FL 410 NA NA Remarks Certification Basis FAR 25, 2014 FAR 25 A 77, 1967/98 Split scimitar winglets. 2016 data. FAR 25, 1999 *Also available with A5 engines; includes 6 additional center tanks Big to bab. **Spec weight. **Spec weight. FAR 25 A TBD All data preliminary. 2016 data. FAR 25 A TBD All data preliminary. 2016 data. All data preliminary. 2016 data. All data preliminary. 2016 data.			Specific Range/Altitude	0.168/FL 510	0.095/FL 410	0.096/FL 410	NA/NA	NA/NA
Missions 3,000 nm Fright limit Fuel lised 0 + 1/2 0 + 5/4 NA NA (8 passengers) 3,000 nm Fuel lised 16,280 29,534 30,070 NA NA 6,000 nm Flight Time Fuel lised 0.184/FL 510 0.102/FL 410 0.100/FL 410 NA/NA NA 6,000 nm Flight Time Fuel lised 12+28 13+34 13+35 NA NA 8,000 nm Flight Time Fuel lised 12+28 13+34 13+35 NA NA 9,000 nm Fuel lised 34,622 63,311 65,528 NA NA 9,000 nm Fuel lised 0.173/FL 510 0.095/FL 410 0.092/FL 410 NA/NA NA/NA 8,000 nm Specific Range/Altitude 0.173/FL 510 0.095/FL 410 0.092/FL 410 NA/NA NA/NA 8,000 nm Certification Basis FAR 25, 2014 FAR 25 A 77, 1967/98 Split scimitar winglets. 2016 data. FAR 25, 1999 *Also available with A5 engines; includes 6 additional center tanks 8 digitional center tanks 8 digitional center tanks 8 digitional center tanks 8 digitional center ta	Missione		Runway	3,591	4,290	4,280	NA	NA
Remarks Certification Basis FAR 25, 2014 FAR 25, 2016 FAR 25, 201	(8 passengers)	3,000 nm	Fiight Llood	16.280	29 52/	0+04 30.070	NA	NA NA
Remarks Certification Basis FAR 25, 2014 FAR 25, 2014 FAR 25 A 77, 1967/98 2016 data. FAR 25 A 77, 1967/98 2016 data. FAR 25 A 78D FAR 25 A 78D FAR 25 A 77, 1967/98 2016 data. FAR 25 A 78D FAR 25 A 78D FAR 25 A 77, 1967/98 2016 data. FAR 25 A 78D FAR 25 A 77, 1967/98 2016 data. FAR 25 A 78D FAR 25 A 78D FAR 25 A 77, 1967/98 2016 data. FAR 25 A 78D FAR 25 A 77D FAR 25 A 77D FAR 25 A 78D FAR 25 A 78D All data preliminary. 2016 data. FAR 25 A 78D			Specific Range/Altitude	0.184/FL 510	0.102/FL 410	0.100/FL 410	NA/NA	NA/NA
6,000 nm Flight Time Fuel Used Specific Range/Altitude 12+28 13+34 13+35 NA NA Remarks Certification Basis FAR 25, 2014 FAR 25, 77, 1967/98 Split scimitar winglets. 2016 data. FAR 25, 77, 1967/98 Split scimitar winglets. 2016 data. FAR 25, 77, 1967/98 FAR 25, 2014 FAR 25, 77, 1967/98 Split scimitar winglets. 2016 data. FAR 25, 1999 *Also available with 26,500-lbf IAEV2527M- A5 engines; includes 6 additional center tanks plus VIP cabin. **Spec weight. FAR 25 A TBD All data preliminary. 2016 data. FAR 25 A TBD All data preliminary. 2016 data. All data preliminary. 2016 data.		6,000 nm	Runwav	5,241	5,855	6,160	NA	NA
Fuel Used Specific Range/Altitude 34,622 63,311 65,528 NA NA Remarks Certification Basis FAR 25, 2014 FAR 25, 77, 1967/98 FAR 25, 77, 1967/98 FAR 25, 2014 FAR 25, 2016 data. FAR 25, 2016 FAR 25,			Flight Time	12+28	13+34	13+35	NA	NA
Specific Range/Altitude 0.173/FL 510 0.095/FL 410 0.092/FL 410 NA/NA Remarks Certification Basis FAR 25, 2014 FAR 25 A 77, 1967/98 FAR 25, 500-1bf IACV2527M-A5 engines; includes 6 additional center tanks 2016 data. FAR 25 A TBD FAR 25 A TBD FAR 25 A TBD Number of the second se			Fuel Used	34,622	63,311	65,528	NA	NA
RemarksCertification BasisFAR 25, 2014FAR 25 A 77, 1967/98 Split scimitar winglets. 2016 data.FAR 25, 1999 *Also available with A5 engines; includes 6 additional center tabin. **Spec weight. **Spec weight.FAR 25 A TBD A1 data preliminary. 2016 data.FAR 25 A TBD AII data preliminary. 2016 data.		Specific Range/Altitude		0.173/FL 510	0.095/FL 410	0.092/FL 410	NA/NA	NA/NA
	Remarks	Certification Basis		FAR 25, 2014	FAR 25 A 77, 1967/98 Split scimitar winglets. 2016 data.	FAR 25, 1999 *Also available with 26,500-lbf IAEV2527M- A5 engines; includes 6 additional center tanks plus VIP cabin. **Spec weight. BCA estimated data.	FAR 25 A TBD All data preliminary. 2016 data.	FAR 25 A TBD All data preliminary. 2016 data.